



SUBSTRUCTURAL
VALUE ENGINEERED DESIGN

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SUBSTRUCTURAL CALCULATIONS

(PREPARED BY JMS CONSULTING ENGINEERS LTD)

**19 Rona Road,
London, NW3 2HY**

Project No: L15/088/04

Revision	Date	Changes
P1	16/07/15	Preliminary Issue for Planning Purposes
P2	24/07/15	Rear extension's steelwork frame amended to suit Architect's new layout
P3	30/07/15	Rear extension's steelwork frame now shown on key plans



General Construction Notes and Guidance on using these Calculations

1. Calculations are not to be used for the purpose of ordering materials and should only be used for Building Regulations submissions. All dimensions should be checked by the contractor on site.
2. All steelwork to be mechanically wire brushed and painted two coats of red oxide. Steelwork located in the cavity or below DPC to be suitably protected with 2 coats of bituminous paint.
3. All steelwork connections to use grade 8.8 bolts unless stated otherwise. These are to be spanner tightened using the appropriate podger spanner (min length 460mm) or suitable power tools in accordance with BS2583. If a torque wrench is used the torque applied should be around 90Nm for M16 bolts, 110Nm for M20 & 130Nm for M24.
4. All timber to be grade C24 (SC4), unless stated otherwise. Preservative treated to Architects details.
5. To be read in conjunction with Architects drawings, any inconsistencies between the drawings should be reported. If any site conditions or existing details are found that may affect the structural design, JMS Consulting Engineers are to be notified immediately.
6. For details of fire protection to steelwork, see Architects drawings.
7. The Contractor is to ensure that all existing construction is adequately supported, using needles and props as required. Where a new beam supports the existing construction, adequate pre-load is to be applied and suitable packs such as driven dry-slate introduced, then pointed up with mortar.
8. All blockwork to be 7.3 N/mm² in class III mortar below DPC in accordance with BS 5628 : Part 3 : 2005 or suitable 7.0 N/mm² foundation quality blocks in class II mortar in accordance with the manufacturer's instructions. All brickwork below DPC to be Engineering Bricks DPC in accordance with BS 5628 : Part 3 : 2005.
9. The project requires the introduction of heavy structural elements such as steel beams or concrete lintels. Although the Construction (Design and Management) Regulation 1994 would not normally apply to this type of construction, the designer still has an obligation to foresee risks and bring to the attention of the builder such risks. In consequence, the builder is to take into consideration the placement of all structural elements, ensuring that the method of lifting and placement is safely carried out. Responsibility for this element lies with the Contractor. As the existing walls need to be propped in order to introduce some of the lintels, this should also be considered in relationship to the risk assessment of the Contractor. Safe working procedures must be adopted. Responsibility for this element lies with the Contractor. Splice details for long-span beams can often be accommodated if required.
10. All construction products should be CE marked in accordance current legislation. This includes all fabricated structural steelwork in accordance with BS EN 1090-1 and BS EN 1090-2. The consequence class is CC2 unless noted otherwise. The service class is SC1 for all buildings, SC2 for all lifting beams, sculptures & fall arrest systems. Production category will be PC1 unless noted otherwise. All site welded items, S355 steelwork & CHS lattice girders will be PC2. As such the execution class for buildings will be EXC2.
11. CLIMATE CHANGE: The Building Research establishment have produced a document CBG 63 "Climate Change: impact on building design and construction". Part of their recommendations are that designers and builders should give consideration to:
 - a. Increased wind loading by providing additional laps and fixings to roof coverings
 - b. Consider foundation depth on shrinkable clays and to avoid future problems, increase the depth above standard requirements if there is a risk. This should be in accordance with the NHBC Standards, Chapter 4.2 Guidance on Building near Trees. If the calculations do not specifically design the depths of the foundations to take into account any local trees, then this should be checked and agreed with the Building Inspector on site.

Party Wall etc. Act 1996

If part of the work is adjacent to the boundary, the adjacent neighbours right to support could be affected; the issues associated with Party Wall Act may need to be considered. This may include providing information to the adjoining owner, giving sufficient notice of works in compliance with the Act. If the following list applies to this project then the Party Wall Act will apply. JMS Engineers can act as Party Wall Surveyors in this instance and should be contacted accordingly.

- Installing a new beam into the shared wall between properties
- Demolishing, building or under-pinning an existing shared wall
- Building a new wall at or on the boundary or junction of two properties
- Damp-proofing all the way through a party wall
- Digging foundations that are within 3m of a Party Wall, where the new foundations are deeper than the existing ones
- Where the new foundations are within 6m and lower than a 45° line from the bottom of the existing foundations



Loading

Roof : Tiles	=	0.65 kN/m ²		<u>Dead</u>	<u>Live</u>
Rafters, felt, insulation etc .	=	0.30 kN/m ²			
		0.95 kN/m ² /Cos 35 =	1.16 kN/m ²		
Plasterboard	=		0.25 kN/m ²		
		TOTAL	=	1.41 kN/m²	
Attic	=		0.25		
Roof snow loading	=	0.6*((60-35)/30)=	0.50 kN/m ²		
		TOTAL	=	0.75 kN/m²	

Roof : Tiles	=	0.65 kN/m ²		<u>Dead</u>	<u>Live</u>
Plasterboard	=	0.25 kN/m ²			
Rafters, felt, insulation etc .	=	0.30 kN/m ²			
		1.20 kN/m ² /Cos 35 =	1.47 kN/m ²		
		TOTAL	=	1.47 kN/m²	
Attic	=		0.25		
Roof snow loading	=	0.6*((60-35)/30)=	0.50 kN/m ²		
		TOTAL	=	0.75 kN/m²	

Roof: Joists & boarding , finishes	=	0.35 kN/m ²			
(flat) Plasterboard	=	0.25 kN/m ²			
		TOTAL	=	0.60 kN/m²	
Imposed	=	0.75 kN/m ²			
		TOTAL	=	0.75 kN/m²	

Floor: PCU's	=	2.50 kN/m ²			
Screed	=	1.80 kN/m ²			
Plasterboard	=	0.25 kN/m ²			
		TOTAL	=	4.55 kN/m²	
Imposed loading	=		1.5		
Partitions loading	=		1.0kN/m ²		
		TOTAL	=	2.5 kN/m²	

Floor: Joists	=	0.15 kN/m ²			
Boarding	=	0.1 kN/m ²			
Plasterboard	=	0.25 kN/m ²			
		TOTAL	=	0.5 kN/m²	
Imposed loading	=		1.5		
Partitions loading	=		0.5kN/m ²		
		TOTAL	=	2.0 kN/m²	

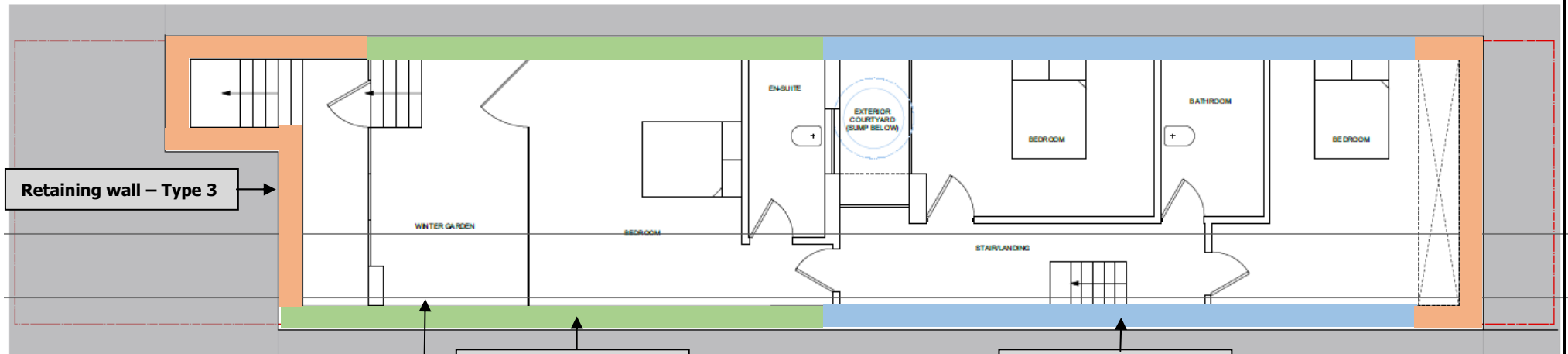
Walls: 2.4m high, 100mm blockwork	=	2.4*1.4	=3.36 kN/m		
Plasterwork both sides	=	2.4*0.25*2	=1.2 kN/m		
		TOTAL	=	4.56 kN/m	
2.4m high, studwork	=	2.4*0.12	=0.29 kN/m		
Plasterwork both sides	=	2.4*0.15*2	=0.72 kN/m		
		TOTAL	=	1.01 kN/m	
2.7m high, 100mm brickwork	=	2.7*2.1	=5.67 kN/m		
Plasterwork both sides	=	2.4*0.25*2	=1.2 kN/m		
		TOTAL	=	6.87 kN/m	
2.7m high, cavity wall blockwork	=	2.7*(2.1+1.4)	=9.45 kN/m		
Plasterwork to one side	=	2.4*0.25	=0.60 kN/m		
		TOTAL	=	10.05 kN/m	
2.7m high, solid brickwork	=	2.7*21*0.215	=12.2 kN/m		
Plasterwork to one side	=	2.4*0.25	=0.60 kN/m		
		TOTAL	=	12.8 kN/m	



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Date : July 2015
Revision : P3

Proposed Basement floor plan foundation layout



Retaining wall – Type 3

Retaining wall – Type 2

Retaining wall – Type 1

Concrete Bridge Above
cantilevered from this wall.

Retaining wall types indicate different
reinforcement schedule depending on the
loadings acting on the retaining walls.

Provide 300mm thick Retaining wall with
300mm thick Base.

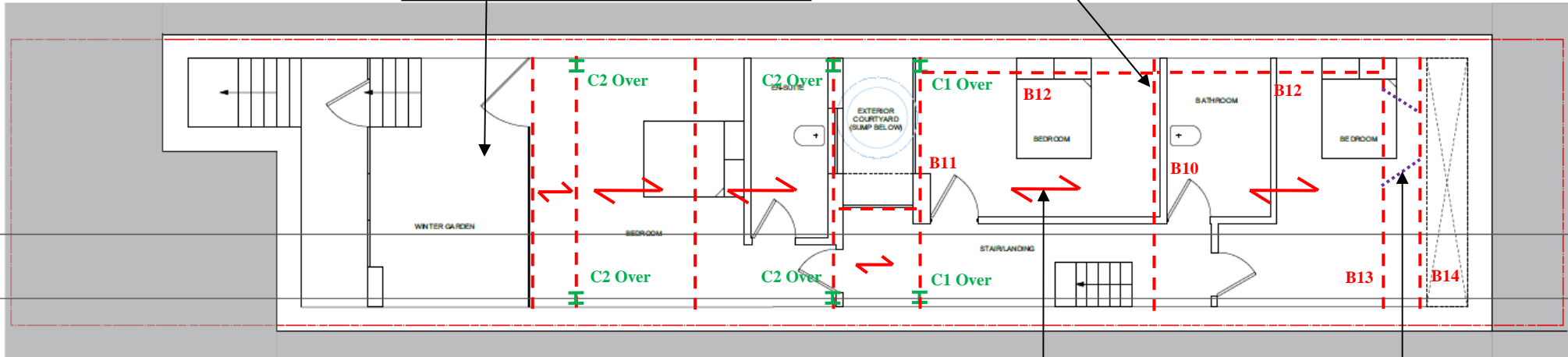
Note:-
Demolition of existing structure to occur in stages
hence, temporary support is not proposed at this stage.



Proposed Basement floor plan showing structure over

Glass Extension:
Glass walls and roof to be designed by Specialist. Allow for area load of 1.5 kN/m² Live Load (Domestic Use)

Steel beams indicated thus.
Refer to the table below for steel section.



Please refer to Pages 33-34 for the Isometric views of the steel frame.

Span of proposed floor joists indicated thus.
Provide 75x225mm deep C24 timber joists at 400mm Centres.

Prestressed Concrete Lintels to support bay window indicated thus.

STEEL SECTIONS

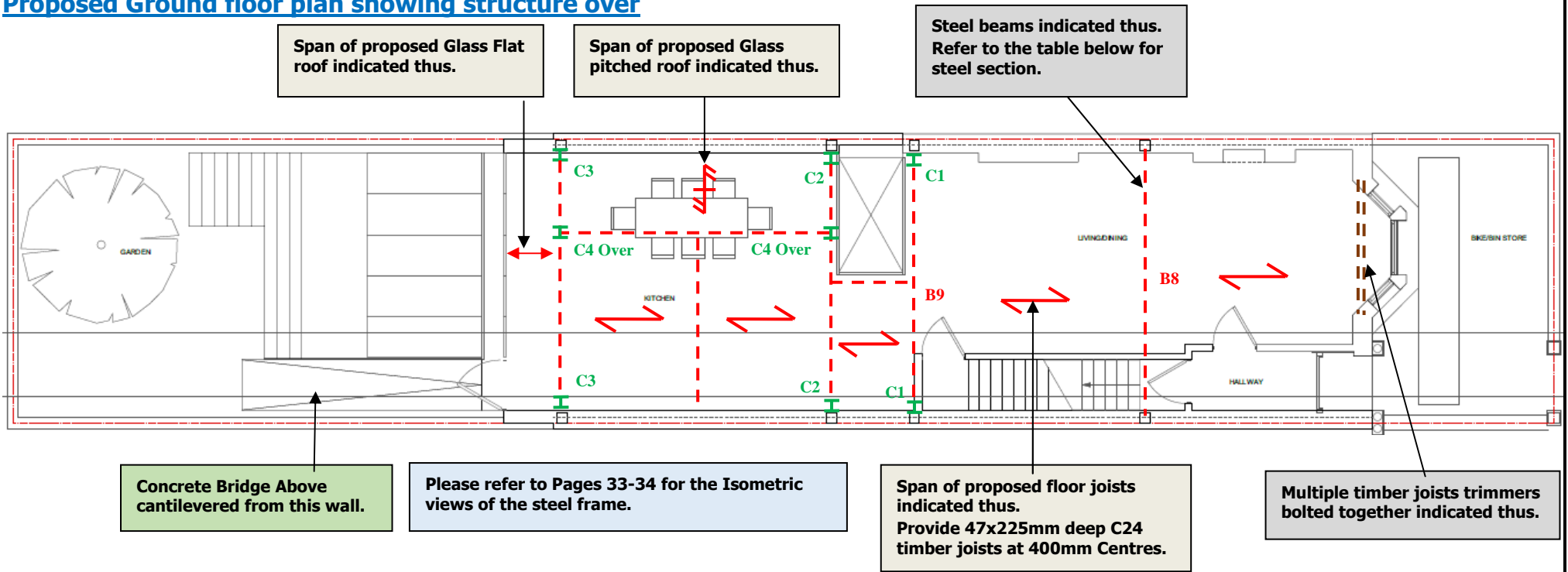
Reference:-	Size:-
B10	203 x 203 UC 60
B11	203 x 203 UC 60
B12	203 x 203 UC 46
B13	254 x 254 UC 89
B14	203 x 203 UC 60



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Proposed Ground floor plan showing structure over



STEEL SECTIONS

Reference:-	Size:-
B8	203 x 203 UC 46
B9	254 x 254 UC 89
C1	203 x 203 UC 46

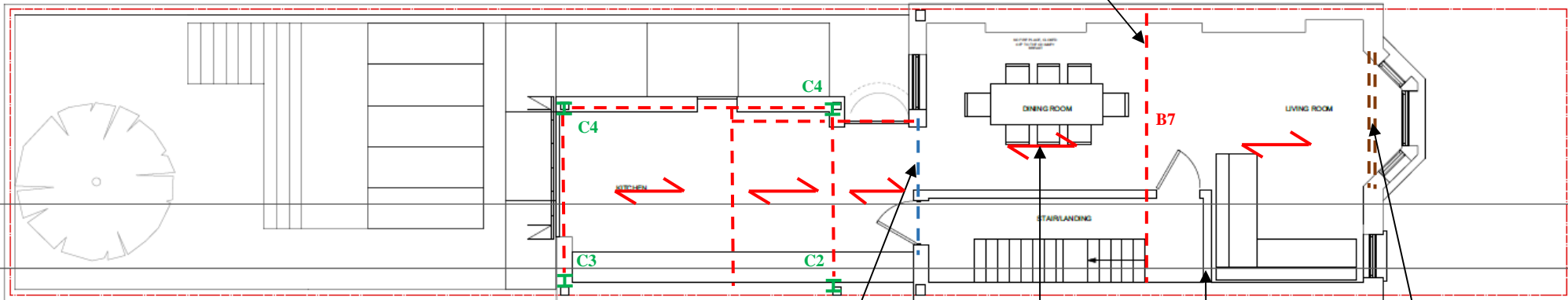


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Proposed First floor plan showing structure over

Steel beam indicated thus. Refer to the table below for steel section.



Please refer to Pages 33-34 for the Isometric view of the steel frame.

Catnic lintel checked indicated thus.

Span of proposed floor joists indicated thus. Provide 47x225mm deep C24 timber joists at 400mm Centres.

Multiple timber joists trimmers bolted together indicated thus.

Internal non load-bearing studwork indicated thus. Provide 38x89mm deep C24 Timber studs at 400mm Centres sheathed using 12mm OSB3 or Tructural Grade Ply.

STEEL SECTIONS	
Reference:-	Size:-
B7	203 x 203 UC 46



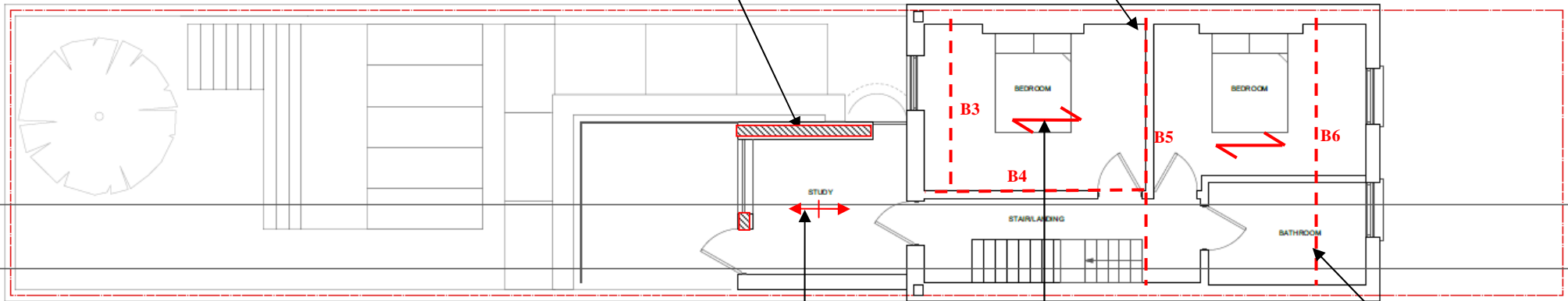
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Proposed Second floor plan showing structure over

External load-bearing studwork indicated thus. Provide 38x140mm deep C24 Timber studs at 400mm Centres sheathed using 12mm OSB3 or Tructural Grade Ply.

Internal non load-bearing studwork indicated thus. Provide 38x89mm deep C24 Timber studs at 400mm Centres sheathed using 12mm OSB3 or Tructural Grade Ply.



Please refer to Pages 33-34 for the Isometric view of the steel frame.

Span of proposed flat roof joists indicated thus. Provide 47x175mm deep C24 timber joists at 400mm Centres

Span of proposed floor joists indicated thus. Provide 47x225mm deep C24 timber joists at 400mm Centres. Floor joists to support internal load-bearing studwork and roof above. Double up joists under dormer cheeks.

Steel beams indicated thus. Refer to the table below for steel sections.

STEEL SECTIONS

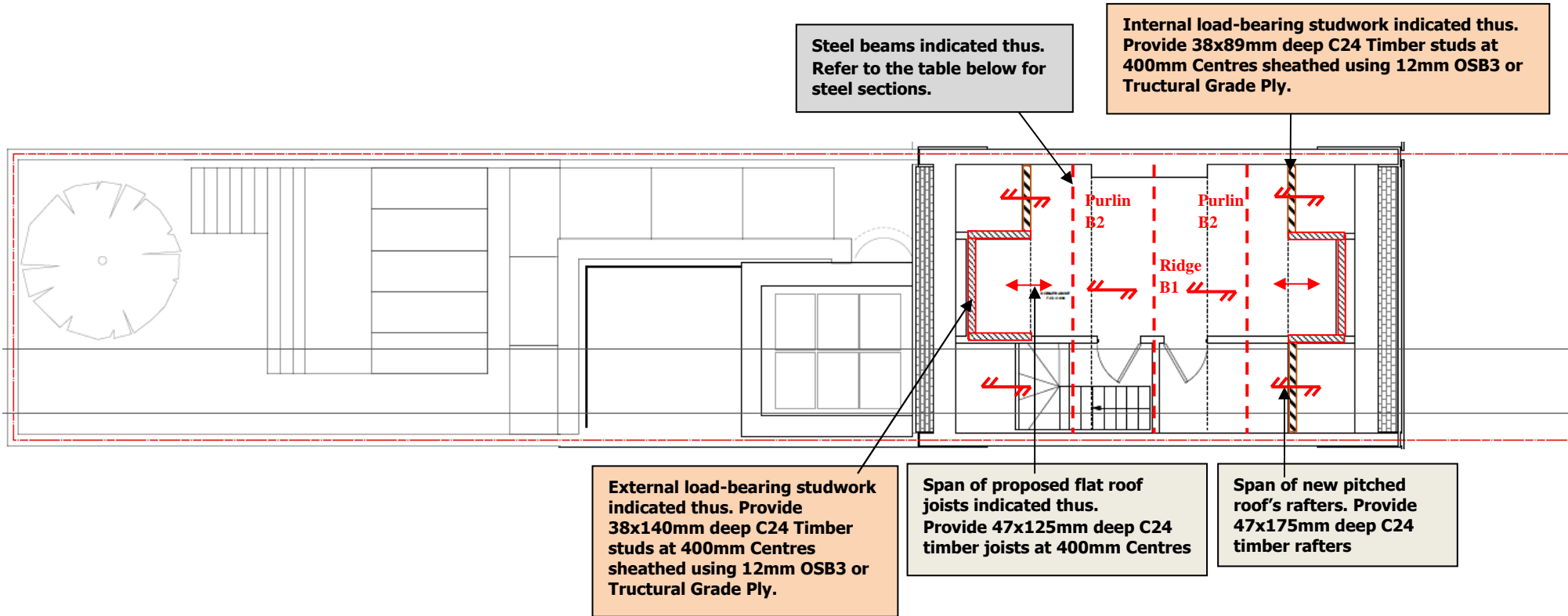
Reference:-	Size:-
B3	203 x 102 UB 23
B4	203 x 102 UB 23
B5	203 x 203 UC 46
B6	203 x 203 UC 46



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Proposed Third floor plan showing structure over



STEEL SECTIONS

Reference:-	Size:-
B1	203 x 102 UB 23
B2	152 x 152 UC 30



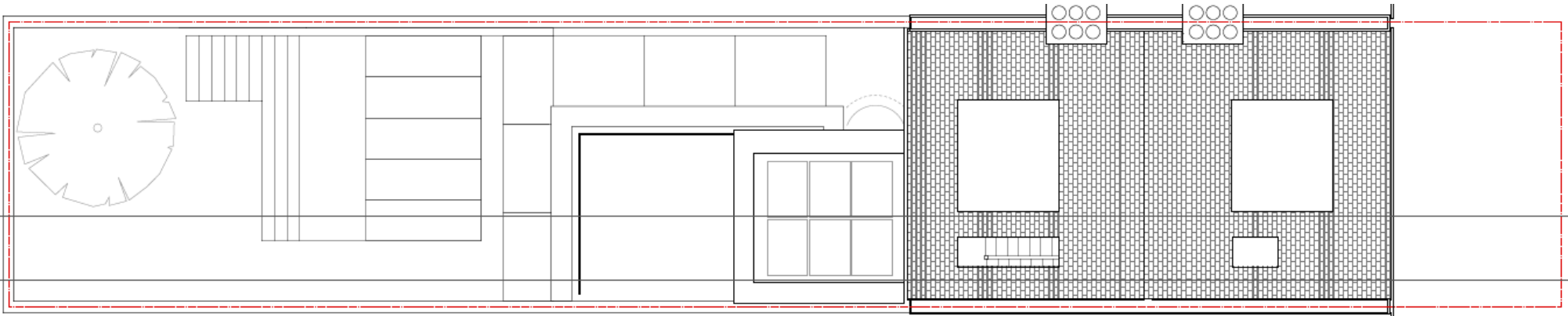
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Date : **July 2015**
Revision : **P3**

Proposed Roof plan





Member Loading (in addition to the area loading as shown at appendix)

Load to B1

Main pitched roof	$4/2 * 1.47 = 2.94$	$4/2 * 0.75 = 1.5$	
Total	2.94 kN/m Dead	1.50 kN/m Live	(Unfactored)

Load to B2 (No dormer)

Main pitched roof	$4.4/2 * 1.47 = 3.24$	$4.4/2 * 0.75 = 1.65$	
Total	3.24 kN/m Dead	1.65 kN/m Live	(Unfactored)

Load to B2 (dormer)

Main pitched roof	$2/2 * 1.47 = 1.47$	$2/2 * 0.75 = 0.75$	
Flat roof	$2.8/2 * 0.6 = 0.84$	$2.8/2 * 0.75 = 1.05$	
Total	2.31 kN/m Dead	1.80 kN/m Live	(Unfactored)

Load to B3, B6

Flat roof	$2.8/2 * 0.6 = 0.84$	$2.8/2 * 0.75 = 1.05$	
External studwork	2.90		
Total	3.74 kN/m Dead	1.05 kN/m Live	(Unfactored)

Load to B4, B5

Internal studwork	1.01		
Total	1.01 kN/m Dead	-	(Unfactored)

Load to B7

Second floor	$8.8/2 * 0.5 = 2.2$	$8.8/2 * 2 = 8.8$	
Internal studwork	1.01		
Total	3.21 kN/m Dead	8.80 kN/m Live	(Unfactored)

Load to B8

First floor	$8.8/2 * 0.5 = 2.2$	$8.3/2 * 2 = 8.8$	
Total	2.20 kN/m Dead	8.8 kN/m Live	(Unfactored)

Load to B9

First floor	$4.5/2 * 0.5 = 1.13$	$4.5/2 * 2 = 4.5$	
Second floor	$4.5/2 * 0.5 = 1.13$	$4.5/2 * 2 = 4.5$	
Third floor	$4.5/2 * 0.5 = 1.13$	$4.5/2 * 2 = 4.5$	
Pitched roof	$2.8/2 * 1.47 = 2.06$	$2.8/2 * 0.75 = 1.05$	
External Wall	30.82		
Total	36.27 kN/m Dead	14.55 kN/m Live	(Unfactored)

Load to B10

Ground floor	$8.8/2 * 2.6 = 11.44$	$8.8/2 * 1.5 = 6.6$	
Total	11.44 kN/m Dead	6.60 kN/m Live	(Unfactored)

Load to B12

Chimney breast	43		
Total	43.00 kN/m Dead	-	(Unfactored)



AXIAL WITH MOMENTS (MEMBER)

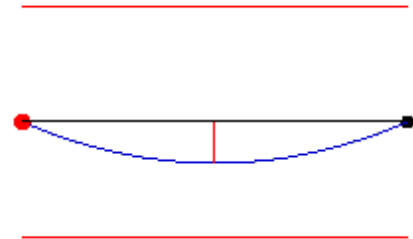
B1

Member 1 (N.1-N.2) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

D1 UDLY -002.940 (kN/m)
L1 UDLY -001.500 (kN/m)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3							
Member No.	Node End1	Node End2	Axial Force (kN)	Shear Force (kN)	Bending Moment (kN.m)	Maximum Moment (kN.m @ m)	Maximum Deflection (mm @ m)
1	1	1	0.087C	17.306	0.000	22.930	10.850
	2	2	0.087C	-17.306	0.000	@ 2.650	@ 2.650

Classification and Effective Area (EN 1993: 2006)

Section (23.07 kg/m) 203x102 UB 23 [S 275]
Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$ 5.47, 31.37, 275, 0.09, 22.93, 0 (Axial: Non-Slender) Class 1
Auto Design Load Cases 1

Local Capacity Check

$V_{y,Ed}/V_{pl,y,Rd}$ 0.003 / 196.416 = 0 Low Shear
 $M_{c,y,Rd} = f_y \cdot W_{pl,y} / \gamma_{M0}$ 275 x 234.1/1 64.378 kN.m
 $N_{pl,Rd} = A_g \cdot f_y / \gamma_{M0}$ 29.39 x 275/1 = 808.225 kN
 $n = N_{Ed}/N_{pl,Rd}$ 0.087 / 808.225 = 0.000 OK
 $W_{pl,N,y} = F_n(W_{pl,y}, A_{vy}, n)$ 234.1, 12.371, 0 234.1 cm³
 $M_{N,y,Rd} = W_{pl,N,y} \cdot f_y / \gamma_{M0}$ 234.1 x 275/1 64.378 kN.m
 $(M_{y,Ed}/M_{N,y,Rd}) + (M_{z,Ed}/M_{N,z,Rd})^2 + (0)^2 = 0.127$ OK

Compression Resistance N.b.Rd

$\lambda_y = \sqrt{A \cdot f_y / N_{cr}}$ $\sqrt{29.39 \times 275 / 1553.76}$ 0.722
 $N_{b,y,Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 29.39 x 0.837 x 275 / 10/1 = 676.770 kN Curve a
 $\lambda_z = \sqrt{A \cdot f_y / N_{crz}}$ $\sqrt{29.39 \times 275 / 121.6}$ 2.576
 $N_{b,z,Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 29.39 x 0.132 x 275 / 10/1 = 106.820 kN Curve b
Let = Kt.Lx 1x5.3 = 5.3
 $\lambda_T = \sqrt{A \cdot f_y / N_{crT}}$ $\sqrt{29.39 \times 275 / 881.43}$ 0.958
 $N_{b,T,Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 29.39 x 0.624 x 275 / 10/1 = 504.438 kN Curve b

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = f_n(M_1, M_2, M_0, \psi, \mu)$ 0.0, 0.0, 22.9, 0.882, 300.000 1.127 Uniform
 $C_{mLT} = 0.95 + 0.05 a_h$ $M_h = 0.02, M_s = 22.93, \psi = 0.882, a_s = 0.001$ 0.95 Table B.3
 $C_{mz} = \text{Max}(0.6 + 0.4\psi, 0.4)$ $M = 0, \psi = 1.000$ 1 Table B.3
 $C_{my} = 0.95 + 0.05 a_h$ $M_h = 0, M_s = 22.93, \psi = 1.000, a_s = 0.000$ 0.95 Table B.3

Lateral Buckling Check M.b.Rd

$l_e = 1.2L + 2D$ 1.2 x 5.3 + 2 x 0.203 = 6.766 m
 $M_{cr} = F_n(C_1, L_e, I_z, I_t, I_w, E)$ 1.127, 6.766, 164.8, 7.019, 0.01537, 210000 24.559 kN.m
 $\lambda_{LT} = \sqrt{W_{pl,y} / M_{cr}}$ $\sqrt{234.1 \times 275 / 24.559}$ 1.619
 $\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$ 1.619, 1.687 0.380 Curve b
 $\chi_{LT,mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$ 0.380, 1.619, 0.942, 1.000 0.380 6.3.2.3
 $M_{b,Rd} = \chi W_{pl,y} \cdot f_y \leq M_{c,y,Rd}$ 0.380 x 234.1 x 275 ≤ 64.378 = 24.440 kN.m

Buckling Resistance

$U_{N,y} = N_{Ed} / (\chi_y \cdot N_{Rk} / \gamma_{M1})$ 0.087 / 676.77 0.000 OK
 $U_{N,z} = N_{Ed} / (\chi_z \cdot N_{Rk} / \gamma_{M1})$ 0.087 / 106.82 0.001 OK
 $U_{M,y} = M_{y,Ed} / (\chi_{LT} \cdot M_{y,Rk} / \gamma_{M1})$ 22.925 / 24.44 0.938 OK
 $U_{M,z} = M_{z,Ed} / (M_{z,Rk} / \gamma_{M1})$ 0 / 13.695 0.000 OK



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$k_{yy}=C_{my}\{1+(\lambda_y-0.2)U_{N,y}\}$		0.950	
$k_{zz}=C_{mz}\{1+1.4U_{N,z}\}$		1.001	
$k_{yz}=0.6 k_{zz}$		0.601	
$k_{zy}=1-\{0.1\lambda_z/(C_{mLT}-0.25)\}U_{N,z}$		1.000	
$U_{Ny}+k_{yy}\cdot U_{M,y}+k_{yz}\cdot U_{M,z}$	0.000+0.950x0.938+0.601x0.000	0.891	OK
$U_{Nz}+k_{zy}\cdot U_{M,y}+k_{zz}\cdot U_{M,z}$	0.001+1.000x0.938+1.001x0.000	0.939	OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams)	$\delta \leq 5300/360 = 14.7$ mm Live (Case 2)	3.48 mm	OK
	$\delta \leq 5300/250 = 21.2$ mm D+L (Case 3)	10.85 mm	OK

Provide Section (23.07 kg/m)

203x102 UB 23 [S 275]

AXIAL WITH MOMENTS (MEMBER)

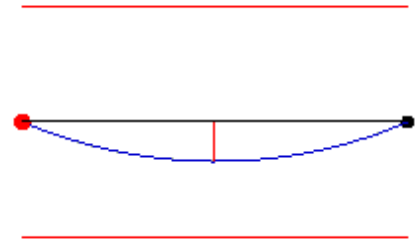
B2

Members 1-3 (N.1-N.2) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

Part 1	
D1 UDLY	-003.240 (kN/m)
L1 UDLY	-001.650 (kN/m)
Part 2	
D1 UDLY	-002.310 (kN/m)
L1 UDLY	-001.800 (kN/m)
Part 3	
D1 UDLY	-003.240 (kN/m)
L1 UDLY	-001.650 (kN/m)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3						
Mem ber No.	Node End1 End2	Axial Force (kN)	Shear Force (kN)	Bending Moment (kN.m)	Maximum Moment (kN.m @ m)	Maximum Deflection (mm @ m)
	1	0.025C	18.061	0.000	23.194	13.243
	4	0.149C	-18.224	0.000	@ 2.660	@ 2.660

Classification and Effective Area (EN 1993: 2006)

Section (30.03 kg/m)	152x152 UC 30 [S 275]		
Class = $f_n(b/T, d/t, f_y, N, M_y, M_z)$	8.13, 19.02, 275, 0.15, 23.19, 0	(Axial: Non-Slender)	Class 1
Auto Design Load Cases	1		

Local Capacity Check

$V_{y,Ed}/V_{pl,y,Rd}$	0.124 / 183.454 =	0.001	Low Shear
$M_{c,y,Rd} = f_y \cdot W_{pl,y} / \gamma_{M0}$	275 x 247.7/1	68.118 kN.m	
$N_{pl,Rd} = A_g \cdot f_y / \gamma_{M0}$	38.26 x 275/1 =	1052.15 kN	
$n = N_{Ed}/N_{pl,Rd}$	0.025 / 1052.15 =	0.000	OK
$W_{pl,N,y} = f_n(W_{pl,y}, A_{vy}, n)$	247.7, 11.555, 0	247.7 cm ³	
$M_{N,y,Rd} = W_{pl,N,y} \cdot f_y / \gamma_{M0}$	247.7 x 275/1	68.118 kN.m	
$(M_{y,Ed}/M_{N,y,Rd}) + (M_{z,Ed}/M_{N,z,Rd})$	$(23.192/68.118)^2 + (0)^1 =$	0.116	OK

Compression Resistance N.b.Rd

$\lambda_y = \sqrt{A \cdot f_y / N_{cr}}$	$\sqrt{38.26 \times 275 / 1290.42}$	0.903	
$N_{b,y,Rd} = \text{Area} \cdot \chi \cdot f_y / \gamma_{M1}$	$38.26 \times 0.659 \times 275 / 10/1 =$	693.565 kN	Curve b
$\lambda_z = \sqrt{A \cdot f_y / N_{crz}}$	$\sqrt{38.26 \times 275 / 414.23}$	1.594	
$N_{b,z,Rd} = \text{Area} \cdot \chi \cdot f_y / \gamma_{M1}$	$38.26 \times 0.286 \times 275 / 10/1 =$	300.842 kN	Curve c
Let = $K_t \cdot L_x$	$1 \times 5.3 =$	5.3	
$\lambda_T = \sqrt{A \cdot f_y / N_{crT}}$	$\sqrt{38.26 \times 275 / 1783.14}$	0.768	
$N_{b,T,Rd} = \text{Area} \cdot \chi \cdot f_y / \gamma_{M1}$	$38.26 \times 0.682 \times 275 / 10/1 =$	717.731 kN	Curve c

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = f_n(M_1, M_2, M_0, \psi, \mu)$	0.0, 0.0, 23.2, 0.941, 300.000	1.127	Uniform
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$C_{mLT}=0.95+0.05a_h$	$M_h = 0.02, M_s = 23.19, \psi = 0.941, a_s = 0.001$	0.95	Table B.3
$C_{mz} = \text{Max}(0.6+0.4\psi, 0.4)$	$M = 0, \psi = 1.000$	1	Table B.3
$C_{my} = 0.95+0.05a_h$	$M_h = 0, M_s = 23.19, \psi = 1.000, a_s = 0.000$	0.95	Table B.3

Lateral Buckling Check M.b.Rd

$l_e = 1.2L+2D$	$1.2 \times 5.3 + 2 \times 0.158 =$	6.675 m	
$M_{cr} = \text{Fn}(C_1, l_e, I_z, I_y, I_w, E)$	$1.127, 6.675, 561.4, 10.52, 0.03075, 210000$	57.376 kN.m	
$\lambda_{LT} = \sqrt{W_{pl,y}/M_{cr}}$	$\sqrt{247.7 \times 275 / 57.376}$	1.090	
$\chi_{LT} = \text{Fn}(\lambda_{LT}, \lambda_{LT5950})$	$1.090, 1.049$	0.645	Curve b
$\chi_{LT,mod} = \text{Fn}(\chi_{LT}, \lambda_{LT}, k_c, f)$	$0.645, 1.090, 0.942, 0.976$	0.661	6.3.2.3
$M_{b,Rd} = \chi W_{pl,y} f_y \leq M_{c,y,Rd}$	$0.661 \times 247.7 \times 275 \leq 68.118 =$	45.015 kN.m	

Buckling Resistance

$U_{N,y} = N_{Ed}/(\chi_y N_{Rk}/\gamma_{M1})$	$0.149 / 693.565$	0.000	OK
$U_{N,z} = N_{Ed}/(\chi_z N_{Rk}/\gamma_{M1})$	$0.149 / 300.842$	0.000	OK
$U_{M,y} = M_{y,Ed}/(\chi_{LT} M_{y,Rk}/\gamma_{M1})$	$23.192 / 45.015$	0.515	OK
$U_{M,z} = M_{z,Ed}/(M_{z,Rk}/\gamma_{M1})$	$0 / 30.69$	0.000	OK
$k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N,y}\}$		0.950	
$k_{zz} = C_{mz} \{1 + 1.4 U_{N,z}\}$		1.001	
$k_{yz} = 0.6 k_{zz}$		0.600	
$k_{zy} = 1 - \{0.1 \lambda_z / (C_{mLT} - 0.25)\} U_{N,z}$		1.000	
$U_{Ny} + k_{yy} U_{My} + k_{yz} U_{Mz}$	$0.000 + 0.950 \times 0.515 + 0.600 \times 0.000$	0.490	OK
$U_{Nz} + k_{zy} U_{My} + k_{zz} U_{Mz}$	$0.000 + 1.000 \times 0.515 + 1.001 \times 0.000$	0.516	OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams)	$\delta \leq 5300/360 = 14.7$ mm Live (Case 2)	4.86 mm	OK
	$\delta \leq 5300/250 = 21.2$ mm D+L (Case 3)	13.24 mm	OK

Provide Section (30.03 kg/m)

152x152 UC 30 [S 275]

AXIAL WITH MOMENTS (MEMBER)

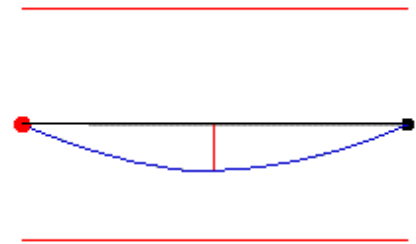
B3

Members 13-18 (N.2-N.3) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

Part 1			
D1 D	078.500	(kN/m ³)	
Part 2			
D1 D	078.500	(kN/m ³)	
Part 3			
D1 D	078.500	(kN/m ³)	
Part 4			
D1 D	078.500	(kN/m ³)	
Part 5			
D1 D	078.500	(kN/m ³)	
D1 UDLY	-003.740	(kN/m)	
L1 UDLY	-001.050	(kN/m)	
Part 6			
D1 D	078.500	(kN/m ³)	
D1 UDLY	-003.740	(kN/m)	
L1 UDLY	-001.050	(kN/m)	



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3										
Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)
				y-y	z-z	y-y	z-z	y-y	z-z	
	2	0.34T	0.00	26.27	0.00	0.00	0.00	25.36	0.00	5.10
	14	0.11T	0.00	-27.20	0.00	0.00	0.00	@ 1.672	@ 0.000	@ 1.740



Classification and Effective Area (EN 1993: 2006)

Section (23.07 kg/m)	203x102 UB 23 [S 275]		
Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$	5.47, 31.37, 275, 0, 25.36, 0	(Axial: Non-Slender)	Class 1
Auto Design Load Cases	1		

Local Capacity Check

$V_{y,Ed}/V_{pl,y,Rd}$	$0.459 / 196.416 =$	0.002	Low Shear
$M_{c,y,Rd} = f_y \cdot W_{pl,y} / \gamma_{M0}$	$275 \times 234.1 / 1$	64.378 kN.m	
$N_{pl,Rd} = A_g \cdot f_y / \gamma_{M0}$	$29.39 \times 275 / 1$	808.225 kN	
$n = N_{Ed} / N_{pl,Rd}$	$-0.343 / 808.225 =$	0.000	OK
$W_{pl,N,y} = F_n(W_{pl,y}, A_{vy}, n)$	$234.1, 12.371, 0$	234.1 cm ³	
$M_{N,y,Rd} = W_{pl,N,y} \cdot f_y / \gamma_{M0}$	$234.1 \times 275 / 1$	64.378 kN.m	
$W_{pl,N,z} = F_n(W_{pl,z}, A_{vz}, n)$	$49.8, 18.935, 0$	49.8 cm ³	
$M_{N,z,Rd} = W_{pl,N,z} \cdot f_y / \gamma_{M0}$	$49.8 \times 275 / 1$	13.695 kN.m	
$(M_{y,Ed} / M_{N,y,Rd}) + (M_{z,Ed} / M_{N,z,Rd})^1 =$	$(25.349 / 64.378)^2 + (0.001 / 13.695)^1 =$	0.155	OK

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = f_n(M_1, M_2, M_0, \psi, \mu)$	0.0, 0.0, 25.3, 0.862, 300.000	1.127	Uniform
$C_{mLT} = 0.95 + 0.05 \alpha_h$	$M_h = 0.03, M_s = 25.35, \psi = 0.862, \alpha_s = 0.001$	0.95	Table B.3
$C_{mz} = \text{Max}(0.6 + 0.4\psi, 0.4)$	$M = 0, \psi = 0.000$	0.6	Table B.3
$C_{my} = 0.95 + 0.05 \alpha_h$	$M_h = 0, M_s = 25.35, \psi = 0.000, \alpha_s = 0.000$	0.95	Table B.3

Lateral Buckling Check M.b.Rd

$l_e = 1.2L + 2D$	$1.2 \times 3.5 + 2 \times 0.203 =$	4.606 m	
$M_{cr} = F_n(C_1, L_e, I_z, I_y, I_w, E)$	1.127, 4.606, 164.8, 7.019, 0.01537, 210000	38.290 kN.m	
$\lambda_{LT} = \sqrt{W_{pl,y} / M_{cr}}$	$\sqrt{234.1 \times 275 / 38.29}$	1.297	
$\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT, S950})$	1.297, 1.350	0.525	Curve b
$\chi_{LT, mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$	0.525, 1.297, 0.942, 0.985	0.533	6.3.2.3
$M_{b,Rd} = \chi W_{pl,y} \cdot f_y \leq M_{c,y,Rd}$	$0.533 \times 234.1 \times 275 \leq 64.378 =$	34.328 kN.m	

Buckling Resistance

$U_{N,y} = N_{Ed} / (\chi_y \cdot N_{Rk} / \gamma_{M1})$	$0 / 752.78$	0.000	OK
$U_{N,z} = N_{Ed} / (\chi_z \cdot N_{Rk} / \gamma_{M1})$	$0 / 224.486$	0.000	OK
$U_{M,y} = M_{y,Ed} / (\chi_{LT} \cdot M_{y,Rk} / \gamma_{M1})$	$25.349 / 34.328$	0.738	OK
$U_{M,z} = M_{z,Ed} / (M_{z,Rk} / \gamma_{M1})$	$0.001 / 13.695$	0.000	OK
$k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N,y}\}$		0.950	
$k_{zz} = C_{mz} \{1 + 1.4 U_{N,z}\}$		0.600	
$k_{yz} = 0.6 k_{zz}$		0.360	
$k_{zy} = 1 - \{0.1 \lambda_z / (C_{mLT} - 0.25)\} U_{N,z}$		1.000	
$U_{Ny} + k_{yy} \cdot U_{M,y} + k_{yz} \cdot U_{M,z}$	$0.000 + 0.950 \times 0.738 + 0.360 \times 0.000$	0.702	OK
$U_{Nz} + k_{zy} \cdot U_{M,y} + k_{zz} \cdot U_{M,z}$	$0.000 + 1.000 \times 0.738 + 0.600 \times 0.000$	0.738	OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams)	$\delta \leq 3500/360 = 9.7$ mm Live (Case 2)	2.27 mm	OK
	$\delta \leq 3500/250 = 14$ mm D+L (Case 3)	5.1 mm	OK

Provide Section (23.07 kg/m)

203x102 UB 23 [S 275]



AXIAL WITH MOMENTS (MEMBER)

B4

Members 34-36 (N.6-N.14) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

Part 1

D1 UDLY -000.450 (kN/m)
L1 UDLY -001.350 (kN/m)
D1 D 078.500 (kN/m³)

Part 2

D1 UDLY -000.450 (kN/m)
L1 UDLY -001.350 (kN/m)
D1 D 078.500 (kN/m³)
D1 UDLY -002.900 (kN/m)

Part 3

D1 UDLY -000.450 (kN/m)
L1 UDLY -001.350 (kN/m)
D1 D 078.500 (kN/m³)
D1 UDLY -001.010 (kN/m)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3										
Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)
				y-y	z-z	y-y	z-z	y-y	z-z	
	6	0.00C	0.00	36.12	0.00	0.00	0.00	18.60	0.05	6.32
	23	0.00T	0.00	-12.72	-0.02	-0.18	0.00	@ 1.530	@ 1.500	@ 2.130

Classification and Effective Area (EN 1993: 2006)

Section (23.07 kg/m) 203x102 UB 23 [S 275]
Class = $f_n(b/T, d/t, f_y, N, M_y, M_z)$ 5.47, 31.37, 275, 0, 18.6, 0.05 (Axial: Non-Slender) Class 1
Auto Design Load Cases 1

Local Capacity Check

$V_{y,Ed}/V_{pl,y,Rd}$ 0.04 / 196.416 = 0 Low Shear
 $M_{C,y,Rd} = f_y \cdot W_{pl,y} / \gamma_{M0}$ 275 x 234.1/1 64.378 kN.m
 $V_{z,Ed}/V_{pl,z,Rd}$ 0.015 / 300.63 = 0 Low Shear
 $M_{C,z,Rd} = f_y \cdot W_{pl,z} / \gamma_{M0}$ 275 x 49.8/1 13.695 kN.m
 $N_{pl,Rd} = A_g \cdot f_y / \gamma_{M0}$ 29.39x275/1 808.225 kN
 $n = N_{Ed}/N_{pl,Rd}$ -0.001 / 808.225 = 0.000 OK
 $W_{pl,N,y} = F_n(W_{pl,y}, A_{vy}, n)$ 234.1, 12.371, 0 234.1 cm³
 $M_{N,y,Rd} = W_{pl,N,y} \cdot f_y / \gamma_{M0}$ 234.1 x 275/1 64.378 kN.m
 $W_{pl,N,z} = F_n(W_{pl,z}, A_{vz}, n)$ 49.8, 18.935, 0 49.8 cm³
 $M_{N,z,Rd} = W_{pl,N,z} \cdot f_y / \gamma_{M0}$ 49.8 x 275/1 13.695 kN.m
 $(M_{y,Ed}/M_{N,y,Rd}) + (M_{z,Ed}/M_{N,z,Rd})$ $(18.596/64.378)^2 + (0.046/13.695)^2 = 0.087$ OK

Compression Resistance N.b.Rd

$\lambda_y = \sqrt{A \cdot f_y / N_{cr}}$ $\sqrt{29.39 \times 275 / 2155.32}$ 0.613
 $N_{b,y,Rd} = A_{eff} \cdot \chi \cdot f_y / \gamma_{M1}$ 29.39x0.885x275/10/1 = 715.373 kN Curve a
 $\lambda_z = \sqrt{A \cdot f_y / N_{crz}}$ $\sqrt{29.39 \times 275 / 168.68}$ 2.187
 $N_{b,z,Rd} = A_{eff} \cdot \chi \cdot f_y / \gamma_{M1}$ 29.39x0.178x275/10/1 = 144.177 kN Curve b
 $l_{et} = K_t \cdot L_x$ 1x4.5 = 4.5
 $\lambda_T = \sqrt{A \cdot f_y / N_{crT}}$ $\sqrt{29.39 \times 275 / 938.31}$ 0.928
 $N_{b,T,Rd} = A_{eff} \cdot \chi \cdot f_y / \gamma_{M1}$ 29.39x0.643x275/10/1 = 519.767 kN Curve b

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = f_n(M_1, M_2, M_0, \psi, \mu)$ 0.0, -0.2, 18.0, -0.205, -108.717 1.147 Uniform
 $C_{mLT} = 0.95 + 0.05 a_h (1 + 2\psi)$ $M_h = -0.17, M_s = 18.03, \psi = -0.205, a_s = -0.009$ 0.95 Table B.3
 $C_{mz} = \text{Max}(0.6 + 0.4\psi, 0.4)$ $M = 0, \psi = 0.000$ 0.6 Table B.3
 $C_{my} = 0.95 + 0.05 a_h$ $M_h = -0.18, M_s = 17.53, \psi = 0.000, a_s = -0.010$ 0.949 Table B.3

Lateral Buckling Check M.b.Rd

$l_e = 1.2L + 2D$ 1.2 x 4.5 + 2 x 0.203 = 5.806 m



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$M_{cr} = F_n(C_1, L_e, I_z, I_y, I_w, E)$	1.147, 5.806, 164.8, 7.019, 0.01537, 210000	29.689 kN.m	
$\lambda_{LT} = \sqrt{W_{fy}/M_{cr}}$	$\sqrt{234.1 \times 275 / 29.689}$	1.473	
$\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$	1.473, 1.547	0.439	Curve b
$\chi_{LT.mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$	0.439, 1.473, 0.934, 0.997	0.441	6.3.2.3
$M_{b,Rd} = \chi W_{pl,y} f_y \leq M_{c,y,Rd}$	$0.441 \times 234.1 \times 275 \leq 64.378 =$	28.368 kN.m	

Buckling Resistance

$U_{N,y} = N_{Ed}/(\chi_y \cdot N_{Rk}/\gamma_{M1})$	0 / 715.373	0.000	OK
$U_{N,z} = N_{Ed}/(\chi_z \cdot N_{Rk}/\gamma_{M1})$	0 / 144.177	0.000	OK
$U_{M,y} = M_{y,Ed}/(\chi_{LT} \cdot M_{y,Rk}/\gamma_{M1})$	18.596 / 28.368	0.656	OK
$U_{M,z} = M_{z,Ed}/(M_{z,Rk}/\gamma_{M1})$	0.046 / 13.695	0.003	OK
$k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N,y}\}$		0.949	
$k_{zz} = C_{mz} \{1 + 1.4 U_{N,z}\}$		0.600	
$k_{yz} = 0.6 k_{zz}$		0.360	
$k_{zy} = 1 - \{0.1 \lambda_z / (C_{mLT} - 0.25)\} U_{N,z}$		1.000	
$U_{Ny} + k_{yy} \cdot U_{M,y} + k_{yz} \cdot U_{M,z}$	$0.000 + 0.949 \times 0.656 + 0.360 \times 0.003$	0.624	OK
$U_{Nz} + k_{zy} \cdot U_{M,y} + k_{zz} \cdot U_{M,z}$	$0.000 + 1.000 \times 0.656 + 0.600 \times 0.003$	0.658	OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams)	$\delta \leq 4500/360 = 12.5$ mm Live (Case 2)	2.65 mm	OK
	$\delta \leq 4500/250 = 18$ mm D+L (Case 3)	6.32 mm	OK

Provide Section (23.07 kg/m)

203x102 UB 23 [S 275]



AXIAL WITH MOMENTS (MEMBER)

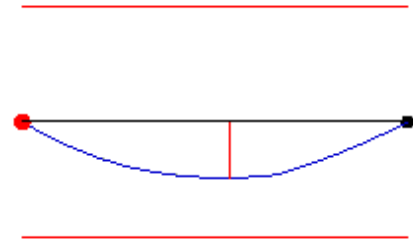
B5

Members 1-12 (N.15-N.16) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

Part 1		
D1 D	078.500	(kN/m ³)
Part 2		
D1 D	078.500	(kN/m ³)
Part 3		
D1 D	078.500	(kN/m ³)
Part 4		
D1 D	078.500	(kN/m ³)
Part 5		
D1 D	078.500	(kN/m ³)
Part 6		
D1 D	078.500	(kN/m ³)
Part 7		
D1 D	078.500	(kN/m ³)
Part 8		
D1 D	078.500	(kN/m ³)
D1 UDLY	-001.010	(kN/m)
Part 9		
D1 D	078.500	(kN/m ³)
D1 UDLY	-001.010	(kN/m)
Part 10		
D1 D	078.500	(kN/m ³)
D1 UDLY	-001.010	(kN/m)
Part 11		
D1 D	078.500	(kN/m ³)
D1 UDLY	-001.010	(kN/m)
Part 12		
D1 D	078.500	(kN/m ³)
D1 UDLY	-001.010	(kN/m)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3										
Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)
				y-y	z-z	y-y	z-z	y-y	z-z	
	15	0.01C	0.00	49.37	0.00	0.00	0.00	66.08	0.00	14.03
	33	0.43C	0.00	-43.69	0.00	0.00	0.00	@ 2.775	@ 0.000	@ 2.658

Classification and Effective Area (EN 1993: 2006)

Section (46.1 kg/m)	203x203 UC 46 [S 275]		
Class = Fn(b/T,d/t,f _y ,N,M _y ,M _z)	9.25, 22.33, 275, 0.43, 66.08, 0	(Axial: Non-Slender)	Class 1
Auto Design Load Cases	1		

Local Capacity Check

$V_{y,Ed}/V_{pl,y,Rd}$	$1.049 / 269.498 =$	0.004	Low Shear
$M_{c,y,Rd} = f_y \cdot W_{pl,y} / \gamma_{M0}$	$275 \times 497.4 / 1$	136.785 kN.m	
$N_{pl,Rd} = A_g \cdot f_y / \gamma_{M0}$	$58.73 \times 275 / 1 =$	1615.075 kN	
$n = N_{Ed} / N_{pl,Rd}$	$0.007 / 1615.075 =$	0.000	OK
$W_{pl,N,y} = F_n(W_{pl,y}, A_{vy}, n)$	$497.4, 16.974, 0$	497.4 cm ³	
$M_{N,y,Rd} = W_{pl,N,y} \cdot f_y / \gamma_{M0}$	$497.4 \times 275 / 1$	136.785 kN.m	
$(M_{y,Ed} / M_{N,y,Rd}) + (M_{z,Ed} / M_{N,z,Rd})$	$(66.039 / 136.785)^2 + (0)^1 =$	0.233	OK

Compression Resistance N.b.Rd

$\lambda_y = \sqrt{A \cdot f_y / N_{cr}}$	$\sqrt{58.73 \times 275 / 3372.49}$	0.692	
$N_{b,y,Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$	$58.73 \times 0.788 \times 275 / 10 / 1 =$	1272.861 kN	Curve b
$\lambda_z = \sqrt{A \cdot f_y / N_{crz}}$	$\sqrt{58.73 \times 275 / 1144.48}$	1.188	
$N_{b,z,Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$	$58.73 \times 0.44 \times 275 / 10 / 1 =$	710.090 kN	Curve c
Let = Kt.Lx	$1 \times 5.3 =$	5.3	



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$\lambda_T = \sqrt{A_f y / N_{crT}}$ $\sqrt{58.73 \times 275 / 2728.77}$ 0.769
 $N_{b,T,Rd} = \text{Area} \cdot \chi_f y / \gamma_{M1}$ $58.73 \times 0.681 \times 275 / 10 / 1 =$ 1100.539 kN Curve c

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = \text{fn}(M_1, M_2, M_0, \psi, \mu)$ 0.0, 0.0, 65.9, 0.854, 300.000 1.127 Uniform
 $C_{mLT} = 0.95 + 0.05 a_h$ $M_h = 0.05, M_s = 65.95, \psi = 0.854, a_s = 0.001$ 0.95 Table B.3
 $C_{mz} = \text{Max}(0.6 + 0.4 \psi, 0.4)$ $M = 0, \psi = 1.000$ 1 Table B.3
 $C_{my} = 0.95 + 0.05 a_h$ $M_h = 0, M_s = 65.96, \psi = 1.000, a_s = 0.000$ 0.95 Table B.3

Lateral Buckling Check M.b.Rd

$l_e = 1.2L + 2D$ $1.2 \times 5.3 + 2 \times 0.203 =$ 6.766 m
 $M_{cr} = \text{Fn}(C_1, l_e, I_z, I_w, E)$ 1.127, 6.766, 1551, 22.15, 0.1429, 210000 147.402 kN.m
 $\lambda_{LT} = \sqrt{W_f y / M_{cr}}$ $\sqrt{497.4 \times 275 / 147.402}$ 0.963
 $\chi_{LT} = \text{Fn}(\lambda_{LT}, \lambda_{LT5950})$ 0.963, 0.922 0.722 Curve b
 $\chi_{LT,mod} = \text{Fn}(\chi_{LT}, \lambda_{LT}, k_{cr}, f)$ 0.722, 0.963, 0.942, 0.973 0.742 6.3.2.3
 $M_{b,Rd} = \chi W_{pl,y} f_y \leq M_{c,y,Rd}$ $0.742 \times 497.4 \times 275 \leq 136.785 =$ 101.552 kN.m

Buckling Resistance

$U_{N,y} = N_{Ed} / (\chi_y \cdot N_{Rk} / \gamma_{M1})$ 0.426 / 1272.861 0.000 OK
 $U_{N,z} = N_{Ed} / (\chi_z \cdot N_{Rk} / \gamma_{M1})$ 0.426 / 710.09 0.001 OK
 $U_{M,y} = M_{y,Ed} / (\chi_{LT} \cdot M_{y,Rk} / \gamma_{M1})$ 66.039 / 101.552 0.650 OK
 $U_{M,z} = M_{z,Ed} / (M_{z,Rk} / \gamma_{M1})$ 0 / 63.498 0.000 OK
 $k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N,y}\}$ 0.950
 $k_{zz} = C_{mz} \{1 + 1.4 U_{N,z}\}$ 1.001
 $k_{yz} = 0.6 k_{zz}$ 0.601
 $k_{zy} = 1 - \{0.1 \lambda_z / (C_{mLT} - 0.25)\} U_{N,z}$ 1.000
 $U_{Ny} + k_{yy} \cdot U_{M,y} + k_{yz} \cdot U_{M,z}$ $0.000 + 0.950 \times 0.650 + 0.601 \times 0.000$ 0.618 OK
 $U_{Nz} + k_{zy} \cdot U_{M,y} + k_{zz} \cdot U_{M,z}$ $0.001 + 1.000 \times 0.650 + 1.001 \times 0.000$ 0.651 OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams) $\delta \leq 5300/360 = 14.7$ mm Live (Case 2) 8.33 mm OK
 $\delta \leq 5300/250 = 21.2$ mm D+L (Case 3) 14.03 mm OK

Provide Section (46.1 kg/m)

203x203 UC 46 [S 275]



AXIAL WITH MOMENTS (MEMBER)

B6

Members 21-32 (N.34-N.36) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

Part 1

D1 UDLY -000.180 (kN/m)
L1 UDLY -000.113 (kN/m)
D1 D 078.500 (kN/m³)

Part 2

D1 UDLY -000.180 (kN/m)
L1 UDLY -000.113 (kN/m)
D1 D 078.500 (kN/m³)

Part 3

D1 UDLY -000.180 (kN/m)
L1 UDLY -000.113 (kN/m)
D1 D 078.500 (kN/m³)

Part 4

D1 UDLY -000.180 (kN/m)
L1 UDLY -000.113 (kN/m)
D1 D 078.500 (kN/m³)

Part 5

D1 UDLY -000.180 (kN/m)
L1 UDLY -000.113 (kN/m)
D1 D 078.500 (kN/m³)
D1 UDLY -003.740 (kN/m)
L1 UDLY -001.050 (kN/m)

Part 6

D1 UDLY -000.180 (kN/m)
L1 UDLY -000.113 (kN/m)
D1 D 078.500 (kN/m³)
D1 UDLY -003.740 (kN/m)
L1 UDLY -001.050 (kN/m)

Part 7

D1 UDLY -000.180 (kN/m)
L1 UDLY -000.113 (kN/m)
D1 D 078.500 (kN/m³)
D1 UDLY -003.740 (kN/m)
L1 UDLY -001.050 (kN/m)

Part 8

D1 UDLY -000.180 (kN/m)
L1 UDLY -000.113 (kN/m)
D1 D 078.500 (kN/m³)

Part 9

D1 UDLY -000.180 (kN/m)
L1 UDLY -000.113 (kN/m)
D1 D 078.500 (kN/m³)

Part 10

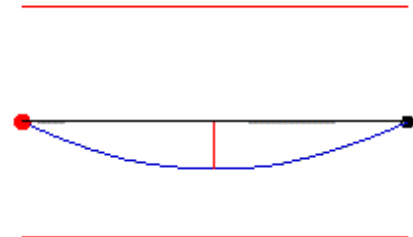
D1 UDLY -000.180 (kN/m)
L1 UDLY -000.113 (kN/m)
D1 D 078.500 (kN/m³)

Part 11

D1 UDLY -000.180 (kN/m)
L1 UDLY -000.113 (kN/m)
D1 D 078.500 (kN/m³)

Part 12

D1 UDLY -000.180 (kN/m)
L1 UDLY -000.113 (kN/m)
D1 D 078.500 (kN/m³)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3



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Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)
				y-y	z-z	y-y	z-z	y-y	z-z	
	34	0.00C	0.00	39.65	0.00	0.00	0.00	55.55	0.00	12.00
	51	0.41C	0.00	-38.71	0.00	0.00	0.00	@ 2.658	@ 0.000	@ 2.645

Classification and Effective Area (EN 1993: 2006)

Section (46.1 kg/m) 203x203 UC 46 [S 275]
Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$ 9.25, 22.33, 275, 0.41, 55.54, 0 (Axial: Non-Slender) Class 1
Auto Design Load Cases 1

Local Capacity Check

$V_{y.Ed}/V_{pl.y.Rd}$ 0.193 / 269.498 = 0.001 Low Shear
 $M_{c.y.Rd} = f_y \cdot W_{pl.y} / \gamma_{M0}$ 275 x 497.4/1 136.785 kN.m
 $N_{pl.Rd} = A_g \cdot f_y / \gamma_{M0}$ 58.73 x 275/1 = 1615.075 kN
 $n = N_{Ed}/N_{pl.Rd}$ 0.004 / 1615.075 = 0.000 OK
 $W_{pl.N.y} = F_n(W_{pl.y}, A_{vy}, n)$ 497.4, 16.974, 0 497.4 cm³
 $M_{N.y.Rd} = W_{pl.N.y} \cdot f_y / \gamma_{M0}$ 497.4 x 275/1 136.785 kN.m
 $(M_{y.Ed}/M_{N.y.Rd}) + (M_{z.Ed}/M_{N.z.Rd})$ (55.544/136.785)² + (0)¹ = 0.165 OK

Compression Resistance N.b.Rd

$\lambda_y = \sqrt{A \cdot f_y / N_{cr}}$ $\sqrt{58.73 \times 275 / 3372.49}$ 0.692
 $N_{b.y.Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 58.73 x 0.788 x 275 / 10 / 1 = 1272.861 kN Curve b
 $\lambda_z = \sqrt{A \cdot f_y / N_{crz}}$ $\sqrt{58.73 \times 275 / 1144.48}$ 1.188
 $N_{b.z.Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 58.73 x 0.44 x 275 / 10 / 1 = 710.090 kN Curve c
 $Let = K_t \cdot L_x$ 1 x 5.3 = 5.3
 $\lambda_T = \sqrt{A \cdot f_y / N_{crT}}$ $\sqrt{58.73 \times 275 / 2728.77}$ 0.769
 $N_{b.T.Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 58.73 x 0.681 x 275 / 10 / 1 = 1100.539 kN Curve c

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = f_n(M_1, M_2, M_0, \psi, \mu)$ 0.0, 0.0, 55.5, 0.946, 300.000 1.127 Uniform
 $C_{mLT} = 0.95 + 0.05 \alpha_h$ $M_h = 0.04, M_s = 55.54, \psi = 0.946, \alpha_s = 0.001$ 0.95 Table B.3
 $C_{mz} = \text{Max}(0.6 + 0.4 \psi, 0.4)$ $M = 0, \psi = 1.000$ 1 Table B.3
 $C_{my} = 0.95 + 0.05 \alpha_h$ $M_h = 0, M_s = 55.55, \psi = 1.000, \alpha_s = 0.000$ 0.95 Table B.3

Lateral Buckling Check M.b.Rd

$Le = 1.2L + 2D$ 1.2 x 5.3 + 2 x 0.203 = 6.766 m
 $M_{cr} = F_n(C_1, Le, I_z, I_y, I_w, E)$ 1.127, 6.766, 1551, 22.15, 0.1429, 210000 147.402 kN.m
 $\lambda_{LT} = \sqrt{W \cdot f_y / M_{cr}}$ $\sqrt{497.4 \times 275 / 147.402}$ 0.963
 $\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$ 0.963, 0.922 0.722 Curve b
 $\chi_{LT.mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$ 0.722, 0.963, 0.942, 0.973 0.742 6.3.2.3
 $M_{b.Rd} = \chi W_{pl.y} \cdot f_y \leq M_{c.y.Rd}$ 0.742 x 497.4 x 275 \leq 136.785 = 101.552 kN.m

Buckling Resistance

$U_{N.y} = N_{Ed} / (\chi_y \cdot N_{Rk} / \gamma_{M1})$ 0.408 / 1272.861 0.000 OK
 $U_{N.z} = N_{Ed} / (\chi_z \cdot N_{Rk} / \gamma_{M1})$ 0.408 / 710.09 0.001 OK
 $U_{M.y} = M_{y.Ed} / (\chi_{LT} \cdot M_{y.Rk} / \gamma_{M1})$ 55.544 / 101.552 0.547 OK
 $U_{M.z} = M_{z.Ed} / (M_{z.Rk} / \gamma_{M1})$ 0 / 63.498 0.000 OK
 $k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N.y}\}$ 0.950
 $k_{zz} = C_{mz} \{1 + 1.4 U_{N.z}\}$ 1.001
 $k_{yz} = 0.6 k_{zz}$ 0.600
 $k_{zy} = 1 - \{0.1 \lambda_z / (C_{mLT} - 0.25)\} U_{N.z}$ 1.000
 $U_{Ny} + k_{yy} \cdot U_{M.y} + k_{yz} \cdot U_{M.z}$ 0.000 + 0.950 x 0.547 + 0.600 x 0.000 0.520 OK
 $U_{Nz} + k_{zy} \cdot U_{M.y} + k_{zz} \cdot U_{M.z}$ 0.001 + 1.000 x 0.547 + 1.001 x 0.000 0.547 OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams) $\delta \leq 5300/360 = 14.7$ mm Live (Case 2) 4.99 mm OK
 $\delta \leq 5300/250 = 21.2$ mm D+L (Case 3) 12 mm OK

Provide Section (46.1 kg/m)

203x203 UC 46 [S 275]



AXIAL WITH MOMENTS (MEMBER)

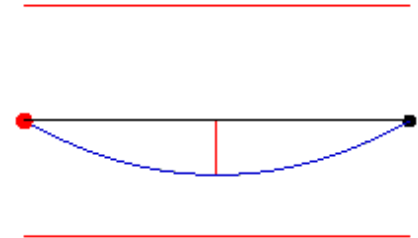
B7

Member 1 (N.1-N.2) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

D1 UDLY -003.210 (kN/m)
L1 UDLY -008.800 (kN/m)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3						
Mem ber No.	Node End1 End2	Axial Force (kN)	Shear Force (kN)	Bending Moment (kN.m)	Maximum Moment (kN.m @ m)	Maximum Deflection (mm @ m)
1	1	0.241C	48.113	0.000	63.750	13.347
	2	0.241C	-48.113	0.000	@ 2.650	@ 2.650

Classification and Effective Area (EN 1993: 2006)

Section (46.1 kg/m) 203x203 UC 46 [S 275]
Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$ 9.25, 22.33, 275, 0.24, 63.75, 0 (Axial: Non-Slender) Class 1
Auto Design Load Cases 1

Local Capacity Check

$V_{y,Ed}/V_{pl,y,Rd}$ 0.003 / 269.498 = 0 Low Shear
 $M_{c,y,Rd} = f_y \cdot W_{pl,y} / \gamma_{M0}$ 275 x 497.4 / 1 136.785 kN.m
 $N_{pl,Rd} = A_g \cdot f_y / \gamma_{M0}$ 58.73 x 275 / 1 = 1615.075 kN
 $n = N_{Ed} / N_{pl,Rd}$ 0.241 / 1615.075 = 0.000 OK
 $W_{pl,N,y} = F_n(W_{pl,y}, A_{vy}, n)$ 497.4, 16.974, 0 497.4 cm³
 $M_{N,y,Rd} = W_{pl,N,y} \cdot f_y / \gamma_{M0}$ 497.4 x 275 / 1 136.785 kN.m
 $(M_{y,Ed} / M_{N,y,Rd}) + (M_{z,Ed} / M_{N,z,Rd})$ (63.747 / 136.785) + (0)¹ = 0.217 OK

Compression Resistance N.b.Rd

$\lambda_y = \sqrt{A \cdot f_y / N_{cr}}$ $\sqrt{58.73 \times 275 / 3372.49}$ 0.692
 $N_{b,y,Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 58.73 x 0.788 x 275 / 10 / 1 = 1272.861 kN Curve b
 $\lambda_z = \sqrt{A \cdot f_y / N_{crz}}$ $\sqrt{58.73 \times 275 / 1144.48}$ 1.188
 $N_{b,z,Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 58.73 x 0.44 x 275 / 10 / 1 = 710.090 kN Curve c
 $l_{et} = K_t \cdot L_x$ 1 x 5.3 = 5.3
 $\lambda_T = \sqrt{A \cdot f_y / N_{crT}}$ $\sqrt{58.73 \times 275 / 2728.77}$ 0.769
 $N_{b,T,Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 58.73 x 0.681 x 275 / 10 / 1 = 1100.539 kN Curve c

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = f_n(M_1, M_2, M_0, \psi, \mu)$ 0.0, 0.0, 63.7, 0.958, 300.000 1.127 Uniform
 $C_{mLT} = 0.95 + 0.05 a_h$ $M_h = 0.05, M_s = 63.75, \psi = 0.958, a_s = 0.001$ 0.95 Table B.3
 $C_{mz} = \text{Max}(0.6 + 0.4 \psi, 0.4)$ $M = 0, \psi = 1.000$ 1 Table B.3
 $C_{my} = 0.95 + 0.05 a_h$ $M_h = 0, M_s = 63.75, \psi = 1.000, a_s = 0.000$ 0.95 Table B.3

Lateral Buckling Check M.b.Rd

$l_e = 1.2L + 2D$ 1.2 x 5.3 + 2 x 0.203 = 6.766 m
 $M_{cr} = F_n(C_1, l_e, I_z, I_T, I_w, E)$ 1.127, 6.766, 1551, 22.15, 0.1429, 210000 147.402 kN.m
 $\lambda_{LT} = \sqrt{W \cdot f_y / M_{cr}}$ $\sqrt{497.4 \times 275 / 147.402}$ 0.963
 $\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$ 0.963, 0.922 0.722 Curve b
 $\chi_{LT,mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$ 0.722, 0.963, 0.942, 0.973 0.742 6.3.2.3
 $M_{b,Rd} = \chi \cdot W_{pl,y} \cdot f_y \leq M_{c,y,Rd}$ 0.742 x 497.4 x 275 ≤ 136.785 = 101.552 kN.m

Buckling Resistance

$U_{N,y} = N_{Ed} / (\chi_y \cdot N_{Rk} / \gamma_{M1})$ 0.241 / 1272.861 0.000 OK
 $U_{N,z} = N_{Ed} / (\chi_z \cdot N_{Rk} / \gamma_{M1})$ 0.241 / 710.09 0.000 OK
 $U_{M,y} = M_{y,Ed} / (\chi_{LT} \cdot M_{y,Rk} / \gamma_{M1})$ 63.747 / 101.552 0.628 OK
 $U_{M,z} = M_{z,Ed} / (M_{z,Rk} / \gamma_{M1})$ 0 / 63.498 0.000 OK
 $k_{yy} = C_{my} \{ 1 + (\lambda_y - 0.2) U_{N,y} \}$ 0.950



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$k_{zz} = C_{mz} \{1 + 1.4 U_{N,z}\}$		1.000	
$k_{yz} = 0.6 k_{zz}$		0.600	
$k_{zy} = 1 - \{0.1 \lambda_z / (C_{mLT} - 0.25)\} U_{N,z}$		1.000	
$U_{N_y} + k_{yy} \cdot U_{M,y} + k_{yz} \cdot U_{M,z}$	0.000 + 0.950 x 0.628 + 0.600 x 0.000	0.597	OK
$U_{N_z} + k_{zy} \cdot U_{M,y} + k_{zz} \cdot U_{M,z}$	0.000 + 1.000 x 0.628 + 1.000 x 0.000	0.628	OK

Deflection Check - Load Case 2

Deflection Limits (Internal Beams)	$\delta \leq 5300/360 = 14.7$ mm Live (Case 2)	9.42 mm	OK
	$\delta \leq 5300/250 = 21.2$ mm D+L (Case 3)	13.35 mm	OK

Provide Section (46.1 kg/m)

203x203 UC 46 [S 275]

AXIAL WITH MOMENTS (MEMBER)

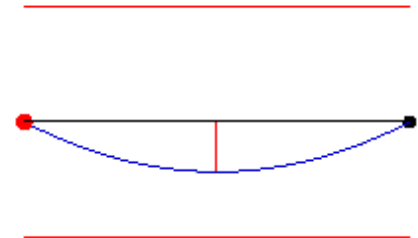
B8

Member 1 (N.1-N.2) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

D1 UDLY -002.200 (kN/m)
L1 UDLY -008.800 (kN/m)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3						
Mem ber No.	Node End1 End2	Axial Force (kN)	Shear Force (kN)	Bending Moment (kN.m)	Maximum Moment (kN.m @ m)	Maximum Deflection (mm @ m)
1	1	0.222C	44.500	0.000	58.962	12.266
	2	0.222C	-44.500	0.000	@ 2.650	@ 2.650

Classification and Effective Area (EN 1993: 2006)

Section (46.1 kg/m) 203x203 UC 46 [S 275]
Class = $f_n(b/T, d/t, f_y, N, M_y, M_z)$ 9.25, 22.33, 275, 0.22, 58.96, 0 (Axial: Non-Slender) Class 1
Auto Design Load Cases 1

Local Capacity Check

$V_{y,Ed} / V_{pl,y,Rd}$ 0.004 / 269.498 = 0 Low Shear
 $M_{c,y,Rd} = f_y \cdot W_{pl,y} / \gamma_{M0}$ 275 x 497.4 / 1 136.785 kN.m
 $N_{pl,Rd} = A_g \cdot f_y / \gamma_{M0}$ 58.73 x 275 / 1 = 1615.075 kN
 $n = N_{Ed} / N_{pl,Rd}$ 0.222 / 1615.075 = 0.000 OK
 $W_{pl,N,y} = F_n(W_{pl,y}, A_{vy}, n)$ 497.4, 16.974, 0 497.4 cm³
 $M_{N,y,Rd} = W_{pl,N,y} \cdot f_y / \gamma_{M0}$ 497.4 x 275 / 1 136.785 kN.m
 $(M_{y,Ed} / M_{N,y,Rd}) + (M_{z,Ed} / M_{N,z,Rd})$ (58.957 / 136.785) + (0)¹ = 0.186 OK

Compression Resistance N.b.Rd

$\lambda_y = \sqrt{A \cdot f_y / N_{cr}}$ $\sqrt{58.73 \times 275 / 3372.49}$ 0.692
 $N_{b,y,Rd} = A_{eff} \cdot \chi_y \cdot f_y / \gamma_{M1}$ 58.73 x 0.788 x 275 / 10 / 1 = 1272.861 kN Curve b
 $\lambda_z = \sqrt{A \cdot f_y / N_{crz}}$ $\sqrt{58.73 \times 275 / 1144.48}$ 1.188
 $N_{b,z,Rd} = A_{eff} \cdot \chi_z \cdot f_y / \gamma_{M1}$ 58.73 x 0.44 x 275 / 10 / 1 = 710.090 kN Curve c
Let = $K_t \cdot L_x$ 1 x 5.3 = 5.3
 $\lambda_T = \sqrt{A \cdot f_y / N_{crT}}$ $\sqrt{58.73 \times 275 / 2728.77}$ 0.769
 $N_{b,T,Rd} = A_{eff} \cdot \chi_T \cdot f_y / \gamma_{M1}$ 58.73 x 0.681 x 275 / 10 / 1 = 1100.539 kN Curve c

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = f_n(M_1, M_2, M_0, \psi, \mu)$ 0.0, 0.0, 58.9, 0.933, 300.000 1.127 Uniform
 $C_{mLT} = 0.95 + 0.05 a_h$ $M_h = 0.05, M_s = 58.96, \psi = 0.933, a_s = 0.001$ 0.95 Table B.3
 $C_{mz} = \text{Max}(0.6 + 0.4 \psi, 0.4)$ $M = 0, \psi = 1.000$ 1 Table B.3



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$C_{my}=0.95+0.05a_h$ $M_h=0, M_s=58.96, \psi=1.000, \alpha_s=0.000$ 0.95 Table B.3

Lateral Buckling Check M.b.Rd

$l_e = 1.2L + 2D$ $1.2 \times 5.3 + 2 \times 0.203 =$ 6.766 m
 $M_{cr} = F_n(C_1, l_e, I_z, I_w, E)$ $1.127, 6.766, 1551, 22.15, 0.1429, 210000$ 147.402 kN.m
 $\lambda_{LT} = \sqrt{W_{pl,y}/M_{cr}}$ $\sqrt{497.4 \times 275 / 147.402}$ 0.963
 $\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$ 0.963, 0.922 0.722
 $\chi_{LT,mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$ 0.722, 0.963, 0.942, 0.973 0.742
 $M_{b,Rd} = \chi W_{pl,y} f_y \leq M_{c,y,Rd}$ $0.742 \times 497.4 \times 275 \leq 136.785 =$ 101.552 kN.m

Curve b
6.3.2.3

Buckling Resistance

$U_{N,y} = N_{Ed}/(\chi_y N_{Rk}/\gamma_{M1})$ 0.222 / 1272.861 0.000 OK
 $U_{N,z} = N_{Ed}/(\chi_z N_{Rk}/\gamma_{M1})$ 0.222 / 710.09 0.000 OK
 $U_{M,y} = M_{y,Ed}/(\chi_{LT} M_{y,Rk}/\gamma_{M1})$ 58.957 / 101.552 0.581 OK
 $U_{M,z} = M_{z,Ed}/(M_{z,Rk}/\gamma_{M1})$ 0 / 63.498 0.000 OK
 $k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N,y}\}$ 0.950
 $k_{zz} = C_{mz} \{1 + 1.4 U_{N,z}\}$ 1.000
 $k_{yz} = 0.6 k_{zz}$ 0.600
 $k_{zy} = 1 - \{0.1 \lambda_z / (C_{mLT} - 0.25)\} U_{N,z}$ 1.000
 $U_{Ny} + k_{yy} U_{My} + k_{yz} U_{Mz}$ 0.000 + 0.950 x 0.581 + 0.600 x 0.000 0.552 OK
 $U_{Nz} + k_{zy} U_{My} + k_{zz} U_{Mz}$ 0.000 + 1.000 x 0.581 + 1.000 x 0.000 0.581 OK

Deflection Check - Load Case 2

Deflection Limits (Internal Beams) $\delta \leq 5300/360 = 14.7$ mm Live (Case 2) 9.42 mm OK
 $\delta \leq 5300/250 = 21.2$ mm D+L (Case 3) 12.27 mm OK

Provide Section (46.1 kg/m)

203x203 UC 46 [S 275]

AXIAL WITH MOMENTS (MEMBER)

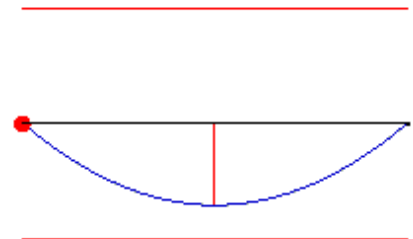
B9

Member 12 (N.27-N.28) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

D1 D 078.500 (kN/m³)
 D1 UDLY -036.270 (kN/m)
 L1 UDLY -014.550 (kN/m)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3											
Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)	
				y-y	z-z	y-y	z-z	y-y	z-z		
12	27	0.00C	0.00	182.13	0.00	-3.00	0.00	227.40	0.00	14.49 @ 2.530 @ 0.000 @ 2.530	
	28	0.00C	0.00	-182.14	0.00	-3.01	0.00	@ 2.530	@ 0.000		

Classification and Effective Area (EN 1993: 2006)

Section (88.95 kg/m) 254x254 UC 89 [S 275]
 Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$ 7.41, 19.45, 265, 0, 227.39, 0 (Axial: Non-Slender) Class 1
 Auto Design Load Cases 1

Moment Capacity Check M.c.y.Rd

$V_{y,Ed}/V_{pl,y,Rd}$ 0.009 / 471.33 = 0 Low Shear
 $M_{c,y,Rd} = f_y W_{pl,y} / \gamma_{M0}$ 265 x 1223.9/1 324.334 kN.m
 $M_{y,Ed}/M_{c,y,Rd}$ 227.392 / 324.334 = 0.701 OK

Equivalent Uniform Moment Factor C1

$C_1 = f_n(M_1, M_2, M_0, \psi, \mu)$ -2.8, -2.8, 230.2, 0.997, -81.463 1.202 Uniform



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Lateral Buckling Check M.b.Rd

$l_e = 1.2L + 2D$	$1.2 \times 5.06 + 2 \times 0.260 =$	6.593 m	
$M_{cr} = F_n(C_1, l_e, I_z, I_y, I_w, E)$	1.202, 6.593, 4864, 102.3, 0.7166, 210000	625.963 kN.m	
$\lambda_{LT} = \sqrt{W_{pl,y}/M_{cr}}$	$\sqrt{1223.9 \times 265 / 625.963}$	0.720	
$\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$	0.720, 0.712	0.860	Curve b
$\chi_{LT,mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$	0.860, 0.720, 0.912, 0.957	0.899	6.3.2.3
$M_{b,Rd} = \chi W_{pl,y} f_y \leq M_{c,y,Rd}$	$0.899 \times 1224 \times 265 \leq 324.334 =$	291.474 kN.m	
$M_{y,Ed}/M_{b,Rd}$	227.392 / 291.474	0.780	OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams)	$\delta \leq 5060/360 = 14.1$ mm Live (Case 2)	4.08 mm	OK
	$\delta \leq 5060/250 = 20.2$ mm D+L (Case 3)	14.49 mm	OK

Provide Section (88.95 kg/m)

254x254 UC 89 [S 275]

AXIAL WITH MOMENTS (MEMBER)

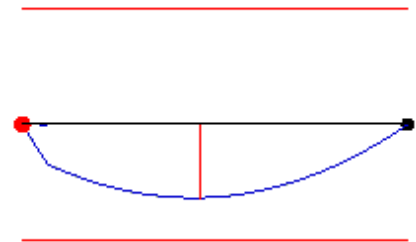
B10

Members 1-2 (N.11-N.12) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

Part 1		
D1 UDLY	-011.440	(kN/m)
L1 UDLY	-006.600	(kN/m)
D1 D	078.500	(kN/m ³)
Part 2		
D1 UDLY	-011.440	(kN/m)
L1 UDLY	-006.600	(kN/m)
D1 D	078.500	(kN/m ³)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3

Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)
				y-y	z-z	y-y	z-z	y-y	z-z	
	11	1.26T	0.00	181.20	0.02	0.00	0.00	113.97	0.01	19.11
	13	0.32T	0.00	-77.21	0.00	0.00	0.00	@ 2.330	@ 0.336	@ 2.578

Classification and Effective Area (EN 1993: 2006)

Section (59.95 kg/m)	203x203 UC 60 [S 275]		
Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$	7.25, 17.11, 275, 0, 113.96, 0.01	(Axial: Non-Slender)	Class 1
Auto Design Load Cases	1		

Local Capacity Check

$V_{y,Ed}/V_{pl,y,Rd}$	$2.575 / 351.748 =$	0.007	Low Shear
$M_{c,y,Rd} = f_y W_{pl,y} / \gamma_{M0}$	$275 \times 656.1 / 1$	180.428 kN.m	
$N_{pl,Rd} = A_g f_y / \gamma_{M0}$	$76.37 \times 275 / 1$	2100.175 kN	
$n = N_{Ed} / N_{pl,Rd}$	$-1.265 / 2100.175 =$	0.001	OK
$W_{pl,N,y} = F_n(W_{pl,y}, A_{vy}, n)$	$656.1, 22.154, 0.001$	656.1 cm ³	
$M_{N,y,Rd} = W_{pl,N,y} f_y / \gamma_{M0}$	$656.1 \times 275 / 1$	180.428 kN.m	
$(M_{y,Ed} / M_{N,y,Rd}) + (M_{z,Ed} / M_{N,z,Rd})$	$(113.843 / 180.428) + (0)^1 =$	0.398	OK

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = f_n(M_1, M_2, M_0, \psi, \mu)$	0.2, 0.1, 112.6, 0.425, 300.000	1.127	Uniform
$C_{mLT} = 0.95 + 0.05 a_h$	$M_h = 0.18, M_s = 112.78, \psi = 0.425, a_s = 0.002$	0.95	Table B.3
$C_{mz} = \text{Max}(0.6 + 0.4 \psi, 0.4)$	$M = 0, \psi = 1.000$	1	Table B.3
$C_{my} = 0.95 + 0.05 a_h$	$M_h = 0, M_s = 112.78, \psi = 1.000, a_s = 0.000$	0.95	Table B.3

Lateral Buckling Check M.b.Rd

$l_e = 1.2L + 2D$	$1.2 \times 5.3 + 2 \times 0.210 =$	6.779 m
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$M_{cr} = F_n(C_1, L_e, I_z, I_y, I_w, E)$	1.127, 6.779, 2068, 47.23, 0.1969, 210000	235.996 kN.m	
$\lambda_{LT} = \sqrt{W_{pl,y}/M_{cr}}$	$\sqrt{656.1 \times 275 / 235.996}$	0.874	
$\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$	0.874, 0.837	0.775	Curve b
$\chi_{LT,mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$	0.775, 0.874, 0.942, 0.971	0.798	6.3.2.3
$M_{b,Rd} = \chi W_{pl,y} f_y \leq M_{c,y,Rd}$	$0.798 \times 656.1 \times 275 \leq 180.428 =$	143.965 kN.m	

Buckling Resistance

$U_{N,y} = N_{Ed}/(\chi_y N_{Rk}/\gamma_{M1})$	0 / 1667.819	0.000	OK
$U_{N,z} = N_{Ed}/(\chi_z N_{Rk}/\gamma_{M1})$	0 / 937.412	0.000	OK
$U_{M,y} = M_{y,Ed}/(\chi_{LT} M_{y,Rk}/\gamma_{M1})$	113.843 / 143.965	0.791	OK
$U_{M,z} = M_{z,Ed}/(M_{z,Rk}/\gamma_{M1})$	0 / 83.958	0.000	OK
$k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N,y}\}$		0.950	
$k_{zz} = C_{mz} \{1 + 1.4 U_{N,z}\}$		1.000	
$k_{yz} = 0.6 k_{zz}$		0.600	
$k_{zy} = 1 - \{0.1 \lambda_z / (C_{mLT} - 0.25)\} U_{N,z}$		1.000	
$U_{Ny} + k_{yy} U_{My} + k_{yz} U_{Mz}$	$0.000 + 0.950 \times 0.791 + 0.600 \times 0.000$	0.751	OK
$U_{Nz} + k_{zy} U_{My} + k_{zz} U_{Mz}$	$0.000 + 1.000 \times 0.791 + 1.000 \times 0.000$	0.791	OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams)	$\delta \leq 5300/360 = 14.7$ mm Live (Case 2)	5.27 mm	OK
	$\delta \leq 5300/250 = 21.2$ mm D+L (Case 3)	19.11 mm	OK

Provide Section (59.95 kg/m)

203x203 UC 60 [S 275]

AXIAL WITH MOMENTS (MEMBER)

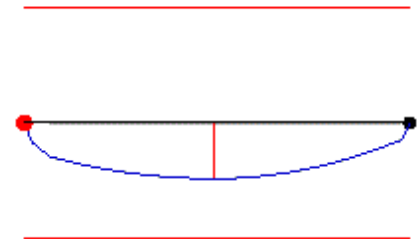
B11

Members 3-7 (N.3-N.4) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

Part 1		
D1 UDLY	-005.720	(kN/m)
L1 UDLY	-003.300	(kN/m)
D1 D	078.500	(kN/m ³)
Part 2		
D1 UDLY	-005.720	(kN/m)
L1 UDLY	-003.300	(kN/m)
D1 D	078.500	(kN/m ³)
Part 3		
D1 UDLY	-005.720	(kN/m)
L1 UDLY	-003.300	(kN/m)
D1 D	078.500	(kN/m ³)
Part 4		
D1 UDLY	-005.720	(kN/m)
L1 UDLY	-003.300	(kN/m)
D1 D	078.500	(kN/m ³)
Part 5		
D1 UDLY	-005.720	(kN/m)
L1 UDLY	-003.300	(kN/m)
D1 D	078.500	(kN/m ³)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3										
Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)
				y-y	z-z	y-y	z-z	y-y	z-z	
	3	0.70T	0.00	277.80	-0.30	0.00	0.00	86.79	-0.13	15.17
	9	0.01T	0.00	-232.87	0.00	0.00	0.00	@ 2.600	@ 0.348	@ 2.626

Classification and Effective Area (EN 1993: 2006)

Section (59.95 kg/m) 203x203 UC 60 [S 275]



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Class = Fn(b/T,d/t,f_y,N,M_y,M_z) 7.25, 17.11, 275, 0, 86.79, 0.13 (Axial: Non-Slender) Class 1
Auto Design Load Cases 1

Local Capacity Check

V _{y.Ed} /V _{pl.y.Rd}	0.027 / 351.748 =	0	Low Shear
M _{c.y.Rd} = f _y ·W _{pl.y} /γ _{M0}	275 x 656.1/1	180.428 kN.m	
V _{z.Ed} /V _{pl.z.Rd}	0.025 / 927.974 =	0	Low Shear
M _{c.z.Rd} = f _y ·W _{pl.z} /γ _{M0}	275 x 305.3/1	83.958 kN.m	
N _{pl.Rd} = A _g ·f _y /γ _{M0}	76.37x275/1	2100.175 kN	
η = N _{Ed} /N _{pl.Rd}	-2.137 / 2100.175 =	0.001	OK
W _{pl.N.y} = Fn(W _{pl.y} , A _{vy} , n)	656.1, 22.154, 0.001	656.1 cm ³	
M _{N.y.Rd} = W _{pl.N.y} ·f _y /γ _{M0}	656.1 x 275/1	180.428 kN.m	
W _{pl.N.z} = Fn(W _{pl.z} , A _{vz} , n)	305.3, 58.447, 0.001	305.3 cm ³	
M _{N.z.Rd} = W _{pl.N.z} ·f _y /γ _{M0}	305.3 x 275/1	83.958 kN.m	
(M _{y.Ed} /M _{N.y.Rd}) + (M _{z.Ed} /M _{N.z.Rd})	(86.792/180.428) ² + (0.065/83.958) ¹ =	0.232	OK

Equivalent Uniform Moment Factors C₁, C_{mLT}, C_{mz}, and C_{my}

C ₁ = fn(M ₁ , M ₂ , M ₀ , ψ, μ)	0.3, 0.2, 86.5, 0.855, 300.000	1.127	Uniform
C _{mLT} = 0.95 + 0.05α _h	M _h = 0.28, M _s = 86.76, ψ = 0.855, α _s = 0.003	0.95	Table B.3
C _{mz} = Max(0.6 + 0.4ψ, 0.4)	M = 0, ψ = 0.000	0.6	Table B.3
C _{my} = 0.95 + 0.05α _h	M _h = 0, M _s = 86.77, ψ = 1.000, α _s = 0.000	0.95	Table B.3

Lateral Buckling Check M_b.Rd

l _e = 1.2L + 2D	1.2 x 5.3 + 2 x 0.210 =	6.779 m	
M _{cr} = Fn(C ₁ , l _e , I _z , I _w , E)	1.127, 6.779, 2068, 47.23, 0.1969, 210000	235.996 kN.m	
λ _{LT} = √ W _f /M _{cr}	√ 656.1 x 275 / 235.996	0.874	
χ _{LT} = Fn(λ _{LT} , λ _{LT5950})	0.874, 0.837	0.775	Curve b
χ _{LT.mod} = Fn(χ _{LT} , λ _{LT} , k _c , f)	0.775, 0.874, 0.942, 0.971	0.798	6.3.2.3
M _{b.Rd} = χ W _{pl.y} ·f _y ≤ M _{c.y.Rd}	0.798 x 656.1 x 275 ≤ 180.428 =	143.965 kN.m	

Buckling Resistance

U _{N.y} = N _{Ed} /(χ _y ·N _{Rk} /γ _{M1})	0 / 1667.819	0.000	OK
U _{N.z} = N _{Ed} /(χ _z ·N _{Rk} /γ _{M1})	0 / 937.412	0.000	OK
U _{M.y} = M _{y.Ed} /(χ _{LT} ·M _{y.Rk} /γ _{M1})	86.792 / 143.965	0.603	OK
U _{M.z} = M _{z.Ed} /(M _{z.Rk} /γ _{M1})	0.065 / 83.958	0.001	OK
k _{yy} = C _{my} {1 + (λ _y - 0.2)U _{N.y} }		0.950	
k _{zz} = C _{mz} {1 + 1.4U _{N.z} }		0.600	
k _{yz} = 0.6 k _{zz}		0.360	
k _{zy} = 1 - {0.1λ _z /(C _{mLT} - 0.25)}U _{N.z}		1.000	
U _{Ny} + k _{yy} ·U _{M.y} + k _{yz} ·U _{M.z}	0.000 + 0.950x0.603 + 0.360x0.001	0.573	OK
U _{Nz} + k _{zy} ·U _{M.y} + k _{zz} ·U _{M.z}	0.000 + 1.000x0.603 + 0.600x0.001	0.603	OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams)	δ ≤ 5300/360 = 14.7 mm Live (Case 2)	4.26 mm	OK
	δ ≤ 5300/250 = 21.2 mm D+L (Case 3)	15.17 mm	OK

Provide Section (59.95 kg/m)

203x203 UC 60 [S 275]



AXIAL WITH MOMENTS (MEMBER)

B12

Members 18-20 (N.5-N.8) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

Part 1

D1 D 078.500 (kN/m³)

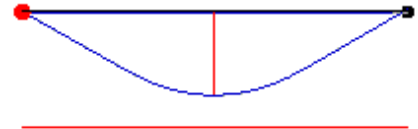
Part 2

D1 D 078.500 (kN/m³)

D1 UDLY -043.000 (kN/m)

Part 3

D1 D 078.500 (kN/m³)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3										
Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)
				y-y	z-z	y-y	z-z	y-y	z-z	
	5	0.44T	0.01	58.77	0.29	-0.43	-0.01	98.34	0.34	14.32
	12	0.44T	-0.01	-60.07	-0.30	-3.29	-0.02	@ 2.180	@ 1.200	@ 2.200

Classification and Effective Area (EN 1993: 2006)

Section (46.1 kg/m) 203x203 UC 46 [S 275]
Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$ 9.25, 22.33, 275, 0, 98.33, 0.34 (Axial: Non-Slender) Class 1
Auto Design Load Cases 1

Local Capacity Check

$V_{y.Ed}/V_{pl.y.Rd}$ 0.657 / 269.498 = 0.002 Low Shear
 $M_{c.y.Rd} = f_y \cdot W_{pl.y} / \gamma_{M0}$ 275 x 497.4 / 1 136.785 kN.m
 $V_{z.Ed}/V_{pl.z.Rd}$ 0.002 / 711.169 = 0 Low Shear
 $M_{c.z.Rd} = f_y \cdot W_{pl.z} / \gamma_{M0}$ 275 x 230.9 / 1 63.498 kN.m
 $N_{pl.Rd} = A_g \cdot f_y / \gamma_{M0}$ 58.73 x 275 / 1 1615.075 kN
 $n = N_{Ed} / N_{pl.Rd}$ -0.442 / 1615.075 = 0.000 OK
 $W_{pl.N.y} = F_n(W_{pl.y}, A_{vy}, n)$ 497.4, 16.974, 0 497.4 cm³
 $M_{N.y.Rd} = W_{pl.N.y} \cdot f_y / \gamma_{M0}$ 497.4 x 275 / 1 136.785 kN.m
 $W_{pl.N.z} = F_n(W_{pl.z}, A_{vz}, n)$ 230.9, 44.792, 0 230.9 cm³
 $M_{N.z.Rd} = W_{pl.N.z} \cdot f_y / \gamma_{M0}$ 230.9 x 275 / 1 63.498 kN.m
 $(M_{y.Ed} / M_{N.y.Rd}) + (M_{z.Ed} / M_{N.z.Rd})$ $(98.329 / 136.785)^2 + (0.341 / 63.498)^2 = 0.522$ OK

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = f_n(M_1, M_2, M_o, \psi, \mu)$ -0.4, -3.2, 100.1, 0.113, -31.009 1.167 Uniform
 $C_{mLT} = 0.95 + 0.05 a_h$ $M_h = -3.23, M_s = 98.33, \psi = 0.113, a_s = -0.033$ 0.948 Table B.3
 $C_{mz} = 0.95 + 0.05 a_h$ $M_h = -0.02, M_s = 0.34, \psi = 0.562, a_s = -0.047$ 0.948 Table B.3
 $C_{my} = 0.95 + 0.05 a_h$ $M_h = -3.29, M_s = 98.33, \psi = 0.129, a_s = -0.033$ 0.948 Table B.3

Lateral Buckling Check M.b.Rd

$l_e = 1.2L + 2D$ 1.2 x 4.4 + 2 x 0.203 = 5.686 m
 $M_{cr} = F_n(C_1, l_e, I_z, I_y, I_w, E)$ 1.167, 5.686, 1551, 22.15, 0.1429, 210000 191.344 kN.m
 $\lambda_{LT} = \sqrt{W_{pl.y} / M_{cr}}$ $\sqrt{497.4 \times 275 / 191.344}$ 0.845
 $\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$ 0.845, 0.823 0.792 Curve b
 $\chi_{LT.mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$ 0.792, 0.845, 0.926, 0.963 0.822 6.3.2.3
 $M_{b.Rd} = \chi_{LT} W_{pl.y} f_y \leq M_{c.y.Rd}$ 0.822 x 497.4 x 275 \leq 136.785 = 112.441 kN.m

Buckling Resistance

$U_{N.y} = N_{Ed} / (\chi_y \cdot N_{Rk} / \gamma_{M1})$ 0 / 1372.147 0.000 OK
 $U_{N.z} = N_{Ed} / (\chi_z \cdot N_{Rk} / \gamma_{M1})$ 0 / 885.187 0.000 OK
 $U_{M.y} = M_{y.Ed} / (\chi_{LT} \cdot M_{y.Rk} / \gamma_{M1})$ 98.329 / 112.441 0.874 OK
 $U_{M.z} = M_{z.Ed} / (M_{z.Rk} / \gamma_{M1})$ 0.341 / 63.498 0.005 OK
 $k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N.y}\}$ 0.948
 $k_{zz} = C_{mz} \{1 + (2\lambda_z - 0.6) U_{N.z}\}$ 0.948
 $k_{yz} = 0.6 k_{zz}$ 0.569
 $k_{zy} = 1 - \{0.1 / (C_{mLT} - 0.25)\} U_{N.z}$ 1.000
 $U_{Ny} + k_{yy} \cdot U_{M.y} + k_{yz} \cdot U_{M.z}$ 0.000 + 0.948 x 0.874 + 0.569 x 0.005 0.832 OK



$U_{Nz} + k_{zy} \cdot U_{M,y} + k_{zz} \cdot U_{M,z}$ 0.000 + 1.000 x 0.874 + 0.948 x 0.005 0.880 OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams) $\delta \leq 4400/250 = 17.6$ mm D+L (Case 3) 14.32 mm OK

Provide Section (46.1 kg/m)

203x203 UC 46 [S 275]

AXIAL WITH MOMENTS (MEMBER)

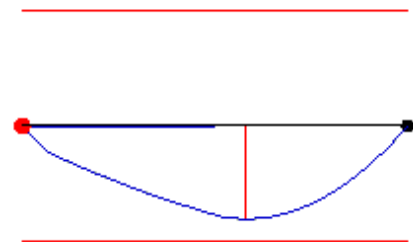
B13

Members 8-11 (N.16-N.17) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

Part 1		
D1 UDLY	-007.295	(kN/m)
L1 UDLY	-003.825	(kN/m)
D1 D	078.500	(kN/m ³)
D1 UDLY	-049.550	(kN/m)
L1 UDLY	-014.550	(kN/m)
Part 2		
D1 UDLY	-007.295	(kN/m)
L1 UDLY	-003.825	(kN/m)
D1 D	078.500	(kN/m ³)
D1 UDLY	-049.550	(kN/m)
L1 UDLY	-014.550	(kN/m)
Part 3		
D1 UDLY	-007.295	(kN/m)
L1 UDLY	-003.825	(kN/m)
D1 D	078.500	(kN/m ³)
Part 4		
D1 UDLY	-007.295	(kN/m)
L1 UDLY	-003.825	(kN/m)
D1 D	078.500	(kN/m ³)
D1 UDLY	-049.550	(kN/m)
L1 UDLY	-014.550	(kN/m)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3										
Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)
				y-y	z-z	y-y	z-z	y-y	z-z	
	16	2.00T	0.00	222.74	0.52	0.00	0.00	260.60	0.18	17.70
	20	0.67T	0.00	-234.50	0.00	0.00	0.00	@ 3.082	@ 0.347	@ 2.750

Classification and Effective Area (EN 1993: 2006)

Section (88.95 kg/m) 254x254 UC 89 [S 275]
Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$ 7.41, 19.45, 265, 0, 260.55, 0.18 (Axial: Non-Slender) Class 1
Auto Design Load Cases 1

Local Capacity Check

$V_{y,Ed}/V_{pl,y,Rd}$ 2.097 / 471.33 = 0.004 Low Shear
 $M_{c,y,Rd} = f_y \cdot W_{pl,y} / \gamma_{M0}$ 265 x 1223.9/1 324.334 kN.m
 $N_{pl,Rd} = A_g \cdot f_y / \gamma_{M0}$ 113.31 x 265/1 3002.715 kN
 $n = N_{Ed}/N_{pl,Rd}$ -1.996 / 3002.715 = 0.001 OK
 $W_{pl,N,y} = F_n(W_{pl,y}, A_{vy}, n)$ 1223.9, 30.806, 0.001 1223.9 cm³
 $M_{N,y,Rd} = W_{pl,N,y} \cdot f_y / \gamma_{M0}$ 1223.9 x 265/1 324.334 kN.m
 $(M_{y,Ed}/M_{N,y,Rd}) + (M_{z,Ed}/M_{N,z,Rd})$ (260.578/324.334)² + (0)¹ = 0.645 OK

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = f_n(M_1, M_2, M_0, \psi, \mu)$ 0.2, 0.2, 248.6, 0.944, 300.000 1.127 Uniform
 $C_{mLT} = 0.95 + 0.05 a_h$ $M_h = 0.23, M_s = 248.83, \psi = 0.944, a_s = 0.001$ 0.95 Table B.3
 $C_{mz} = 0.95 + 0.05 a_h$ $M_h = 0, M_s = 0.01, \psi = 0.000, a_s = 0.111$ 0.956 Table B.3



T 02076 085740

$C_{my}=0.95+0.05a_h$ $M_h=0, M_s=248.83, \psi=1.000, a_s=0.000$ 0.95 Table B.3

Lateral Buckling Check M.b.Rd

$l_e = 1.2L+2D$ $1.2 \times 5.3 + 2 \times 0.260 =$ 6.881 m
 $M_{cr} = F_n(C_1, L_e, I_z, I_y, I_w, E)$ 1.127, 6.881, 4864, 102.3, 0.7166, 210000 555.377 kN.m
 $\lambda_{LT} = \sqrt{W_{pl,y}/M_{cr}}$ $\sqrt{1223.9 \times 265 / 555.377}$ 0.764
 $\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$ 0.764, 0.732 0.837
 $\chi_{LT,mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$ 0.837, 0.764, 0.942, 0.971 0.861
 $M_{b,Rd} = \chi W_{pl,y} f_y \leq M_{c,y,Rd}$ $0.861 \times 1224 \times 265 \leq 324.334 =$ 279.397 kN.m

Curve b
6.3.2.3

Buckling Resistance

$U_{N,y} = N_{Ed}/(\chi_y \cdot N_{Rk}/\gamma_{M1})$ 0 / 2608.696 0.000 OK
 $U_{N,z} = N_{Ed}/(\chi_z \cdot N_{Rk}/\gamma_{M1})$ 0 / 1773.616 0.000 OK
 $U_{M,y} = M_{y,Ed}/(\chi_{LT} \cdot M_{y,Rk}/\gamma_{M1})$ 260.578 / 279.397 0.933 OK
 $U_{M,z} = M_{z,Ed}/(M_{z,Rk}/\gamma_{M1})$ 0 / 152.455 0.000 OK
 $k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N,y}\}$ 0.950
 $k_{zz} = C_{mz} \{1 + (2\lambda_z - 0.6) U_{N,z}\}$ 0.956
 $k_{yz} = 0.6 k_{zz}$ 0.573
 $k_{zy} = 1 - \{0.1 / (C_{mLT} - 0.25)\} U_{N,z}$ 1.000
 $U_{Ny} + k_{yy} \cdot U_{M,y} + k_{yz} \cdot U_{M,z}$ $0.000 + 0.950 \times 0.933 + 0.573 \times 0.000$ 0.886 OK
 $U_{Nz} + k_{zy} \cdot U_{M,y} + k_{zz} \cdot U_{M,z}$ $0.000 + 1.000 \times 0.933 + 0.956 \times 0.000$ 0.933 OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams) $\delta \leq 5300/360 = 14.7$ mm Live (Case 2) 3.87 mm OK
 $\delta \leq 5300/250 = 21.2$ mm D+L (Case 3) 17.7 mm OK

Provide Section (88.95 kg/m)

254x254 UC 89 [S 275]

AXIAL WITH MOMENTS (MEMBER)

B14

Members 13-15 (N.21-N.22) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

Part 1

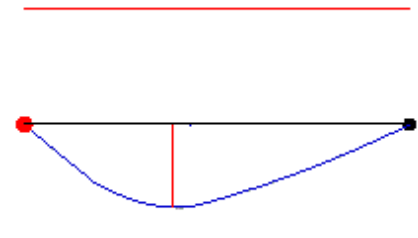
D1 UDLY -002.700 (kN/m)
L1 UDLY -000.900 (kN/m)
D1 TRY -001.063 0.000 0.450 (kN, m, m)
L1 TRY -000.354 0.000 0.450 (kN, m, m)
D1 D 078.500 (kN/m³)

Part 2

D1 UDLY -002.700 (kN/m)
L1 UDLY -000.900 (kN/m)
D1 TRY -001.063 0.000 0.450 (kN, m, m)
L1 TRY -000.354 0.000 0.450 (kN, m, m)
D1 D 078.500 (kN/m³)
D1 UDLY -033.420 (kN/m)
L1 UDLY -002.640 (kN/m)

Part 3

D1 UDLY -002.700 (kN/m)
L1 UDLY -000.900 (kN/m)
D1 TRY -001.063 0.000 0.450 (kN, m, m)
L1 TRY -000.354 0.000 0.450 (kN, m, m)
D1 D 078.500 (kN/m³)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3

Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)
				y-y	z-z	y-y	z-z	y-y	z-z	
	21	0.94T	0.00	97.11	-0.10	0.00	0.00	129.15	-0.10	19.63
	25	0.07T	0.00	-55.65	0.00	0.00	0.00	@ 2.146	@ 0.900	@ 2.484



Classification and Effective Area (EN 1993: 2006)

Section (59.95 kg/m)	203x203 UC 60 [S 275]		
Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$	7.25, 17.11, 275, 0, 129.14, 0.1	(Axial: Non-Slender)	Class 1
Auto Design Load Cases	1		

Local Capacity Check

$V_{y.Ed}/V_{pl.y.Rd}$	$5.779 / 351.748 =$	0.016	Low Shear
$M_{c.y.Rd} = f_y \cdot W_{pl.y} / \gamma_{M0}$	$275 \times 656.1/1$	180.428 kN.m	
$V_{z.Ed}/V_{pl.z.Rd}$	$0.073 / 927.974 =$	0	Low Shear
$M_{c.z.Rd} = f_y \cdot W_{pl.z} / \gamma_{M0}$	$275 \times 305.3/1$	83.958 kN.m	
$N_{pl.Rd} = A_g \cdot f_y / \gamma_{M0}$	$76.37 \times 275/1$	2100.175 kN	
$n = N_{Ed}/N_{pl.Rd}$	$-0.945 / 2100.175 =$	0.000	OK
$W_{pl.N.y} = F_n(W_{pl.y}, A_{vy}, n)$	$656.1, 22.154, 0$	656.1 cm ³	
$M_{N.y.Rd} = W_{pl.N.y} \cdot f_y / \gamma_{M0}$	$656.1 \times 275/1$	180.428 kN.m	
$W_{pl.N.z} = F_n(W_{pl.z}, A_{vz}, n)$	$305.3, 58.447, 0$	305.3 cm ³	
$M_{N.z.Rd} = W_{pl.N.z} \cdot f_y / \gamma_{M0}$	$305.3 \times 275/1$	83.958 kN.m	
$(M_{y.Ed}/M_{N.y.Rd}) + (M_{z.Ed}/M_{N.z.Rd})$	$(128.855/180.428)^2 + (0.012/83.958)^1 =$	0.51	OK

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = f_n(M_1, M_2, M_0, \psi, \mu)$	0.1, 0.1, 116.8, 0.613, 300.000	1.127	Uniform
$C_{mLT} = 0.95 + 0.05\alpha_h$	$M_h = 0.09, M_s = 116.86, \psi = 0.613, \alpha_s = 0.001$	0.95	Table B.3
$C_{mz} = \text{Max}(0.6 + 0.4\psi, 0.4)$	$M = 0, \psi = 0.000$	0.6	Table B.3
$C_{my} = 0.95 + 0.05\alpha_h$	$M_h = 0, M_s = 116.86, \psi = 1.000, \alpha_s = 0.000$	0.95	Table B.3

Lateral Buckling Check M.b.Rd

$l_e = 1.2L + 2D$	$1.2 \times 5.3 + 2 \times 0.210 =$	6.779 m	
$M_{cr} = F_n(C_1, l_e, I_z, I_y, I_w, E)$	1.127, 6.779, 2068, 47.23, 0.1969, 210000	235.996 kN.m	
$\lambda_{LT} = \sqrt{W_{pl.y}/M_{cr}}$	$\sqrt{656.1 \times 275 / 235.996}$	0.874	
$\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$	0.874, 0.837	0.775	Curve b
$\chi_{LT.mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$	0.775, 0.874, 0.942, 0.971	0.798	6.3.2.3
$M_{b.Rd} = \chi W_{pl.y} \cdot f_y \leq M_{c.y.Rd}$	$0.798 \times 656.1 \times 275 \leq 180.428 =$	143.965 kN.m	

Buckling Resistance

$U_{N.y} = N_{Ed}/(\chi_y \cdot N_{Rk}/\gamma_{M1})$	0 / 1667.819	0.000	OK
$U_{N.z} = N_{Ed}/(\chi_z \cdot N_{Rk}/\gamma_{M1})$	0 / 937.412	0.000	OK
$U_{M.y} = M_{y.Ed}/(\chi_{LT} \cdot M_{y.Rk}/\gamma_{M1})$	128.855 / 143.965	0.895	OK
$U_{M.z} = M_{z.Ed}/(M_{z.Rk}/\gamma_{M1})$	0.012 / 83.958	0.000	OK
$k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N.y}\}$		0.950	
$k_{zz} = C_{mz} \{1 + 1.4 U_{N.z}\}$		0.600	
$k_{yz} = 0.6 k_{zz}$		0.360	
$k_{zy} = 1 - \{0.1 \lambda_z / (C_{mLT} - 0.25)\} U_{N.z}$		1.000	
$U_{Ny} + k_{yy} \cdot U_{M.y} + k_{yz} \cdot U_{M.z}$	$0.000 + 0.950 \times 0.895 + 0.360 \times 0.000$	0.850	OK
$U_{Nz} + k_{zy} \cdot U_{M.y} + k_{zz} \cdot U_{M.z}$	$0.000 + 1.000 \times 0.895 + 0.600 \times 0.000$	0.895	OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams)	$\delta \leq 5300/360 = 14.7$ mm Live (Case 2)	2.27 mm	OK
	$\delta \leq 5300/250 = 21.2$ mm D+L (Case 3)	19.63 mm	OK

Provide Section (59.95 kg/m)

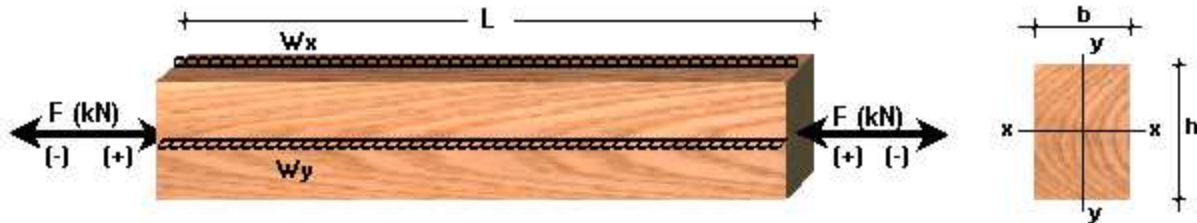
203x203 UC 60 [S 275]



MASTERKEY : TIMBER DESIGN

AXIAL LOAD WITH MOMENT DESIGN TO BS EN 1995-1-1:2004 + A1:2008

Third Floor's Joists



Summary Design Data

Eurocode National Annex	Using UK values
Strength class code	BS EN 338:2009
Design Cases Covered	1.35 D1 + 1.5 L1
Deflection Cases Covered	1.0 L1, 1.0 D1 + 1.0 L1
Section Size	b = 47, h 225 Regularized Section in Strength Class C24
Section Properties (cm ² , cm ³ , cm)	Area 105.8, W _{el,y} 396.6, W _{el,z} 82.8, i _y 6.5, i _z 1.36
Specification	1 : Internal use in continuously heated building
Integrated Design	Long Term loading
Member Details	Critical Case 1 N _{Ed} = 0.0 kN, L _y = 4.1 m, L _z = 4.1 m, L _{cr,y} = 1.0 L _y , L _{cr,z} = 1.0 L _z Bearing length 75, Distance to Bearing 150 mm

Grade and Admissible Stresses (Strength Class C24)

$f_{m,y,d} = K_{mod} \cdot K_{hy} \cdot K_{sys} \cdot f_{m,k} / \gamma_m$	$0.70 \times 1.00 \times 1.00 \times 24.00 / 1.3$	12.92 N/mm ²	
$f_{m,z,d} = K_{mod} \cdot K_{hz} \cdot K_{sys} \cdot f_{m,k} / \gamma_m$	$0.70 \times 1.26 \times 1.00 \times 24.00 / 1.3$	16.30 N/mm ²	
$f_{c,90,d} = K_{mod} \cdot K_{c,90} \cdot K_{sys} \cdot f_{c,90,k} / \gamma_m$	$0.70 \times 1.50 \times 1.00 \times 2.50 / 1.3$	2.02 N/mm ²	
$f_{v,d} = K_{mod} \cdot K_{sys} \cdot f_{v,k} / \gamma_m$	$0.70 \times 1.00 \times 4.00 / 1.3$	2.15 N/mm ²	
E _{mean}	Instantaneous Deflection	11000 N/mm ²	Deflection

Axial Load with Moments Check

Critical Design Location	X = 1.610		
$\sigma_{m,y,d} = M_y / W_{el,y}$	4.889 / 396.56 ≤ 12.92	12.33 N/mm ²	OK
$\sigma_{m,z,d} = M_z / W_{el,z}$	0.022 / 82.84 ≤ 16.30	0.27 N/mm ²	OK
$U_t = \sigma_{c,0,d} / f_{t,0,d}$	0.000 / 7.538	0.000	OK
$U_{m,y} = \sigma_{m,y,d} / f_{m,y,d}$	12.330 / 12.923	0.954	OK
$U_{m,z} = \sigma_{m,z,d} / f_{m,z,d}$	0.270 / 16.299	0.017	OK
$U_t + U_{m,y} + k_m \cdot U_{m,z}$	0.000 + 0.954 + 0.7 × 0.017	0.966	OK
$U_t + k_m \cdot U_{m,y} + U_{m,z}$	0.000 + 0.7 × 0.954 + 0.017	0.684	OK

Shear and Bearing Check

Critical Design Location	X = 0.000		
$\tau_a = 1.5 \sqrt{(V_{y,Ed}^2 + V_{z,Ed}^2)} / \text{Area} / k_{cr}$	$1.5 \sqrt{(5.155^2 + 0.025^2)} / 106 / 0.67 \leq 2.15$	1.09 N/mm ²	OK
$\sigma_{c,\perp ax} = V_{y,Ed} / (b \cdot l_y)$	$5.155 / (47 \times 75) \leq 2.02$	1.46 N/mm ²	OK
$\sigma_{c,\perp ay} = V_{z,Ed} / (h \cdot l_z)$	$0.025 / (225 \times 75) \leq 2.02$	0.00 N/mm ²	OK

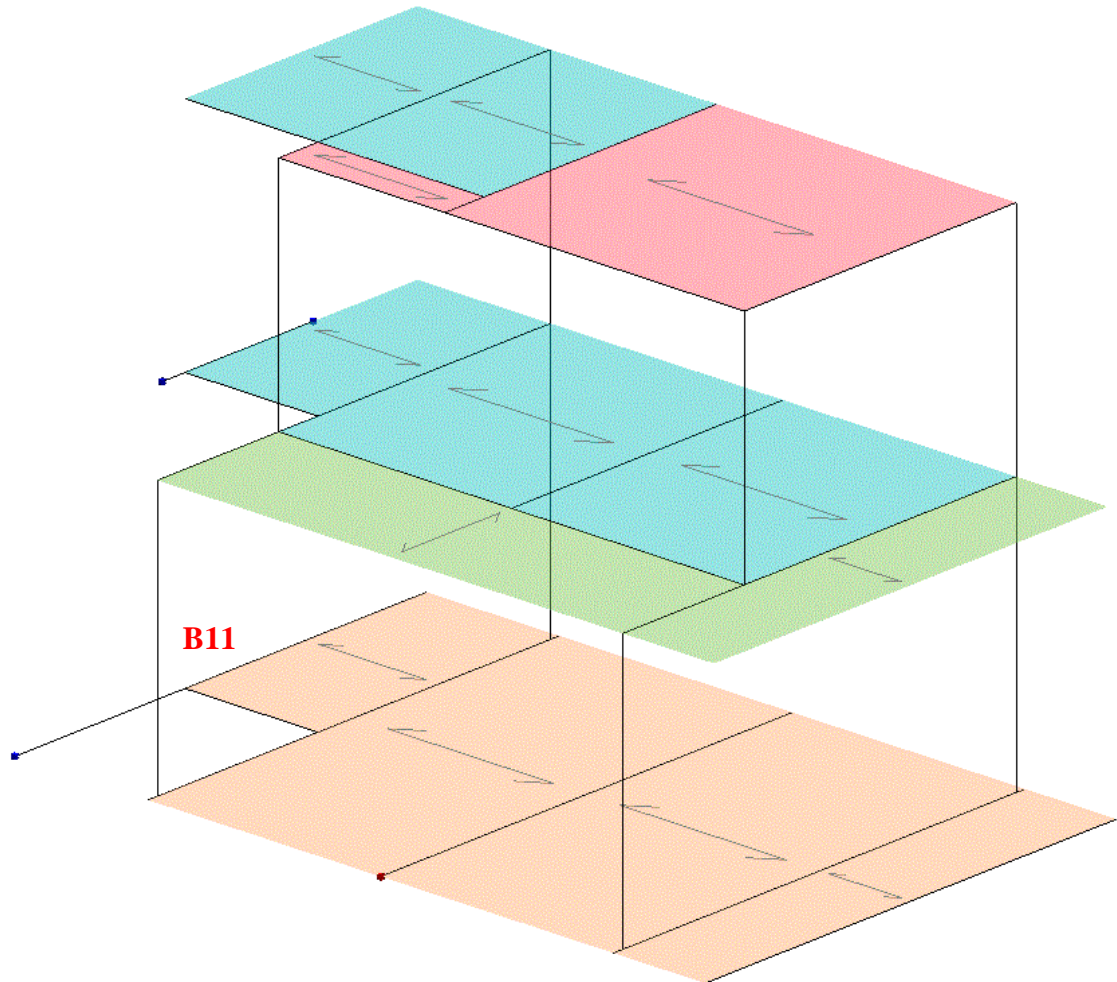
Deflection Check (Shear Deflection NOT Included)

Critical Load Case 003 : Dead Plus Live (Serviceability)			
$\delta = \delta_m$	$12.06 \leq L / 250$	12.06 mm	OK

Provide C24 Timber 47x225mm deep floor joists at 400mm centres



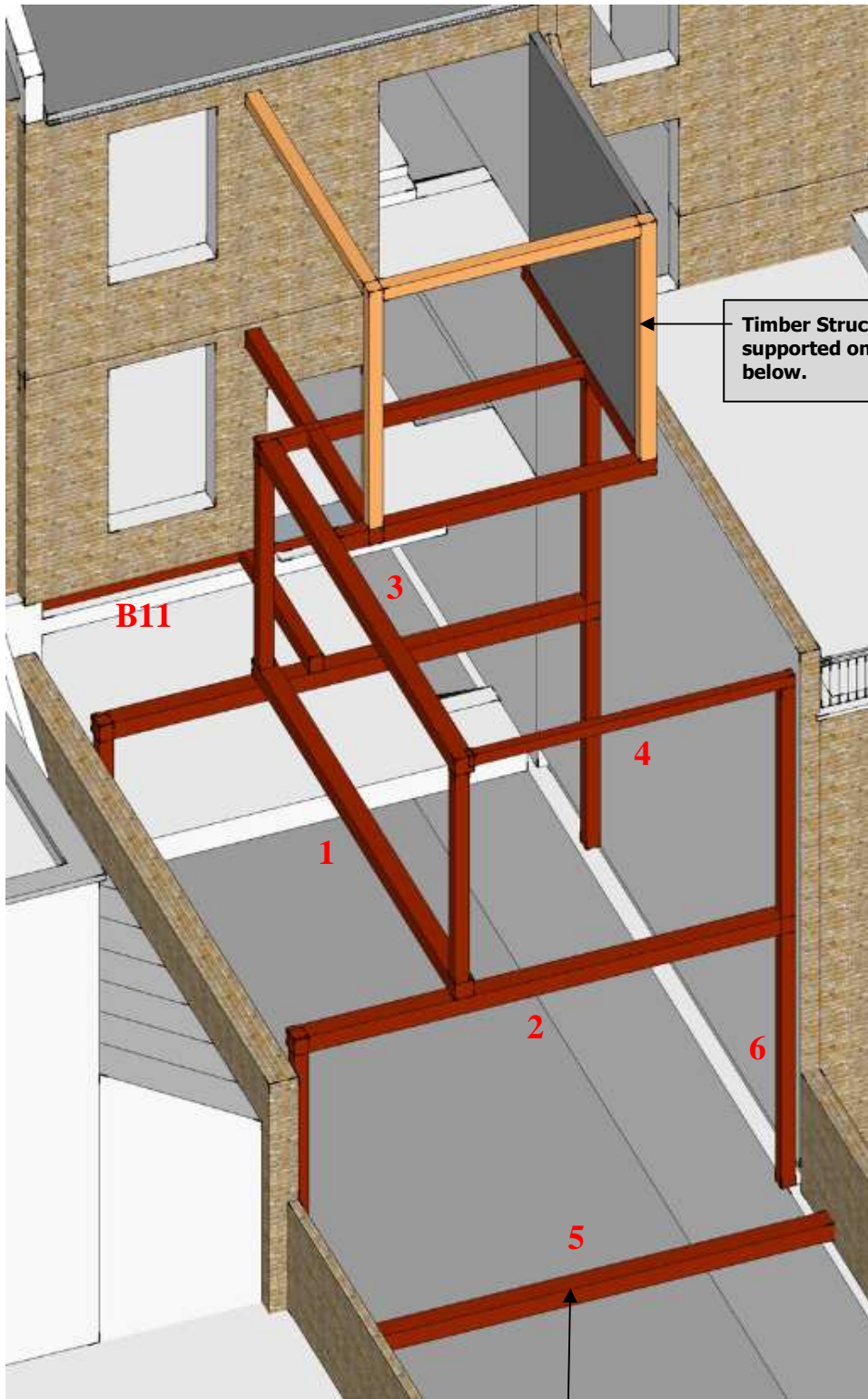
Isometric view of the Rear Extension's frame showing Area Loadings



Dead D	Live L
2.600	1.500
0.500	1.500
1.000	1.500
0.600	0.750



Isometric view of the Rear Extension's frame showing Steel Sections



Timber Structure to be supported on the steelwork below.

Numbered members' detailed designs are shown at the following pages.



AXIAL WITH MOMENTS (MEMBER)

1

Members 42-43 (N.23-N.25) @ Level 1 in Load Case 1

Member Loading and Member Forces

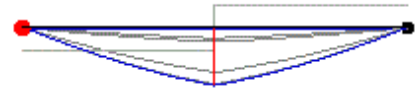
Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

Part 1

D1 UDLY -000.604 (kN/m)
L1 UDLY -000.604 (kN/m)
D1 D 078.500 (kN/m³)
L1 UDLY -002.900 (kN/m)

Part 2

D1 UDLY -000.604 (kN/m)
L1 UDLY -000.604 (kN/m)
D1 D 078.500 (kN/m³)
L1 UDLY -002.900 (kN/m)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 9										
Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)
				y-y	z-z	y-y	z-z	y-y	z-z	
	23	0.88T	27.54	34.11	0.14	-1.46	-0.01	67.80	0.38	13.69
	27	0.88T	27.54	-33.98	-0.13	-1.11	0.00	@ 2.800	@ 2.800	@ 2.800

Classification and Effective Area (EN 1993: 2006)

Section (46.1 kg/m) 203x203 UC 46 [S 275]
Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$ 9.25, 22.33, 275, 0, 67.8, 0.38 (Axial: Non-Slender) Class 1
Auto Design Load Cases 1 and 4-7

Local Capacity Check

$V_{y,Ed}/V_{pl,y,Rd}$ 15.366 / 269.498 = 0.057 Low Shear
 $M_{c,y,Rd} = f_y \cdot W_{pl,y} / \gamma_{M0}$ 275 x 497.4/1 136.785 kN.m
 $V_{z,Ed}/V_{pl,z,Rd}$ 0.137 / 711.169 = 0 Low Shear
 $M_{c,z,Rd} = f_y \cdot W_{pl,z} / \gamma_{M0}$ 275 x 230.9/1 63.498 kN.m
 $N_{pl,Rd} = A_g \cdot f_y / \gamma_{M0}$ 58.73 x 275/1 1615.075 kN
 $n = N_{Ed}/N_{pl,Rd}$ -0.877 / 1615.075 = 0.001 OK
 $W_{pl,N,y} = F_n(W_{pl,y}, A_{vy}, n)$ 497.4, 16.974, 0.001 497.4 cm³
 $M_{N,y,Rd} = W_{pl,N,y} \cdot f_y / \gamma_{M0}$ 497.4 x 275/1 136.785 kN.m
 $W_{pl,N,z} = F_n(W_{pl,z}, A_{vz}, n)$ 230.9, 44.792, 0.001 230.9 cm³
 $M_{N,z,Rd} = W_{pl,N,z} \cdot f_y / \gamma_{M0}$ 230.9 x 275/1 63.498 kN.m
 $(M_{y,Ed}/M_{N,y,Rd}) + (M_{z,Ed}/M_{N,z,Rd})$ $(67.804/136.785)^2 + (0.376/63.498)^2 = 0.252$ OK

Lateral Buckling Check M.b.Rd

$M_{b,Rd} = M_{c,y,Rd}$ Fully Restrained 136.785 kN.m

Buckling Resistance

$U_{N,y} = N_{Ed}/(\chi_y \cdot N_{Rk}/\gamma_{M1})$ 0 / 1236.682 0.000 OK
 $U_{N,z} = N_{Ed}/(\chi_z \cdot N_{Rk}/\gamma_{M1})$ 0 / 1615.075 0.000 OK
 $U_{M,y} = M_{y,Ed}/(\chi_{LT} \cdot M_{y,Rk}/\gamma_{M1})$ 67.804 / 136.785 0.496 OK
 $U_{M,z} = M_{z,Ed}/(M_{z,Rk}/\gamma_{M1})$ 0.376 / 63.498 0.006 OK
 $k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N,y}\}$ 0.898
 $k_{zz} = C_{mz} \{1 + (2\lambda_z - 0.6) U_{N,z}\}$ 0.901
 $k_{yz} = 0.6 k_{zz}$ 0.541
 $k_{zy} = 0.6 k_{yy}$ 0.539
 $U_{Ny} + k_{yy} \cdot U_{M,y} + k_{yz} \cdot U_{M,z}$ 0.000 + 0.898 x 0.496 + 0.541 x 0.006 0.448 OK
 $U_{Nz} + k_{zy} \cdot U_{M,y} + k_{zz} \cdot U_{M,z}$ 0.000 + 0.539 x 0.496 + 0.901 x 0.006 0.272 OK

Deflection Check - Load Case 2

Deflection Limits (Internal Beams) $\delta \leq 5600/360 = 15.6$ mm Live (Case 2) 10.14 mm OK
 $\delta \leq 5600/250 = 22.4$ mm D+L (Case 3) 13.69 mm OK

Provide Section (46.1 kg/m)

203x203 UC 46 [S 275]



AXIAL WITH MOMENTS (MEMBER)

2

Members 15-16 (N.20-N.23) @ Level 1 in Load Case 1

Member Loading and Member Forces

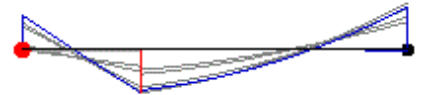
Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

Part 1

D1 UDLY -000.413 (kN/m)
L1 UDLY -000.413 (kN/m)
D1 D 078.500 (kN/m³)

Part 2

D1 UDLY -000.413 (kN/m)
L1 UDLY -000.413 (kN/m)
D1 D 078.500 (kN/m³)
D1 UDLY -002.900 (kN/m)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 9										
Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)
				y-y	z-z	y-y	z-z	y-y	z-z	
	20	23.90C	0.02	58.43	-0.01	-40.81	0.00	50.03	-0.02	5.80
	24	11.43C	0.00	-44.88	0.00	-50.63	0.01	@ 1.594	@ 1.530	@ 2.222

Classification and Effective Area (EN 1993: 2006)

Section (46.1 kg/m) 203x203 UC 46 [S 275]
Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$ 9.25, 22.33, 275, 23.9, 50.93, 0.02 (Axial: Non-Slender) Class 1
Auto Design Load Cases 1 and 4-7

Local Capacity Check

$V_{y,Ed}/V_{pl,y,Rd}$ 55.532 / 269.498 = 0.206 Low Shear
 $M_{c,y,Rd} = f_y \cdot W_{pl,y} / \gamma_{M0}$ 275 x 497.4/1 136.785 kN.m
 $V_{z,Ed}/V_{pl,z,Rd}$ 0.007 / 711.169 = 0 Low Shear
 $M_{c,z,Rd} = f_y \cdot W_{pl,z} / \gamma_{M0}$ 275 x 230.9/1 63.498 kN.m
 $N_{pl,Rd} = A_g \cdot f_y / \gamma_{M0}$ 58.73 x 275/1 = 1615.075 kN
 $n = N_{Ed}/N_{pl,Rd}$ 11.432 / 1615.075 = 0.007 OK
 $W_{pl,N,y} = F_n(W_{pl,y}, A_{vy}, n)$ 497.4, 16.974, 0.007 497.4 cm³
 $M_{N,y,Rd} = W_{pl,N,y} \cdot f_y / \gamma_{M0}$ 497.4 x 275/1 136.785 kN.m
 $W_{pl,N,z} = F_n(W_{pl,z}, A_{vz}, n)$ 230.9, 44.792, 0.007 230.9 cm³
 $M_{N,z,Rd} = W_{pl,N,z} \cdot f_y / \gamma_{M0}$ 230.9 x 275/1 63.498 kN.m
 $(M_{y,Ed}/M_{N,y,Rd}) + (M_{z,Ed}/M_{N,z,Rd}) = (50.927/136.785)^2 + (0.016/63.498)^1 = 0.139$ OK

Compression Resistance N.b.Rd

$\lambda_y = \sqrt{A \cdot f_y / N_{cr}}$ $\sqrt{58.73 \times 275 / 3490.01}$ 0.68
 $N_{b,y,Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 58.73 x 0.795 x 275 / 10 / 1 = 1283.423 kN Curve b
 $\lambda_z = \sqrt{A \cdot f_y / N_{crz}}$ $\sqrt{58.73 \times 275 / 1184.36}$ 1.168
 $N_{b,z,Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 58.73 x 0.45 x 275 / 10 / 1 = 726.039 kN Curve c
 $l_{et} = K_t \cdot L_x$ 1 x 5.21 = 5.21
 $\lambda_T = \sqrt{A \cdot f_y / N_{crT}}$ $\sqrt{58.73 \times 275 / 2764.03}$ 0.764
 $N_{b,T,Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 58.73 x 0.685 x 275 / 10 / 1 = 1105.527 kN Curve c

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = f_n(M_1, M_2, M_0, \psi, \mu)$ -40.8, -50.6, 78.7, 0.806, -1.555 1.943 Uniform
 $C_{mLT} = \text{Max}(0.1 - 0.8a_s, 0.4)$ $M_h = -50.59, M_s = 33, \psi = 0.806, a_s = -0.652$ 0.622 Table B.3
 $C_{mz} = \text{Max}(0.6 + 0.4\psi, 0.4)$ $M = 0.01, \psi = -0.714$ 0.4 Table B.3
 $C_{my} = \text{Max}(0.1 - 0.8a_s, 0.4)$ $M_h = -50.63, M_s = 33, \psi = 0.806, a_s = -0.652$ 0.621 Table B.3

Lateral Buckling Check M.b.Rd

$l_e = 1.2L + 2D$ 1.2 x 5.21 + 2 x 0.203 = 6.658 m
 $M_{cr} = F_n(C_1, l_e, I_z, I_w, E)$ 1.943, 6.658, 1551, 22.15, 0.1429, 210000 259.392 kN.m
 $\lambda_{LT} = \sqrt{W_{pl,y} / M_{cr}}$ $\sqrt{497.4 \times 275 / 259.392}$ 0.726
 $\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$ 0.726, 0.913 0.856 Curve b
 $\chi_{LT,mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$ 0.856, 0.726, 0.717, 0.860 0.996 6.3.2.3
 $M_{b,Rd} = \chi \cdot W_{pl,y} \cdot f_y \leq M_{c,y,Rd}$ 0.996 x 497.4 x 275 \leq 136.785 = 136.174 kN.m



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Job ref : L15/088/04
Sheet : **Structure / 37 -**
Made By : Chris Papadakis
Date : **July 2015**
Revision : **P3**

Buckling Resistance

$U_{N,y} = N_{Ed}/(\chi_y \cdot N_{Rk}/\gamma_{M1})$	23.898 / 1283.423	0.019	OK
$U_{N,z} = N_{Ed}/(\chi_z \cdot N_{Rk}/\gamma_{M1})$	23.898 / 726.039	0.033	OK
$U_{M,y} = M_{y,Ed}/(\chi_{LT} \cdot M_{y,Rk}/\gamma_{M1})$	50.927 / 136.174	0.374	OK
$U_{M,z} = M_{z,Ed}/(M_{z,Rk}/\gamma_{M1})$	0.016 / 63.498	0.000	OK
$k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N,y}\}$		0.627	
$k_{zz} = C_{mz} \{1 + 1.4 U_{N,z}\}$		0.418	
$k_{yz} = 0.6 k_{zz}$		0.251	
$k_{zy} = 1 - \{0.1 \lambda_z / (C_{mLT} - 0.25)\} U_{N,z}$		0.990	
$U_{Ny} + k_{yy} \cdot U_{M,y} + k_{yz} \cdot U_{M,z}$	0.019 + 0.627 x 0.374 + 0.251 x 0.000	0.253	OK
$U_{Nz} + k_{zy} \cdot U_{M,y} + k_{zz} \cdot U_{M,z}$	0.033 + 0.990 x 0.374 + 0.418 x 0.000	0.403	OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams)	$\delta \leq 5210/360 = 14.5$ mm Live (Case 2)	2.93 mm	OK
	$\delta \leq 5210/250 = 20.8$ mm D+L (Case 3)	5.59 mm	OK

Provide Section (46.1 kg/m)

203x203 UC 46 [S 275]



AXIAL WITH MOMENTS (MEMBER)

3

Members 39-40 (N.34-N.36) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

Part 1

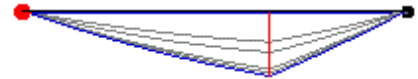
 D1 D 078.500 (kN/m³)

D1 UDLY -001.500 (kN/m)

Part 2

 D1 D 078.500 (kN/m³)

D1 UDLY -001.500 (kN/m)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 9											
Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)	
				y-y	z-z	y-y	z-z	y-y	z-z		
	34	0.87C	0.00	17.87	0.12	-0.77	-0.01	47.81	0.45	15.17	
	37	0.84C	0.00	-26.88	-0.23	-1.08	-0.01	@ 3.600	@ 3.600	@ 2.952	

Classification and Effective Area (EN 1993: 2006)

 Section (29.99 kg/m) 203x133 UB 30 [S 275]
 Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$ 6.97, 26.94, 275, 0.87, 47.82, 0.45 (Axial: Non-Slender) Class 1
 Auto Design Load Cases 1 and 4-7

Local Capacity Check

 $V_{y.Ed}/V_{pl.y.Rd}$ 22.016 / 231.406 = 0.095 Low Shear
 $M_{C.y.Rd} = f_y \cdot W_{pl.y} / \gamma_{M0}$ 275 x 314.4 / 1 86.46 kN.m
 $V_{z.Ed}/V_{pl.z.Rd}$ 0.229 / 408.182 = 0.001 Low Shear
 $M_{C.z.Rd} = f_y \cdot W_{pl.z} / \gamma_{M0}$ 275 x 88.2 / 1 24.255 kN.m
 $N_{pl.Rd} = A_g \cdot f_y / \gamma_{M0}$ 38.21 x 275 / 1 = 1050.775 kN
 $n = N_{Ed} / N_{pl.Rd}$ 0.842 / 1050.775 = 0.001 OK
 $W_{pl.N.y} = F_n(W_{pl.y}, A_{vy}, n)$ 314.4, 14.575, 0.001 314.4 cm³
 $M_{N.y.Rd} = W_{pl.N.y} \cdot f_y / \gamma_{M0}$ 314.4 x 275 / 1 86.46 kN.m
 $W_{pl.N.z} = F_n(W_{pl.z}, A_{vz}, n)$ 88.2, 25.709, 0.001 88.2 cm³
 $M_{N.z.Rd} = W_{pl.N.z} \cdot f_y / \gamma_{M0}$ 88.2 x 275 / 1 24.255 kN.m
 $(M_{y.Ed} / M_{N.y.Rd}) + (M_{z.Ed} / M_{N.z.Rd})$ $(47.809 / 86.46)^2 + (0.448 / 24.255)^2 = 0.324$ OK

Compression Resistance N.b.Rd

 $\lambda_y = \sqrt{A \cdot f_y / N_{cr}}$ $\sqrt{38.21 \times 275 / 1914.33}$ 0.741
 $N_{b.y.Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 38.21 x 0.828 x 275 / 10 / 1 = 869.891 kN Curve a
 $\lambda_z = \sqrt{A \cdot f_y / N_{crz}}$ $\sqrt{38.21 \times 275 / 254.78}$ 2.028
 $N_{b.z.Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 38.21 x 0.204 x 275 / 10 / 1 = 214.622 kN Curve b
 $Let = Kt \cdot Lx$ 1 x 5.6 = 5.6
 $\lambda_T = \sqrt{A \cdot f_y / N_{crT}}$ $\sqrt{38.21 \times 275 / 1255.13}$ 0.915
 $N_{b.T.Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 38.21 x 0.652 x 275 / 10 / 1 = 684.626 kN Curve b

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

 $C_1 = f_n(M_1, M_2, M_0, \psi, \mu)$ -0.7, -1.1, 40.6, 0.708, -38.493 1.196 Uniform
 $C_{mLT} = 0.95 + 0.05 a_h$ $M_h = -1.06, M_s = 39.75, \psi = 0.708, a_s = -0.027$ 0.949 Table B.3
 $C_{mz} = 0.95 + 0.05 a_h$ $M_h = -0.01, M_s = 0.34, \psi = 0.700, a_s = -0.030$ 0.949 Table B.3
 $C_{my} = 0.95 + 0.05 a_h$ $M_h = -1.08, M_s = 39.75, \psi = 0.707, a_s = -0.027$ 0.949 Table B.3

Lateral Buckling Check M.b.Rd

 $Le = 1.00 L$ 1 x 5.6 = 5.6 m
 $M_{cr} = F_n(C_1, Le, I_z, I_y, I_w, E)$ 1.196, 5.600, 385.5, 10.3, 0.03734, 210000 62.687 kN.m
 $\lambda_{LT} = \sqrt{W_{pl.y} / M_{cr}}$ $\sqrt{314.4 \times 275 / 62.687}$ 1.174
 $\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$ 1.174, 1.237 0.594 Curve b
 $\chi_{LT.mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$ 0.594, 1.174, 0.915, 0.969 0.613 6.3.2.3
 $M_{b.Rd} = \chi \cdot W_{pl.y} \cdot f_y \leq M_{C.y.Rd}$ 0.613 x 314.4 x 275 $\leq 86.460 = 52.997$ kN.m

Buckling Resistance

 $U_{N.y} = N_{Ed} / (\chi_y \cdot N_{Rk} / \gamma_{M1})$ 0.868 / 869.891 0.001 OK



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$U_{N,z} = N_{Ed}/(\chi_z \cdot N_{Rk}/\gamma_{M1})$	0.868 / 214.622	0.004	OK
$U_{M,y} = M_{y,Ed}/(\chi_{LT} \cdot M_{y,Rk}/\gamma_{M1})$	47.809 / 52.997	0.902	OK
$U_{M,z} = M_{z,Ed}/(M_{z,Rk}/\gamma_{M1})$	0.448 / 24.255	0.018	OK
$k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N,y}\}$		0.949	
$k_{zz} = C_{mz} \{1 + 1.4 U_{N,z}\}$		0.954	
$k_{yz} = 0.6 k_{zz}$		0.572	
$k_{zy} = 1 - \{0.1 \lambda_z / (C_{mLT} - 0.25)\} U_{N,z}$		0.999	
$U_{Ny} + k_{yy} \cdot U_{M,y} + k_{yz} \cdot U_{M,z}$	0.001 + 0.949x0.902 + 0.572x0.018	0.868	OK
$U_{Nz} + k_{zy} \cdot U_{M,y} + k_{zz} \cdot U_{M,z}$	0.004 + 0.999x0.902 + 0.954x0.018	0.923	OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams)	$\delta \leq 5600/360 = 15.6$ mm Live (Case 2)	4.88 mm	OK
	$\delta \leq 5600/250 = 22.4$ mm D+L (Case 3)	15.17 mm	OK

Provide Section (29.99 kg/m)

203x133 UB 30 [S 275]



AXIAL WITH MOMENTS (MEMBER)

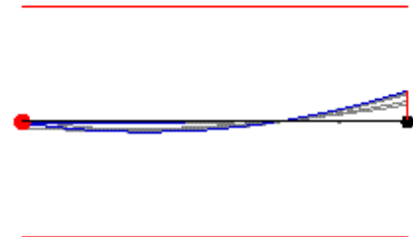
4

Member 19 (N.34-N.35) @ Level 1 in Load Case 6

Member Loading and Member Forces

Loading Combination : 1 UT + 1.25 D1 + 1.5 L1 + 0.75 W1

D1 UDLY -001.800 (kN/m)
L1 UDLY -002.700 (kN/m)
D1 D 078.500 (kN/m³)
D1 UDLY -001.500 (kN/m)



Member Forces in Load Case 6 and Maximum Deflection from Load Case 9											
Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)	
				y-y	z-z	y-y	z-z	y-y	z-z		
19	34	12.83C	0.00	9.82	0.00	0.50	0.01	6.18		0.64	
	35	12.83C	0.00	-20.74	0.00	-19.15	0.00	@ 1.152		@ 1.332	

Classification and Effective Area (EN 1993: 2006)

Section (25.09 kg/m) 203x133 UB 25 [S 275]
Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$ 8.54, 30.25, 275, 12.83, 19.15, 0.01 (Axial: Non-Slender) Class 1
Auto Design Load Cases 1 and 4-7

Local Capacity Check

$V_{y.Ed}/V_{pl.y.Rd}$ 20.738 / 203.402 = 0.102 Low Shear
 $M_{c.y.Rd} = f_y \cdot W_{pl.y} / \gamma_{M0}$ 275 x 257.7/1 70.868 kN.m
 $V_{z.Ed}/V_{pl.z.Rd}$ 0.003 / 329.914 = 0 Low Shear
 $M_{c.z.Rd} = f_y \cdot W_{pl.z} / \gamma_{M0}$ 275 x 70.9/1 19.498 kN.m
 $N_{pl.Rd} = A_g \cdot f_y / \gamma_{M0}$ 31.96 x 275/1 = 878.9 kN
 $n = N_{Ed}/N_{pl.Rd}$ 12.827 / 878.9 = 0.015 OK
 $W_{pl.N.y} = F_n(W_{pl.y}, A_{vy}, n)$ 257.7, 12.811, 0.015 257.7 cm³
 $M_{N.y.Rd} = W_{pl.N.y} \cdot f_y / \gamma_{M0}$ 257.7 x 275/1 70.868 kN.m
 $W_{pl.N.z} = F_n(W_{pl.z}, A_{vz}, n)$ 70.9, 20.779, 0.015 70.9 cm³
 $M_{N.z.Rd} = W_{pl.N.z} \cdot f_y / \gamma_{M0}$ 70.9 x 275/1 19.498 kN.m
 $(M_{y.Ed}/M_{N.y.Rd}) + (M_{z.Ed}/M_{N.z.Rd})$ $(19.15/70.868)^2 + (0.003/19.498)^2 = 0.073$ OK

Compression Resistance N.b.Rd

$\lambda_y = \sqrt{A \cdot f_y / N_{cr}}$ $\sqrt{31.96 \times 275 / 3743.98}$ 0.484
 $N_{b.y.Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ $31.96 \times 0.929 \times 275 / 10/1 = 816.529$ kN Curve a
 $\lambda_z = \sqrt{A \cdot f_y / N_{crz}}$ $\sqrt{31.96 \times 275 / 493.37}$ 1.333
 $N_{b.z.Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ $31.96 \times 0.411 \times 275 / 10/1 = 361.352$ kN Curve b
 $Let = Kt \cdot Lx$ $1 \times 3.6 = 3.6$
 $\lambda_T = \sqrt{A \cdot f_y / N_{crT}}$ $\sqrt{31.96 \times 275 / 1146.18}$ 0.876
 $N_{b.T.Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ $31.96 \times 0.677 \times 275 / 10/1 = 594.828$ kN Curve b

Equivalent Uniform Moment Factors C1, C.mLT, C.mz, and C.my

$C_1 = fn(M_1, M_2, M_0, \psi, \mu)$ 0.5, -19.1, 13.7, -0.027, -0.718 3.737 Uniform
 $C_{mLT} = \text{Max}(0.1(1-\psi) - 0.8\alpha_s, 0.4)$ $M_h = -19.13, M_s = 4.42, \psi = -0.027, \alpha_s = -0.231$ 0.4 Table B.3
 $C_{mz} = \text{Max}(0.6 + 0.4\psi, 0.4)$ $M = 0.01, \psi = -0.500$ 0.4 Table B.3
 $C_{my} = \text{Max}(0.1(1-\psi) - 0.8\alpha_s, 0.4)$ $M_h = -19.15, M_s = 4.42, \psi = -0.026, \alpha_s = -0.231$ 0.4 Table B.3

Lateral Buckling Check M.b.Rd

$Le = 1.2L + 2D$ $1.2 \times 3.6 + 2 \times 0.203 = 4.726$ m
 $M_{cr} = Fn(C_1, Le, I_z, I_y, I_w, E)$ 3.737, 4.726, 308.5, 5.964, 0.02933, 210000 173.568 kN.m
 $\lambda_{LT} = \sqrt{W_{pl.y} / M_{cr}}$ $\sqrt{257.7 \times 275 / 173.568}$ 0.639
 $\chi_{LT} = Fn(\lambda_{LT}, \lambda_{LT5950})$ 0.639, 1.189 0.899 Curve b
 $\chi_{LT.mod} = Fn(\chi_{LT}, \lambda_{LT}, k_c, f)$ 0.899, 0.639, 0.517, 0.771 1.000 6.3.2.3
 $M_{b.Rd} = \chi W_{pl.y} \cdot f_y \leq M_{c.y.Rd}$ $1.000 \times 257.7 \times 275 \leq 70.868 = 70.868$ kN.m

Buckling Resistance

$U_{N,y} = N_{Ed} / (\chi_y \cdot N_{Rk} / \gamma_{M1})$ 12.827 / 816.529 0.016 OK



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$U_{N,z} = N_{Ed}/(\chi_z \cdot N_{Rk}/\gamma_{M1})$	12.827 / 361.352	0.035	OK
$U_{M,y} = M_{y,Ed}/(\chi_{LT} \cdot M_{y,Rk}/\gamma_{M1})$	19.15 / 70.868	0.270	OK
$U_{M,z} = M_{z,Ed}/(M_{z,Rk}/\gamma_{M1})$	0.003 / 19.498	0.000	OK
$k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N,y}\}$		0.402	
$k_{zz} = C_{mz} \{1 + 1.4 U_{N,z}\}$		0.420	
$k_{yz} = 0.6 k_{zz}$		0.252	
$k_{zy} = 0.6 k_{yy}$		0.241	
$U_{Ny} + k_{yy} \cdot U_{M,y} + k_{yz} \cdot U_{M,z}$	0.016 + 0.402 x 0.270 + 0.252 x 0.000	0.124	OK
$U_{Nz} + k_{zy} \cdot U_{M,y} + k_{zz} \cdot U_{M,z}$	0.035 + 0.241 x 0.270 + 0.420 x 0.000	0.101	OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams)	$\delta \leq 3600/360 = 10 \text{ mm Live (Case 2)}$	0.19 mm	OK
	$\delta \leq 3600/250 = 14.4 \text{ mm D+L (Case 3)}$	0.54 mm	OK

Provide Section (25.09 kg/m)

203x133 UB 25 [S 275]



AXIAL WITH MOMENTS (MEMBER)

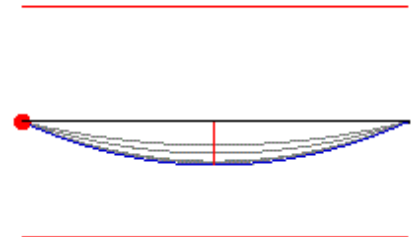
5

Member 27 (N.3-N.4) @ Level 1 in Load Case 1

Member Loading and Member Forces

Loading Combination : 1 UT + 1.35 D1 + 1.5 L1

D1 UDLY -001.430 (kN/m)
L1 UDLY -000.825 (kN/m)
D1 D 078.500 (kN/m³)
D1 UDLY -002.950 (kN/m)
L1 UDLY -000.750 (kN/m)



Member Forces in Load Case 1 and Maximum Deflection from Load Case 3											
Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)	
				y-y	z-z	y-y	z-z	y-y	z-z		
27	3	0.00C	0.00	23.65	0.00	0.00	0.00	32.23		11.81 @ 2.725	
	4	0.00C	0.00	-23.65	0.00	0.00	0.00	@ 2.725			

Classification and Effective Area (EN 1993: 2006)

Section (29.99 kg/m) 203x133 UB 30 [S 275]
Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$ 6.97, 26.94, 275, 0, 32.22, 0 (Axial: Non-Slender) Class 1
Auto Design Load Cases 1 and 4-7

Moment Capacity Check M.c.y.Rd

$V_{y.Ed}/V_{pl.y.Rd}$ 0.003 / 231.406 = 0 Low Shear
 $M_{c.y.Rd} = f_y \cdot W_{pl.y} / \gamma_{M0}$ 275 x 314.4 / 1 86.46 kN.m
 $M_{y.Ed}/M_{c.y.Rd}$ 32.224 / 86.46 = 0.373 OK

Equivalent Uniform Moment Factor C1

$C_1 = f_n(M_1, M_2, M_o, \psi, \mu)$ 0.0, 0.0, 32.2, 0.875, 300.000 1.127 Uniform

Lateral Buckling Check M.b.Rd

$l_e = 1.2L + 2D$ 1.2 x 5.45 + 2 x 0.207 = 6.954 m
 $M_{cr} = F_n(C_1, l_e, I_z, I_t, I_w, E)$ 1.127, 6.954, 385.5, 10.3, 0.03734, 210000 45.638 kN.m
 $\lambda_{LT} = \sqrt{W_{pl.y} / M_{cr}}$ $\sqrt{314.4 \times 275 / 45.638}$ 1.376
 $\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$ 1.376, 1.408 0.484 Curve b
 $\chi_{LT.mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$ 0.484, 1.376, 0.942, 0.990 0.489 6.3.2.3
 $M_{b.Rd} = \chi W_{pl.y} \cdot f_y \leq M_{c.y.Rd}$ 0.489 x 314.4 x 275 \leq 86.460 = 42.288 kN.m
 $M_{y.Ed}/M_{b.Rd}$ 32.224 / 42.288 0.762 OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams) $\delta \leq 5450/360 = 15.1$ mm Live (Case 2) 2.97 mm OK
 $\delta \leq 5450/250 = 21.8$ mm D+L (Case 3) 11.81 mm OK

Provide Section (29.99 kg/m)

203x133 UB 30 [S 275]



AXIAL WITH MOMENTS (MEMBER)

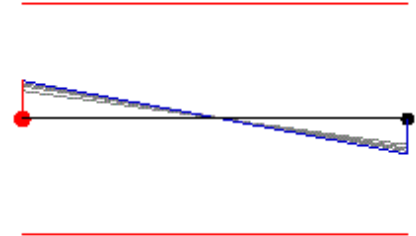
6

Member 59 (N.7-N.24) @ Level 1 in Load Case 6

Member Loading and Member Forces

Loading Combination : 1 UT + 1.25 D1 + 1.5 L1 + 0.75 W1

D1 D 078.500 (kN/m³)



Member Forces in Load Case 6 and Maximum Deflection from Load Case 9											
Mem ber No.	Node End1 End2	Axial Force (kN)	Torque Moment (kN.m)	Shear Force (kN)		Bending Moment (kN.m)		Maximum Moment (kN.m @ m)		Maximum Deflection (mm @ m)	
				y-y	z-z	y-y	z-z	y-y	z-z		
59	7	70.27C	0.00	28.78	0.00	-44.83	0.00		0.00	0.59 @ 0.720	
	24	68.55C	0.00	28.78	0.00	41.52	-0.01	@ 0.000			

Classification and Effective Area (EN 1993: 2006)

Section (46.1 kg/m) 203x203 UC 46 [S 275]
Class = $F_n(b/T, d/t, f_y, N, M_y, M_z)$ 9.25, 22.33, 275, 70.27, 44.83, 0.01 (Axial: Non-Slender) Class 1
Auto Design Load Cases 1 and 4-7

Local Capacity Check

$V_{y.Ed}/V_{pl.y.Rd}$ 28.784 / 269.498 = 0.107 Low Shear
 $M_{c.y.Rd} = f_y \cdot W_{pl.y} / \gamma_{M0}$ 275 x 497.4/1 136.785 kN.m
 $V_{z.Ed}/V_{pl.z.Rd}$ 0.001 / 711.169 = 0 Low Shear
 $M_{c.z.Rd} = f_y \cdot W_{pl.z} / \gamma_{M0}$ 275 x 230.9/1 63.498 kN.m
 $N_{pl.Rd} = A_g \cdot f_y / \gamma_{M0}$ 58.73 x 275/1 = 1615.075 kN
 $n = N_{Ed}/N_{pl.Rd}$ 68.546 / 1615.075 = 0.042 OK
 $W_{pl.N.y} = F_n(W_{pl.y}, A_{vy}, n)$ 497.4, 16.974, 0.042 497.4 cm³
 $M_{N.y.Rd} = W_{pl.N.y} \cdot f_y / \gamma_{M0}$ 497.4 x 275/1 136.785 kN.m
 $W_{pl.N.z} = F_n(W_{pl.z}, A_{vz}, n)$ 230.9, 44.792, 0.042 230.9 cm³
 $M_{N.z.Rd} = W_{pl.N.z} \cdot f_y / \gamma_{M0}$ 230.9 x 275/1 63.498 kN.m
 $(M_{y.Ed}/M_{N.y.Rd}) + (M_{z.Ed}/M_{N.z.Rd})^2 + (0.004/63.498)^2 = 0.107$ OK

Compression Resistance N.b.Rd

$\lambda_y = \sqrt{A \cdot f_y / N_{cr}}$ $\sqrt{58.73 \times 275 / 10525.9}$ 0.392
 $N_{b.y.Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 58.73 x 0.929 x 275 / 10/1 = 1500.914 kN Curve b
 $\lambda_z = \sqrt{A \cdot f_y / N_{crz}}$ $\sqrt{58.73 \times 275 / 3572.04}$ 0.672
 $N_{b.z.Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 58.73 x 0.742 x 275 / 10/1 = 1197.991 kN Curve c
 $Let = Kt \cdot Lx$ 1x3 = 3
 $\lambda_T = \sqrt{A \cdot f_y / N_{crT}}$ $\sqrt{58.73 \times 275 / 4874.8}$ 0.576
 $N_{b.T.Rd} = Area \cdot \chi \cdot f_y / \gamma_{M1}$ 58.73 x 0.8 x 275 / 10/1 = 1291.661 kN Curve c

Equivalent Uniform Moment Factor C1

$C_1 = f_n(M_1, M_2, \psi)$ -44.8, 41.5, -0.926 2.550 Not Loaded
 $C_{mLT} = \text{Max}(0.6 + 0.4\psi, 0.4)$ M = -44.81, $\psi = -0.926$ 0.4 Table B.3
 $C_{m2} = \text{Max}(0.6 + 0.4\psi, 0.4)$ M = -0.01, $\psi = 0.500$ 0.8 Table B.3
 $C_{my} = \text{Max}(0.6 + 0.4\psi, 0.4)$ M = -44.83, $\psi = -0.926$ 0.4 Table B.3

Lateral Buckling Check M.b.Rd

$Le = 1.2L + 2D$ 1.2 x 3 + 2 x 0.203 = 4.006 m
 $M_{cr} = F_n(C_1, Le, I_z, I_t, I_w, E)$ 2.550, 4.006, 1551, 22.15, 0.1429, 210000 687.999 kN.m
 $\lambda_{LT} = \sqrt{W_{pl.y} / M_{cr}}$ $\sqrt{497.4 \times 275 / 687.999}$ 0.446
 $\chi_{LT} = F_n(\lambda_{LT}, \lambda_{LT5950})$ 0.446, 0.642 0.982 Curve b
 $\chi_{LT.mod} = F_n(\chi_{LT}, \lambda_{LT}, k_c, f)$ 0.982, 0.446, 0.626, 0.860 1.000 6.3.2.3
 $M_{b.Rd} = \chi \cdot W_{pl.y} \cdot f_y \leq M_{c.y.Rd}$ 1.000 x 497.4 x 275 \leq 136.785 = 136.785 kN.m

Buckling Resistance

$U_{N,y} = N_{Ed} / (\chi_y \cdot N_{Rk} / \gamma_{M1})$ 70.275 / 1500.914 0.047 OK



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Job ref : L15/088/04
Sheet : **Structure / 44 -**
Made By : Chris Papadakis
Date : **July 2015**
Revision : **P3**

$U_{N,z} = N_{Ed}/(\chi_z \cdot N_{Rk}/\gamma_{M1})$	70.275 / 1197.991	0.059	OK
$U_{M,y} = M_{y,Ed}/(\chi_{LT} \cdot M_{y,Rk}/\gamma_{M1})$	44.834 / 136.785	0.328	OK
$U_{M,z} = M_{z,Ed}/(M_{z,Rk}/\gamma_{M1})$	0.004 / 63.498	0.000	OK
$k_{yy} = C_{my} \{1 + (\lambda_y - 0.2) U_{N,y}\}$		0.404	
$k_{zz} = C_{mz} \{1 + (2\lambda_z - 0.6) U_{N,z}\}$		0.835	
$k_{yz} = 0.6 k_{zz}$		0.501	
$k_{zy} = 0.6 k_{yy}$		0.242	
$U_{Ny} + k_{yy} \cdot U_{M,y} + k_{yz} \cdot U_{M,z}$	0.047 + 0.404 x 0.328 + 0.501 x 0.000	0.179	OK
$U_{Nz} + k_{zy} \cdot U_{M,y} + k_{zz} \cdot U_{M,z}$	0.059 + 0.242 x 0.328 + 0.835 x 0.000	0.138	OK

Deflection Check - Load Case 3

Deflection Limits (Internal Beams)	$\delta \leq 3000/360 = 8.3 \text{ mm Live (Case 2)}$	0.18 mm	OK
	$\delta \leq 3000/250 = 12 \text{ mm D+L (Case 3)}$	0.53 mm	OK

Provide Section (46.1 kg/m)

203x203 UC 46 [S 275]