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PROJECT: **1578**
BRITISH MUSEUM
STRUCTURAL CALCULATIONS

CLIENT: **THE BRITISH MUSEUM**
Great Russell Street
London
WC1B 3DG

ARCHITECT: **PMT Purcell Miller Tritton**
The Clove Building
4 Maguire Street
Butlers Wharf
London
SE1 2NQ

ENGINEER: **ATELIER ONE**
4 Goodge Place
London
W1T 4SB

February 2007

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INTRODUCTION

These calculations relate to the structural design of a temporary platform proposed to be constructed within the Round Reading Room at the centre of the British Museum, London. The platform is to be used for major temporary exhibitions of antiquities.

The platform is to consist of a two layers of 12.5mm thick plywood bearing on cold formed channel beams. These beams are generally arranged in a radial and circumferential pattern. Support is provided by a series of columns fixed to spreader beams placed on the existing ground floor slab. The area at the centre of the platform and immediately to the north is referred to as the 'keyhole' area. The deck is supported by a series of trusses spanning between supporting columns in this region.

Stability is provided by an arrangement of bracing members located on both radial and circumferential lines. These members have designed to withstand a notional sway loads related to the magnitude of dead and live loads applied to the deck. Two sway load conditions have been considered separately acting in two directions.

Support is provided to the platform by the existing structure which consists of a number of radial walls and arch structures spanning between the radial walls. Limitations on the applied loads within these areas have been provided in a separate report prepared by Alan Baxter & Associates. This report provides limitations on the load that can be applied to the radial walls as point loads of 30kN at a minimum spacing of 1m and point loads on the arch structures of 3kN. The maximum allowable load within the 'keyhole' area has been set as 5kN/m². The structural scheme has been developed using spreader beams to apply load directly onto the radial walls and avoid applying load onto the fragile arches.

The magnitude of the live load allowance has been selected as 4kN/m² in accordance with BS6399: Part 1 for the purpose of these calculations, although this has been reduced to 3kN/m² in the keyhole area to avoid overloading the existing structure. However, the client has advised that access to the exhibitions will be controlled using ticket and gate keeping facilities limiting the occupancy to 400 people at any one time. Additionally, schedules of weights and the layout of exhibits have been provided by the client. These calculations also include an assessment of these loads to provide a comparison to the uniformly distributed live load allowance. This assessment indicates that the applied loads are generally substantially lower than design allowances of 3 and 4kN/m².

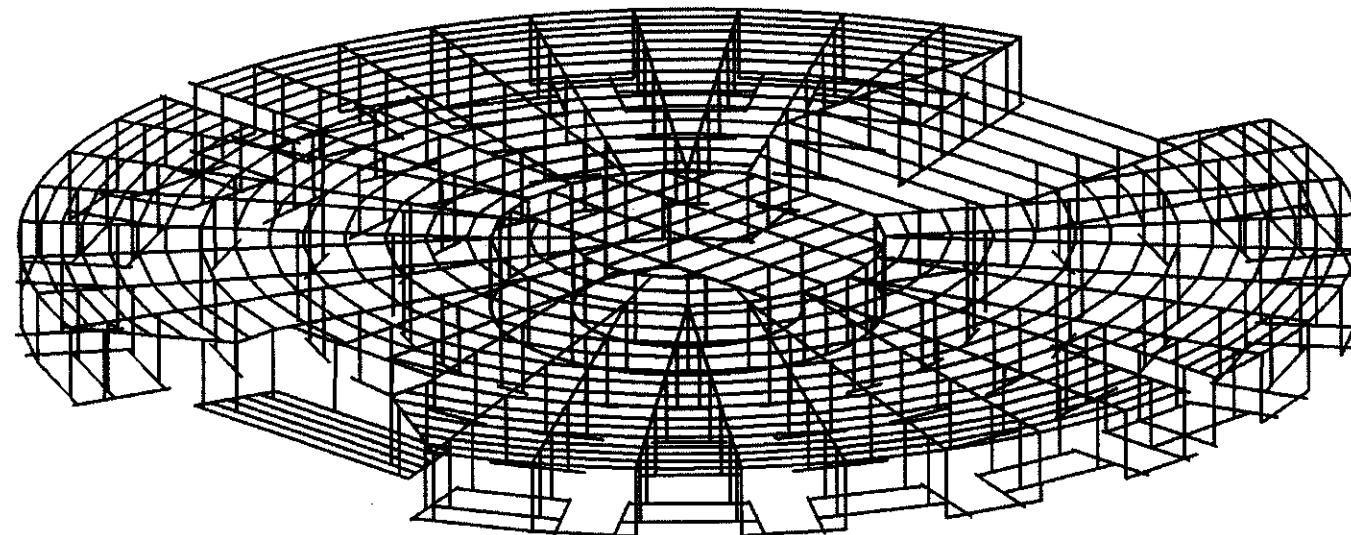
The entire platform is to be surrounded by a fabric wall and calculations are also included for the design of this element and its supporting framing.

atelier one**REFERENCE DOCUMENTS****BRITISH STANDARDS INSTITUTION**

- BS 2569:** Structural use of timber. Part 2: Code of practice for permissible stress design, material and workmanship.
- BS 5950 :** Structural use of steelwork in building. Part 1: 1990 Code of practice for design in simple and continuous construction: hot rolled sections. Part 5: 1987 Code of practice for design of cold formed sections.
- BS 6399 :** Loading for buildings. Part 1: 1996 Code of practice for dead and imposed loads + AMD 13669

View - Cases: 1 (Dead Load)

GENERAL VIEW



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Notes

MODEL INPUT DATA

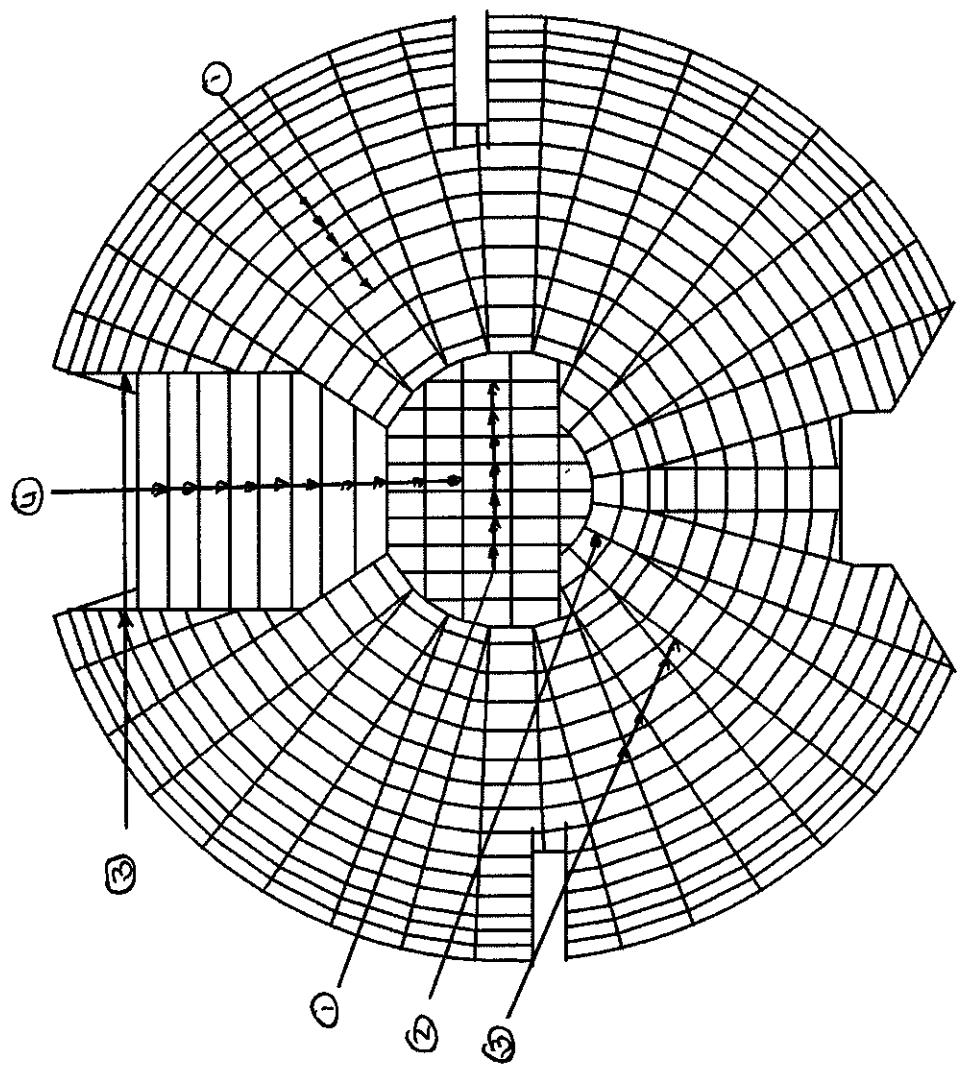
- MEMBER PROPERTIES

- Platform Level
- Ground level
- Columns

- SUPPORTS LAYOUT

- LOADING

View:2 - Cases: 1 (Dead Load) PLATFORM BEAMS & TRUSSES



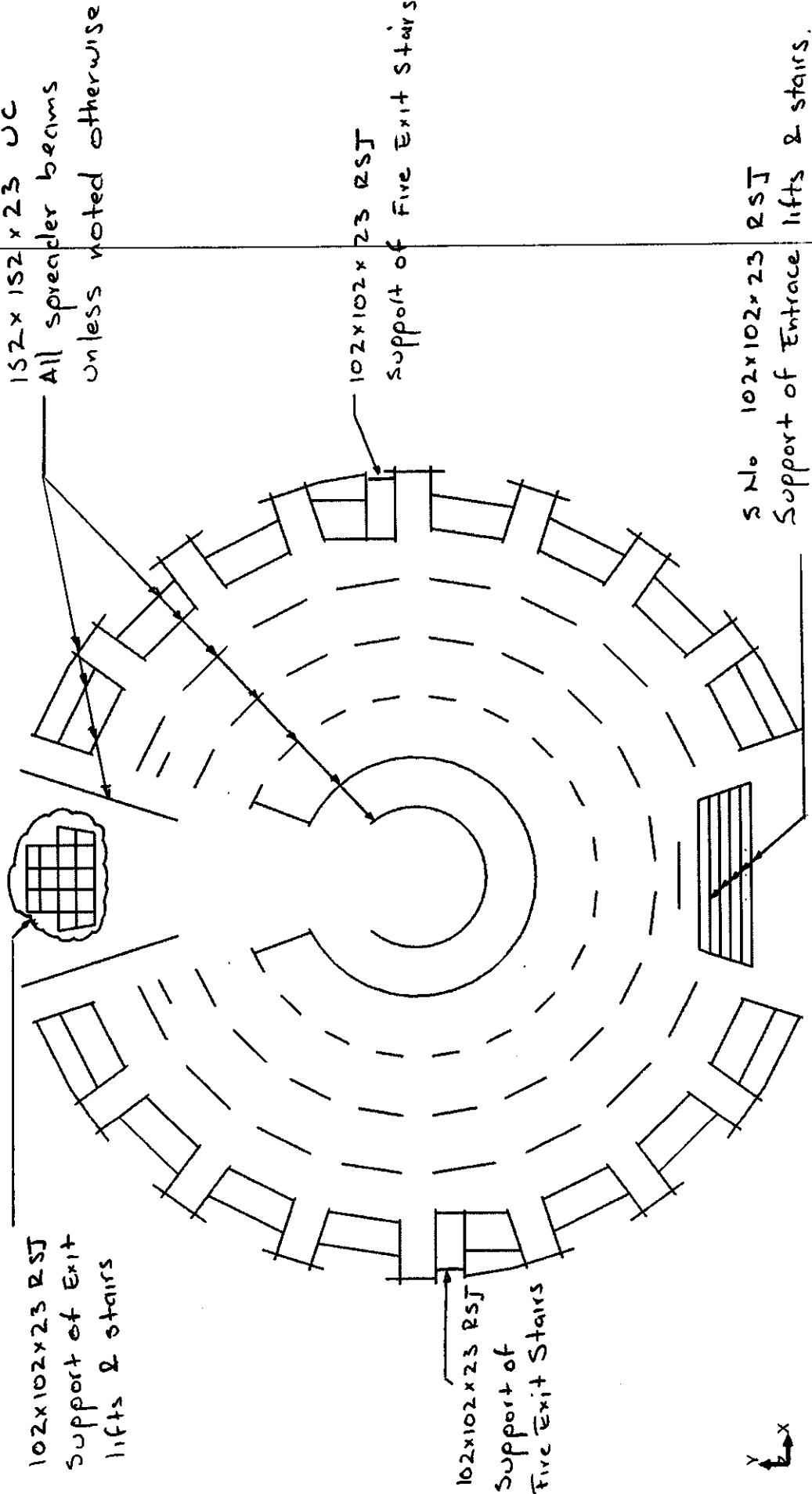
- ① Secondary circumferential beams & between trusses
Cold Form DPFCM 17.5 x 65 x 1.4
- ② Inner radial beams
- ③ Radial Beams & keyhole area
Double Cold Form
DPFC 17.5 x 65 x 2.5
Back to Back
- ④ Keyhole Area including
Platform Centre
Truss Type A, B, C & D

View:1 - Cases: 1 (Dead Load)

GROUND SPREADER BEAMS

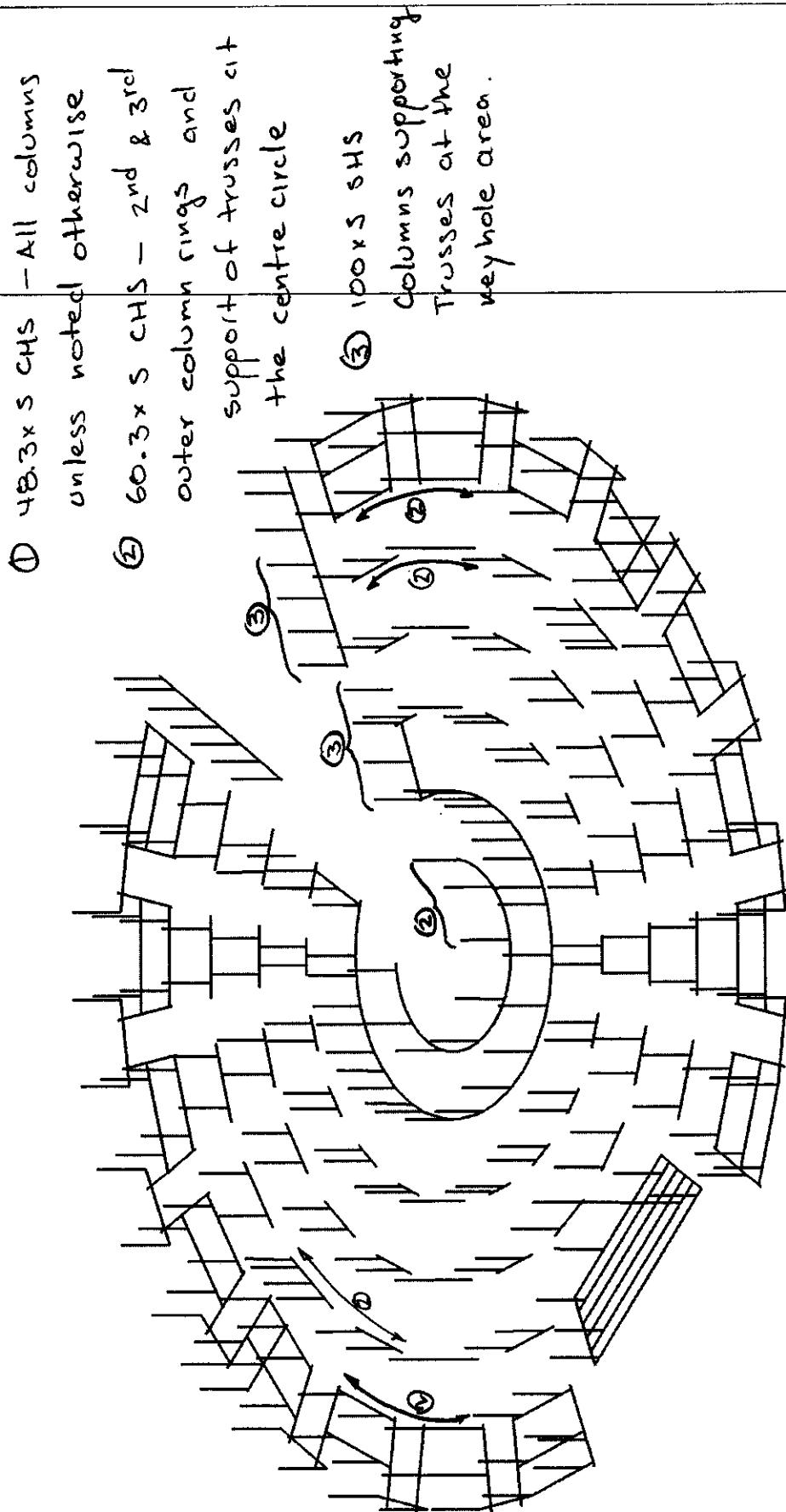
$102 \times 102 \times 23$ RSJ
Support of Exit
lifts & stairs

$152 \times 152 \times 23$ UC
All spreader beams
unless noted otherwise



1578/2/3

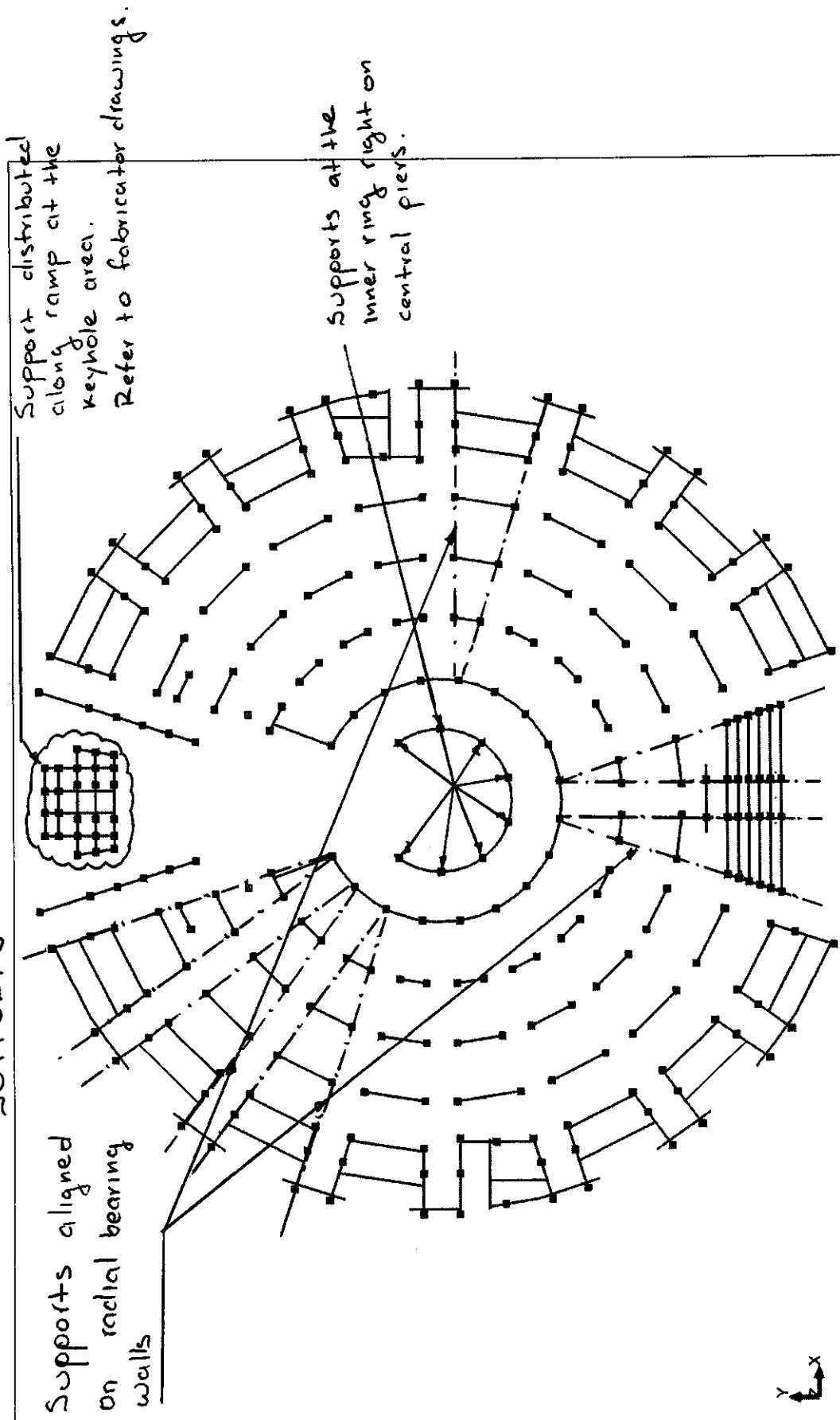
View:2 - Cases: 51 (1.0DL + 1.0LL) Columns



x
y
z

View:1 - Cases: 1 (Dead Load) SUPPORTS

Supports aligned
on radial bearing
walls

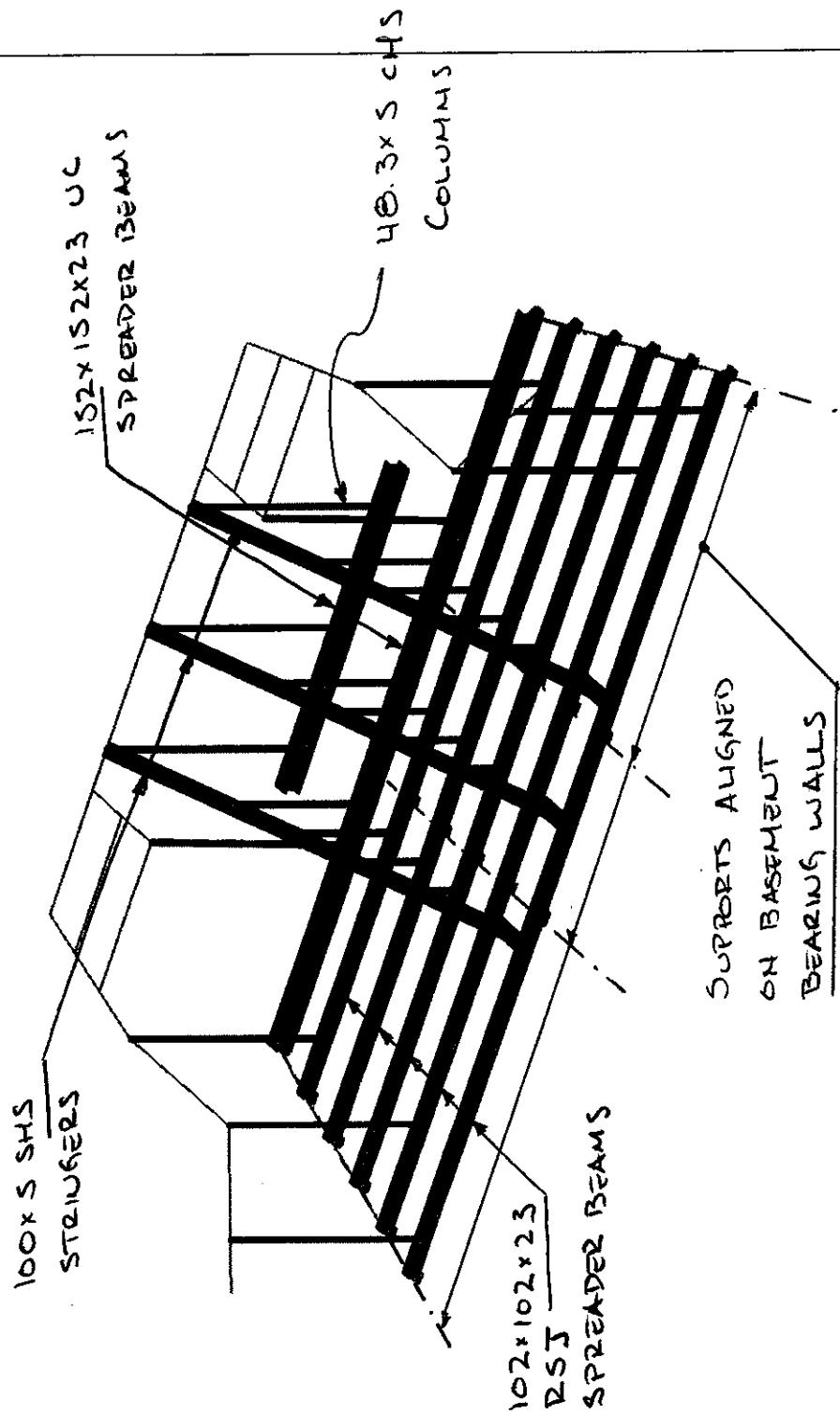


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Address:

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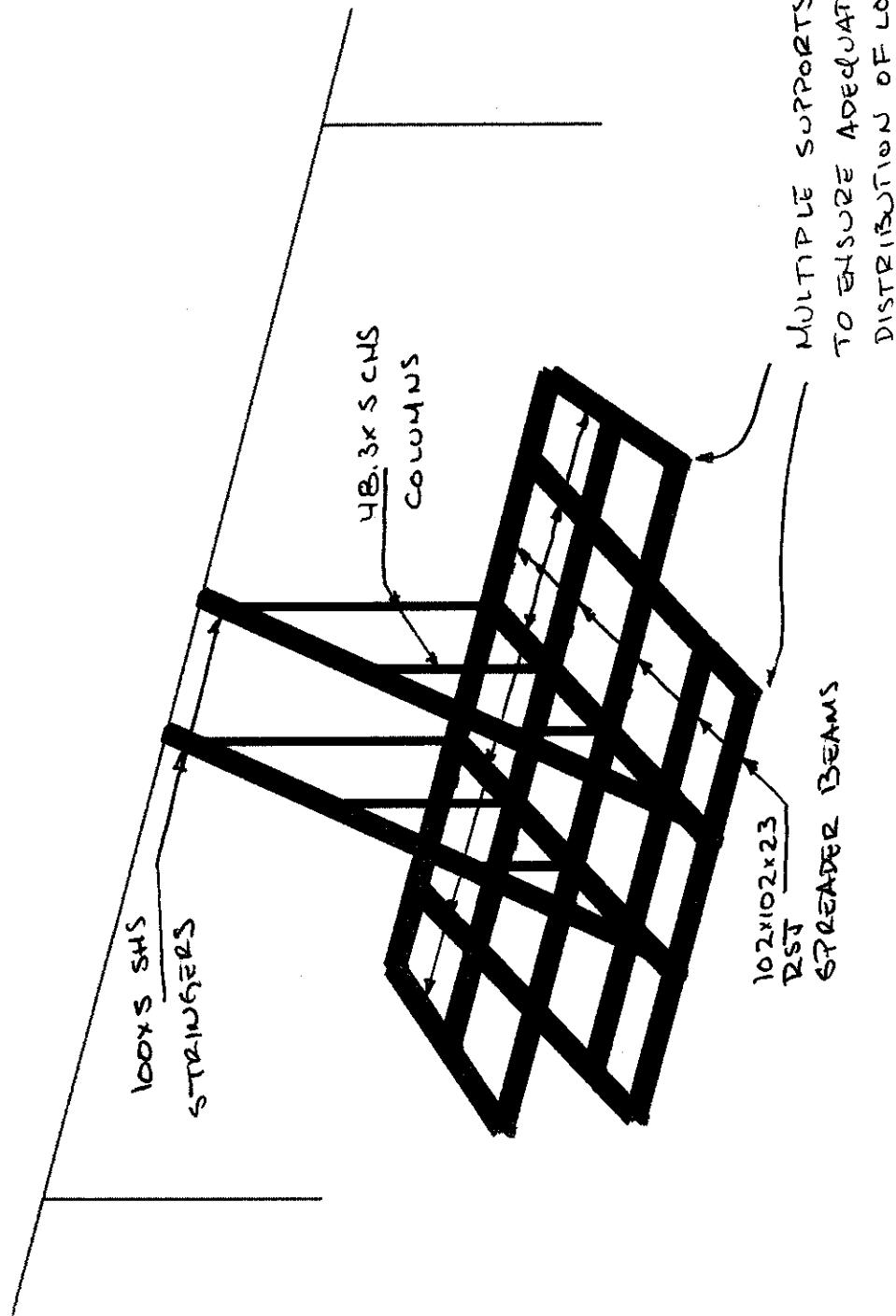
View:1 - Cases: 1 (Dead Load)

ENTRANCE - STAIR CASE STRUCTURE



View:1 - Cases: 5 (Lift& staircase)

Exit - STAIR CASE & LIFT'S SUPPORT STRUCTURE



1578/2/7

REFER TO
DWG
1578-1214

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Notes

LOADING

Dead Load:

$$2 \text{ layers } 12.5 \text{ mm of play} = 0.18 \text{ kN/m}^2$$

$$\begin{array}{l} \text{Carpet \& finishes} \\ \hline 0.09 \text{ kN/m}^2 \\ \hline 0.27 \text{ kN/m}^2 \end{array}$$

Fabric Wall: along perimeter

$$- \text{Fabric wall \& handrail} = 0.4 \text{ kN/m}$$

$$- \text{Plywood wall below deck} = 0.36 \text{ kN/m}$$

$$\begin{array}{l} - \text{Lighting Perimetre} \\ \hline 0.23 \text{ kN/m} \\ \hline 0.99 \text{ kN/m} \end{array}$$

Live Load

$$4.00 \text{ kN/m}^2$$

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Notes

STAIRCASE SELF-WEIGHT

2.6 m wide.

4mm thick Plate

Load per
stringer.

0.43 kN/m^2

$$\begin{array}{c} (\times 1.175) \\ \hline \text{entrance middle} \\ \text{stringer.} \end{array}$$

0.51 kN/m

$$\begin{array}{c} (\times 0.713) \\ \hline \text{entrance side} \\ \text{stringer} \end{array}$$

0.31 kN/m

Balustrade

0.5 kN/m

LOADING

Adopt

$$\begin{array}{l} \text{DL } \left\{ \begin{array}{l} \text{Side stringer} \\ \text{(entrance & exit)} \end{array} \right. \text{ --- } 0.81 \text{ kN/m} \\ \left. \begin{array}{l} \text{Middle stringer} \\ \text{(entrance only)} \end{array} \right. \text{ --- } 0.51 \text{ kN/m} \end{array}$$

$$\begin{array}{l} \text{LL } \left\{ \begin{array}{l} \text{Side stringer} \\ \text{(\mathbf{4.0} \times 0.713)} \end{array} \right. \text{ --- } 2.9 \text{ kN/m} \\ \left. \begin{array}{l} \text{Middle stringer} \\ \text{(\mathbf{4.0} \times 1.175)} \end{array} \right. \text{ --- } 4.7 \text{ kN/m} \end{array}$$

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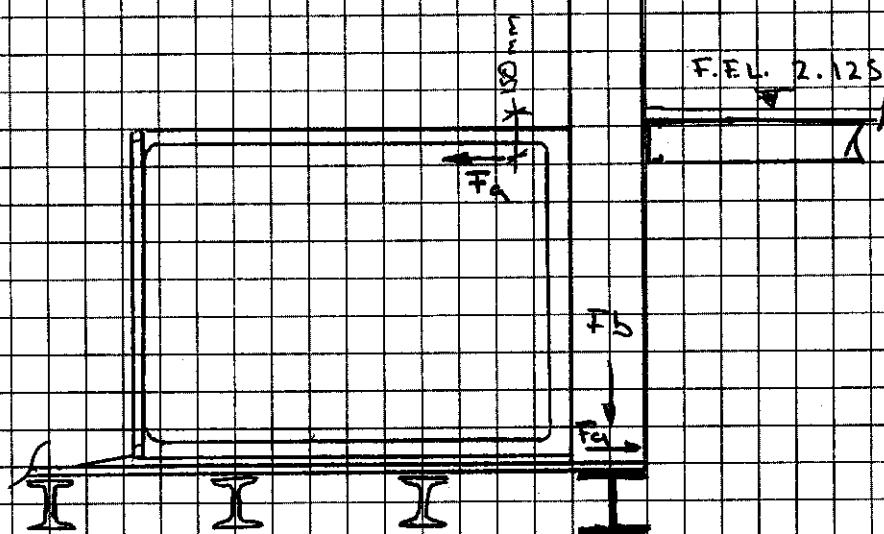
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Notes

LIFT SELF-WEIGHT & LIVE LOAD

$$F_b = 10 \text{ kN}$$

$$F_a = 5.5 \text{ kN}$$



$$F_b : D_L = 7.0 \text{ kN}$$

$$L_L = 3.0 \text{ kN}$$

$$F_a : D_L = 5.5 \text{ kN}$$

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Notes

NATIONAL HORIZONTAL SWAY LOAD

Platform Area = 1085 m²

Dead Load

Plywood 25 mm thick 27 kg/m² — 293 kN

DPFC 175x65x2.4 5668 kg — 56.7 kN

PFCH 175x65x2.4 96 kg — 1.0 kN

PFCH 175x65x1.4 7776 kg — 77.8 kN

TRUSS 27 kg/m 3259 — 32.6 kN

Perimeter load — L = 122 m

↳ { Fabric wall 4830 — 48.8 kN

| Lighting 2810 — 28.1 kN

538 kN

Live Load

4 kN/m² — 434000 kg — 4340 kN

4340 kN.

$$\text{National Load} = (1.4 \times 538) \times (1.6 \times 4340) \times 0.5 \cdot \%$$

$$= 38.5 \text{ kN}$$

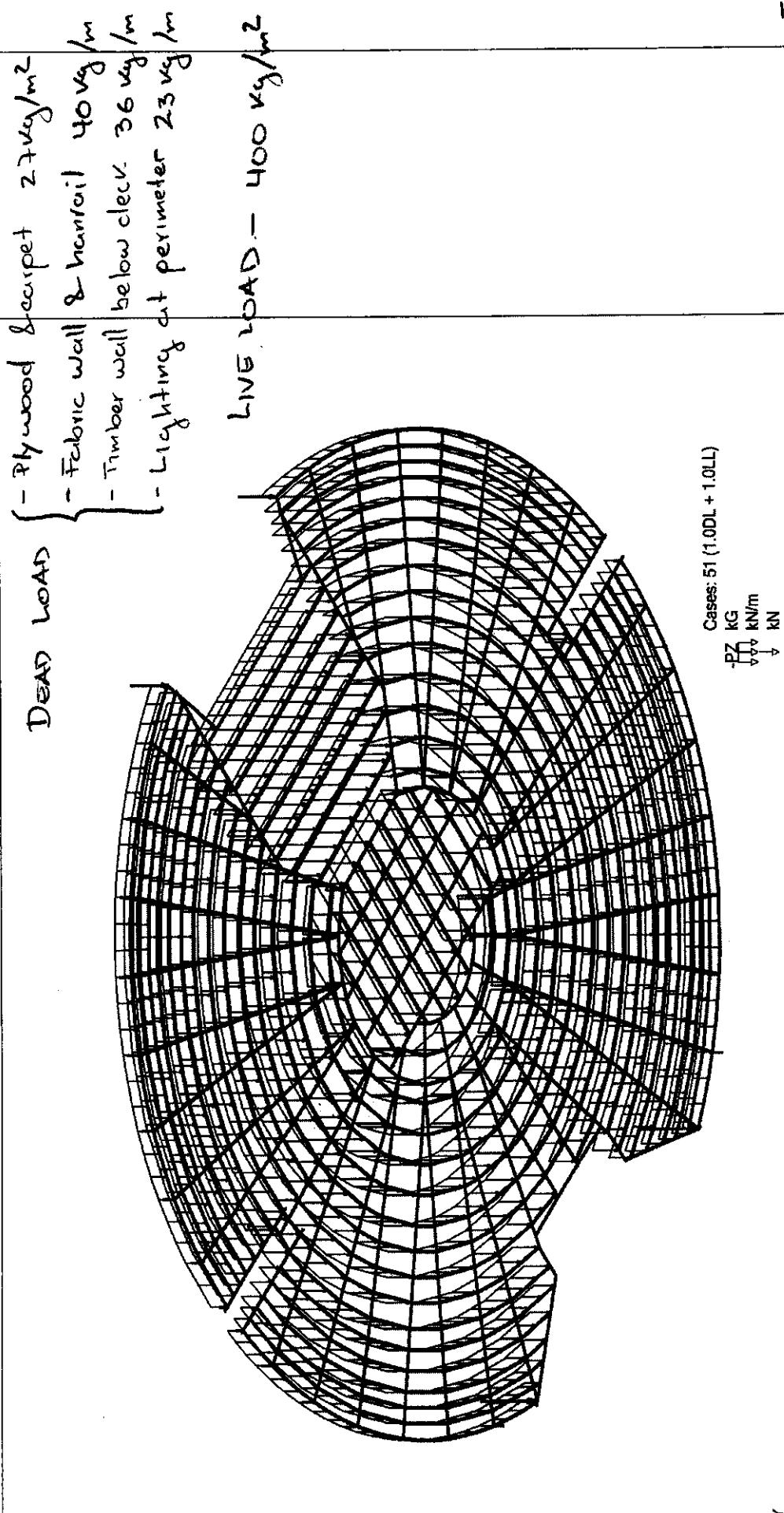
L = 518.2 m

L = 16.9 m

L = 2437 m

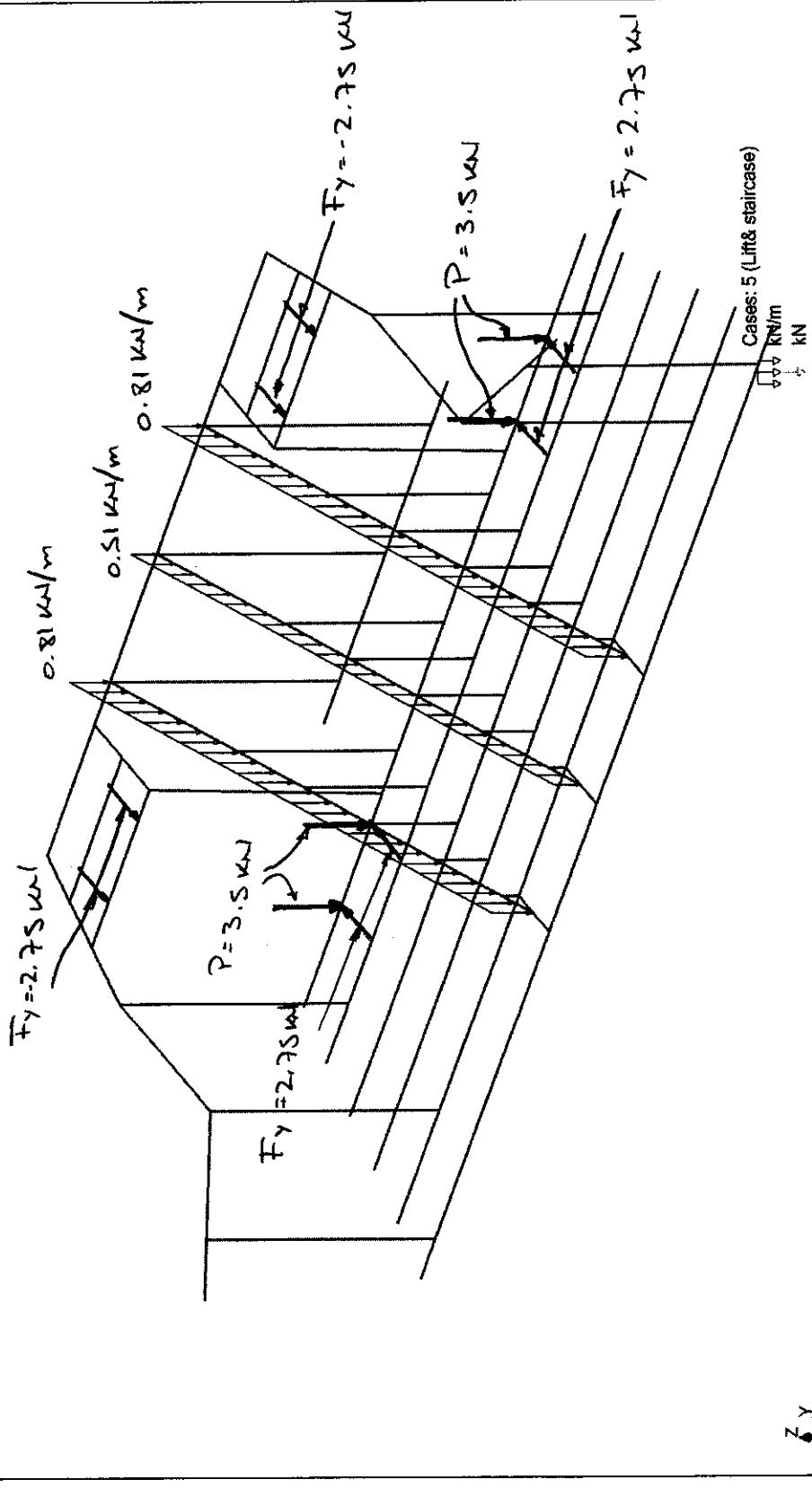
L = 120.7 m

View:2 - Cases: 51 (1.0DL + 1.0LL) LOADW6



View:2 - Cases: 5 (Lift& staircase)

LOAD CASE 5 : DEAD LOAD STAIRCASE & LIFTS



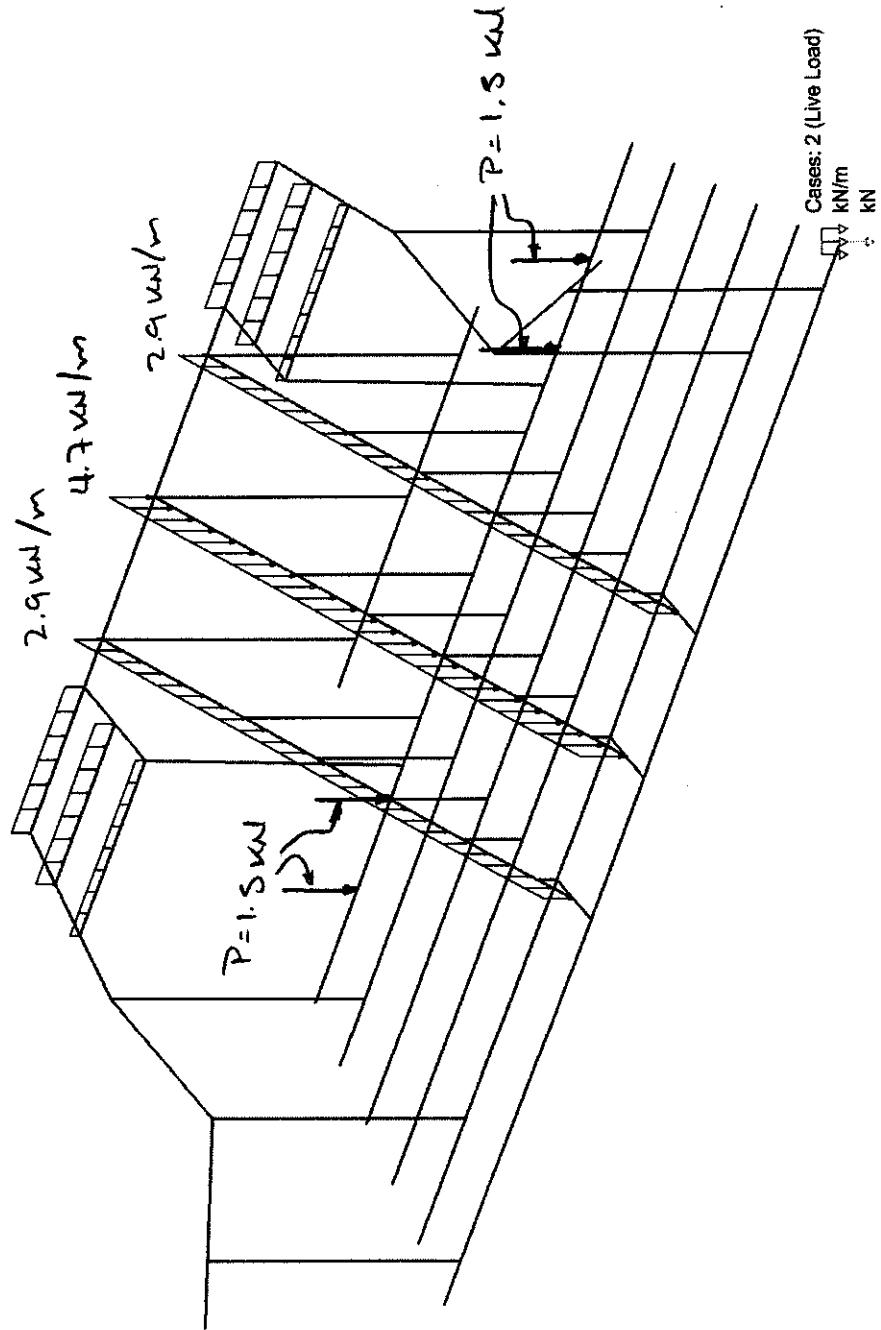
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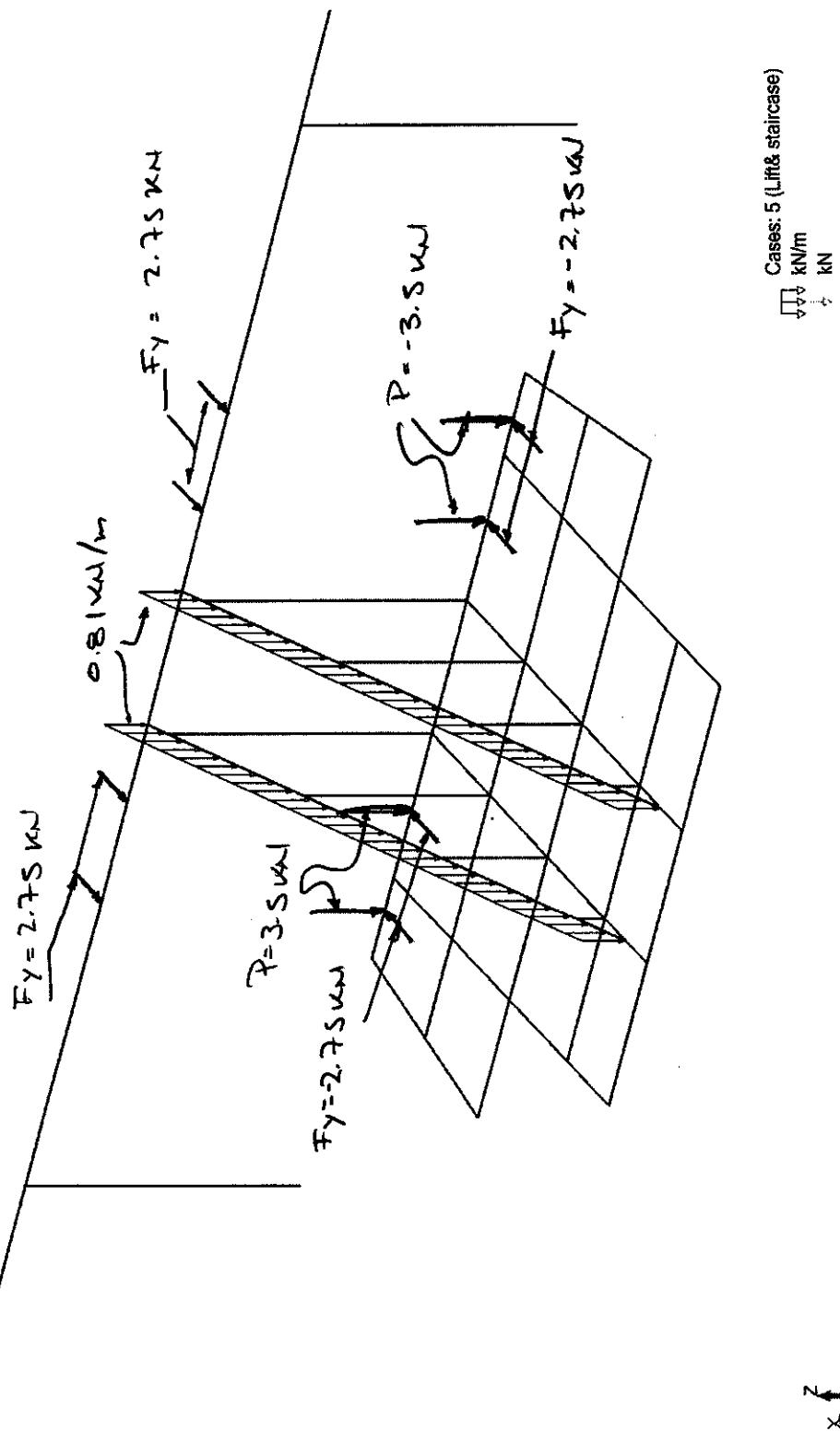
View:2 - Cases: 2 (Live Load)

LOAD CASE 2 : LIVE LOAD

ENTRANCE STAIRCASE & LIFTS.



View:1 - Cases: 5 (Lift& staircase) LOAD CASE S : DEAD LOAD STAIRCASE & LIFTS

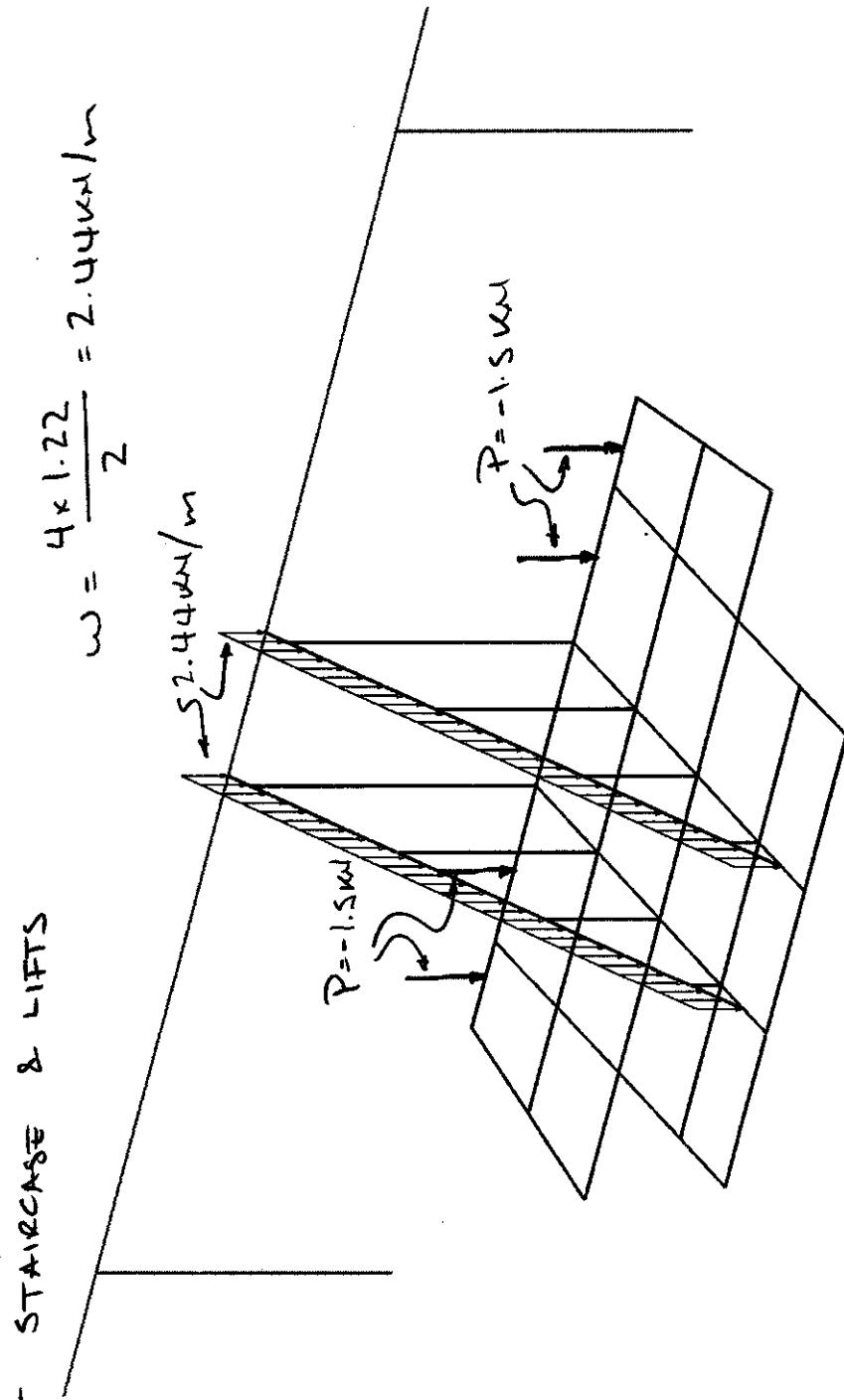


View:1 - Cases: 2 (Live Load)

LOAD CASE 2: LIVE LOAD

EXIT STAIRCASE & LIFTS

$$\omega = \frac{4 \times 1.22}{2} = 2.44 \text{ kN/m}$$

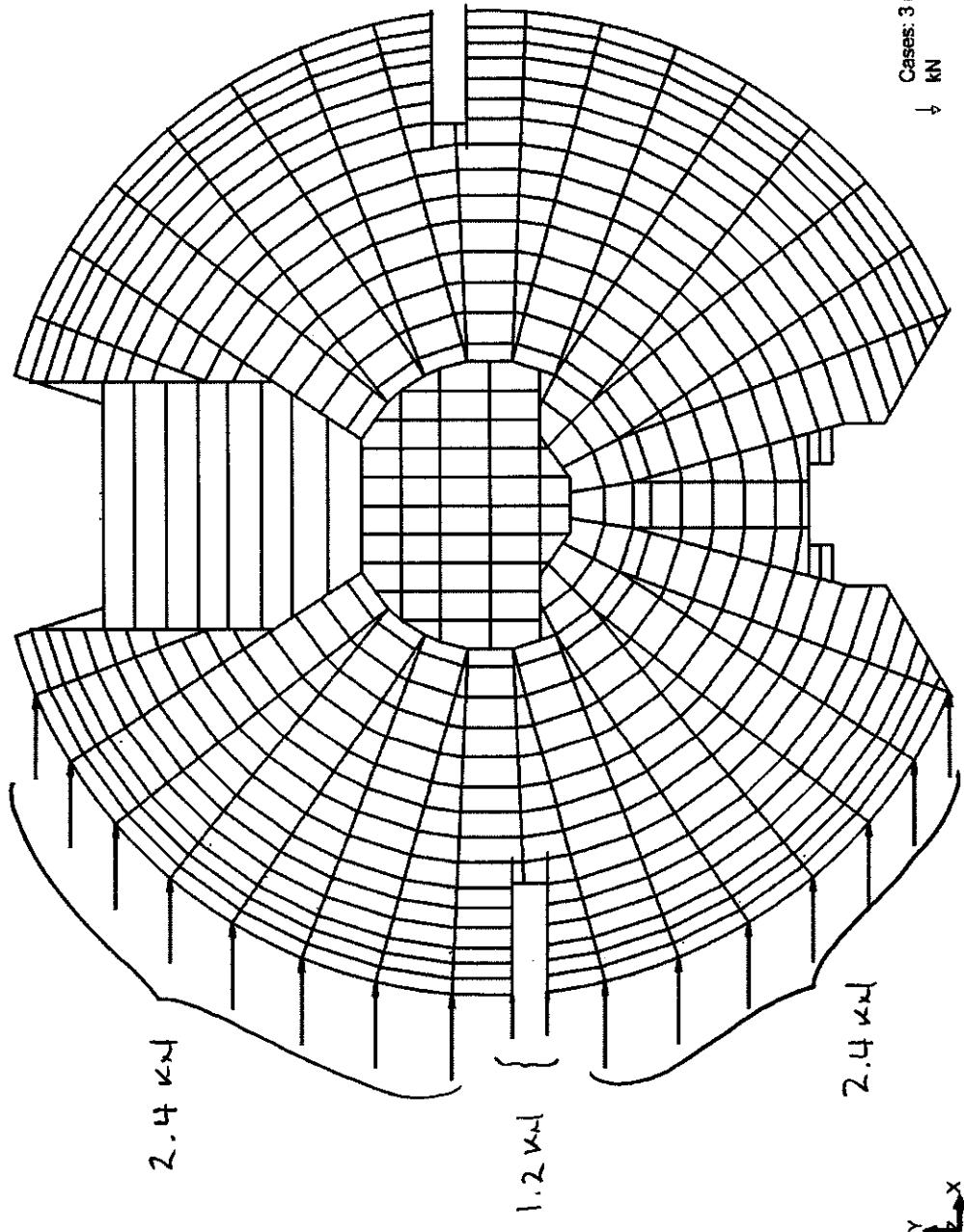

 Cases: 2 (Live Load)

 ↓ N/m
 ↓ N

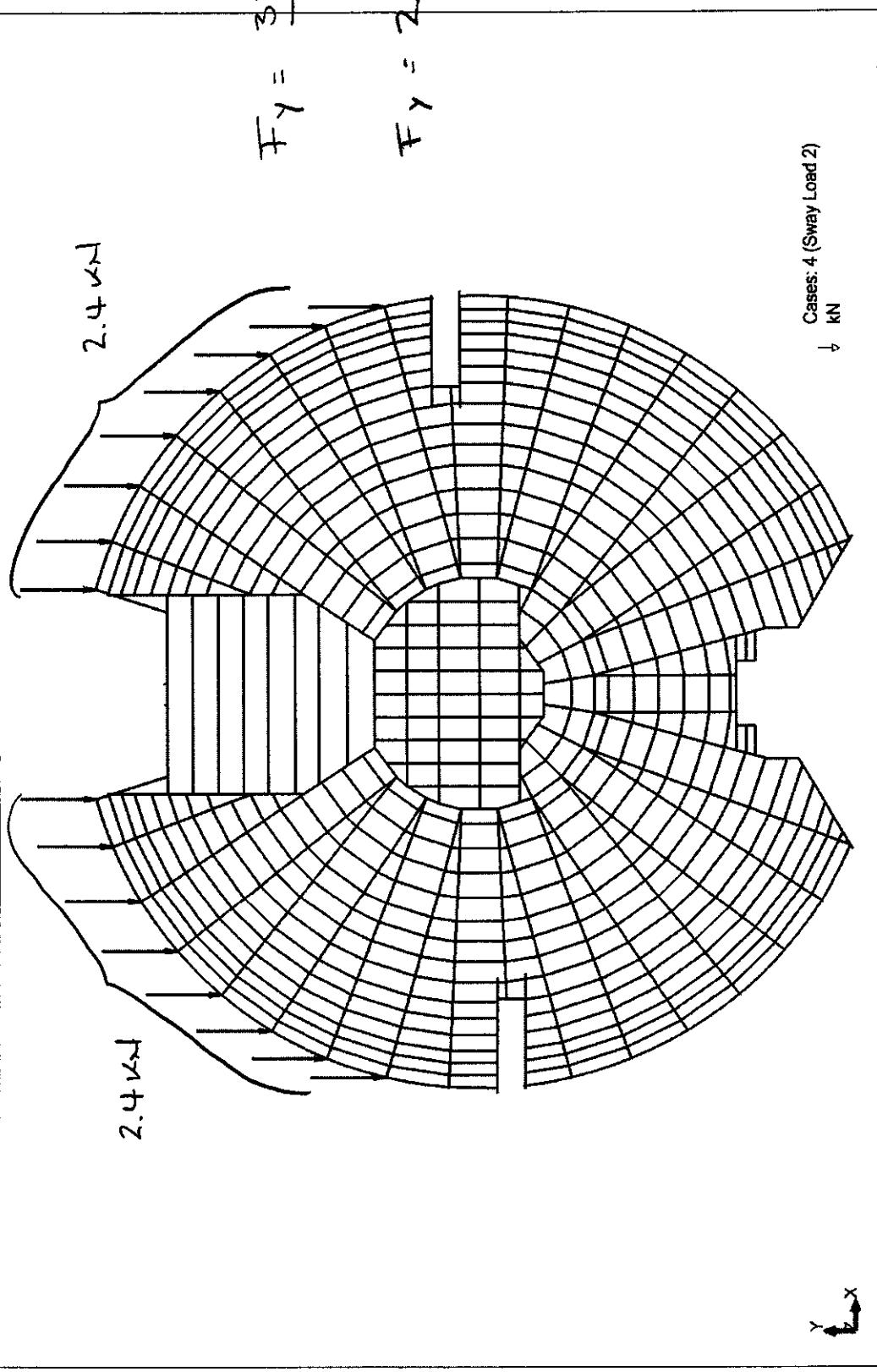
 Z
 X

View:1 - Cases: 3 (Sway Load 1)

LOAD CASE 3 : SWAY LOAD 1 — X direction . Allows -3850 kN



View:1 - Cases: 4 (Sway Load 2) LOAD CASE 4: SWAY LOAD 2 - Y direction Allow 3850 kN





Load Combination - Cases: 51to67 101to104

Values

1

LOAD COMBINATIONS

- Cases: 51to67 101to104

Filtering	Combinations
Full list	51to67 101to104
Selection	51to67 101to104
Total number	21
Selected number	21

- Cases: 51to67 101to104

Combinations	Name	Analysis type	Combination	Case nature	Definition
51 (C)	1.0DL + 1.0LL	Linear Combination	SLS	dead	(1+2+5)*1.00
52 (C)	1.0DL + 1.0LL + 1.0Sway 1	Linear Combination	SLS	dead	(1+2+3)*1.00
53 (C)	1.0DL + 1.0LL - 1.0Sway 1	Linear Combination	SLS	dead	(1+2)*1.00+3*-1.00
54 (C)	1.0DL + 1.0LL + 1.0Sway 2	Linear Combination	SLS	dead	(1+2+4)*1.00
55 (C)	1.0DL + 1.0LL - 1.0Sway 2	Linear Combination	SLS	dead	(1+2)*1.00+4*-1.00
56 (C)	1.0DL + 1.0LL + 1.0Sway 1 + Lift	Linear Combination	SLS	dead	(1+2+3+5)*1.00
57 (C)	1.0DL + 1.0LL - 1.0Sway 1 + Lift	Linear Combination	SLS	dead	(1+2+5)*1.00+3*-1.00
58 (C)	1.0DL + 1.0LL + 1.0Sway 2 + Lift	Linear Combination	SLS	dead	(1+2+4+5)*1.00
59 (C)	1.0DL + 1.0LL - 1.0Sway 2 + Lift	Linear Combination	SLS	dead	(1+2+5)*1.00+4*-1.00
60 (C)	1.0DL + 1.0Sway 1	Linear Combination	SLS	dead	(1+3)*1.00
61 (C)	1.0DL - 1.0Sway 1	Linear Combination	SLS	dead	1*1.00+3*-1.00
62 (C)	1.0DL + 1.0Sway 2	Linear Combination	SLS	dead	(1+4)*1.00
63 (C)	1.0DL - 1.0Sway 2	Linear Combination	SLS	dead	1*1.00+4*-1.00
64 (C)	1.0DL + 1.0Sway 1 + Lift	Linear Combination	SLS	dead	(1+3+5)*1.00
65 (C)	1.0DL - 1.0Sway 1 + Lift	Linear Combination	SLS	dead	(1+5)*1.00+3*-1.00
66 (C)	1.0DL + 1.0Sway 2 + Lift	Linear Combination	SLS	dead	(1+4+5)*1.00
67 (C)	1.0DL - 1.0Sway 2 + Lift	Linear Combination	SLS	dead	(1+5)*1.00+4*-1.00
101 (C)	1.4DL	Linear Combination	ULS	dead	(1+5)*1.40
102 (C)	1.4DL + 1.6LL	Linear Combination	ULS	dead	(1+5)*1.40+2*1.60
103 (C)	1.4DL + 1.6LL + 1.0Sway 1	Linear Combination	ULS	dead	(1+5)*1.40+2*1.60+3*1.00
104 (C)	1.4DL + 1.6LL + 1.0Sway 2	Linear Combination	ULS	dead	(1+5)*1.40+2*1.60+4*1.00

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Notes

RESULTS:

F_x — Axial Force (+) Compression

(-) Tension

F_y — Shear Force minor axis

F_z — Shear Force major axis

M_x — Torsional Moment

M_y — Bending Moment major axis

M_z — Bending Moment minor axis

Forces - Cases: 1to4 51 101to104
Global extremes

1

- Cases: 1to4 51 101to104

Filtering	Bar	Case
Full list	1to81 84to87 92t 11to4 51 101to10	
Selection	2001 2004to200 1to4 51 101to10	
Total number	2765	9
Selected number	1153	9

GLOBAL EXTREME FORCES

- Cases: 1to4 51 101to104

Case : All Load Combinations
Bar : DPFC 175x65x2.4 Back to Back.

- Cases: 1to4 51 101to104

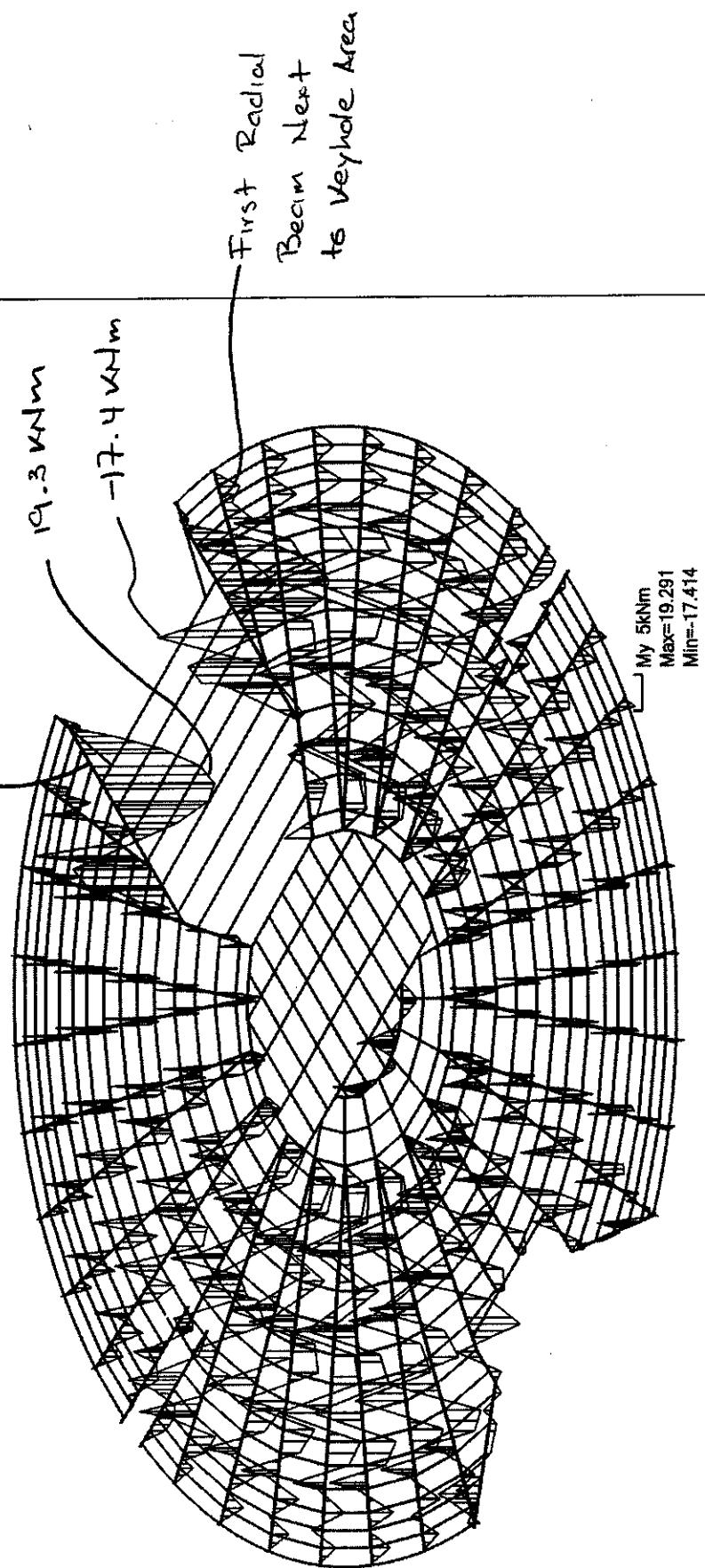
	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX						
Bar	2047	2344	3086	2570	2708	2324
Node	2225	2888	3334	2014	2415	2134
Case	104 (C)	102 (C)	104 (C)	104 (C)	102 (C)	103 (C)
MIN						
Bar	2409	2317	2697	2571	3101	2566
Node	2256	2659	3326	2017	2099	2306
Case	103 (C)	102 (C)	104 (C)	103 (C)	102 (C)	103 (C)

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View:1 - MY, Cases: 102 (1.4DL + 1.6LL)
 DPFC 175x65x2.4

BMD — LOAD COMBINATION 1.4 DL + 1.6 LL

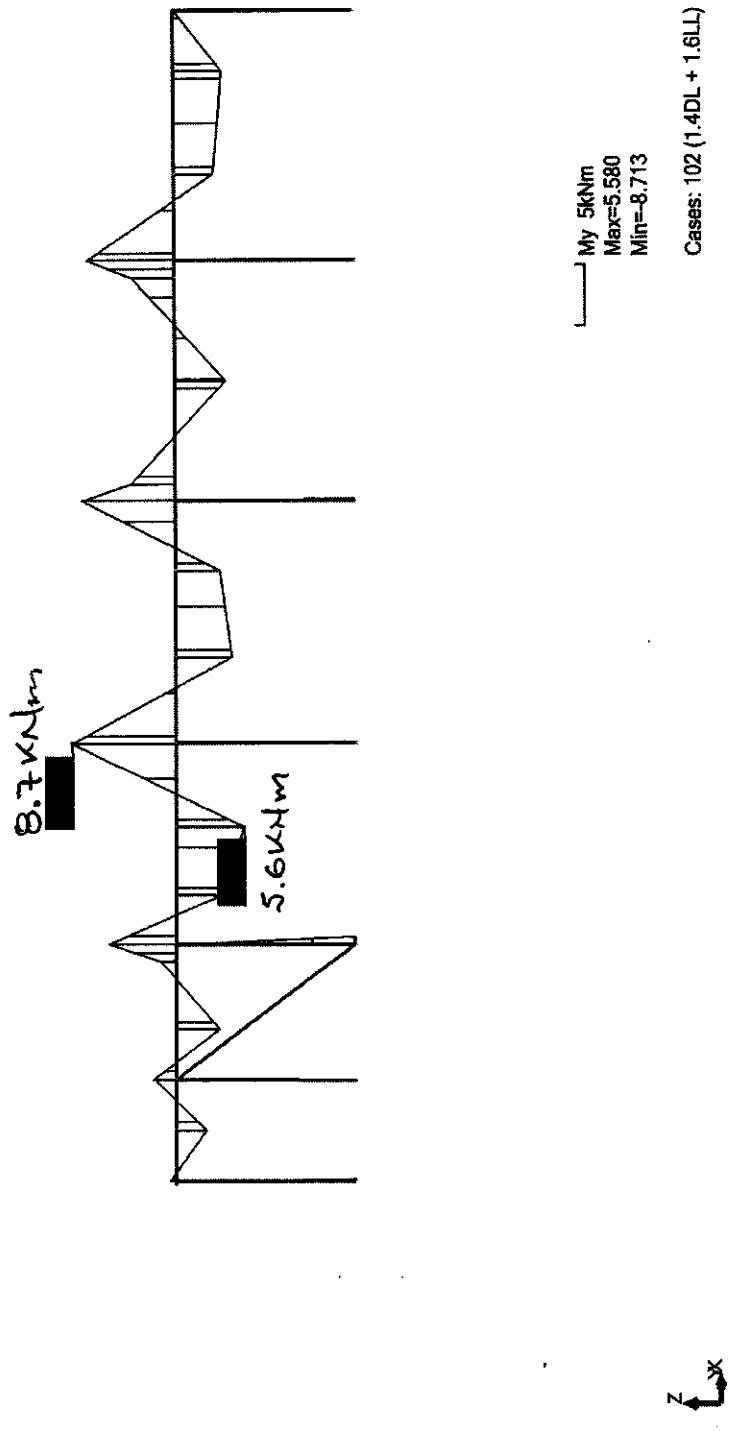
Beam Along Keyhole



1578/3/4

View:2 - MY, Cases: 102 (1.4DL + 1.6LL) BMD - LOAD COMBINATION 1.4 DL + 1.6 LL

— TYPICAL RADIAL BEAM DPFC 175x65x2.4

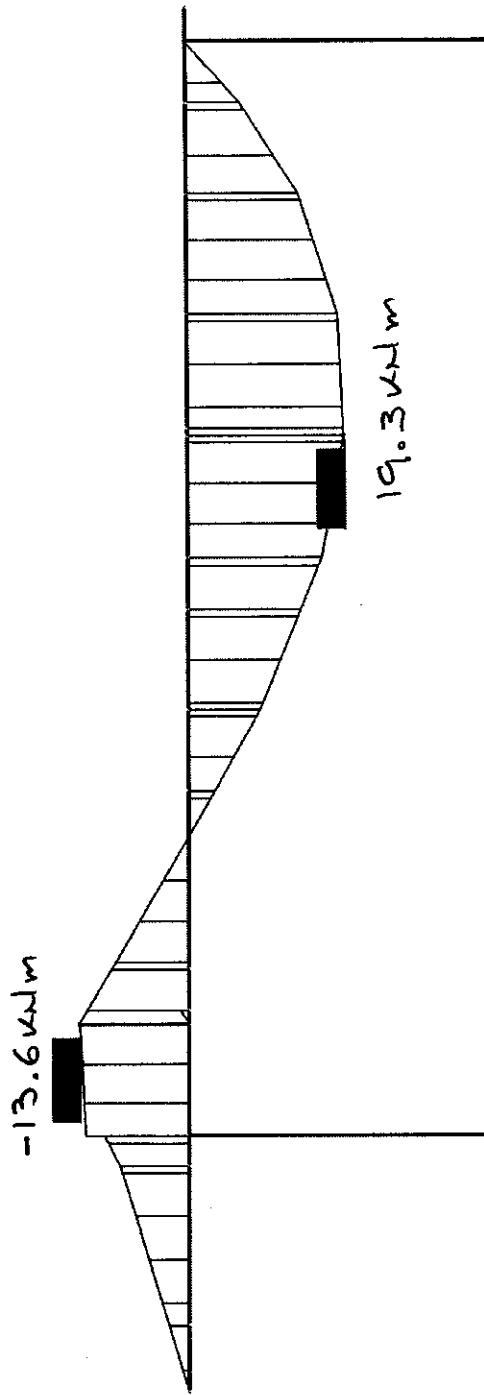


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View:2 - MY, Cases: 102 (1.4DL + 1.6LL) BMD - LOAD COMBINATION 1.4 DL + 1.6LL

BEAM ALONG KEYHOLE AREA



Z
Y

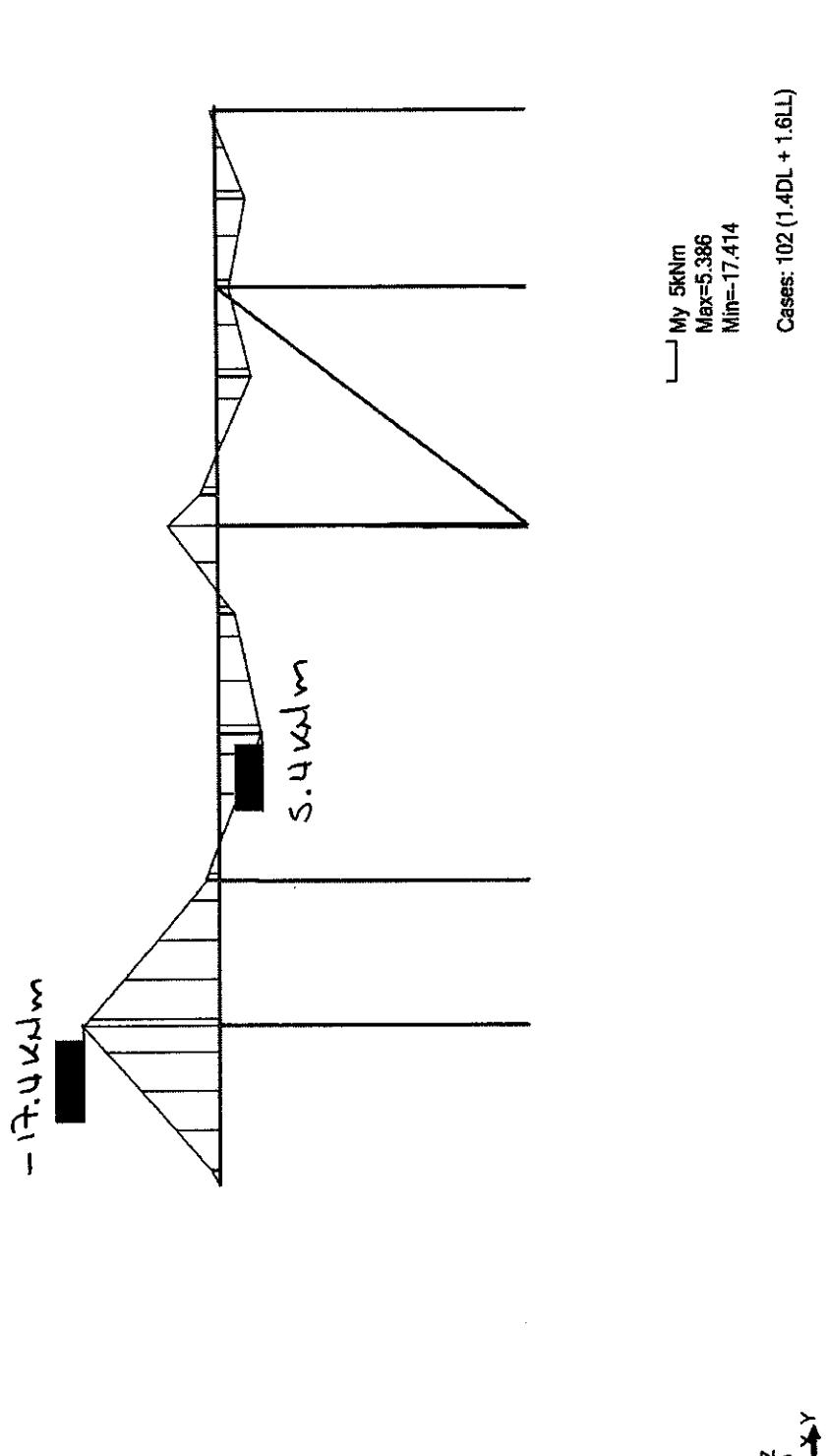
My 5kNm
Max=19.291
Min=-13.598
Cases: 102 (1.4DL + 1.6LL)

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Address:

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View:2 - MY, Cases: 102 (1.4DL + 1.6LL) BMD - LOAD COMBINATION 1.4DL + 1.6LL

FIRST RADIAL BEAM NEXT TO KEYHOLE AREA



Forces - Cases: 1to4 51 101to104
 Global extremes
 1

- Cases: 1to4 51 101to104

Filtering	Bar	Case
Full list	1to81 84to87 92t 1to4 51 101to10	
Selection	4059 4060 4072	1to4 51 101to10
Total number	2765	9
Selected number	16	9

GLOBAL EXTREME FORCES

- Cases: 1to4 51 101to104

Case : All Load Combinations

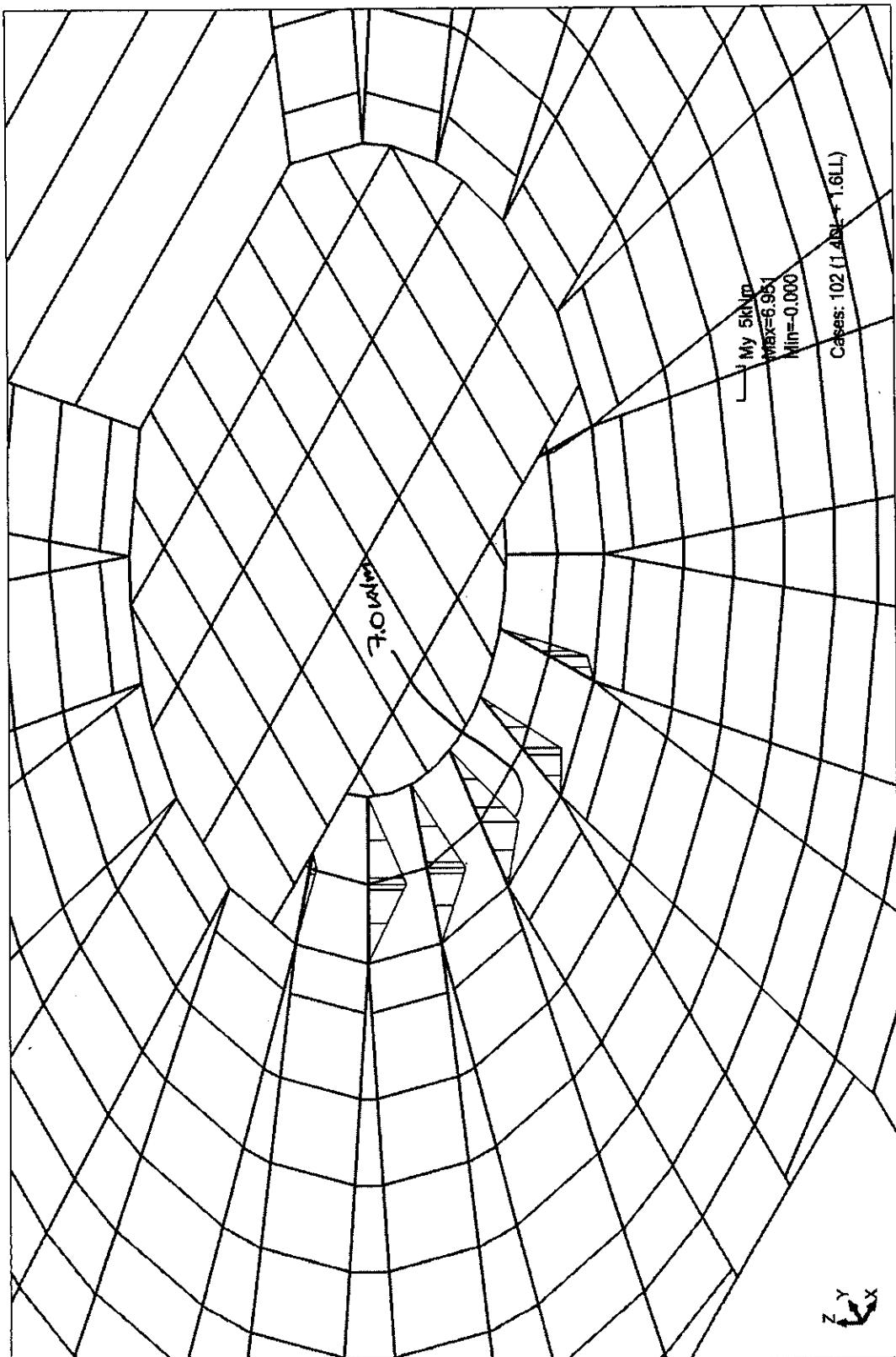
Bar: PTCH 175x65x2.4

- Cases: 1to4 51 101to104

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX	1.576	0.449	0.449	0.379	0.379	0.379
Bar	4060	4060	4182	4655	4182	4059
Node	2341	2341	2241	2242	2345	2341
Case	3	103 (C)	102 (C)	102 (C)	102 (C)	103 (C)
MIN	-0.423	-0.423	-0.423	-0.379	-0.379	-0.379
Bar	4649	4059	4656	4183	4060	4073
Node	2343	2302	2002	2345	2341	2342
Case	103 (C)	103 (C)	102 (C)	104 (C)	4	103 (C)

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View:1 - MY, Cases: 102 (1.4DL + 1.6LL) BMD — LOAD COMBINATION 1.4 DL+1.6 LL — PFC H 175x65x2.4



Forces - Cases: 1to4 51 101to104
Global extremes

- Cases: 1to4 51 101to104

Filtering	Bar	Case
Full list	1to81 84to87 92t 11to4 51 101to10	
Selection	9 19to21 23 414 1to4 51 101to10	
Total number	2765	9
Selected number	69	9

GLOBAL EXTREME FORCES

- Cases: 1to4 51 101to104

Filtering	Bar	Case
Full list	1to81 84to87 92t 11to4 51 101to10	
Selection	9 19to21 23 414 1to4 51 101to10	
Total number	2765	9
Selected number	69	9

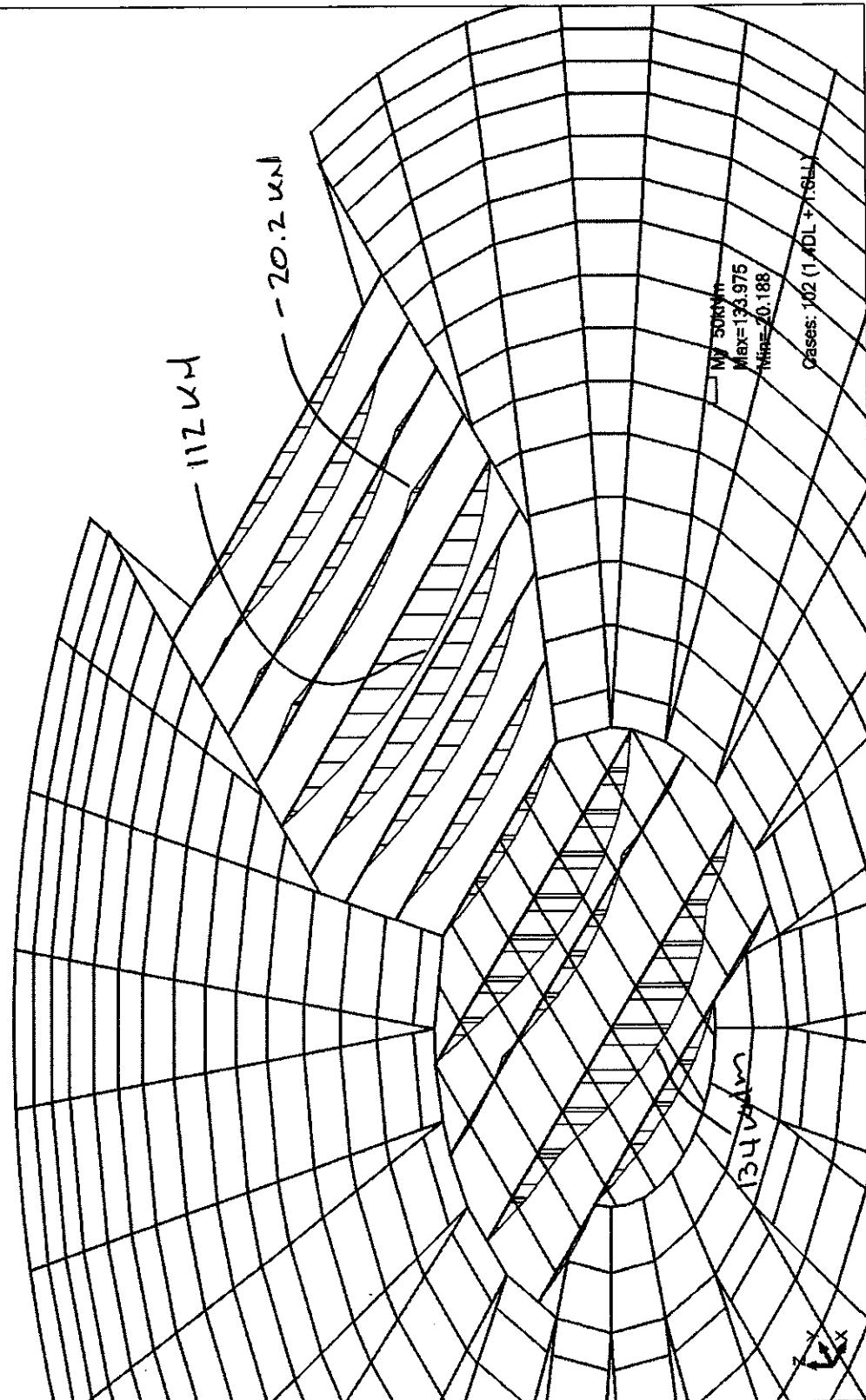
- Cases: 1to4 51 101to104

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX						
Bar	4235	5040	5030	4234	4151	5047
Point	2320	2106	2393	2316	2298	2428
Case	3	103 (C)	103 (C)	104 (C)	104 (C)	103 (C)
MIN						
Bar	4235	5045	5032	5030	5028	5041
Point	2320	2431	2394	2393	2458	2427
Case	104 (C)	103 (C)	102 (C)	103 (C)	102 (C)	103 (C)

1578/3/A

1578/3/10

View:1 - MY, Cases: 102 (1.4DL + 1.6LL) BMD - LOAD COMBINATION 1.4DL + 1.6LL - TRUSS (see Drawg
1578-135)



Forces - Cases: 51to67 101to104
 Global extremes
 1

- Cases: 51to67 101to104

Filtering	Bar	Case
Full list	1to147 150 151	1to5 51to67 101to104
Selection	85to67 92to97 151to67 101to104	
Total number	2842	26
Selected number	33	21

GLOBAL EXTREME FORCES

Case: All Load Combination

Bar: 48.3x5 CHS Bracing

- Cases: 51to67 101to104

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX						
Bar	6300	6307	6297	6307	6297	6300
Point	1211	1356	1292	1356	1121	2654
Case	103 (C)	102 (C)	102 (C)	102 (C)	102 (C)	103 (C)
MIN						
Bar	129	6300	6297	6300	6299	6307
Point	3229	1211	2784	1211	1243	2561
Case	102 (C)	103 (C)	101 (C)	104 (C)	103 (C)	102 (C)

Forces - Cases: 51to67 101to104
Global extremes

Columns

- Cases: 51to67 101to104

Filtering	Bar	Case
Full list	11to147 1001to1011to51to67 101to6004to6008 60151to67 101to104	
Selection		
Total number	2842	26
Selected number	172	21

- Cases: 51to67 101to104

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX	39.82	0.19	0.67			
Bar	6203	6144	6239	6405	6190	6144
Node	1570	1158	2257	3468	1180	1158
Case	102 (C)	104 (C)	103 (C)	103 (C)	104 (C)	104 (C)
MIN	-23.8	-0.61	-0.61			
Bar	6156	7	6235	6190	6239	8
Node	2657	1	1592	1180	2257	3
Case	101 (C)	104 (C)	102 (C)	104 (C)	103 (C)	104 (C)

- Cases: 51to67 101to104

	Bar	Case
Case	All Load Combinations	Bar : 48.3x S chs columns - except columns under entrance & exit stairs.
Node		
Bar		
Node		
Case		

Forces - Cases: 1to4 51 101to104
Global extremes

1

- Cases: 1to4 51 101to104

Filtering	Bar	Case
Full list	1to81 84to87 921	1to4 51 101to10
Selection	1to6 6002 6003	1to4 51 101to10
Total number	2765	9
Selected number	84	9

GLOBAL EXTREME FORCES

- Cases: 1to4 51 101to104

Case: All Local Combinations

Box: 60.3x5 C115 Columns

- Cases: 1to4 51 101to104

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX	55297	0.35	8159	9012	960	1885
Bar	6	4	6228	6240	3	6045
Point	1734	2402	1004	1561	2404	1428
Case	103 (C)	103 (C)	103 (C)	103 (C)	103 (C)	104 (C)
MIN	1032	1032	1032	1032	1032	1032
Bar	6233	6046	3	6264	6173	6046
Point	1200	1431	1731	1757	1537	1431
Case	3	103 (C)	103 (C)	103 (C)	102 (C)	103 (C)

1578/3/13

Forces - Cases: 51to67 101to104
Global extremes

1

- Cases: 51to67 101to104

Filtering	Bar	Case
Full list	1to147 100to101to5 51to67 101to104	
Selection	26 27 100 106 151to67 101to104	
Total number	2842	26
Selected number	29	21

GLOBAL EXTREME FORCES

- Cases: All Local Combinations

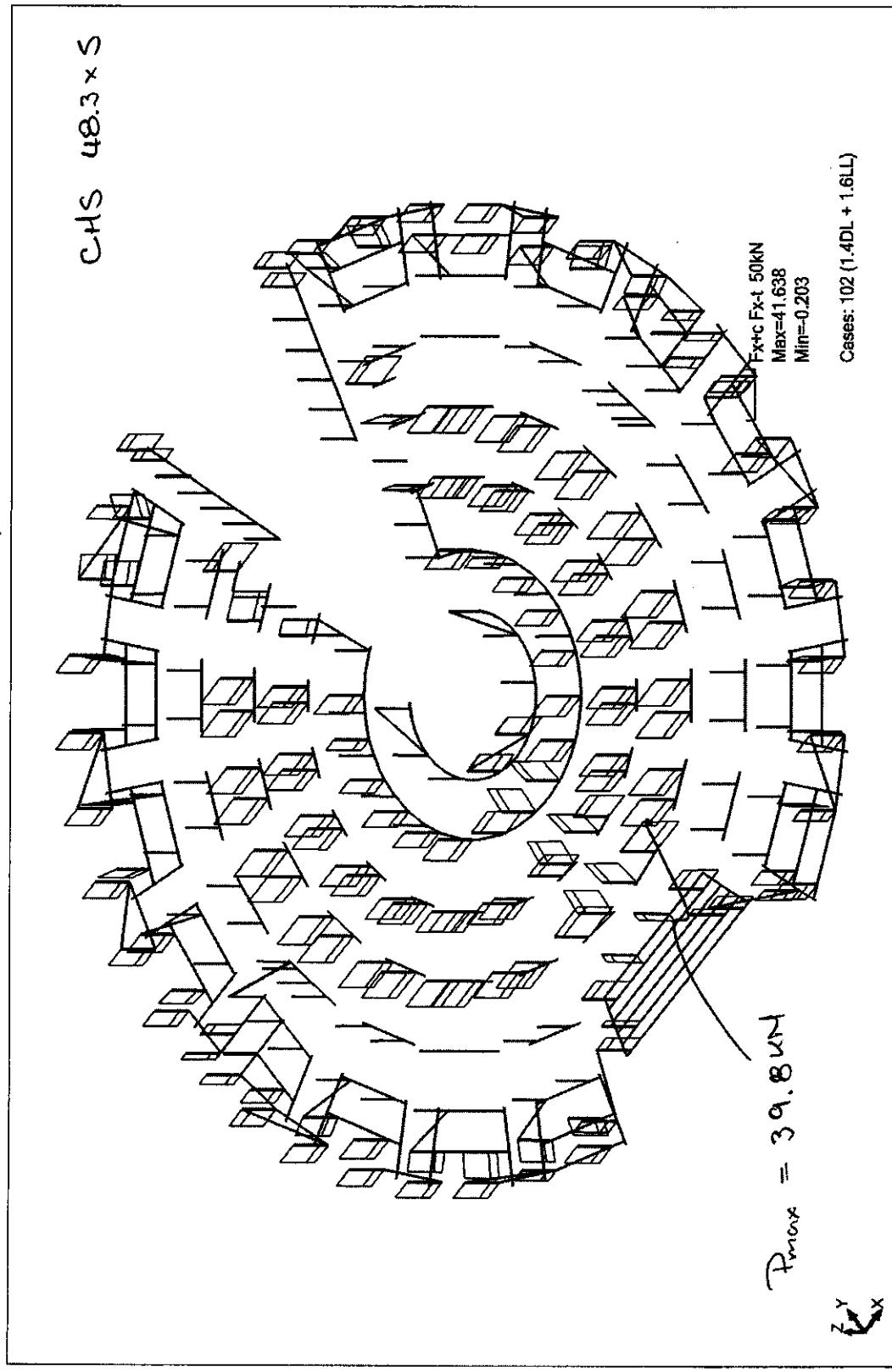
Bar: 100x5 SHS Columns

- Cases: 51to67 101to104

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX	49225	6167	6167			
Bar	6259	6197	100	6263	6259	6197
Node	3351	1553	99	3352	3351	1553
Case	103 (C)	104 (C)	54 (C)	103 (C)	103 (C)	104 (C)
MIN	-33888	-5167	-5167			
Bar	6399	6401	6400	6400	6256	6197
Node	3426	3451	3425	1975	1559	2106
Case	104 (C)	104 (C)	104 (C)	103 (C)	104 (C)	103 (C)

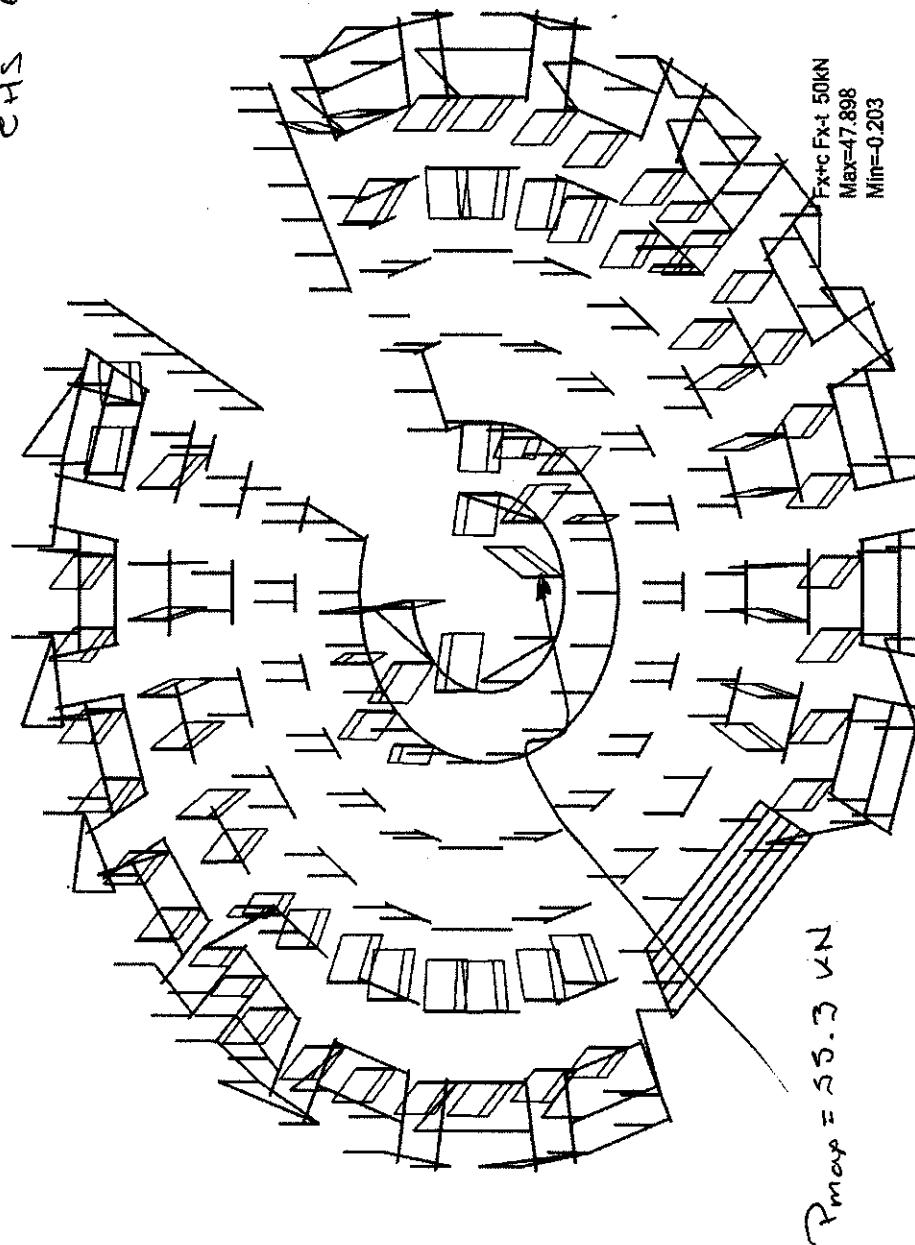
1578/3A4

View:2 - FX, Cases: 102 (1.4DL + 1.6LL) Axial Force Diagram - LOAD COMBINATIONS 1.4DL + 1.6LL



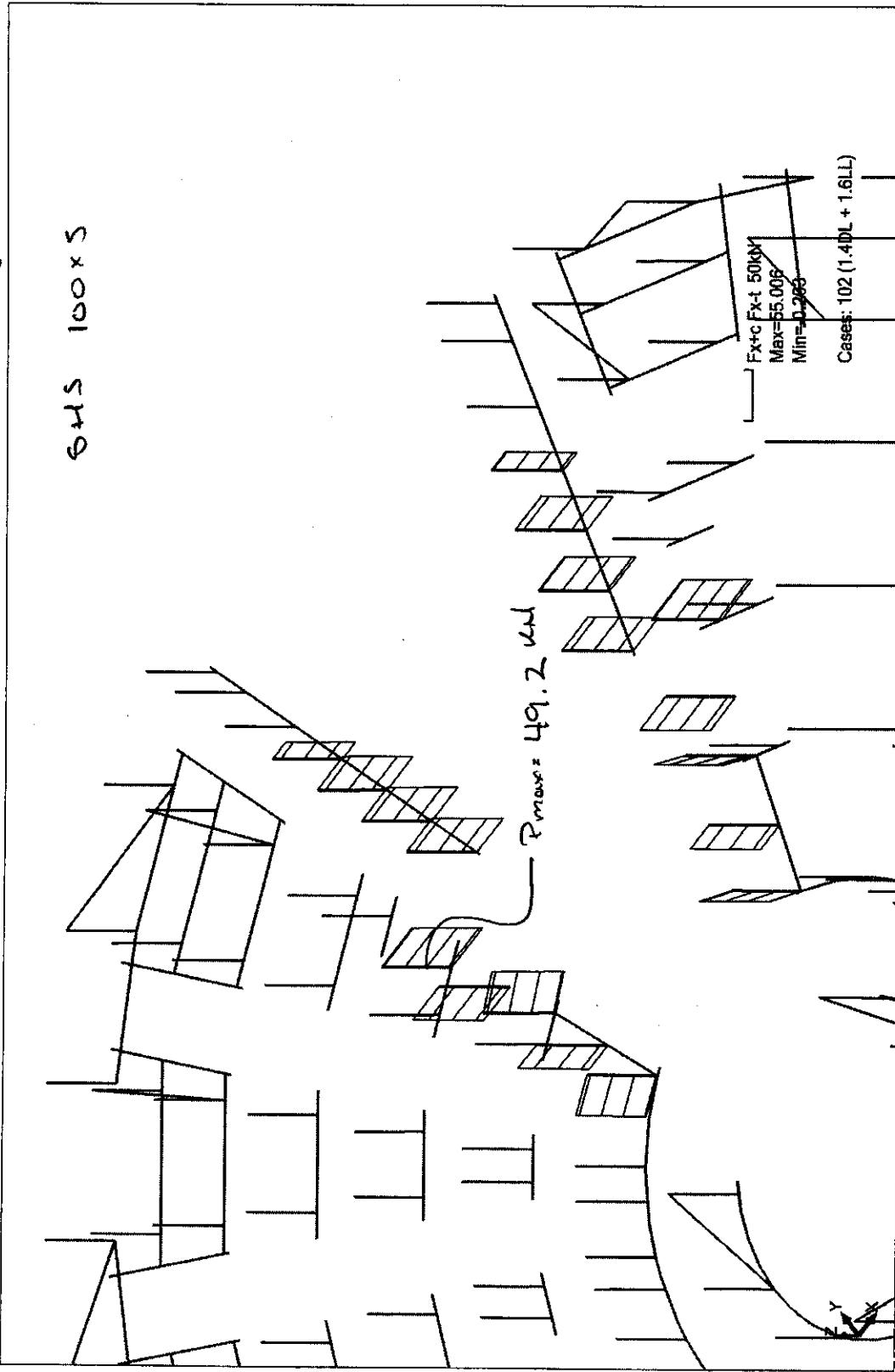
View:2 - FX, Cases: 102 (1.4DL + 1.6LL) AXIAL Force Diagram - LOAD COMBINATION 1.4 DL + 1.6 LL

C4S 60.3x5



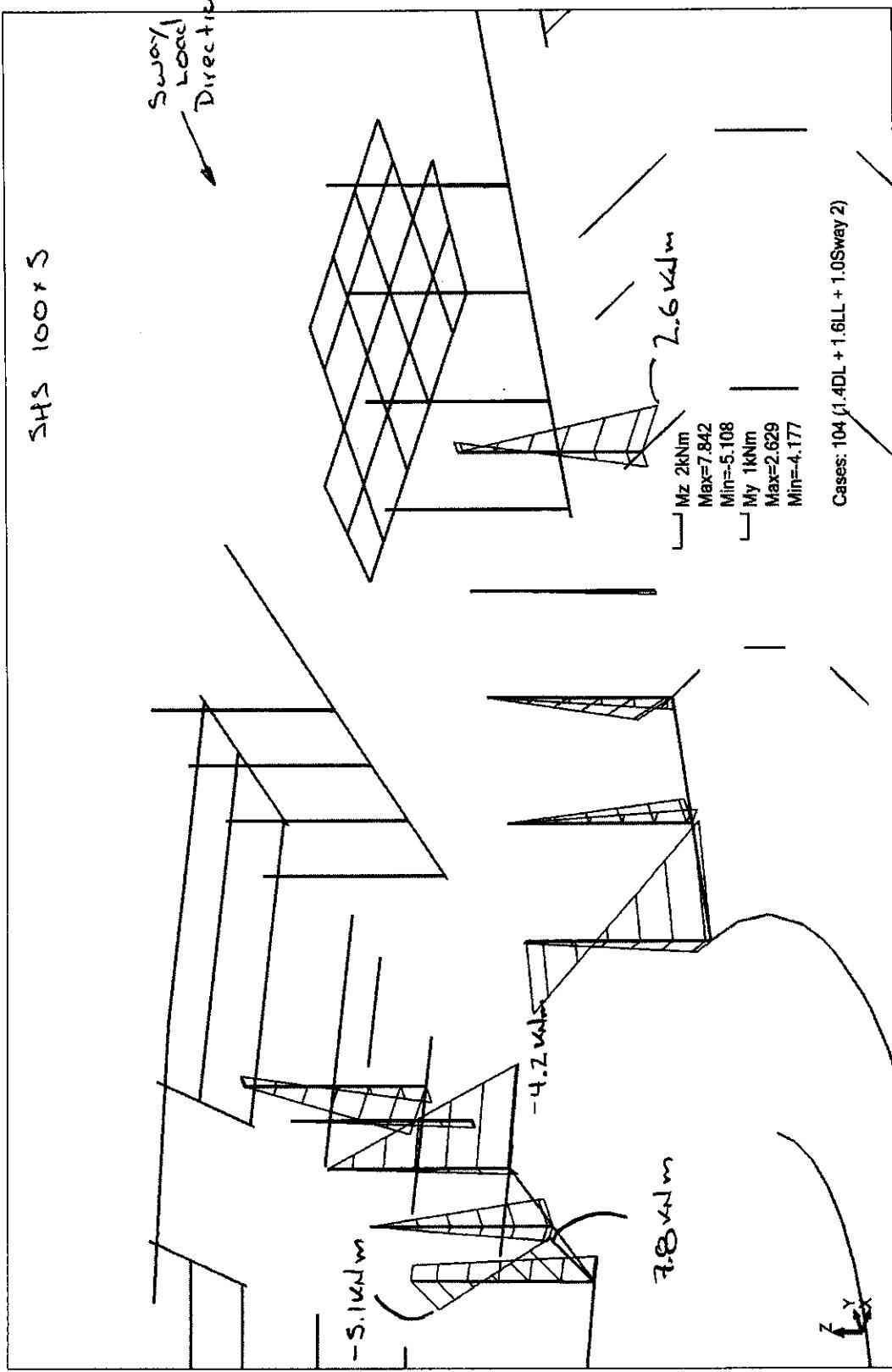
1578/3/17

View:2 - FX, Cases: 102 (1.4DL + 1.6LL) Axial Force Diagram - LOAD COMBINATION 1.4DL + 1.6LL



1578/3/18

View:1 - MY,MZ, Cases: 104 (1.4DL + 1.6LL + 1.0Sway 2) BMD - LOAD COMBINATION 1.4DL+1.6LL+1.0 Sway load



Forces - Cases: 1to4 51 101to104
Global extremes
1

- Cases: 1to4 51 101to104

Filtering	Bar	Case
Full list	1to88 92to97 10 1to4 51 101to10	
Selection	28 29 55 57 58 7 1to4 51 101to10	
Total number	2766	9
Selected number	294	9

- Cases: 1to4 51 101to104

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX	1496	1178	1300	1513	1264	1248
Bar	1496	1178	1300	1513	1264	1248
Point	1050	1235	1162	1087	5111	1142
Case	104 (C)	103 (C)	102 (C)	103 (C)	103 (C)	103 (C)
MIN	1173	1248	1264	1520	1112	1178
Bar	1173	1248	1264	1520	1112	1178
Point	1021	1152	1147	911	1015	1020
Case	103 (C)	103 (C)	103 (C)	102 (C)	103 (C)	103 (C)

1578/3/19

GLOBAL EXTREME FORCES

Case: All Load Combinations

Bar: 152x152x23 UC Spreader Beam

$b_e \leq 2300$ mm

Forces - Cases: 1to4 51 101to104
Global extremes
1

- Cases: 1to4 51 101to104

Filtering	Bar	Case
Full list	1to88 92to97 10 1to4 51 101to10	
Selection	59to67 1049to10 1to4 51 101to10	
Total number	2766	9
Selected number	267	9

- Cases: 1to4 51 101to104

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX						
Bar	1133	1141	1131	1100	1394	1141
Point	1448	1225	1241	1003	1485	1225
Case	103 (C)	103 (C)	103 (C)	103 (C)	104 (C)	103 (C)
MIN						
Bar	1216	1387	1426	67	64	1165
Point	1462	1306	1175	35	39	1225
Case	103 (C)	104 (C)	104 (C)	104 (C)	104 (C)	103 (C)

GLOBAL EXTREME FORCES

Cases : All Load Combinations

Bar : 152x152x23 JC spreader beam

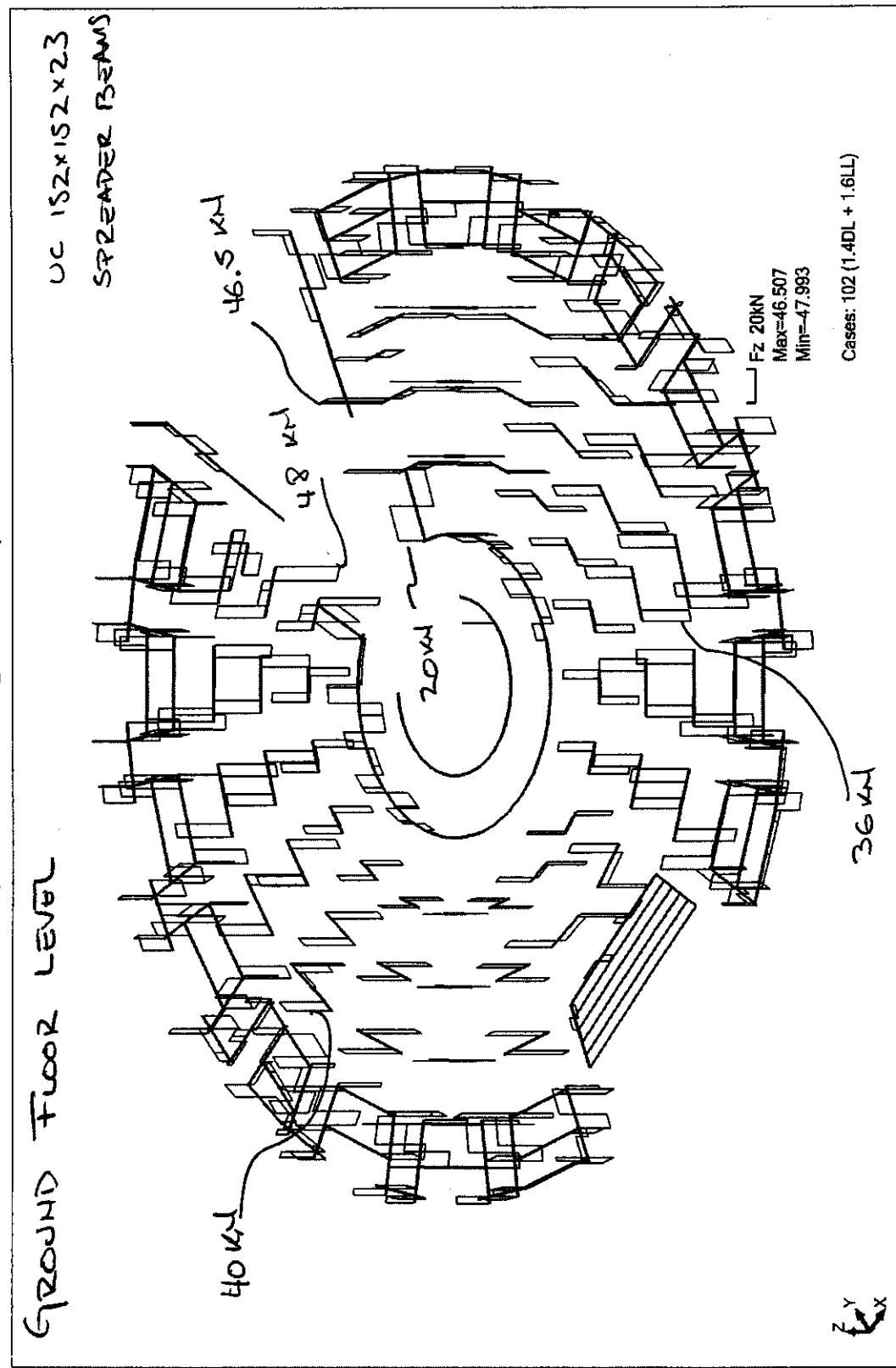
4300mm ≥ Le ≥ 2300 mm

1578/3/21

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View:2 - FZ, Cases: 102 (1.4DL + 1.6LL) SHEAR FORCE DIAGRAM - LOAD COMBINATION 1.4DL + 1.6LL



1578/3/22



ROBOT v 19.0.7

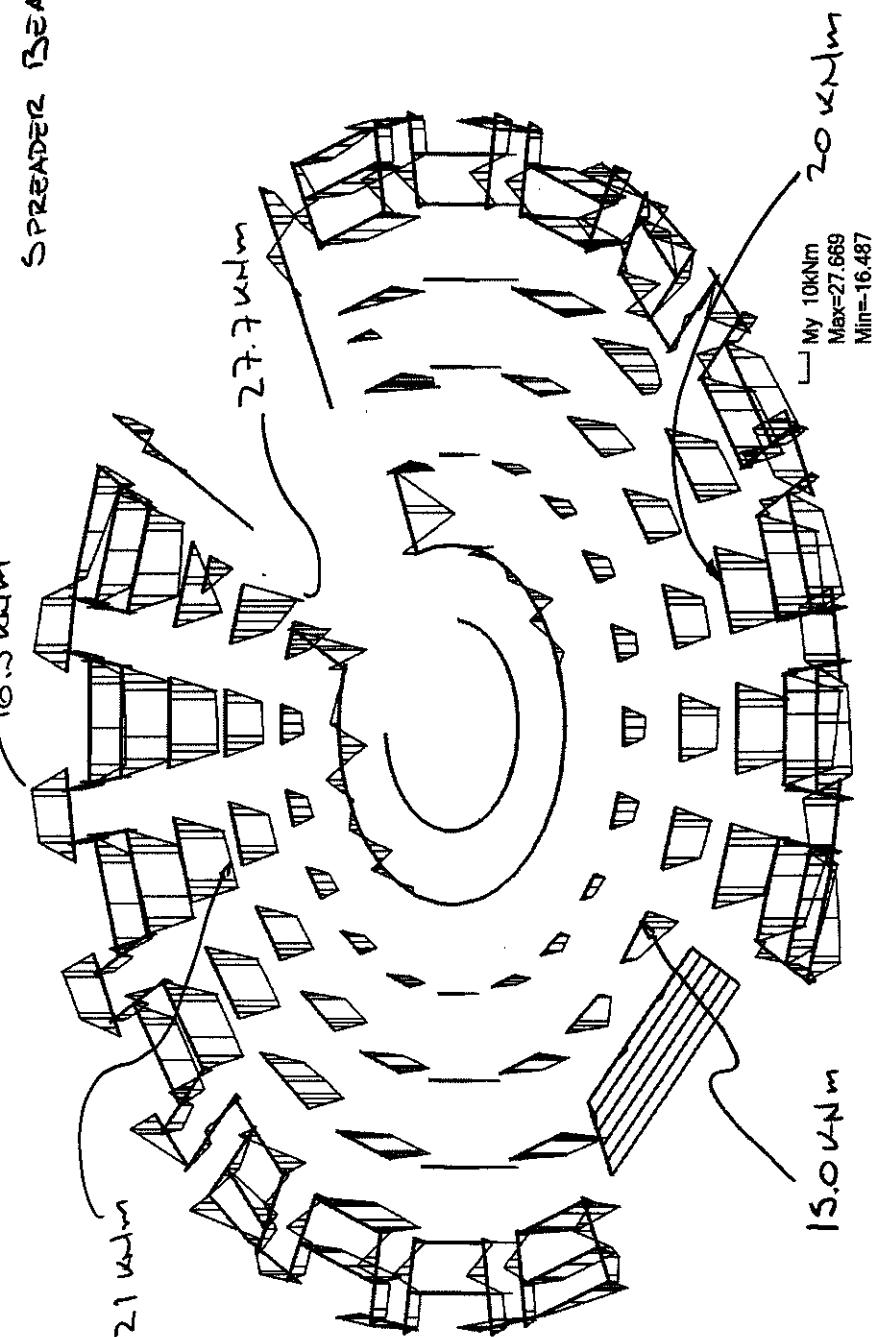
Author:
Address:

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Project: British Museum Platform

View:2 - MY, Cases: 102 (1.4DL + 1.6LL) BMD - LOAD COMBINATION 1.4DL + 1.6 LL

UC 152x152x23
SPREADER BEAMS

GROUND FLOOR LEVEL



Date : 05/02/07

Page : 1

Forces - Cases: 51to67 101to104
Global extremes
1

- Cases: 51to67 101to104

Filtering	Bar	Case
Full list	1to147 150 151 1to5 51to67 101to67 131to138 6370to51to67 101to104	
Selection		
Total number	2842	26
Selected number		21

STAIRCASE MEMBERS

GLOBAL EXTREME FORCES

- Cases: 51to67 101to104

Case: All load Combinations
Bar: 48.3x5 CHS Columns Under Entrance & Exit Staircases.

- Cases: 51to67 101to104

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX	1052					
Bar	6395	6372	6373	6395	6393	6377
Node	3476	3427	3426	2282	3472	3457
Case	102 (C)	103 (C)	103 (C)	102 (C)	103 (C)	102 (C)
MIN	2595	2595	2595	2595	2595	2595
Bar	6370	6381	6381	6393	6370	6371
Node	3434	3451	3461	2283	3429	3433
Case	55 (C)	55 (C)	104 (C)	102 (C)	103 (C)	103 (C)

Forces - Cases: 51to67 101to104

Global extremes

1

- Cases: 51to67 101to104

Filtering	Bar	Case
Full list	1to147 150 151 1to5 51to67 101to104	
Selection	100 106 140to145 1to67 101to104	
Total number	2842	26
Selected number	5	21

- Cases: 51to67 101to104

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX	10/106	10/224	10/257	10/243	10/224	10/256
Bar	140	142	141	142	140	142
Point	13/21	20/21	1976	2282	10/21	19/21
Case	104 (C)	104 (C)	104 (C)	101 (C)	104 (C)	104 (C)
MIN	8/21	20/21	7/21	20/21	7/21	11/21
Bar	142	140	141	140	141	140
Point	55 (C)	103 (C)	102 (C)	103 (C)	102 (C)	103 (C)
Case						

GLOBAL EXTREME FORCES

Case: All Load Combinations

Bar: 100x5 SHS STRINGERS EXIT & ENTRANCE

Forces - Cases: 51to67 101to104
Global extremes
1

- Cases: 51to67 101to104

Filtering	Bar	Case
Full list	1to147 150 151	1to5 51to67 101t
Selection	10 22 24 25 30to51to67	101to104
Total number	2842	26
Selected number	85	21

CASE: All Load Combinations

102x102x23 RSJ SPREADER BEAMS.

- EXIT

- ENTRANCE

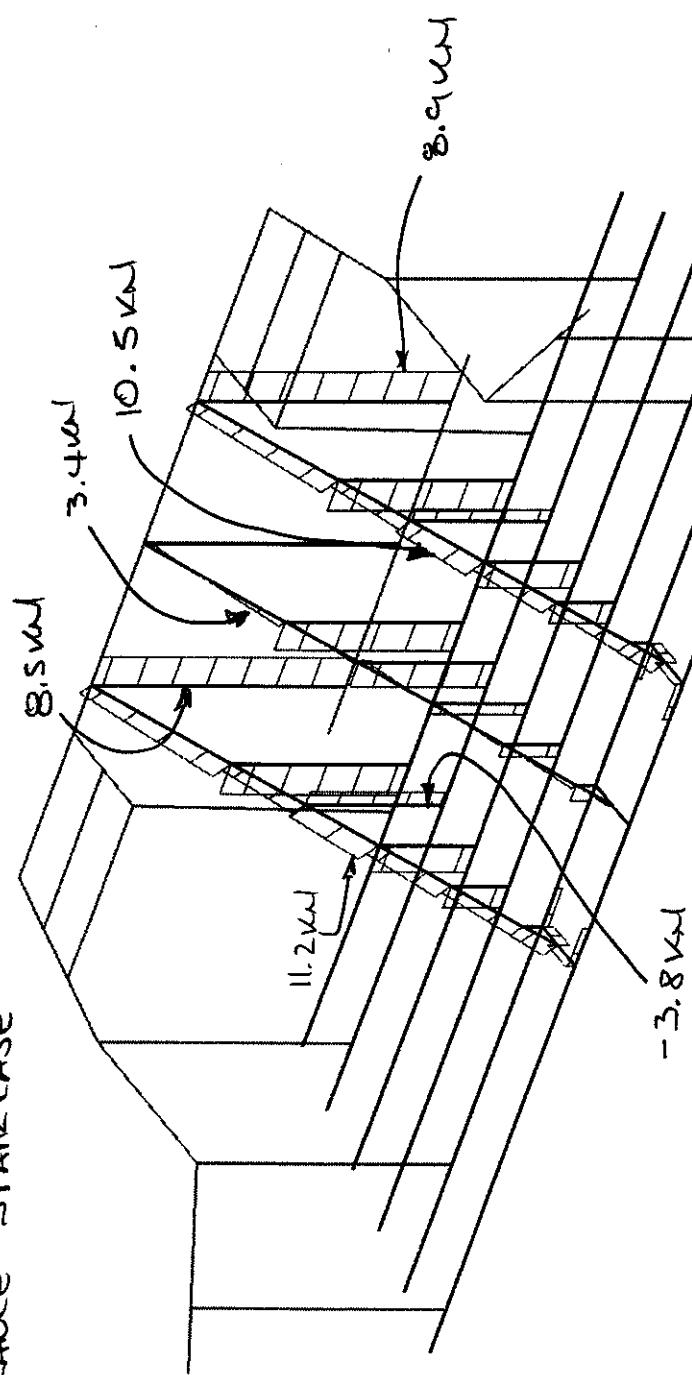
GLOBAL EXTREME FORCES

- Cases: 51to67 101to104

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX						
Bar	48	6324	41	6328	44	6329
Node	3377	1069	1069	1072	27	1068
Case	101 (C)	101 (C)	102 (C)	103 (C)	103 (C)	101 (C)
MIN						
Bar	22	6329	45	6323	44	6324
Node	11	1070	1070	1071	26	1067
Case	101 (C)	101 (C)	103 (C)	102 (C)	103 (C)	101 (C)

View:1 - FX, Cases: 104 (1.4DL + 1.6LL + 1.0Sway 2) Axial Force Diagram - LOAD COMB: 1.4DL+1.6LL+1.0Sway 2

ENTRANCE STAIR CASE



Legend: $F_x+c F_x \cdot t$ 10kN
Max=11.172
Min=-3.840
Cases: 104 (1.4DL + 1.6LL + 1.0Sway 2)



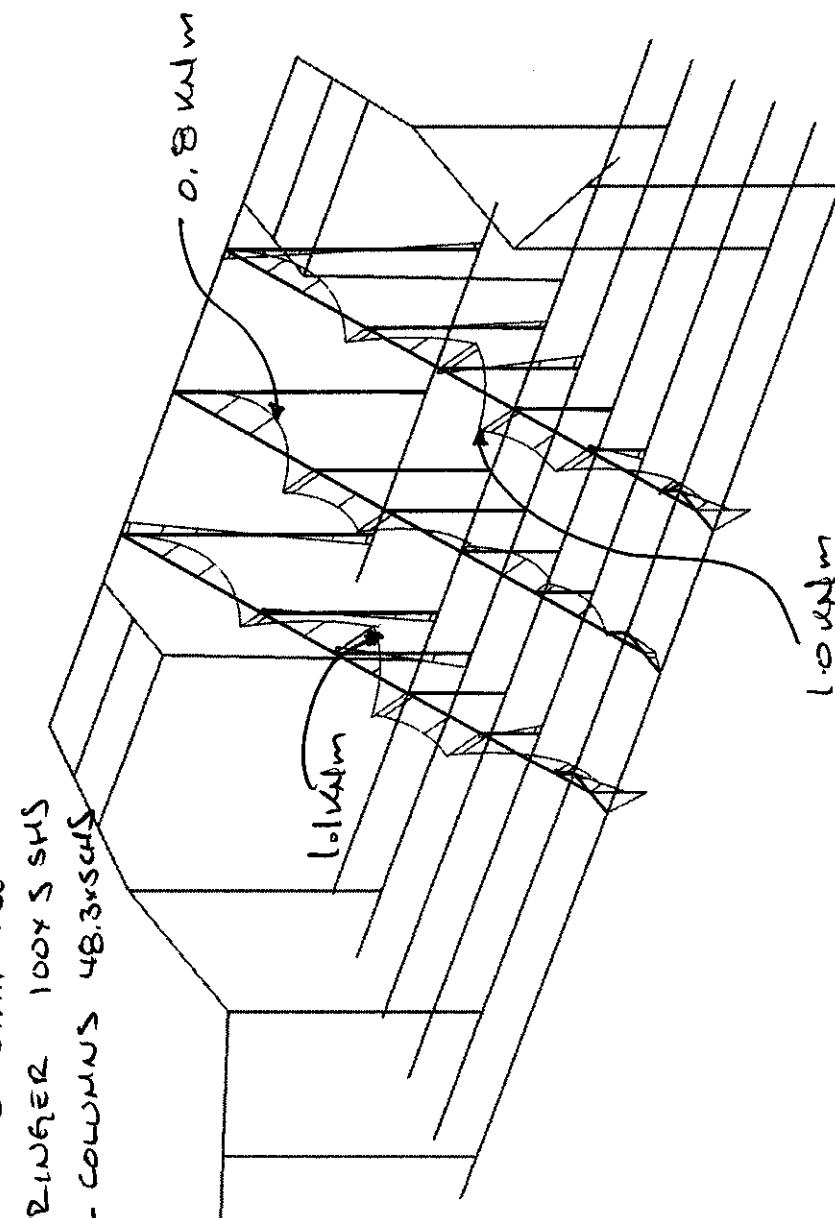
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View:1 - MY, Cases: 104 (1.4DL + 1.6LL + 1.0Sway 2) BMD - LOAD COMBINATIONS 1.4DL + 1.6LL + 1.0 Sway 2

ENTRANCE STAIRCASE:

STRUCTURAL 100x5 SHS
2 COLUMNS 48.3x5x4.5



Z Y X

Cases: 104 (1.4DL + 1.6LL + 1.0Sway 2)

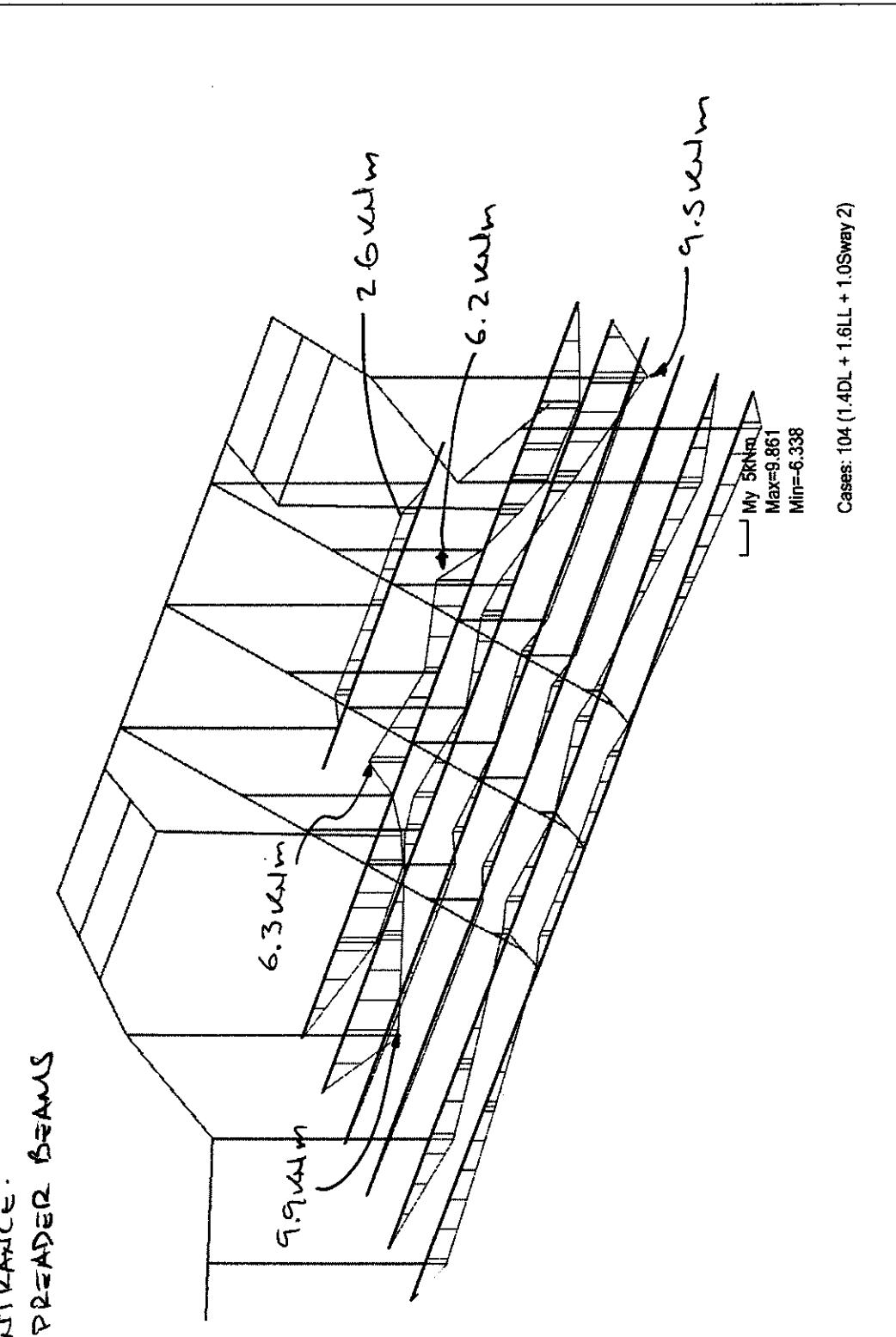
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View:1 - MY, Cases: 104 (1.4DL + 1.6LL + 1.0Sway 2) BMD - LOAD COMBINATIONS 1.4DL+1.6LL+1.0 Sway 2

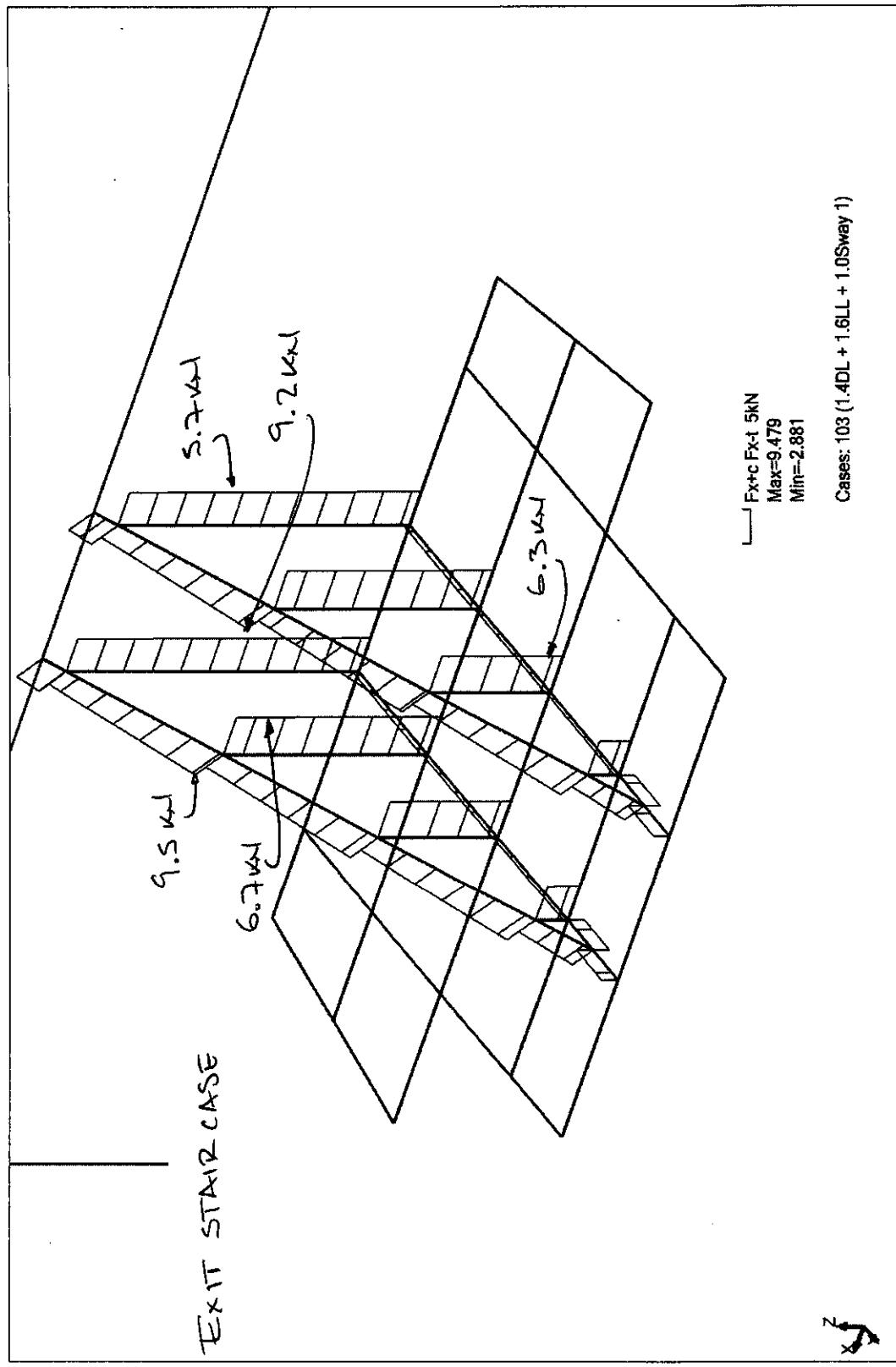
ENTRANCE:
SPREADER BEAMS



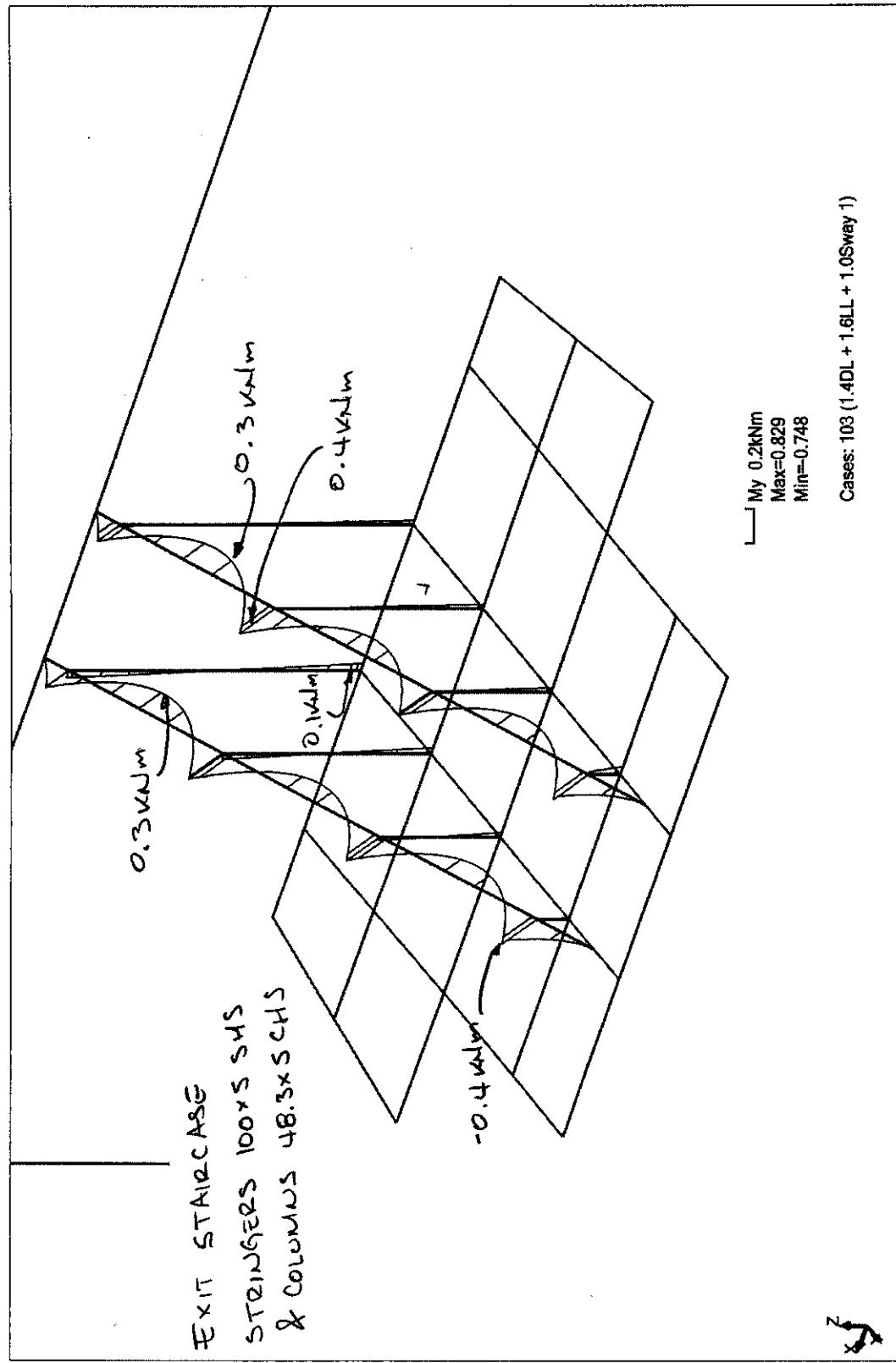
Date : 19/02/07

Page : 1

View:2 - FX, Cases: 103 (1.4DL + 1.6LL + 1.0Sway 1) Axial Diagram Force - LOAD Combination 1.4DL+1.6LL+1.0SL



View:2 - MY, Cases: 103 (1.4DL + 1.6LL + 1.0Sway 1) BMD - LOAD COMBINATION 1.4DL + 1.6LL + 1.0 Sway 1

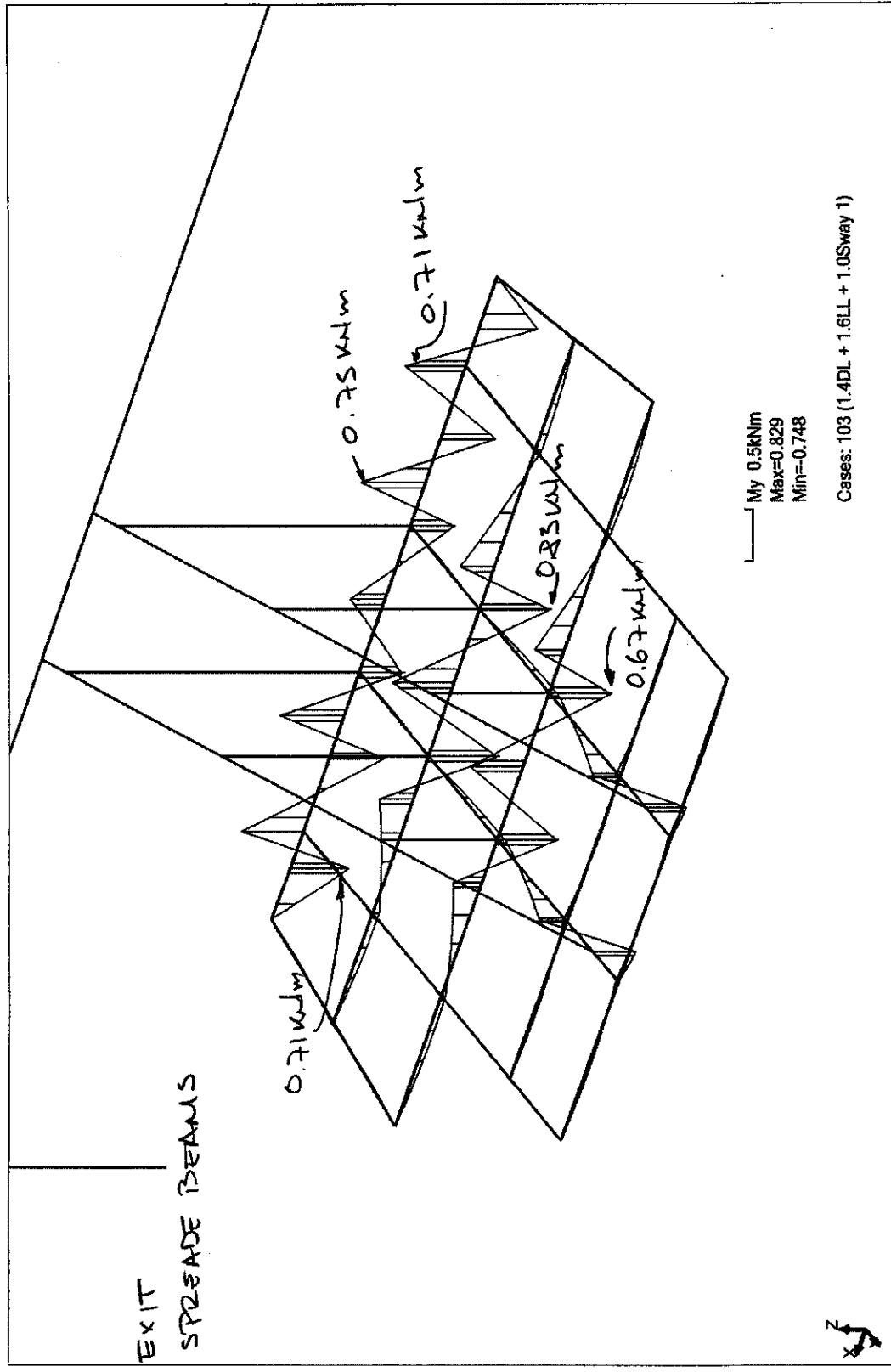


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View:2 - MY, Cases: 103 (1.4DL + 1.6LL + 1.0Sway 1)

BMD - LOAD COMBINATION 1.4DL + 1.6LL + 1.0 Sway 1



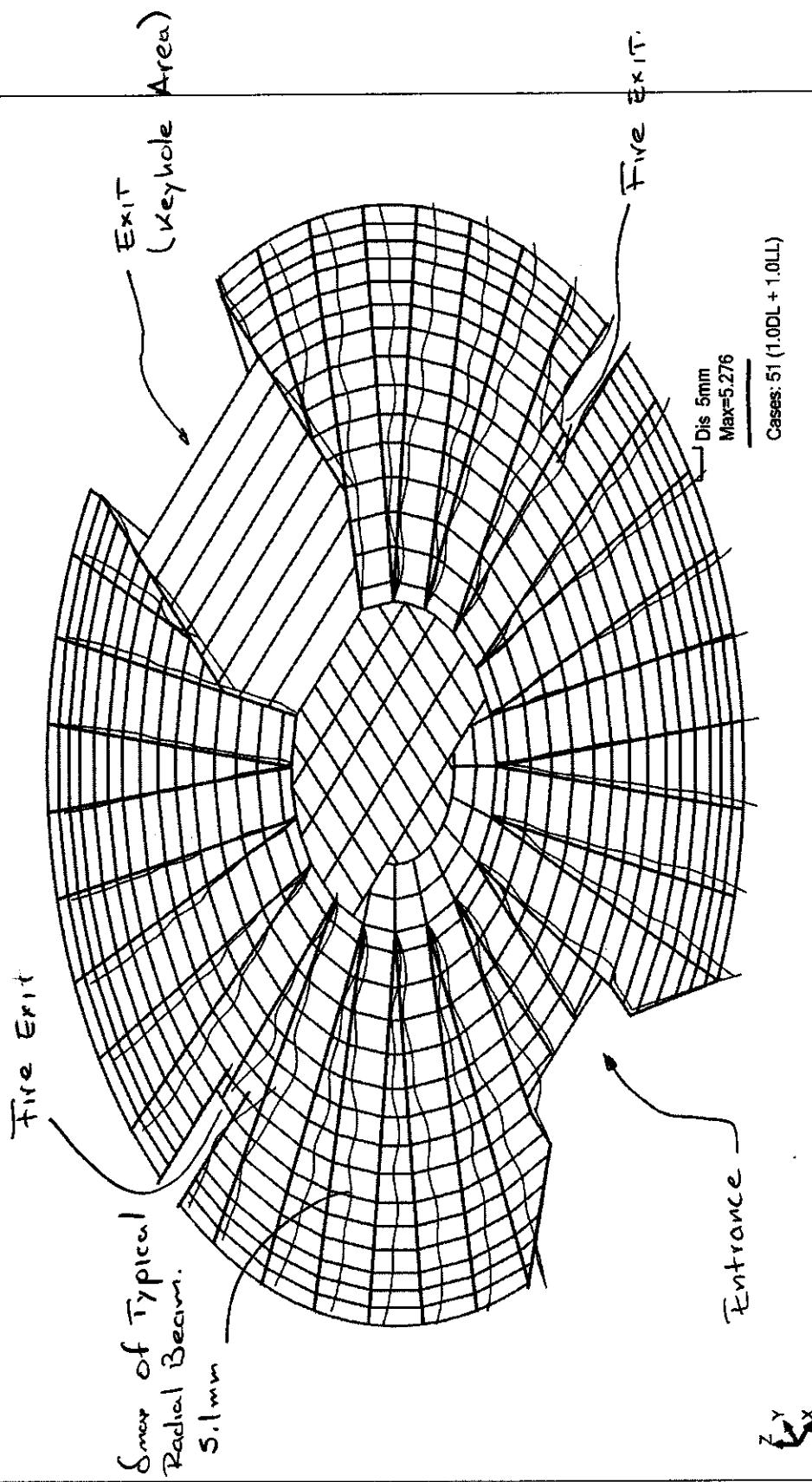
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File: British Museum Platform.rtd
Project: British Museum Platform

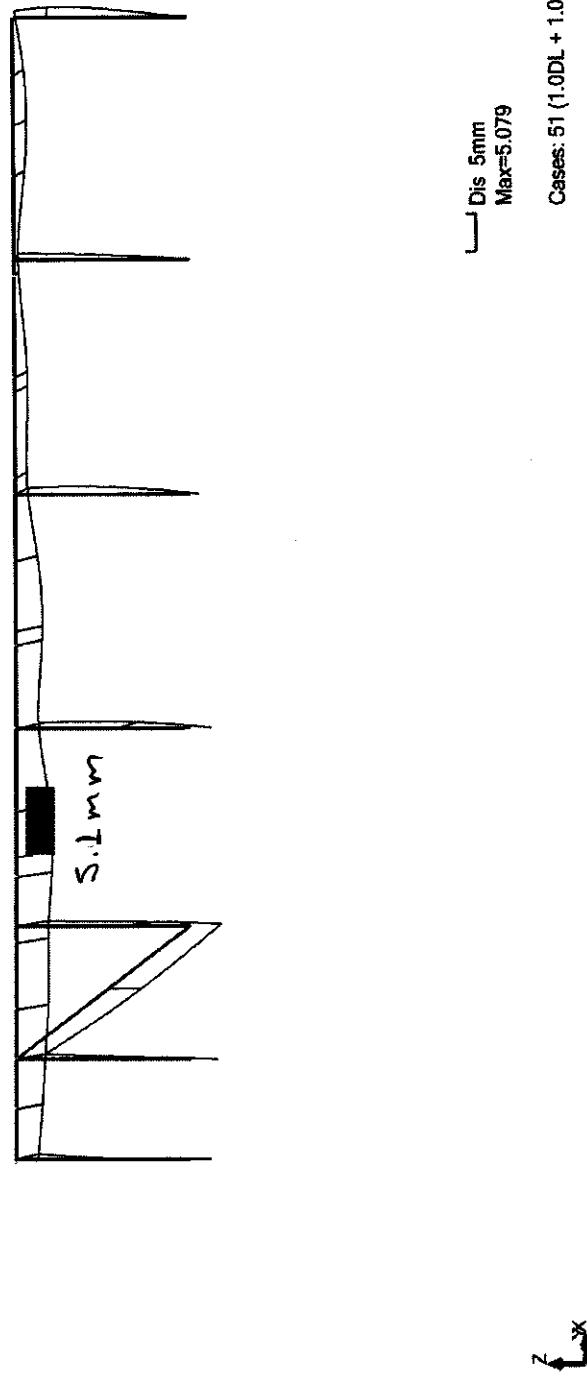
View:1 - Exact deformation(s), Cases: 51 (1.0DL + 1.0LL) DEFLECTIONS - LOAD COMBINATION 1.0DL+1.0LL

Radial Beams DPFC 175x65x2.4



View:2 - Exact deformation(s), Cases: 51 (1.0DL + 1.0LL) DEFLECTION - LOAD COMBINATION 1.0DL + 1.0LL

Typical Radial Beam DPFC 175x65x2.4



Cases: 51 (1.0DL + 1.0LL)

Dis 5mm
 Max=5.079

atelier one

Project BRITISH MUSEUM

No. 1578/4/1

Date 02/07 By LF

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Notes

MEMBER CAPACITY CHECK

& DEFLECTION CHECK

atelier one

Project BRITISH MUSEUM

No 1578/4/2

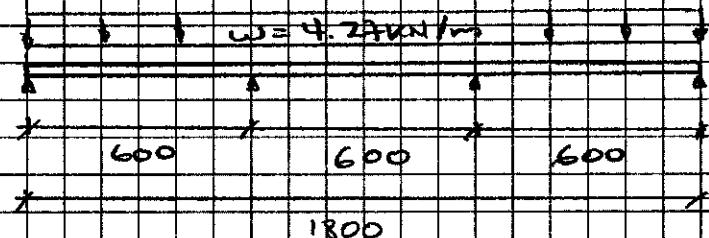
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Notes

TIMBER FLOOR CHECK

2 layers of ply 12.5mm thick.



$$\text{Span} = 600 \text{ mm}$$

Use ply 1.8m min length to be distributed over 3 spans

Bending Check

$$M = \frac{wL^2}{10} = \frac{4.27 \times 0.6^2}{10} \therefore M = 0.15 \text{ kNm/m}$$

$$I = \frac{1000 \times 12.5^3}{6}$$

$$I = 26042 \text{ mm}^3$$

$$\sigma_{\text{m/1}} = \frac{0.15 \times 10^6}{26042} \therefore \sigma_{\text{m/1}} = 5.9 \text{ N/mm}^2$$

$$\sigma_{\text{m/1}} = 10.4 \text{ N/mm}^2 > \sigma_{\text{m/1}}$$

\hookrightarrow 12.5mm Canadian Douglas fir plywood

V.O.K.

atelier one

Project BRITISH MUSEUM

No 1578/4/3

Date 02/07 By LF

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Notes

Deflection check

$$w = 4.27 \text{ kN/m}^2$$

$$\delta = \frac{2.1 w l^4}{384 EI}$$

two layers 12.5 mm

$$w = 2.14 \text{ kN/m per layer}$$

$$I = \frac{1000 \times 12.5^3}{12}$$

$$I = 162760 \text{ mm}^4$$

$E = 5490 \text{ N/mm}^2$ Canadian Douglas fir
plywood

$$\delta = \frac{2.1 \times 2.14 \times 0.64}{384 \times 0.893} = 0.0017$$

$$\delta = 1.7 \text{ mm}$$

$$\delta_{adim} = 600 \times 0.003$$

$$\delta_{adim} = 1.8 \text{ mm} \geq \delta = 1.7 \text{ mm}$$

✓OK

atelier one

Project BRITISH MUSEUM

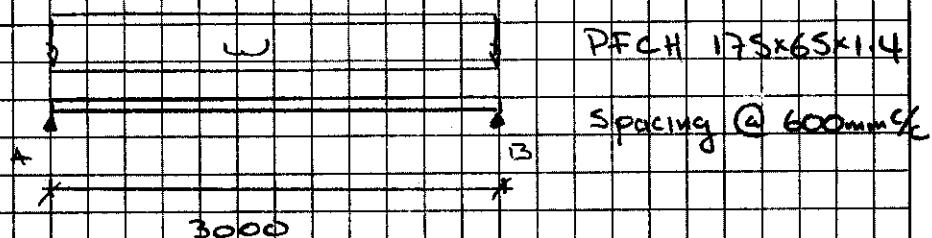
No. 1578/4/4

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Notes

Secondary steel beam - Circumferential Beams.



w : Dead Load (inc. $5w$) — 0.2 kN/m

Live Load (4.0×0.6) — 2.4 kN/m

SLS — $w = 2.6 \text{ kN/m}$

ULS — $w = 4.12 \text{ kN/m}$

Bending Moment

$$M = \frac{wL^2}{8} \quad | \text{ SLS} \Rightarrow M = \frac{2.6 \times 3.0^2}{8}$$

$$\text{ULS} \Rightarrow M = \frac{4.12 \times 3.0^2}{8} \quad | \quad M = 2.93 \text{ kNm}$$

$$M = 4.64 \text{ kNm}$$

Shear Force

$$R_A = R_B = \frac{w \times L}{2}$$

$$\text{SLS} \Rightarrow R_A = R_B = \frac{2.6 \times 3}{2}$$

$$R_A = R_B = 3.9 \text{ kN}$$

$$| \quad \text{ULS} \Rightarrow R_A = R_B = \frac{4.12 \times 3}{2}$$

$$R_A = R_B = 6.2 \text{ kN}$$

atelier one

Project BRITISH MUSEUM

No. 1578/4/S

Date 02/07 By LF

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Notes

Use 175x65x1.4 PFCI Channel Section

Material Properties

$$E = 205 \text{ kN/mm}^2$$

$$\rho_y = 280 \text{ N/mm}^2$$

Section Properties

$$t = 1.4 \text{ mm}$$

$$I_{xx} = 196.7 \text{ cm}^4$$

$$D = 175 \text{ mm}$$

$$f_y = 20 \text{ mm}$$

$$I_{x1} = 159.5 \text{ cm}^4$$

$$Z_{x1} = 1822.5 \text{ mm}^3$$

Moment Resistance

Compression flange fully restrained.

$$M_{cr} = Z_{x1} \rho_y$$

$$M_{cr} = 1822.5 \times 280 / 10^6$$

$$M_{cr} = 5.1 \text{ kNm} > 4.64 \text{ kNm}$$

J.O.U.

Shear Resistance

Shear yield strength.

$$\rho_y = 0.6 \rho_y = 0.6 \times 280 = 168 \text{ N/mm}^2$$

atelier one

Project BRITISH MUSEUM

No. 1578/4/6

Date 02/07 By LF

Checked _____

Notes

* Shear buckling strength

$$q_{cr} = \left(\frac{1000t}{D} \right)^2 = \left(\frac{1000 \times 1.4}{175} \right)^2$$

$q_{cr} = 64 \text{ N/mm}^2$ — which is more critical

* Max Shear Force, $F_v \text{ max} = 6.2 \text{ kN}$

$$\text{Shear area} \quad 175 \times 1.4 = 245 \text{ mm}^2$$

* Average shear stress,

$$f_v = \frac{6.2 \times 10^3}{245} = 25.3 \text{ N/mm}^2 < q_{cr}$$

✓ O.K.

Connection check

$$R_A = R_B = 6.2 \text{ kN}$$

M12 Shear Capacity

$$P_v = 31.6 \text{ kN} \quad \text{— Grade 8.8}$$

✓ O.K.

1.4 mm thick Web Bearing Cap

$$P_{bs} = 0.65 (280 + 360)$$

$$P_{bs} = 416 \text{ N/mm}^2$$

$$P_{bs} = c \times t \times P_{bs} \therefore P_{bs} = 12 \times 1.4 \times 416 / 10^3$$

$$P_{bs} = 7.0 \text{ kN} \geq 6.2 \text{ kN}$$

✓ O.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/7

Date 02/07 By LF

Checked _____

Notes

Deflection check

$$\delta_{adm} = \frac{le}{360} = \frac{3000}{360}$$

$$\delta_{adm} = 8.3 \text{ mm}$$

$$S_m = \frac{5 \times w L^4}{384 EI_{av}}$$

$$I_{av} = \frac{I_{xx} + I_{yy}}{2} = \frac{196.7 + 159.5}{2}$$

$$I_{av} = 173.1 \times 10^4 \text{ mm}^4$$

$$EI_{av} = 365.1 \text{ kNm}^2$$

$$S_m = \frac{5 \times 2.6 \times 3^4}{384 \times 365.1} = 7.5 \times 10^{-3} \text{ m}$$

$$\delta_m = 7.5 \text{ mm} \leq \delta_{adm} = 8.3 \text{ mm}$$

✓ O.K.

atelier one

Project BRITISH MUSEUM

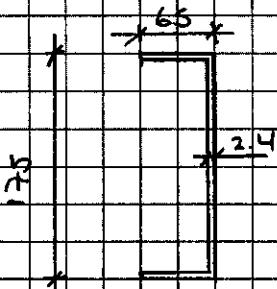
No. 1578/4/8

Date 02/07 By LF

Checked _____

Notes

SINGLE PECH 175x65x2.4



Material Properties

$$E = 205000 \text{ N/mm}^2$$

$$\rho_y = 280 \text{ N/mm}^2$$

Section Properties

$$t = 2.4 \text{ mm}$$

$$I_{xx} = 3309901 \text{ mm}^4$$

$$D = 175 \text{ mm}$$

$$r_y = 19.8 \text{ mm}$$

$$I_{xr} = 3278082 \text{ mm}^4$$

$$Z_{xr} = 37437 \text{ mm}^3$$

Bending Moment Resistance — Restrained Beam

$$M_{cr} = Z_{xr} \times \rho_y$$

$$M_{cr} = 37437 \times 280 / 10^6$$

$$M_{cr} = 10.5 \text{ kNm} > 7.2 \text{ kNm}$$

✓OK.

Shear Resistance

* Shear yield strength

$$\rho_v = 0.6 \rho_y = 0.6 \times 280 = 168 \text{ N/mm}^2$$

atelier one

Project BRITISH MUSEUM

No. 1578/4/9

Date 02/07 By LF

Checked _____

Notes

* Shear buckling strength

$$q_{ter} = \left(\frac{1000t}{D} \right)^2 = \left(\frac{1000 \times 2.4}{175} \right)^2$$

$$q_{ter} = 188.1 \text{ N/mm}^2 \Rightarrow f_v = 168 \text{ N/mm}^2 -$$

more critical

* Max Shear Force

$$F_v, \text{max} = 6.4 \text{ kN}$$

$$\text{Shear area} - A_v = 175 \times 2.4 = 420 \text{ mm}^2$$

* Average Shear stress,

$$f_v = \frac{6.4 \times 10^3}{420} = 15.2 \text{ N/mm}^2 \leq f_v$$

J.O.K.

Connection Check

$$F_v = 6.4 \text{ kN} < 7.0 \text{ kN} \text{ Bearing Cap.}$$

1.4 mm thick Channel steel

J.O.K.

Check by inspection

atelier one

Project BRITISH MUSEUM

No. 1578/4/10

Date 02/07 By LF

Checked _____

Notes

Deflection Check

$$L_e = 2370 \text{ mm}$$

$$\delta_{adm} = \frac{2370}{360}$$

$$\delta_{adm} = 6.6 \text{ mm}$$

$$\delta_m = 5.5 \text{ mm} \leq \delta_{adm} 6.6 \text{ mm}$$

From Results
Summary
V.O.K.

atelier one

Project BRITISH MUSEUM

No. 15784/11

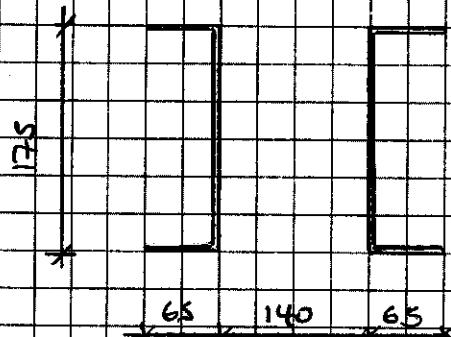
Date 02/07 By LF

Checked _____

Notes

RADIAL BEAMS

DOUBLE CHANNELS



DPC 175x65x2.4

2No 175x65x2.4 PFCH

Back to Back

Material Properties

$$E = 205 \text{ kN/mm}^2$$

$$P_y = 280 \text{ N/mm}^2$$

Section Properties

$$t = 2.4 \text{ mm}$$

$$I_{xx} = 6619802 \text{ mm}^4$$

$$D = 175 \text{ mm}$$

$$C_y = 19.3 \text{ mm}$$

$$I_{x,r} = 6556164 \text{ mm}^4 \quad Z_{x,r} = 74878 \text{ mm}^3$$

Bending Moment Resistance

The compression flange is fully restrained

over the sagging moment region but is unrestrained

over the hogging moment region, that is, over

atelier one

Project BRITISH MUSEUM

No. 1578/4/12

Date 02/07 By LF

Checked _____

Notes

the internal supports Unrestrained 1.2 m long

The moment resistance of the restrained section

$$M_{cy} = Z_c \times P_y$$

$$M_{cy} = 74878 \times 285 / 10^6$$

$$\begin{aligned} M_{cy} &= 21.0 \text{ kNm} &> 12.5 \text{ kNm} \\ &&> 19.3 \text{ KNm} \end{aligned}$$

Buckling Resistance Moment of Unrestrained Sect.

$L_e = 1250 \text{ mm}$ — Effective length

$$\frac{L_e}{r_y} = \frac{1250}{121.93} = 63.1$$

$$\lambda_{LT} = \frac{\mu_{uv} \frac{L_e}{r_y}}{\lambda_y}$$

where $\mu_{uv} = 0.65$

$$\lambda_y = 85$$

$$\lambda_{LT} = \frac{0.65 \times 63.1}{85}$$

$$\lambda_{LT} = 0.48 \quad \therefore M_b = 1.0 \text{ My}$$

$$\underline{M_{cy} = M_b = 21.0 \text{ kNm}}$$

S.O.K.
(keyhole area)

atelier one

Project BRITISH MUSEUM

No. 1578/4/13

Date 02/07 By LF

Checked _____

Notes

Shear Resistance

* Shear yield strength

$$P_V = 0.6 P_y = 0.6 \times 280 = 168 \text{ N/mm}^2$$

* Shear buckling strength

$$q_{fcr} = \left(\frac{1000t}{D} \right)^2 = \left(\frac{1000 \times 2.4}{175} \right)^2$$

$$q_{fcr} = 182.1 \text{ N/mm}^2 \geq P_V = 168 \text{ N/mm}^2 \text{ ---}$$

* Max Shear Force

$$F_V = 270 \text{ kN}$$

$$\text{Shear area} = 2 \times 175 \times 2.4 = 840 \text{ mm}^2$$

* Average shear stress.

$$\bar{f}_V = \frac{270 \times 10^3}{840} = 32.1 \text{ N/mm}^2 \leq P_V$$

more critical

J.O.K.

atelier one

Project BRITISH MUSEUM

No 1578/4/14

Date 02/07 By LF

Checked _____

Notes _____

Web Crushing at End Support

check the limits of the formulae.

$$\frac{D}{t} = \frac{175}{2.4} = 73 \leq 200 \quad \text{Y.O.K.}$$

- Bearing length $N = 50 \text{ mm}$

For $c=0$, $N/t = 20.8$ and restrained section,

$$P_w = t^2 C_7 \rho_y (8.8 + 1.11 \sqrt{N/t})$$

$$C_7 = 1 + \frac{D/t}{750} = 1.1$$

$$P_w = 2.4^2 \cdot 1.1 \times 280 (8.8 + 1.11 \sqrt{20.8}) \times 10^{-3}$$

$P_w = 24.5 \text{ kN}$ — per channel section

$$P_w = 2 \times 24.5 = 49.0 \text{ kN} > R_{max} = 23.8 \text{ kN}$$

✓ O.K.

Web Crushing at Internal Support

At internal support, the bearing length, N , is 50mm

For $c > 1.5 D$ $N/t = 20.8$ and restrained section,

$$P_w = t^2 C_5 C_6 \rho_y (13.2 + 2.87 \sqrt{N/t})$$

$$K = \frac{280}{228} = 1.23$$

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Project BRITISH MUSEUM

No. 1578/4/15

Date 02/07 By LF

Checked _____

Notes _____

$$C_s = 1.49 - 0.53 \times 4 = 0.84$$

$$C_b = 0.88 + 0.12 \times (2.4/1.9) = 1.03$$

$$P_w = 2.4^2 \times 0.84 \times 1.03 \times 280 (13.2 + 2.87 \sqrt{20.8})$$

$$P_w = 36.7 \text{ kN} \quad \text{per channel section}$$

$$P_w = 2 \times 36.7 = 73.4 \text{ kN} < 44.3 \text{ kN}$$

V.O.K.

Combined Bending and Web Crushing at Internal Support

$$1.1 \left(\frac{F_w}{P_w} \right) + \left(\frac{M_x}{M_{cr}} \right) \leq 1.5$$

$$F_w = 44.3 \text{ kN} \quad P_w = 73.4 \text{ kN}$$

$$\frac{F_w}{P_w} \leq 1.0 \quad \text{J.o.k.}$$

$$M_x = 12.5 \text{ kNm} \quad M_c = 21.0 \text{ kNm}$$

$$\frac{M_x}{M_c} \leq 1.0$$

$$1.1 \left(\frac{44.3}{73.4} \right) + \left(\frac{12.5}{21.0} \right) = 1.26 \leq 1.5$$

V.O.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/16

Date 02/07 By LF

Checked _____

Notes

keyhole area check

$$1.1 \left(\frac{F_w}{P_w} \right) + \left(\frac{M_x}{M_c} \right) \leq 1.5$$

$$F_w = 27.1 \text{ kN} \quad P_w = 73.4 \text{ kN}$$

$$\frac{F_w}{P_w} \leq 1.0 \quad \text{OK.}$$

$$M_x = 19.2 \text{ kNm} \quad M_c = 21.0 \text{ kNm}$$

$$\frac{M_x}{M_c} \leq 1.0$$

$$1.1 \left(\frac{27.1}{73.4} \right) + \left(\frac{19.2}{21.0} \right) = 1.32 \leq 1.5$$

OK.

atelier one

Project BRITISH MUSEUM

No. 1578/4/17

Date 02/07 By LF

Checked _____

Notes _____

Deflection Check

Radial Beams

$$\delta_{max} = 7.5 \text{ mm}$$

Load Combination 100% L1 + 100% L2

Relative deflection

$$① \delta_m = 7.5 - \left(\frac{5.8 + 7.3}{2} \right)$$

$$\delta_m = 1.0 \text{ mm}$$

$$L_e = 1501$$

$$\frac{L_e}{360} = \frac{1501}{360} = 4.1 \text{ mm} \geq 1.0 \text{ mm}$$

$$② \delta_m = 7.3 - 4.4 = 2.9 \text{ mm}$$

$$L_e = 1182$$

$$\frac{L_e}{360} = \frac{1182}{360} = 3.3 \text{ mm} \geq 2.9 \text{ mm}$$

Results
Summary

✓ O.K.

✓ O.K.

atelier one

Project BRITISH MUSEUM

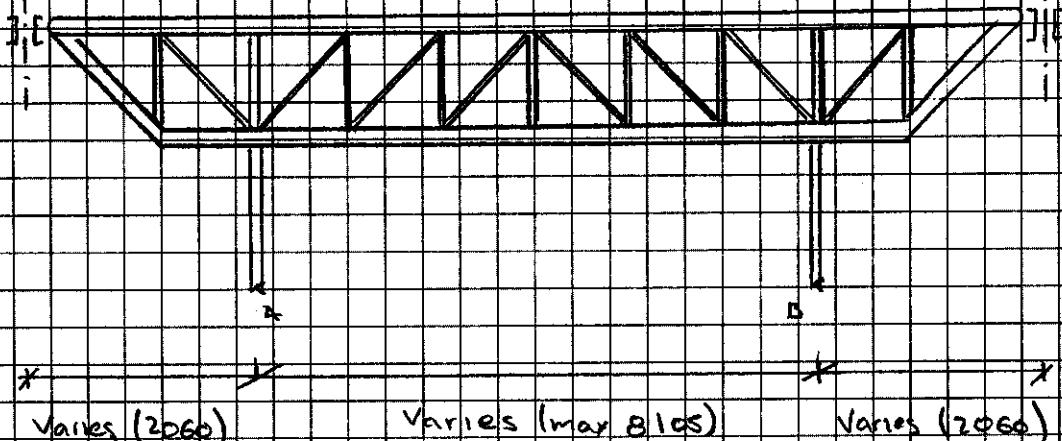
No. 1578/4/18

Date 02/07 By LF

Checked _____

Notes

TRUSS TYPE A'



Beam self weight.

Section	Mass (kg/m)	Length (m)	Mass (kg)
80x3.6 SHS (chords)	8.53	20.4	174
80x3.6 SHS (uprights)	8.53	1.0	8.5
60x40x3.2 RHS (uprights)	4.35	5.5	23.9
60x40x3.2 RHS (diag.)	4.35	10.1	43.9
			250.3

$$w = \frac{250.3}{9.42} = 26.6 \approx 27 \text{ kg/m}$$

atelier one

Project BRITISH MUSEUM

No. 1578/4/19

Date 02/07 By LF

Checked _____

Notes _____

Load

Dead Load (inc. sw) 0.67 kN/m

Live Load ————— 5.2 kN/m

SLS ————— $w = 5.87 \text{ kN/m}$

ULS ————— $w = 9.26 \text{ kN/m}$

Bending Moment

Max. sagging moment $M_{max} = 75.1 \text{ kNm}$

Max hogging moment $M_{max} = 20.2 \text{ kNm}$

Shear Force

$V_{max} = 38.5 \text{ kN}$

$R_A = R_B = 46.2 \text{ kN}$

Material Properties

Hot rolled steel — grade S275

$E = 205 \text{ kN/mm}^2$

$\sigma_y = 275 \text{ N/mm}^2$

Results
Summary

atelier one

Project BRITISH MUSEUM

No. 1578/4/20

Date 02/07 By LF

Checked _____

Notes

Moment Resistance

Sagging

$$T = C = \frac{M_{max}}{d} = \frac{75.1 \times 10^3}{500}$$

$$T = C = 150.2 \text{ kN.}$$

Compression Top chord — fully restrained

Tension bottom chord.

Compression capacity of top chord is
equal to tension capacity

$$P_f = P_c = 299 \text{ kN} \geq 150.2 \text{ kN}$$

V.O.K.

Hogging

$$T = C = \frac{M_{max}}{d} = \frac{20.2 \times 10^3}{500}$$

$$T = C = 40.4 \text{ kN.}$$

Compression bottom chord — L.EF 5570×0.7

$$\text{L.E} = 3900 \text{ mm}$$

$$P_c = 114 \text{ kN} \geq 40.4 \text{ kN}$$

V.O.K.

Tension Top Chord $P_f = 299 \text{ kN} \geq 40.4 \text{ kN}$

V.O.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/21

Date 02/07 By LF

Checked _____

Notes

Shear Resistance

$$\theta = 53.4^\circ$$

$$62.9 \text{ kN}$$

37.5 kN - Reaction - touching between supports.

Max. Compression 60x40x3.0 RHS diagonal

$$C = 62.9 \text{ kN}$$

$$L_E = 840 \text{ mm}$$

$$P_c = 130 \text{ kN} \geq 62.9 \text{ kN}$$

V.O.K.

80x80x3.6 SHS

$$P_v = 0.6 p_y A_v \quad A_v = 0.5 A = 0.5 \times 1090$$

$$P_v = 0.6 \times 275 \times 545 / 10^3$$

$$P_v = 89.9 \text{ kN} \geq 46.2 \text{ kN} = P_A = P_R$$

V.O.K.

atelier one

Project BRITISH MUSEUM

No. 15784/22

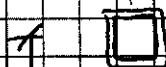
Date. 02/07 By LF

Checked _____

Notes

Deflection check

$$E = 205000 \text{ N/mm}^2$$



$$I_o = \frac{b h^3}{12} = \frac{105 \times 10^4 \times 250^3}{12}$$

$$I_o = 13835 \times 10^4 \text{ mm}^4$$

$$500 \text{ kN} \rightarrow I_{xx} = I_o / (1 + F)$$

$$\text{where } F = k I_o / A s L^2$$



$$k = 9.6 ; L = 8.1 \text{ m}$$

$$\frac{I}{A s} = \sec \varphi \left(\frac{1}{A D \sin^2 \varphi} + \frac{\sin \varphi}{A B} \right)$$

$$A D = 554 \text{ mm}^2 = A B \Rightarrow \text{Area of } 60 \times 40 \times 3.0 \text{ RHS}$$

$$\varphi = 36.6^\circ$$

$$\frac{I}{A s} = \sec 36.6 \left(\frac{1}{554 \times \sin^2 36.6} + \frac{\sin 36.6}{554} \right)$$

$$\frac{I}{A s} = 7.66 \times 10^{-3} / \text{mm}^2$$

$$A s = 130.5 \text{ mm}^2$$

$$F = 9.6 \times 13835 \times 10^4 / 130.5 \times 8105^2$$

$$F = 0.155$$

$$I_{xx} = 13835 \times 10^4 / (1.155)$$

$$I_{xx} = 11978 \times 10^4 \text{ mm}^4$$

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Project

BRITISH MUSEUM

No 1578/4/23

Date 02/07 By LF

Checked

Notes

$$\Delta m = \frac{5 \times 5.87 \times 8.14}{384 \times 24556} \times 10^3$$

$$\Delta m = 13.4 \text{ mm}$$

$$\Delta adm = \frac{8105}{360} = 22.5 \text{ mm} \geq 13.4 \text{ mm}$$

O.K.

atelier one

Project BRITISH MUSEUM

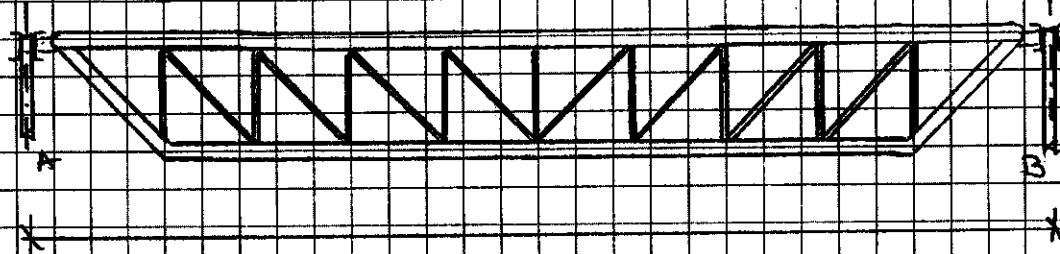
No. 1578/4/24

Date 02/07 By LF

Checked _____

Notes

TRUSS TYPE B & C



Varies (max 9.69)

Loads

Dead Load (inc sw) 0.67 kN/m

Live Load 5.2 kN/m

SUS $\text{w} = 5.87 \text{ kN/m}$

ULS $\text{w} = 9.26 \text{ kN/m}$

Bending Moment

$M_{\text{max}} = 111.6 \text{ kNm}$ — sagging moment only.

Shear Force

$$V_{\text{max}} = R_A = R_B = \frac{9.26 \times 9.69}{2}$$

$$R_A = R_B = 44.9 \text{ kN}$$

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Project BRITISH MUSEUM

No. 1578/4/25

Date 02/07 By LF

Checked _____

Notes _____

Moment Resistance

$$T = C = \frac{N_{max}}{d} = \frac{111.6 \times 10^3}{500}$$

$$T = C = 223.2 \text{ kN}$$

Compression Top chord - fully restrained

Tension bottom chord

Compression capacity of top chord is
equal to tension capacity

$$P_T = P_C = 299 \text{ kN} \geq 223.2 \text{ kN}$$

Shear Resistance

Max Tension 60x40x30 RHS

$$F = 44.9 / \cos 53.1^\circ$$

$$F = 75.3 \text{ kN} \geq 152 \text{ kN} = P_T$$

Max Compression

$$F = 37.4 \text{ kN} \geq 130 \text{ kN} = P_C$$

✓ O.K.

P_C & P_T
From Blue Book

✓ O.K.

✓ O.K.

atelier one

Project BRITISH MUSEUM

No. 15784/26

Date 02/07 By LF

Checked _____

Notes

Bearing on web check

PFCH 175x65x2.4

$$P_A = P_B = 44.9 \text{ kN}$$

Use 4N M12

$$P_b = t \times D \times p_{bs} \quad | \quad \text{where } t = 2.4 \text{ mm}$$

$$P_b = 4 \times 2.4 \times 12 \times 416 / 10 =$$

$$D = 12 \text{ mm}$$

$$P_{bs} = 0.65 (280 + 360)$$

$$P_b = 47.9 \text{ kN}$$

$$p_{bs} = 416 \text{ N/mm}^2$$

$$P_b = 47.9 \text{ kN} \geq 44.9 \text{ kN} = P_A = P_B$$

Deflection Check

$$I_{xx} = 13835 \times 10^4 \text{ mm}^4$$

$$\delta_m = \frac{5 \times 5.87 \times 9.62^4}{384 \times 24556}$$

$$\delta_m = 26.6 \text{ mm}$$

$$\delta_{adlm} = \frac{9.690}{360} = 26.9 \text{ mm} \geq 26.6 \text{ mm}$$

See sketch
following
page.

✓ o.k.

See truss
Type A

✓ o.k.

atelier one

Project BRITISH MUSEUM

No. 1578/4/27

Date 02/07 By LF

Checked _____

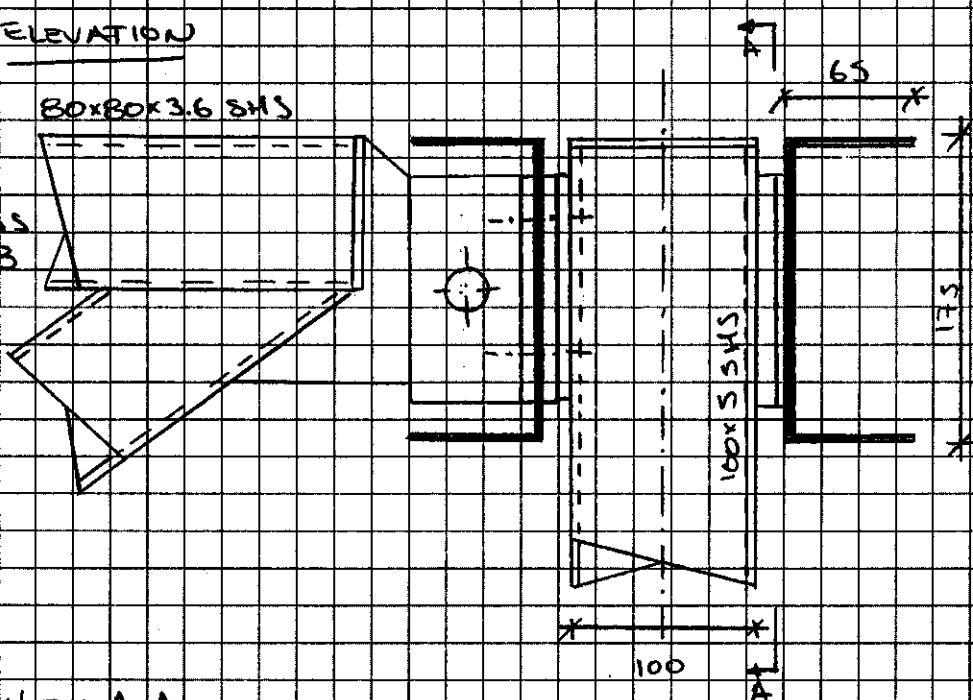
Notes

Connection Detail Truss - Channel - Column

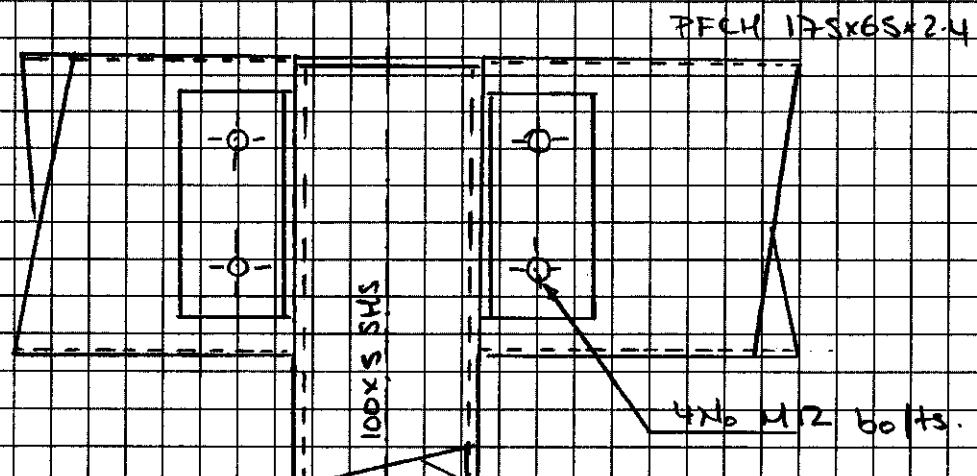
ELEVATION

80x80x3.6 SHS

TRUSS
TYPE B



VIEW A-A



atelier one

Project BRITISH MUSEUM

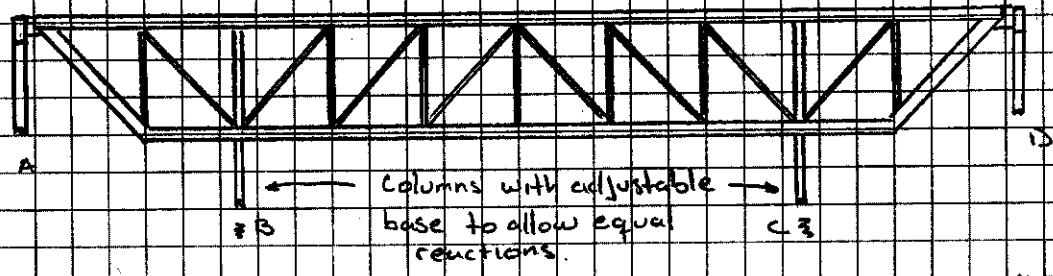
No. 1578/4/28

Date 02/07 By LF

Checked _____

Notes

TRUSS TYPE D



Varies
(Max 2640)

Varies (Max 6700)

Varies
(Max 2640)

Loads

Dead Load (inc snow) — 0.01 kN/m

Live Load — 7.88 kN/m

SLS — $w = 8.8 \text{ kN/m}$

ULS — $w = 13.9 \text{ kN/m}$

CENTRAL RING

OPTION 1 2640 5420 2640

OPTION 2 2350 6700 2350

OPTION 3 2640 5420 2640

1762

1969

1762

atelier one

Project BRITISH MUSEUM

No. 15784/29

Date 02/07 By LF

Checked _____

Notes

Adjustable Column Base

Use spring of stiffness K (kN/m) in the mid-supports to distribute reaction loads to the outer supports.

OPTION 1 — $K = 9000 \text{ kN/m}$

OPTION 2 — $K = 1300 \text{ kN/m}$

OPTION 3 — $K = 6000 \text{ kN/m}$

Bending Moment

$M_{\max} = 128.3 \text{ kNm}$ — OPTION 2
sagging moment

$M_{\max} = -3.7 \text{ kNm}$ — OPTION 3
hogging moment

Shear Force

$V_{\max} = 39.8 \text{ kN}$ — OPTION 2

$R_{\max} = 52.3 \text{ kN}$ — OPTION 1

atelier one

Project BRITISH MUSEUM

No. 1578/4/30

Date 02/07/05 By LF

Checked _____

Notes

Moment Resistance

$$T = C = \frac{M_{max}}{c_l} = \frac{128.3 \times 10^3}{500}$$

$$T = C = 257 \text{ kN}$$

Sagging moment | - Top chord in compression
 ↳ Fully restrained

- Bottom chord in tension

Compression Capacity is equal to Tension

Capacity

$$P_T = P_C = 299 \text{ kN} \geq 257 \text{ kN}$$

✓.O.K.

Shear Resistance

- Max compression $60 \times 40 \times 3.0$ RHS

$$F = 39.8 / \cos 53.4$$

$$F = 66.8 \text{ kN}$$

$$L_e = 840 \text{ mm}$$

$$P_C = 130 \text{ kN} \geq 66.8 \text{ kN}$$

- $80 \times 80 \times 3.6$ SHS

$$P_V = 89.9 \text{ kN} \geq 66.8 \text{ kN}$$

✓.O.K.

See Truss
Type A
✓.O.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/31

Date 02/07 By LF

Checked _____

Notes

Deflection check

Critical deflection - OPTION 2.

* Consider mid-span and overall deflection

* Relative deflection = mid-span

$$\delta_m = 13.9 \text{ mm}$$

$$L_e = 6700 \text{ mm}$$

$$\delta_{adm} = \frac{6700}{360} = 18.6 \text{ mm} \geq 13.9 \text{ mm}$$

V.O.K.

* Overall deflection

$$\delta_{max} = 32.4 \text{ mm}$$

$$L_e = 11400 \text{ mm}$$

$$\delta_{adm} = \frac{11400}{360} = 31.7 \text{ mm exceeds.}$$

V.O.U.

atelier one

Project BRITISH MUSEUM

No. 1578/4/32

Date 02/07 By L.F.

Checked _____

Notes

BRACING CAPACITY CHECK

48.3 x 5 CHS — S355

Effective Length = 3.7 m.

$P_c = 19.5 \text{ kN}$ — Compression capacity

$M_c + M_b = 2.85 \text{ kNm}$

$P_y z_y = 2.37 \text{ kNm}$

$P_y = 241 \text{ kN}$ — Tension capacity

From
Blue
Book

Results Summary

$F_{max\ T} = 14.5 \text{ kN} \leq P_y = 241 \text{ kN}$

$F_{max\ C} = 11.3 \text{ kN} \leq P_c = 19.5 \text{ kN}$

$V_{max} = 0.11 \text{ kN}$

✓.K.

✓.K.

Shear capacity

$$P_v = 0.6 \times P_y \times A_v \quad | \quad A_s = 0.6 A = 0.6 \times 680$$

$$R_v = 0.6 \times 355 \times 408 / 10^3 \quad | \quad A_v = 408 \text{ mm}^2$$

$$P_v = 86.9 \text{ kN}$$

$$V_{max} \leq 0.6 P_v$$

✓.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/33

Date 02/07 By LF

Checked _____

Notes

COLUMN CAPACITY CHECK

CHS 48.3 x 5 - S335

$P_{max} = 39.8 \text{ kN}$ Max Compression Bar 6203

Nominal Bending Moment

$$M = \frac{P_{max}}{2} \times e$$

assume $e = 70 \text{ mm}$

$$M = \frac{39.8}{2} \times 0.07$$

$$M = 1.39 \text{ kNm}$$

} where only
one side of
beam supported
is loaded

Compression Capacity

$$L_e = 1350$$

$$P_c = 87.7 \text{ kN} \geq 41.7 \text{ kN}$$

Bending Capacity & Buckling

$$M_c = M_b = 2.85 \text{ kNm}$$

From
Blue Book
J.O.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/34

Date 02/07 By LF

Checked _____

Notes

Overall Buckling Check

$$\frac{P_{\text{max}}}{P_c} + \frac{M}{M_b} \leq 1.0 \quad \text{Nominal Bending Moment}$$

$$\frac{30.8}{87.7} + 1.39 \\ 0.354 + 1.39 \\ 2.85$$

$$0.454 + 0.488 = 0.94 \leq 1.0$$

V.O.K.

Coexistant Forces from Analysis Model

Bar 7 — Load Combination 103: 1.4DL + 1.6CL
+ 1.0 Sway L

$$F_x = 27.4 \text{ kN}$$

$$M_x = 1.09 \text{ kNm} \quad \text{— major axis}$$

$$M_y = 0.23 \text{ kNm} \quad \text{— minor axis}$$

$$\frac{F_x}{P_c} + \frac{M_x}{M_b} + \frac{M_y}{M_b} \leq 1.0 \quad P_y z_y = 2.37 \text{ kNm}$$

$$\frac{25.7}{87.7} + \frac{1.09}{2.85} + \frac{0.23}{2.37}$$

$$0.293 + 0.382 + 0.382 = 0.74 \leq 1.0$$

V.O.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/35

Date 02/07 By LF

Checked _____

Notes

CHS 60.3 x 5 — 5353

$P_{max} = 55.3 \text{ kN}$ Max Compression Bar 6

$M_{max} = 1.46 \text{ kNm}$ Bar 3

Nominal Bending Moment

$$N = \frac{55.3 \times 0.07}{2}$$

$$M = 1.94 \text{ kNm} \geq M_{max} = 1.46 \text{ kNm}$$

More critical.

Compression Capacity

$$L_e = 1573 \text{ mm}$$

$$P_c = 204 \text{ kN} \geq 55.3 \text{ kN}$$

S.O.U.

Bending & Buckling Capacity

$$M_c = M_b = 4.73 \text{ kNm} \geq 1.94 \text{ kNm}$$

S.O.U.

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Project BRITISH MUSEUM

No. 1578/4/36

Date 02/07 By LF

Checked _____

Notes

Overall Buckling Check

$$\frac{P}{P_c} + \frac{M_y}{M_b} \leq 1.0$$

Bar 6 - Coexistent Forces

$$\frac{55.3}{204} + \frac{1.04}{4.73}$$

$$0.271 + 0.410 = 0.681 \leq 1.0$$

Jo.K.

Non coexistent Forces

$$P = 55.3 \text{ kN}$$

$$M_y = 1.46 \text{ kNm}$$

$$M_z = 1.07 \text{ kNm}$$

Results from
Robot Model

$$\frac{P}{P_c} + \frac{M_y}{M_b} + \frac{M_z}{P_y z_y} \leq 1.0$$

$$\frac{55.3}{204} + \frac{1.46}{4.73} + \frac{1.07}{3.94}$$

$$0.271 + 0.309 + 0.272 = 0.85 \leq 1.0$$

Jo.K.

atelier one

Project BRITISH MUSEUM

No. 15784/37

Date 02/07 By LF

Checked _____

Notes

$$\text{SHTS } 100 \times 5 = \underline{\underline{\text{S355}}}$$

$$P_{\max} = 49.3 \text{ kN} \quad \text{Bar } 625A$$

$$M_{\text{major}} = 7.84 \text{ kNm} \quad \text{Bar } 619A$$

$$M_{\text{minor}} = 4.18 \text{ kNm} \quad \text{Bar } 625G$$

Nominal Bending Moment

$$M = \frac{55.0}{2} \times 0.07$$

$$M = 1.93 \text{ kNm} \leq M_{\text{major}} \text{ & } M_{\text{minor}}$$

Compression Capacity

$$L_c = 1573 \text{ mm}$$

$$P_c = 618 \text{ kN} \geq 49.3 \text{ kN}$$

Bending & Buckling Capacity

$$M_c = M_b = 23.6 \text{ kNm} \geq 7.84 \text{ kNm}$$

J.O.K.

V.O.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/38

Date 02/07 By LF

Checked _____

Notes

Overall Buckling Check

Non coexistent Forces

$$\frac{P}{P_c} + \frac{M_y}{M_b} \rightarrow \frac{M_2}{M_y z_y} \leq 1.0$$

$$\frac{49.3}{618} + \frac{7.34}{23.6} = \frac{4.18}{19.8}$$

$$0.080 + 0.332 + 0.211 = 0.623 \leq 1.0$$

V.O.K.

atelier one

Project BRITISH MUSEUM

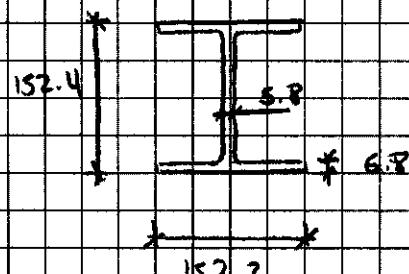
No. 1578/4/39

Date 02/07 By LF

Checked _____

Notes

SPREADER BEAM CAPACITY CHECK



UC 152x152x23 - 275

$$B = 152.2 \text{ mm}$$

$$D = 152.4 \text{ mm}$$

$$T = 6.8 \text{ mm}$$

$$t = 5.8 \text{ mm}$$

$$\frac{b}{T} = 1.2$$

$$A = 2920 \text{ mm}^2$$

$$c_l = 21.3$$

$$I_{xx} = 1250 \times 10^4 \text{ mm}^4 \quad r_{xx} = 65.4 \text{ mm}$$

$$t$$

$$I_{yy} = 400 \times 10^4 \text{ mm}^4 \quad r_{yy} = 37.0 \text{ mm}$$

$$S_{xx} = 182 \times 10^3 \text{ mm}^3 \quad Z_{xx} = 164 \times 10^3 \text{ mm}^3$$

$$S_{yy} = 80.2 \times 10^3 \text{ mm}^3 \quad Z_{yy} = 52.6 \times 10^3 \text{ mm}^3$$

Semi-compact section

Shear Capacity

$$P_v = 0.6 \times P_y \cdot A_v = 0.6 \times 275 \times 5.8 \times 152.4 / 10^3$$

$$P_v = 145.8 \text{ kN}$$

$$F_{vmax} = 58.7 \text{ kN} \leq 0.6 \cdot P_v$$

atelier one

Project BRITISH MUSEUM

No. 1578/4/40

Date 02/07 By LF

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Notes

Bending Capacity

$$M_{cx} = p_y Z_{xx} = 275 \times 164 \times 10^3 / 10^6$$

$$M_{cx} = 45.1 \text{ kNm} \geq 26.7 \text{ kNm}$$

$$M_{cy} = p_y Z_{yy} = 275 \times 52.6 \times 10^3 / 10^6$$

$$M_{cy} = 14.5 \text{ kNm} \geq 0.9 \text{ kNm}$$

Jo.K.

Jo.K.

Buckling Capacity

* Simply supported beams : $L_e = 1.0L + 2D$

* Cantilever $L_e = 3.0L$

Take three effective lengths.

$$\textcircled{1} \quad L_e = 2004 + (2 \times 152.4) = 2308 \text{ mm}$$

Use 2300 mm Inner 4 concentric
spreader beams.

$$\textcircled{2} \quad L_e = 4000 + (2 \times 152.4) = 4303 \text{ mm}$$

Use 4300 mm Outer 3 concentric
spreader beams
except for cantilever
beams

$$\textcircled{3} \quad L_e = 3.0 \times 774 = 2322 \text{ mm}$$

Use 2300 mm For cantilever beams
Outer spreader ring

Table 9
BS 5950

atelier one

Project BRITISH MUSEUM

No. 1578/4/41

Date 02/07 LF By

Checked _____

Notes

Case ① & ⑤

$$\lambda = \frac{2300}{37.0}$$

$$\lambda = 62$$

$$\lambda_{LT} = n \nu \lambda \quad \text{where:}$$

$$\left\{ \begin{array}{l} \lambda/x = 2.77 \\ \nu = 0.9 \\ n = 1.0 \end{array} \right.$$

$$\lambda_{LT} = 0.9 \times 1.0 \times 0.92 \times 62$$

$$\nu = 0.92$$

$$\lambda_{LT} = 51$$

$$\rho_b = 235 \text{ N/mm}^2$$

$$M_b = \rho_b \times S_{xx} = 235 \times 182 \times 10^{-3}$$

$$M_b = 42.83 \text{ kNm} \geq 26.7 \text{ kNm}$$

V.O.K.

Case ②

$$\lambda = \frac{4300}{37.0}$$

$$\lambda = 116$$

$$\lambda_{LT} = n \nu \lambda$$

$$\lambda_{LT} = 0.9 \times 1.0 \times 0.81 \times 116$$

$$\lambda_{LT} = 85$$

$$\rho_b = 154 \text{ N/mm}^2 \quad \therefore M_b = 28 \text{ kNm} \geq 23.4 \text{ kNm}$$

V.O.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/42

Date 02/07 By LF

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Notes

Overall Buckling Check

$$\rho_y z_y = 14.3 \text{ kNm}$$

$L_e \leq 2300 \text{ mm}$ Non co-existent forces

$$\frac{P}{P_c} + \frac{M_y}{M_b} + \frac{M_z}{\rho_y z_y} \leq 1.0$$

$$\frac{7.9}{543} + \frac{26.7}{42.8} + \frac{0.9}{14.3}$$

$$0.015 + 0.624 + 0.062 = 0.70 \leq 1.0$$

V.O.K.

$4300 \text{ mm} \geq L_e \geq 2300 \text{ mm}$ Non co-existent forces

$$\frac{P}{P_c} + \frac{M_y}{M_b} + \frac{M_z}{\rho_y z_y} \leq 1.0$$

$$\frac{3.6}{23.6} + \frac{23.4}{28.0} + \frac{1.5}{14.3}$$

$$0.15 + 0.835 + 0.103 = 0.954 \leq 1.0$$

V.O.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/43

Date 02/07 By LF

Checked _____

Notes

Deflection Check

$$\delta_{max} = 9.2 \text{ mm}$$

See Robot Results.

$$\delta_{udm} = \frac{le}{360} \cdot \frac{4000}{360}$$

$$\delta_{udm} = 11.1 \text{ mm} \geq 9.2 \text{ mm}$$

J.O.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/44

Date 02/07 By LF

Checked _____

Notes

STAIRCASE MEMBERS CAPACITY CHECK

48.3x5 CHS Columns.

Overall Buckling check - S353

$$P_c = 87.7 \text{ kNm}$$

$$M_c = M_b = 2.85 \text{ kNm}$$

$$\phi_y z_y = 2.37 \text{ kNm}$$

$$\frac{F_x}{P_c} + \frac{M_y}{M_b} + \frac{M_z}{\phi_y z_y} \leq 1.0$$

} From
Blue
Book.

Non coexistent Forces - From Results
Summary

$$\frac{10.1}{87.7} + \frac{0.24}{2.85} + \frac{0.11}{2.37}$$

$$0.115 + 0.084 + 0.046 = 0.24 \leq 1.0$$

✓ O.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/45

Date 02/07 By LF

Checked _____

Notes

100x5 RHS STAINLESS - S355

Overall Buckling Check

$$P_c = 301 \text{ kN} \quad (L_e = 4000 \text{ mm})$$

$$M_e = M_b = 23.6 \text{ kNm}$$

$$P_y z_y = 19.8 \text{ kNm}$$

From
Blue
Book

$$\frac{F_x}{P_c} + \frac{M_y}{M_b} + \frac{M_z}{P_y z_y} \leq 1.0$$

Non coexistent Forces - From Results Summary

$$\frac{10.2}{301} + \frac{1.03}{23.6} + \frac{0.55}{19.8}$$

$$0.034 + 0.044 + 0.028 = 0.105 \leq 1.0$$

V.O.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/46

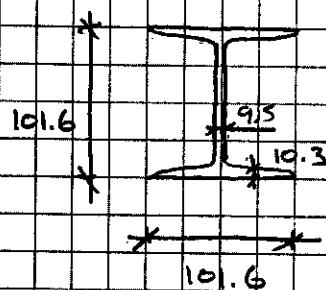
Date 22/07 By LF

Checked _____

Notes

RST 102x102x23

$$\rho_y = 275 \text{ N/mm}^2$$



$$D = 101.6 \text{ mm} = B$$

$$t = 9.5 \text{ mm} ; T = 10.3$$

$$\frac{b}{T} = 4.93 ; \frac{d}{t} = 5.81$$

$$A = 2930 \text{ mm}^2$$

$$I_{xx} = 486 \times 10^4 \text{ mm}^4$$

$$I_{yy} = 154 \times 10^4 \text{ mm}^4$$

$$r_{xx} = 40.7 \text{ mm}$$

$$r_{yy} = 22.9 \text{ mm}$$

$$Z_{xx} = 95.6 \times 10^3 \text{ mm}^3$$

$$Z_{yy} = 30.3 \times 10^3 \text{ mm}^3$$

$$S_{xx} = 113 \times 10^3 \text{ mm}^3$$

$$S_{yy} = 50.6 \times 10^3 \text{ mm}^3$$

Shear Capacity

$$P_v = 0.6 \rho_y A_v$$

$$A_v = t D = 965 \text{ mm}^2$$

$$P_v = 0.6 \times 275 \times 965 / 10^3$$

$$P_v = 159 \text{ kN} > 0.6 F_{v\max}$$

$$F_{v\max} = 18.3 \text{ kN}$$

Plastic.

✓ O.K.

atelier one

Project BRITISH MUSEUM

No. 1578/4/47

Date 02/07 By LF

Checked _____

Notes

Bending Capacity

$$M_{cx} = S_{xx} \rho_y = 113 \times 10^3 \times 275 / 10^6$$

$$\underline{M_{cx} = 31.1 \text{ kNm}} \geq 3.6 \text{ kN}$$

$$M_{cy} = 1.2 \times Z_{yy} \rho_y = 1.2 \times 30.3 \times 10^3 \times 275 / 10^6$$

$$\underline{M_{cy} = 10 \text{ kNm}} \geq 2.0 \text{ kN}$$

✓.O.K.

✓.O.K.

Buckling Capacity

$$L_e = 1.0L + 2D = 3524 + 2 \cdot 101.6 = 1946$$

$$L_e = 3750$$

$$\lambda = \frac{3750}{72.9} = 164$$

$$x = \frac{9}{7} = 9.9$$

$$\lambda_{LT} = n u \sqrt{\lambda} \quad | \quad \frac{\lambda}{x} = 16.5 ; u = 0.51$$

$$\lambda_{LT} = 0.836 \times 0.51 \times 164 \quad | \quad n = 1.0$$

$$\lambda_{LT} = 70 \quad | \quad u = 0.836$$

$$\rho_b = 188 \text{ N/mm}^2$$

$$M_b = S_{xx} \rho_b = 113 \times 10^3 \times 188 / 10^6$$

$$\underline{M_b = 21.2 \text{ kNm}} \geq 3.6 \text{ kNm}$$

✓.O.K.

atelier one

Project BRITISH MUSEUM

No. 157814/48

Date 02/07 By LF

Checked _____

Notes

SPREADER BEAM UNDER STAIR & LIFTS CHECK

Overall Buckling check

Non Coaxistant Forces

$$\frac{M_y}{M_b} + \frac{M_z}{P_y z_y} \leq 1.0 \quad P_y z_y = 3.3 \text{ kNm}$$

$$\frac{3.6}{21.2}, \frac{2.0}{8.3}$$

$$0.17 + 0.24 = 0.41 \leq 1.0$$

S.O.U.

atelier one

Project BRITISH MUSEUM

No 1578/S/1

Date 02/07 By LF

Checked _____

Notes

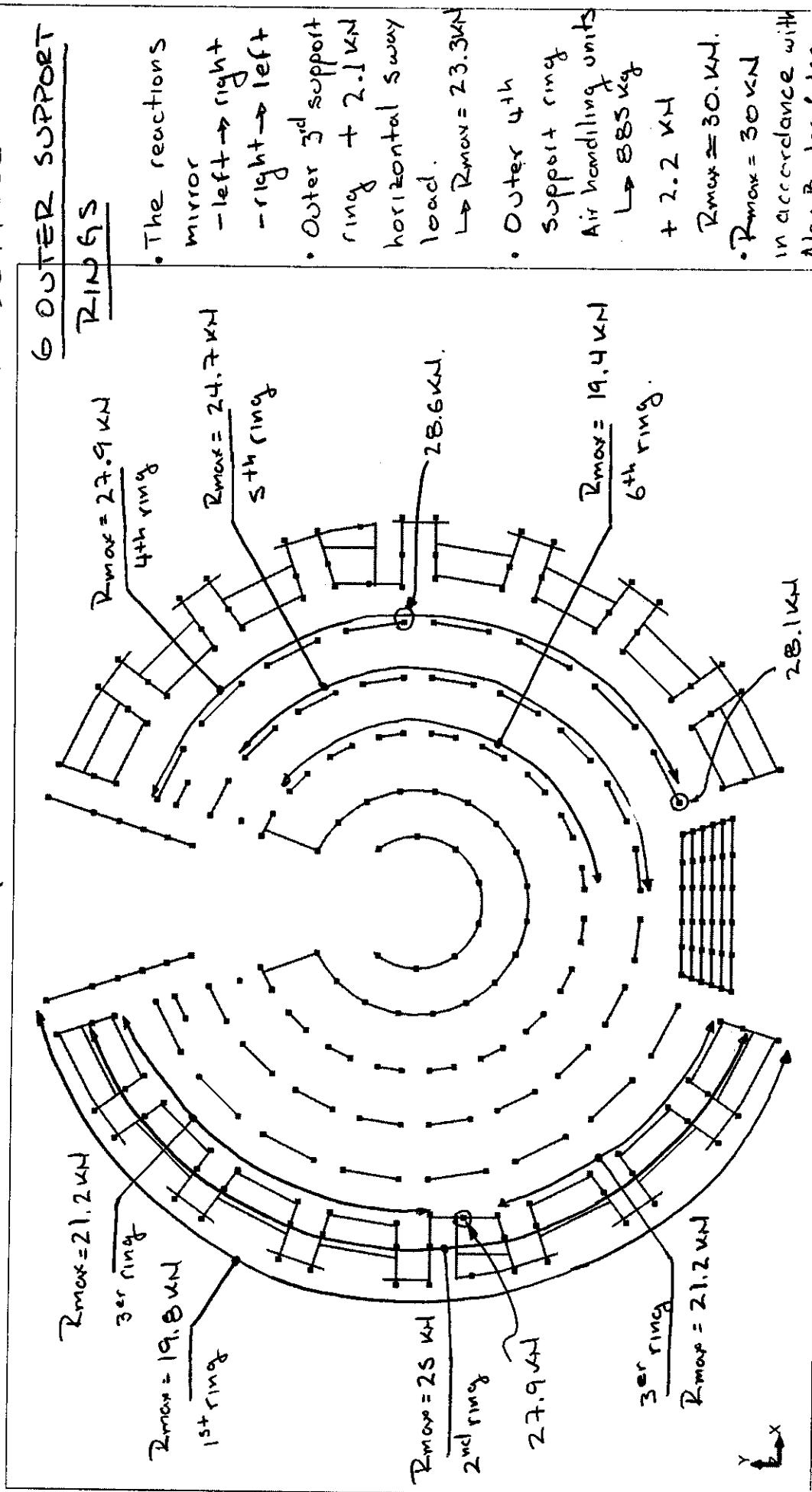
REACTIONS CAPACITY

The Round Reading Room floor
structure has been assessed by
Alan Baxter & Associates

As a result of this investigation
a list of loading constraints to
the existing floor structure have
been established

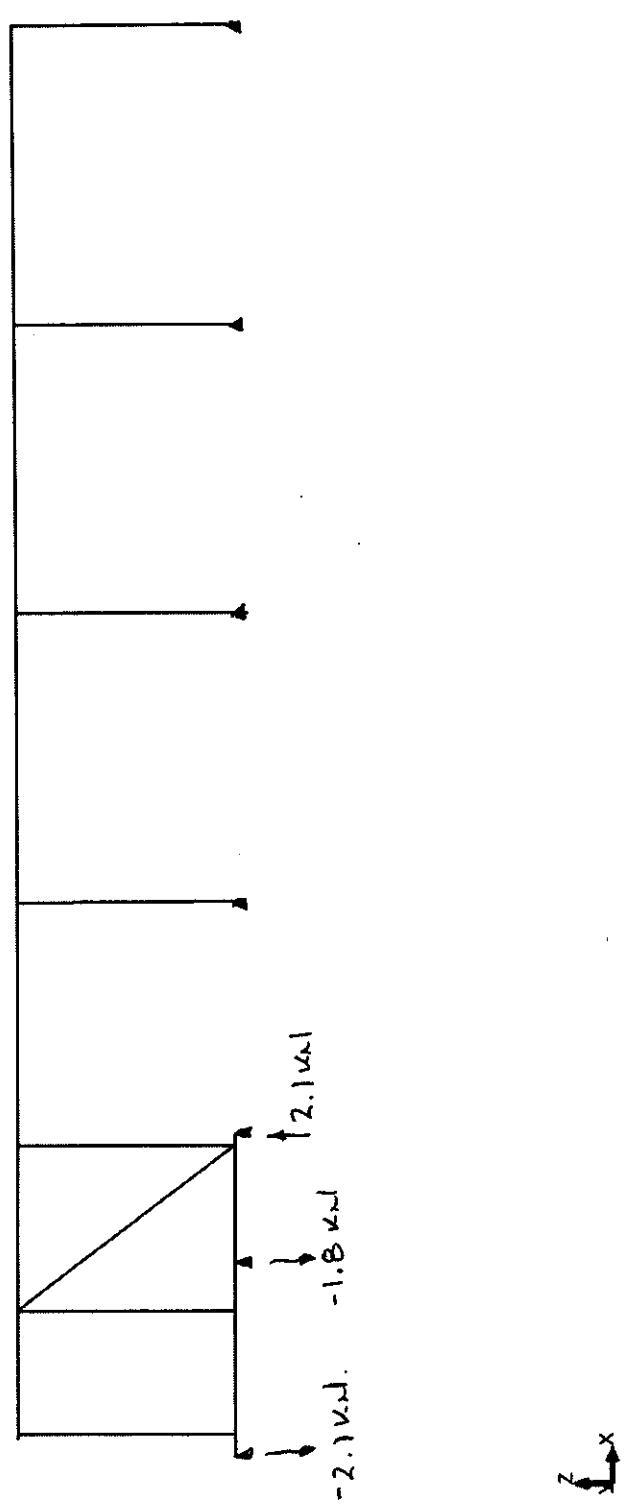
These constraints are explained in
Alan Baxter & Associates Drawings
0699/301/20 Rev A and
0699/301/21

View2 - Cases: 1 () REACTIONS DIAGRAM - LOAD COMBINATIONS 1.0 DL + 1.0 LL



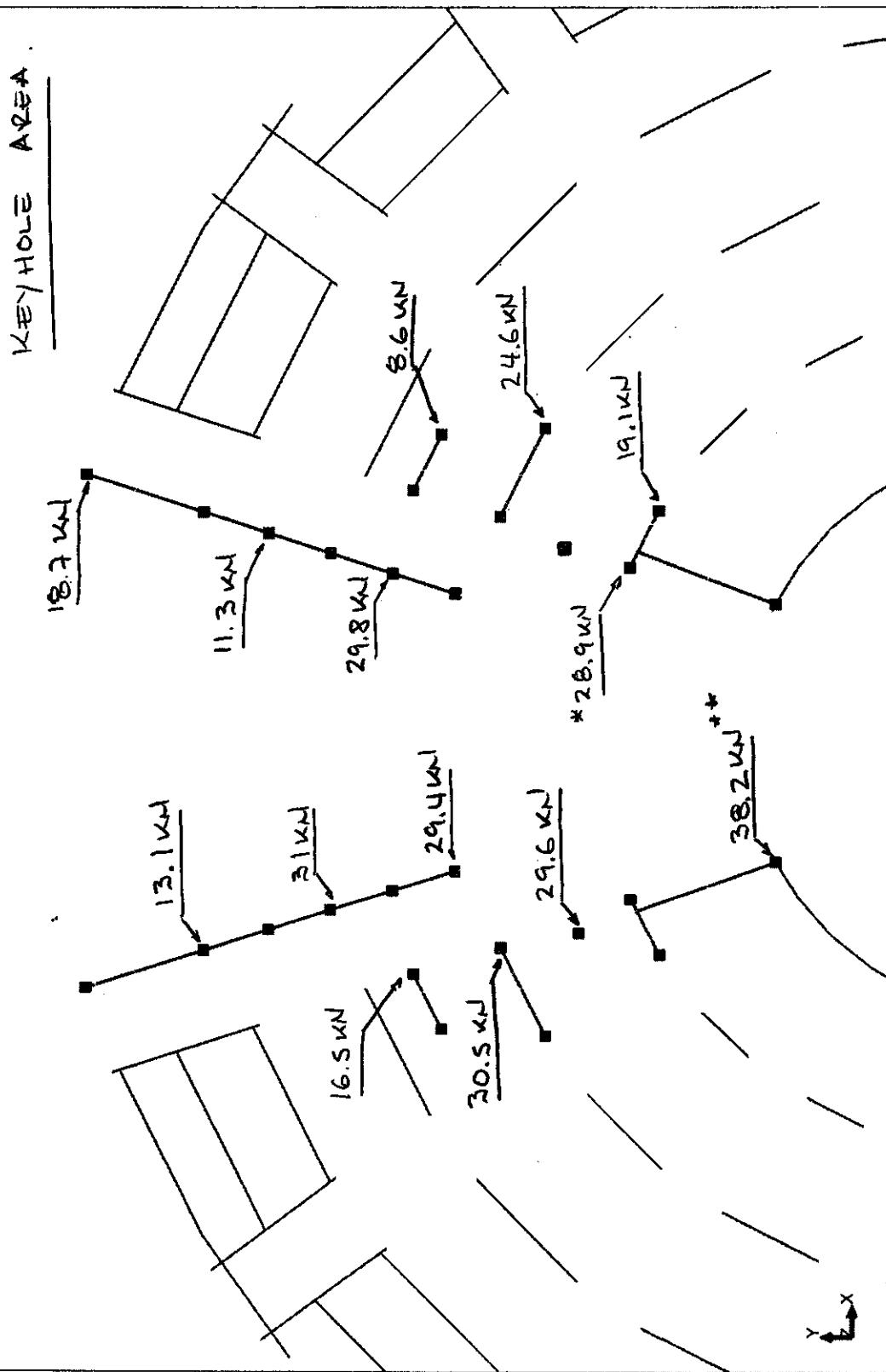
View:2 - Cases: 3 (Sway Load 1) HORIZONTAL SWAY LOAD

REACTIONS TYPICAL RADIAL BEAM WITH BRACING.



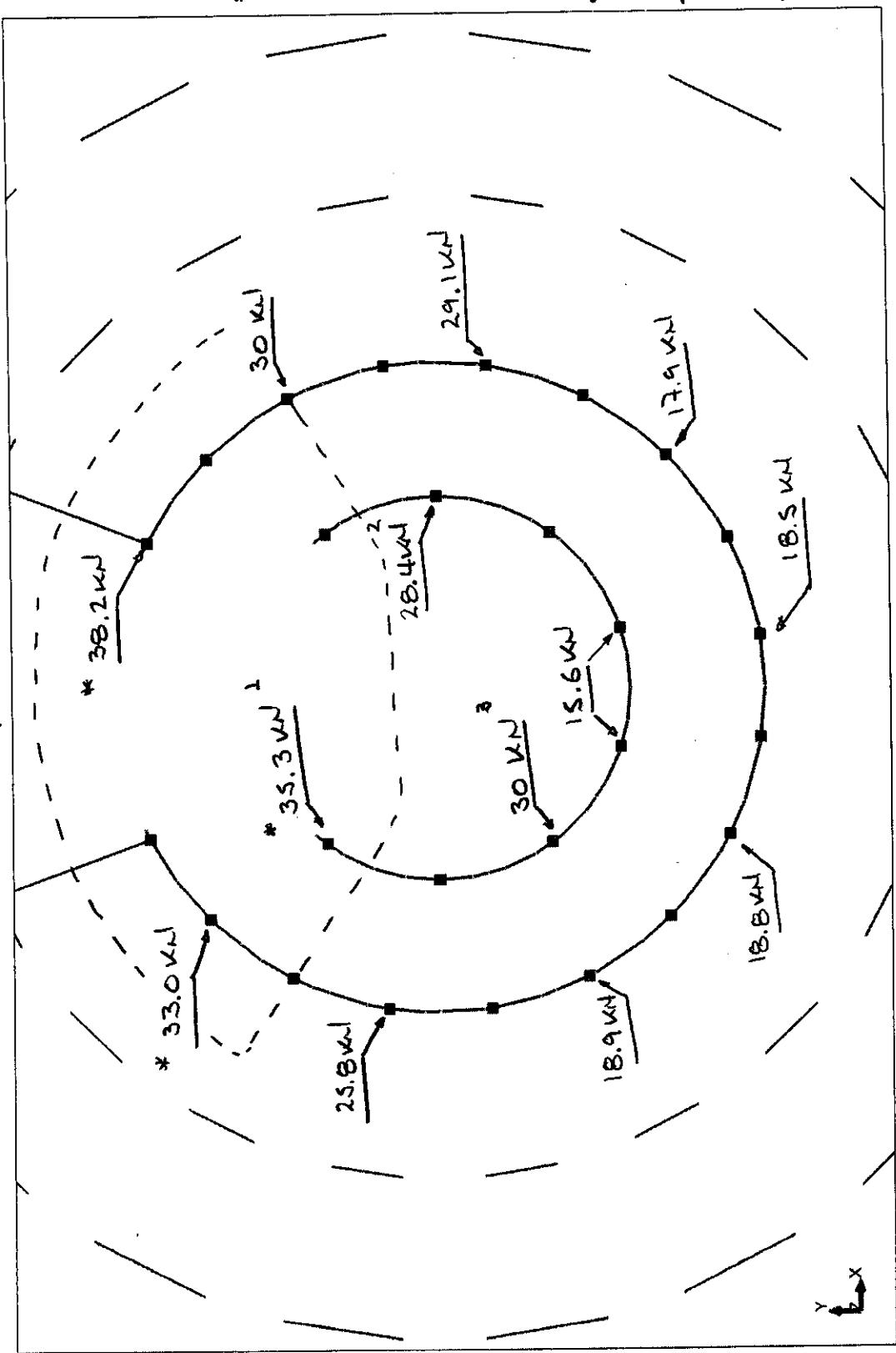
View:2 - Cases: 51 (1.0DL + 1.0LL)

REACTIONS DIAGRAM - LOAD COMBINATIONS 1.0 DL + 1.0 LL



View:1 - Cases: 1 (Dead Load)

REACTIONS DIAGRAM - LOAD COMBINATION 1.0 DL + 1.0 LL

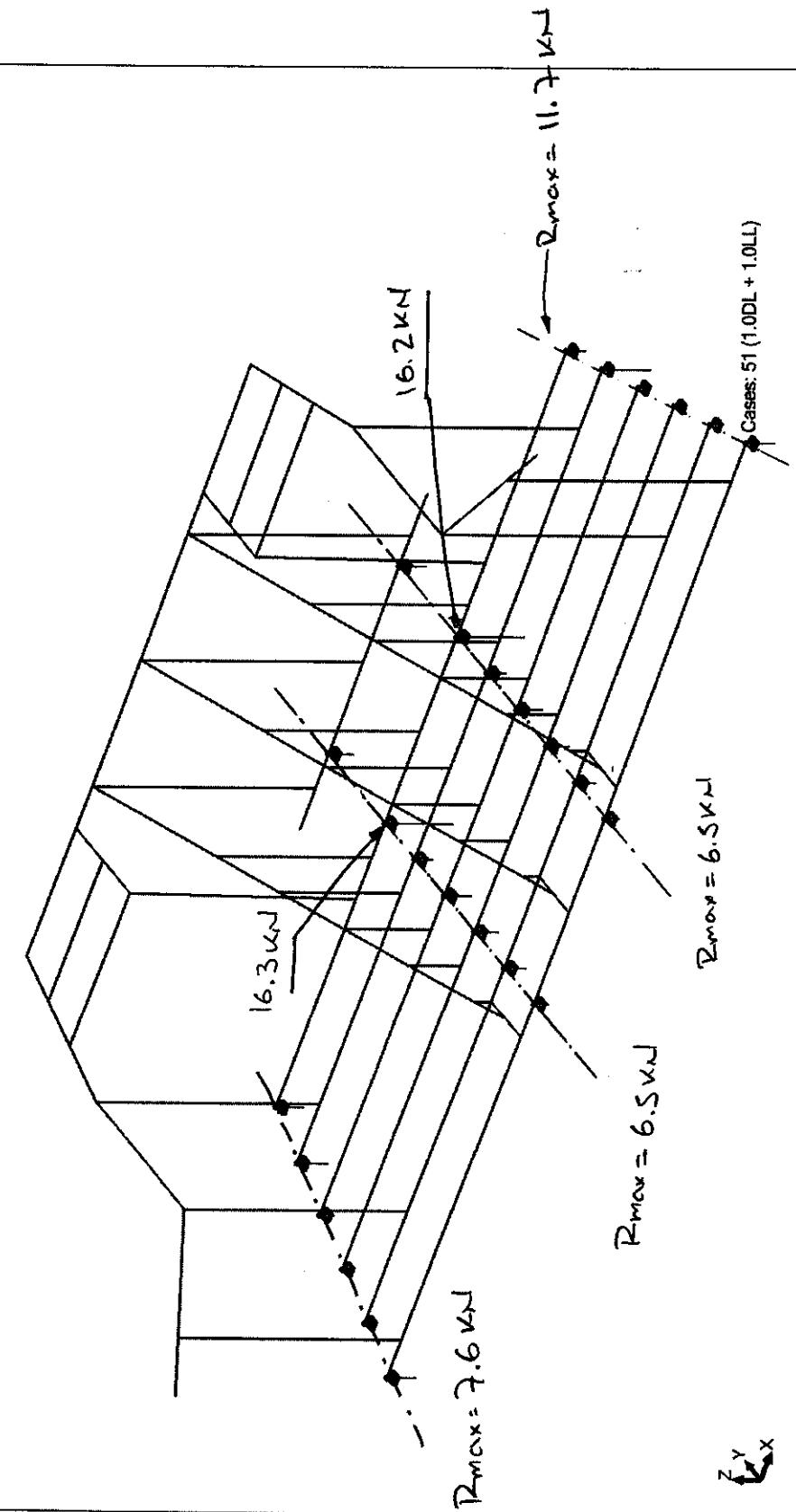


- * Reactions mirror symmetrically
- * Reactions exceeds the 30 kN limit.
↳ Decrease live load allowance to 0
- 300 kg/m² within dotted line.
- * Adopt spring supports to aid load distribution.
 - 1 Spring - $K = 9000 \text{ N/m}$
 - 2 Spring - $K = 13000 \text{ N/m}$
 - 3 Spring - $K = 60000 \text{ N/m}$

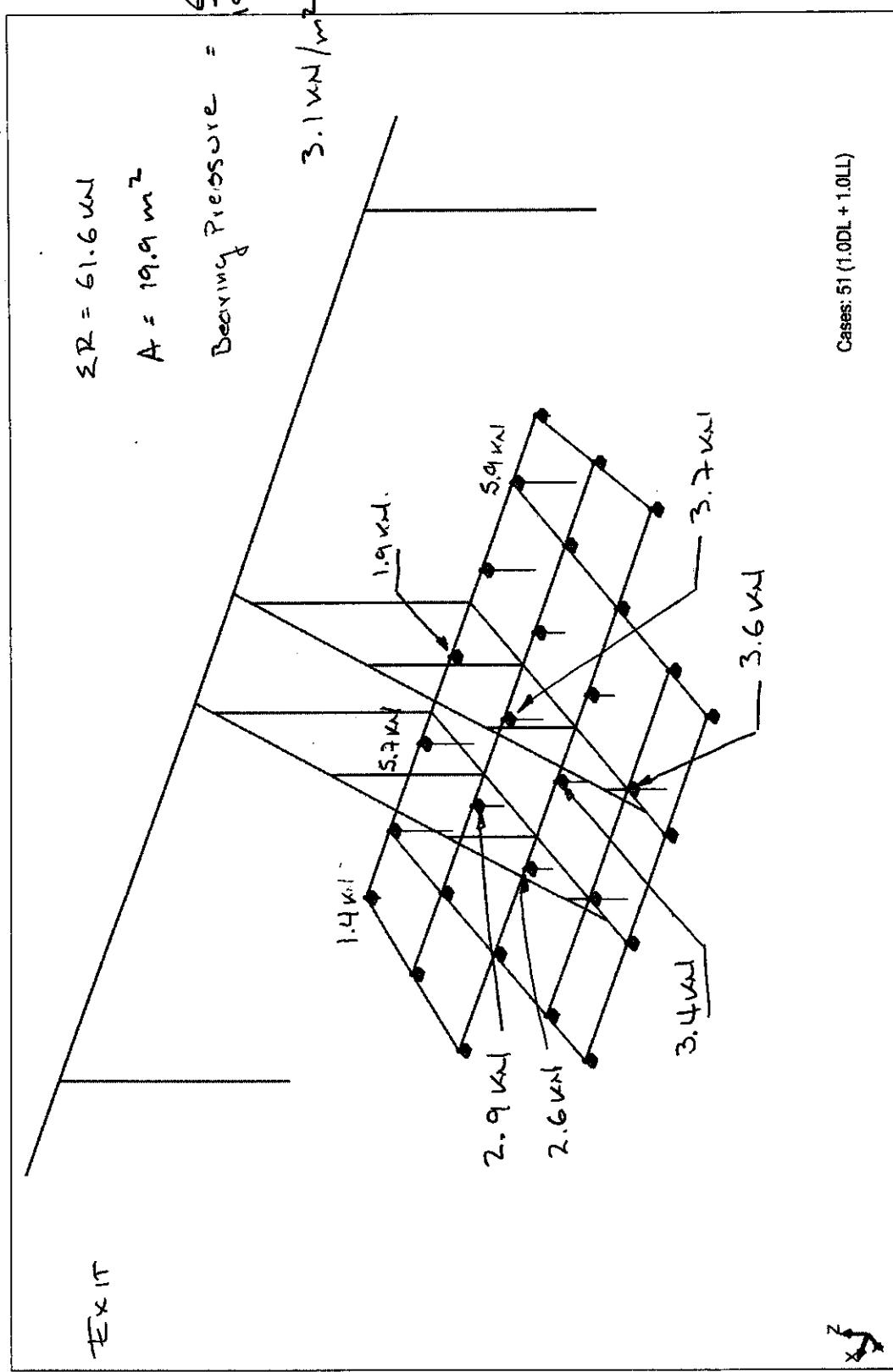
1578/5/S

View:1 - Reaction forces(kN), Cases: 51 (1.0DL + 1.0LL) REACTIONS - LOAD COMBINATION 1.0DL+1.0LL

ENTRANCE



View:2 - Reaction forces(kN), Cases: 51 (1.0DL + 1.0LL) REACTIONS - LOAD COMBINATIONS 1.0 DL + 1.0 LL



atelier one

Project BRITISH MUSEUM

No. 1578/S/8

Date 02/07 By LF

Checked _____

Notes

REACTION'S CHECK

LOADING CONSTRAINTS

- 30 kN point load min spacing 1 metre
on Basement Bearing walls and
on the middle of central piers
- 5 kN/m² Crooked Load Keyhole Area.

Refer to Alan Baxter & Associates Report.

Reactions along platform are not greater than 30 kN with exception of dotted section at the centre of the deck.

Decrease live load allowance from 400 kg/m² to 300 kg/m² in this area

$$\begin{aligned} P_{max} &= 38.4 \text{ kN} & (P_{LL} = 32.5 \text{ kN} \quad (400 \text{ kg/m}^2)) \\ &\hookrightarrow 400 \text{ kg/m}^2 & | P_{SL} = 5.9 \text{ kN} \end{aligned}$$

atelier one

Project BRITISH MUSEUM

No. 1578/5/9

Date 02/07 By LF

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Notes

$$R_w = 32.5 \text{ kN} \quad \text{Live Load Allowance} \\ 400 \text{ kg/m}^2$$

→ Decrease LL. to 300 kg/m^2

$$\therefore R_w = \frac{32.5 \times 3}{4} = 24.4 \text{ kN}$$

$$R_L = 24.4 \text{ kN}$$

$$R_D = 5.5 \text{ kN}$$

$$R_{MAX} = 30.3 \text{ kN} \approx 30 \text{ kN}$$

V.O.K.

atelier one

Project BRITISH MUSEUM

No 1578k/10

Date 02/07 By LF

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Notes

Reactions Summary

- 1st Outer ring - bracing resisting Sway load

$$R_{max} = 19.8 \text{ kN} = 1.0DL + 1.0LL$$

$$R_{max} = 4.9 \text{ kN} = 1.0 \text{ Sway Load}$$

$$\therefore R_T = 24.7 \text{ kN}$$

$$R_T \leq R_{adm} = 30 \text{ kN}$$

- 2nd Outer ring

$$R_{max} = 25 \text{ kN}$$

$$R_T = 25 \text{ kN}$$

$$R_T \leq R_{adm} = 30 \text{ kN}$$

So.k.

- 3rd Outer ring - bracing resisting Sway Load

$$\textcircled{1} \quad R_{max} = 27.9 \text{ kN} = 1.0DL + 1.0LL$$

$$\textcircled{2} \quad R_{max} = 21.2 \text{ kN} = 1.0DL + 1.0LL$$

$$R_{max} = 2.1 \text{ kN} = 1.0 \text{ Sway Load}$$

$$\therefore R_T = 21.2 + 2.1 = 23.3 \text{ kN}$$

$$R_{max} = 27.9 \leq R_{adm} = 30 \text{ kN}$$

Including
Fabric wall

So.k.

without
bracing
with
bracing

So.k.

atelier one

Project BRITISH MUSEUM

No. 1578/5/11

Date 02/07 By LF

Checked _____

Notes

• 4th Outer ring — spreader beams carrying have
of Air handling units

$w = 885 \text{ kg}$ — Air handling unit weight

$\rightarrow 443 \text{ kg}$ carried by spreader

$$R = \frac{443 \times 10}{2 \times 1000} \therefore R = 2.2 \text{ kN}$$

$$R_{adm} = 30 - 2.2 = 27.8 \text{ kN}$$

$$R_{max} = 27.9 \text{ kN} \leq R_{adm}$$

✓.K.

• 5th Outer ring — spreader beams carrying have
of Air handling units

$w = 885 \text{ kg} \Rightarrow R = 2.2 \text{ kN}$ as above

$$R_{adm} = 30 - 2.2 = 27.8 \text{ kN}$$

$$R_{max} = 24.7 \text{ kN} \leq R_{adm}$$

✓.K.

• 6th Outer ring

$$R_{max} = 19.4 \text{ kN} \leq R_{adm}$$

$$R_{adm} = 30 \text{ kN}$$

✓.K.

atelier one

Project BRITISH MUSEUM

No. 1578/81

Date 02/07 By LF

Checked _____

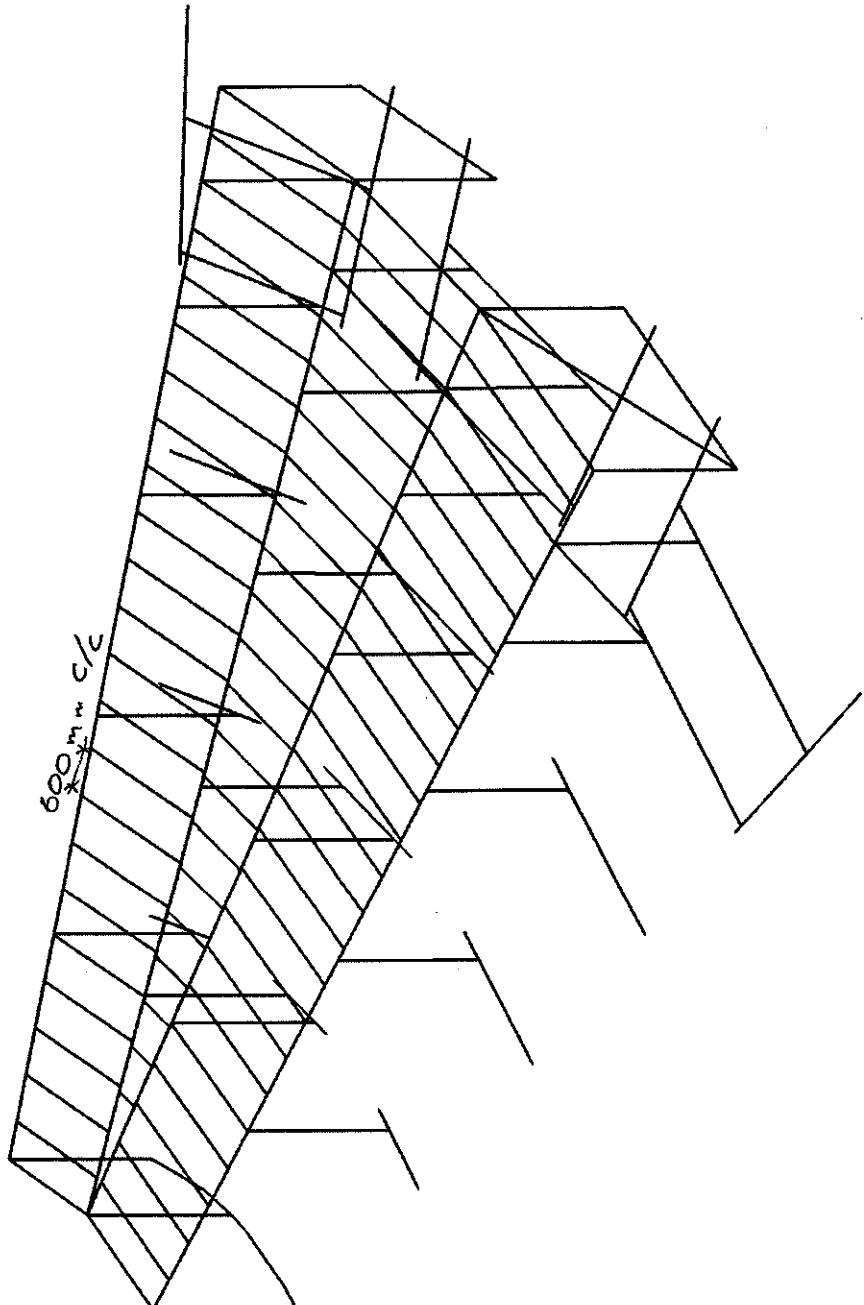
Notes

SECTIONAL MODEL - PLATFORM SLICE

Analysis model made for the platform is based with a variable secondary beams spacing, however the fabricators have suggested an even distribution of 600 mm c/c.

The Sectional Model is to compare the results of a typical bay with distribution of secondary beams according to sheetfubs drawings.

View - Cases: 1 (Dead Load) ANALYSIS - PLATFORM SLICE WITH SECONDARY BEAMS @ 600mm c/c



Z
X

View - Cases: 1 (Dead Load)

MEMBER PROPERTIES & STRUCTURE SUPPORTS

Secondary beams
PFC-H

175x65x1.4

DPFC 175x65x2.4
Radial Beams

152x152x2.3
UC
Spreader Beams

48.3x5 CHS
All Columns unless
noted otherwise

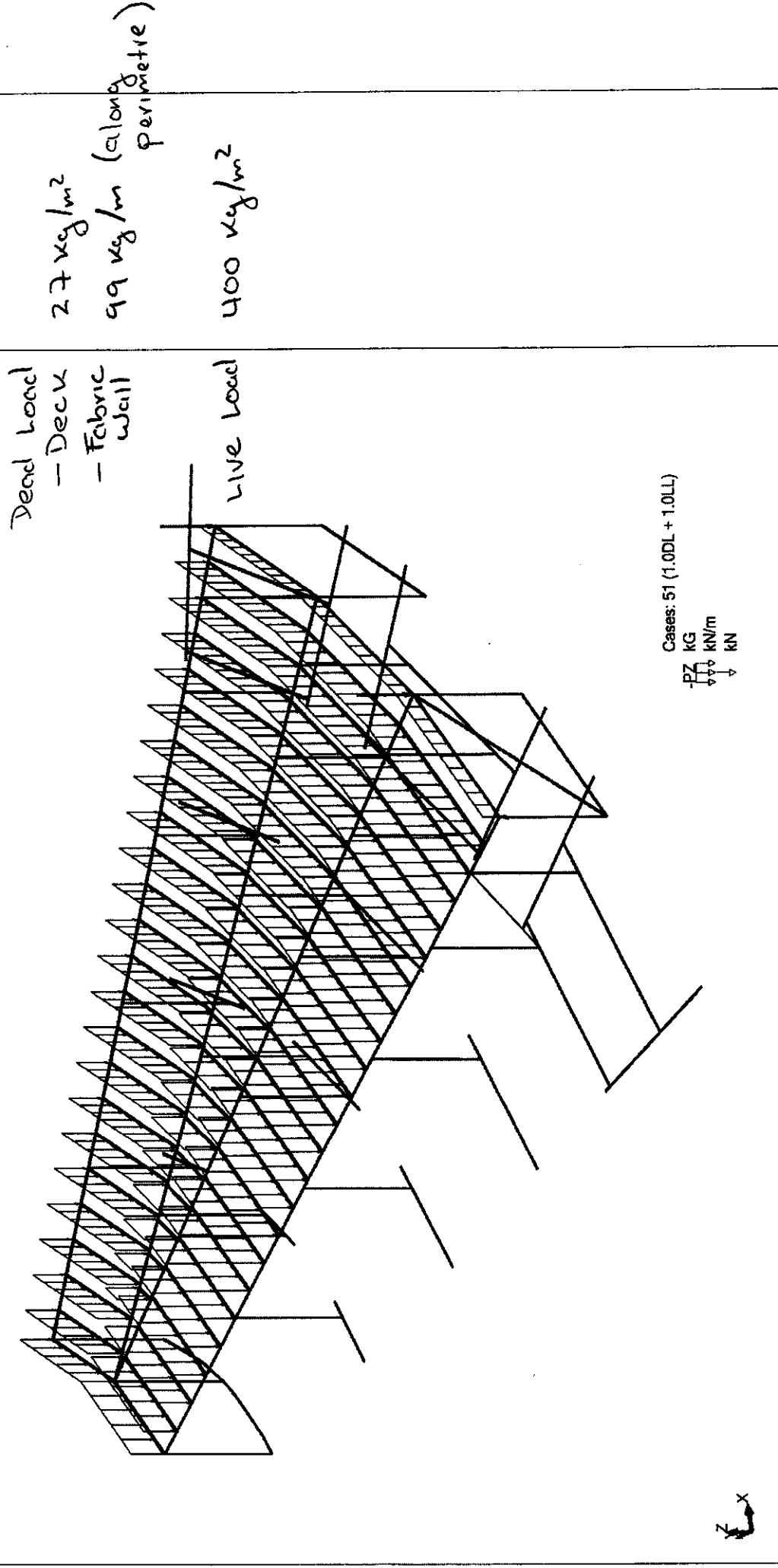
60.3x5 CHS
Columns

48.3x5 CHS
Bracing

Z
X

1578/16/3

View - Cases: 51 (1.0DL + 1.0LL) LOADS — DEAL & LIVE LOAD



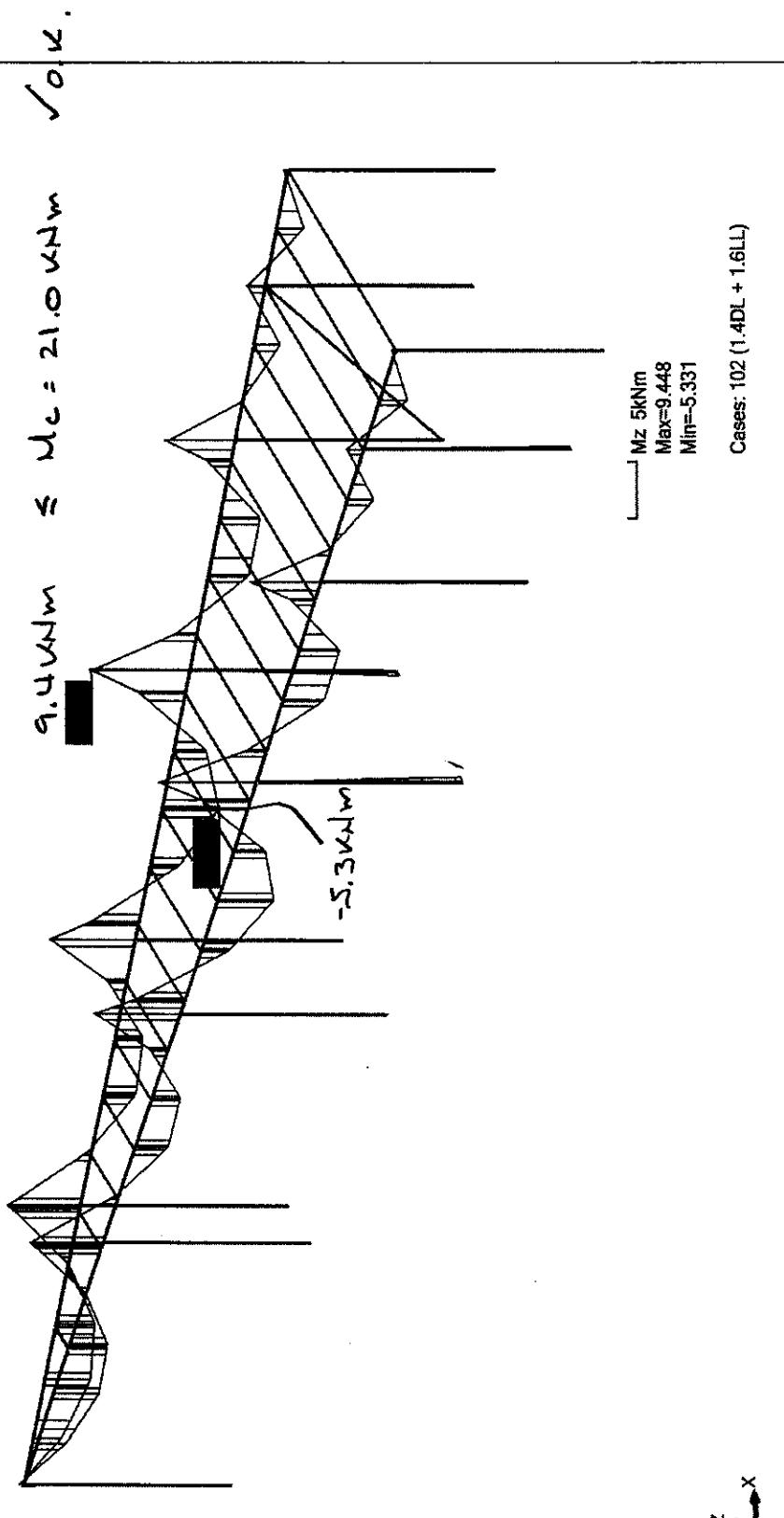
BAI ROBOT v 19.0.7
Author:
Address:

© RoboBAT 1996-2006
File: Slice.rtd
Project: Slice

View2 - MZ, Cases: 102 (1.4DL + 1.6LL)

BMD - LOAD COMBINATION 1.4DL + 1.6LL

Slice of Platform with Secondary Beams @ 600 mm c/c



View:2 - Exact deformation(s), Cases: 51 (1.0DL + 1.0LL) DEFLECTIONS - LOAD COMBINATION 1.0DL + 1.0LL

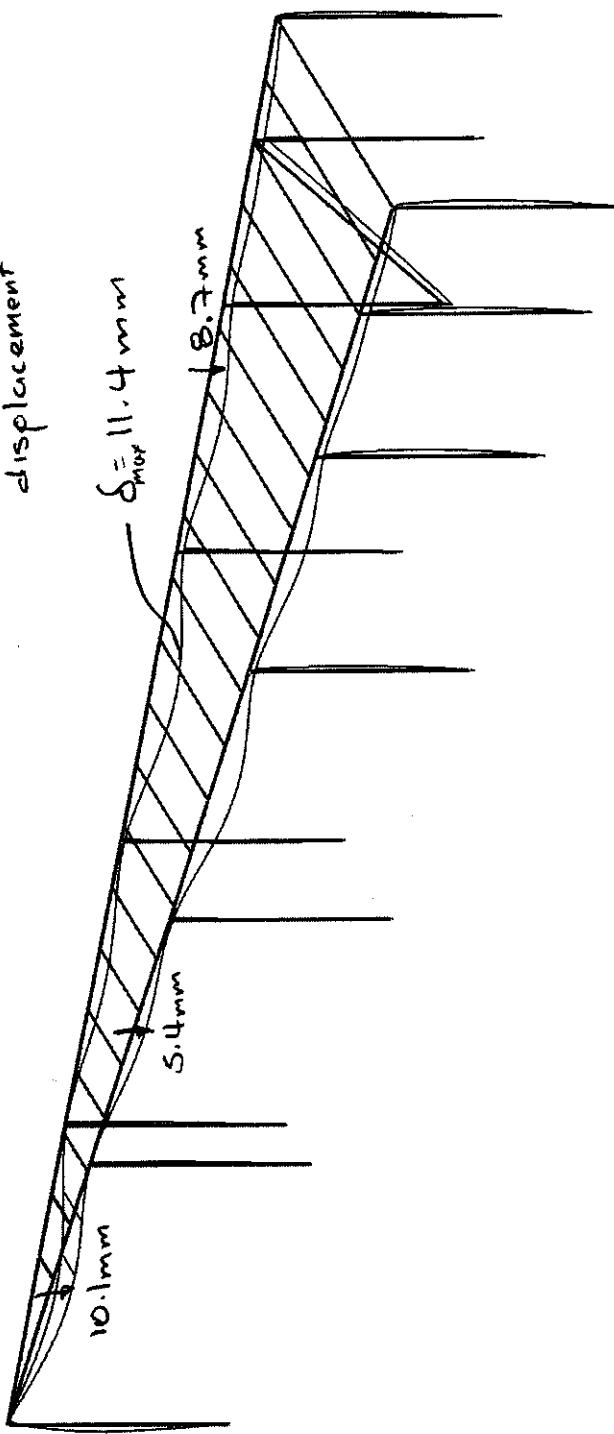
Slice OF PLATFORM WITH SECONDARY BEAMS @ 600 mm c/c

$$L_e = 2811$$

$$\text{max. relative displacement} \Rightarrow \Delta \delta = 9.0 \text{ mm}$$

$$\delta_{\text{down}} = \frac{2811}{360}$$

$$\delta_{\text{down}} = 7.8 \text{ mm}$$



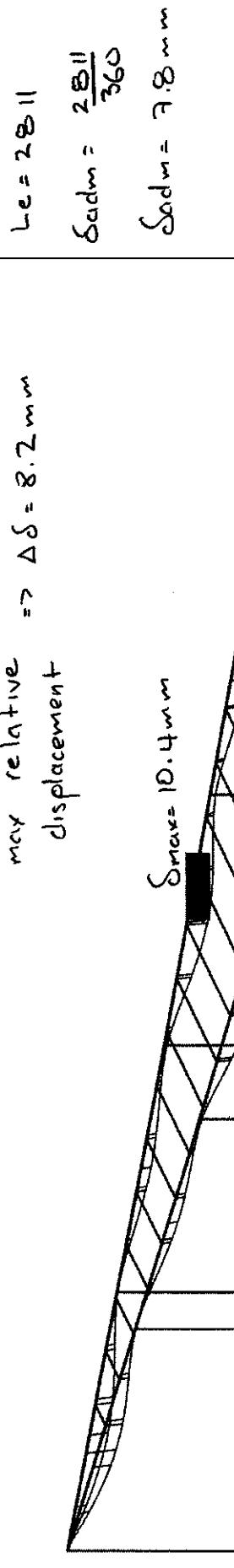
Dis 20mm
Max=11.444

Cases: 51 (1.0DL + 1.0LL)



View:1 - Exact deformation(s), Cases: 2 (Live Load) DEFLECTIONS - LOAD CASE 2: LIVE LOAD

max relative displacement $\Rightarrow \Delta \delta = 8.2 \text{ mm}$



Note

Base on LL = 400 kg/m²
Actual Live Load is 400 people
plus exhibition which is
far less than 400 kg/m²

Deflection is
OK by inspection.

Deflection is
OK by inspection.

1578/6/7

atelier one

Project BRITISH MUSEUM

No. 1578/6/8

Date 02/07 By LF

Checked _____

Notes

Sectional Analysis Model checked in
Deflection and Bending.

Therefore 600 mm c/c secondary beams
spacing is permissible.

atelier one

Project BRITISH MUSEUM

No. 1578/7/1

Date 01-02-07 By VB

Checked _____

Notes

FABRIC WALL

ANALYSIS MODEL:

The steel structure was analysed using Robot computer program.

STRUCTURE DESCRIPTION

The structure follow a circular geometry but in the model, we considered it faceted.

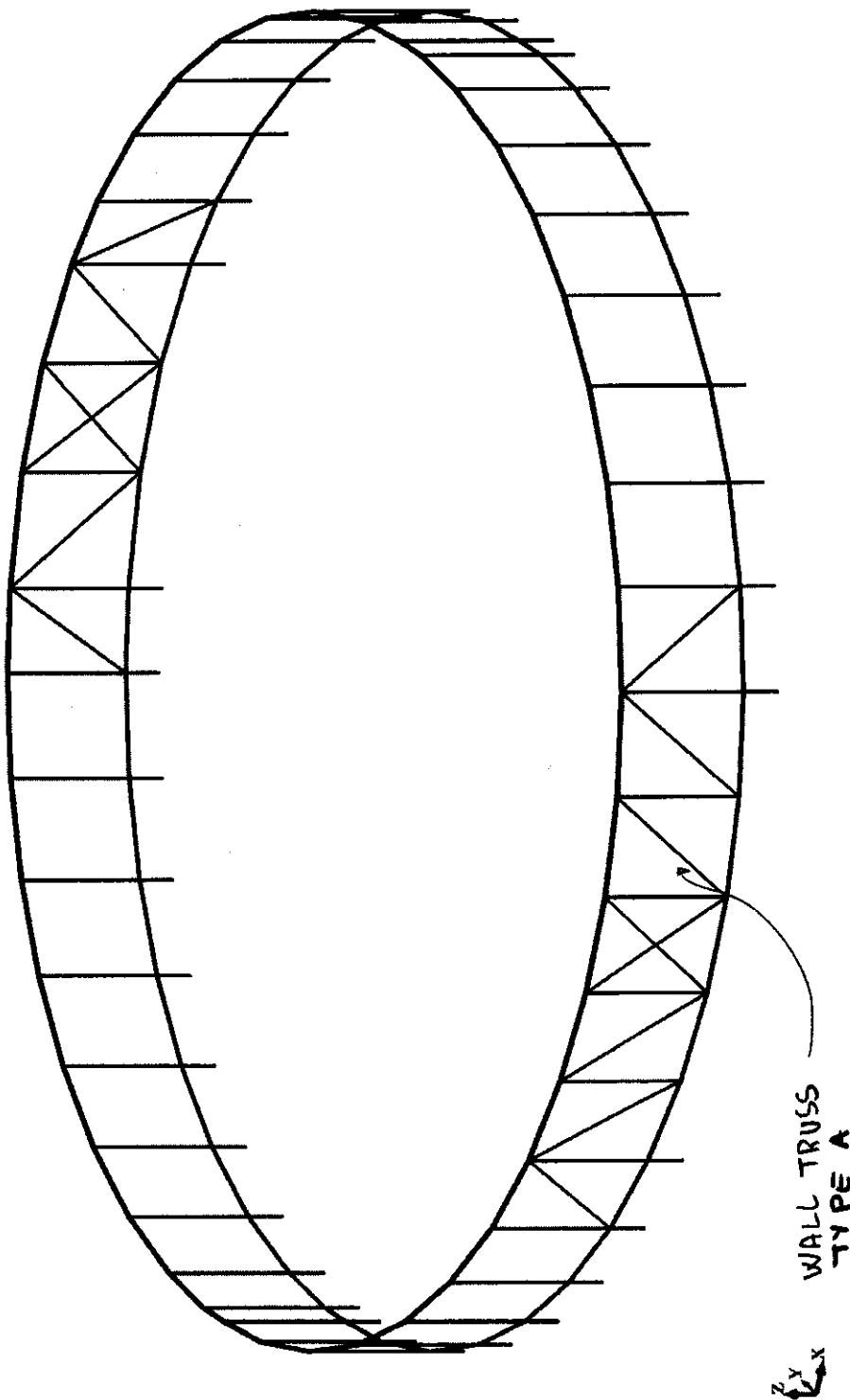
The Fabric is carried by rails to columns to deck.

At the access area we have two particular situations : 2 walls truss

View - Cases: 1 (Self Weight)

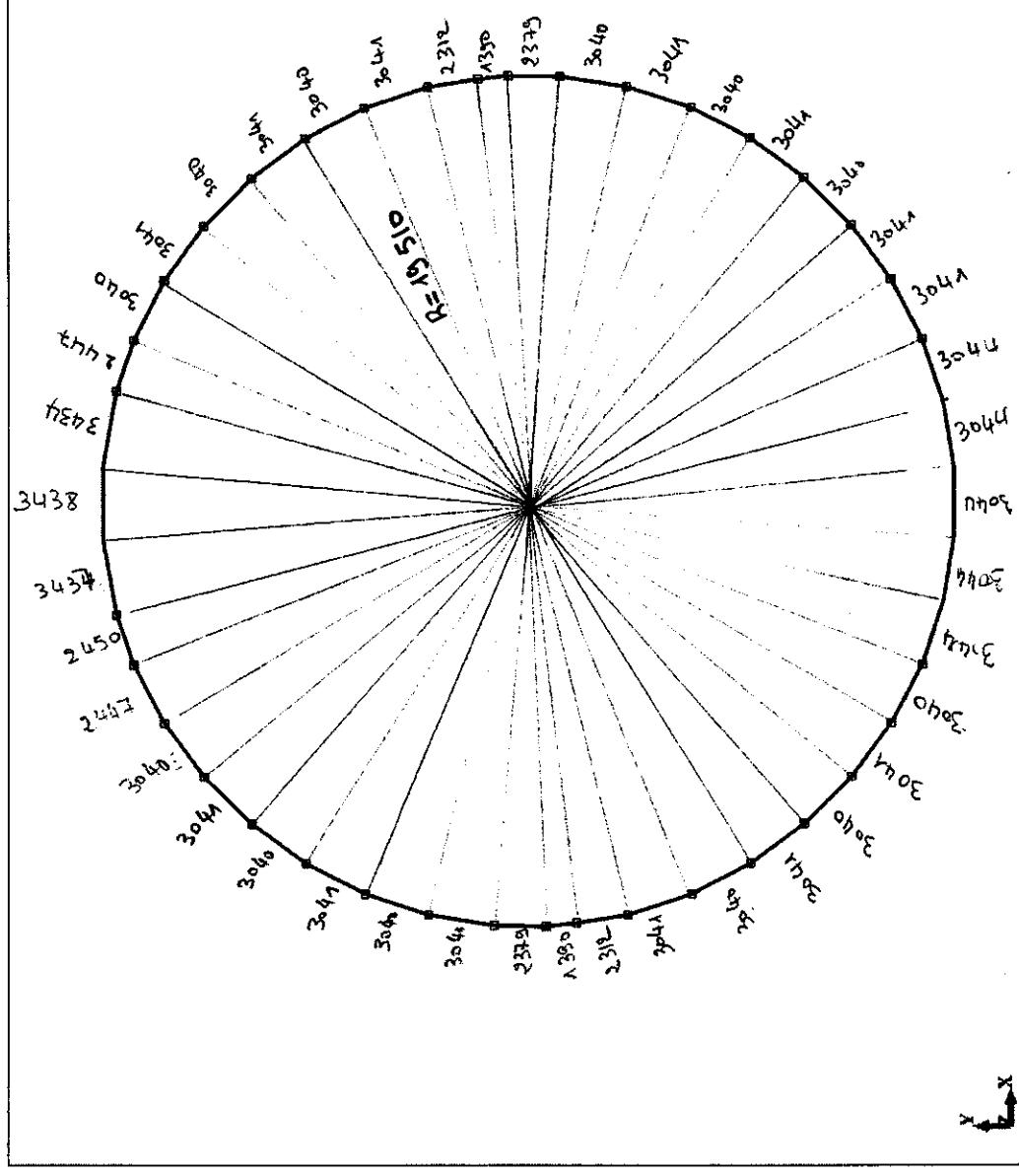
GENERAL VIEW

columns : CHS 60.3 x 4
 Top Rail : SHS 80 x 5
 Lower Rail : SHS 60 x 5
 Bracing : CHS 483 x 5



View - Cases: 1 (Self Weight)

VIEW IN PLAN

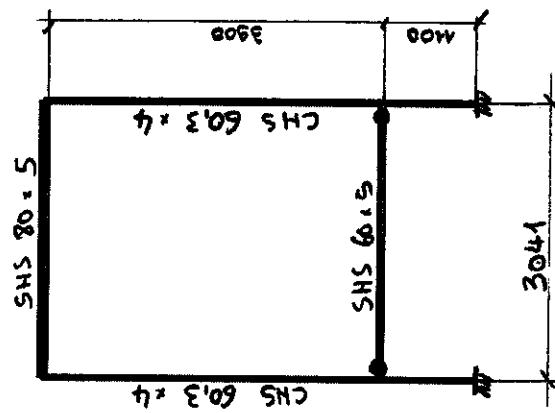


View:3 - Cases: 1 (Self Weigh)

TYPICAL SPAN

Note:

- The Top Rail is continuous
- The Lower Rail is pinned to the columns



Key:

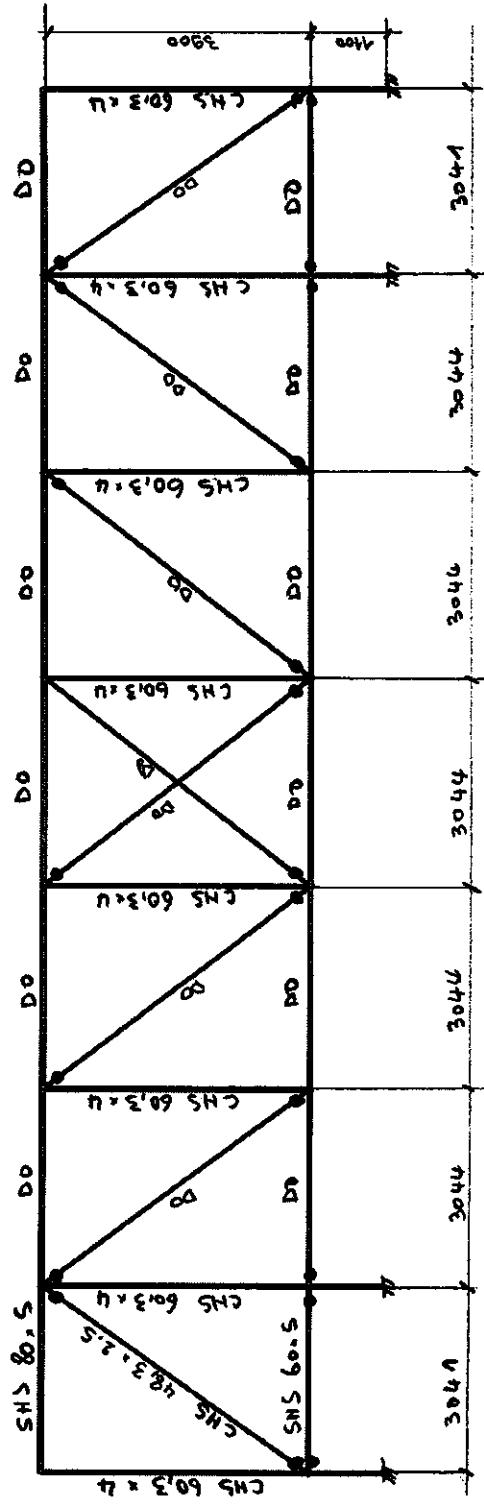
- † : Column fixed
- ◊ : Pinned member

View:2 - Cases: 1 (Self Weight)

WALL TRUSS TYPE A

Note:

- Top Rail is continuous
- Lower Rail is continuous between the columns



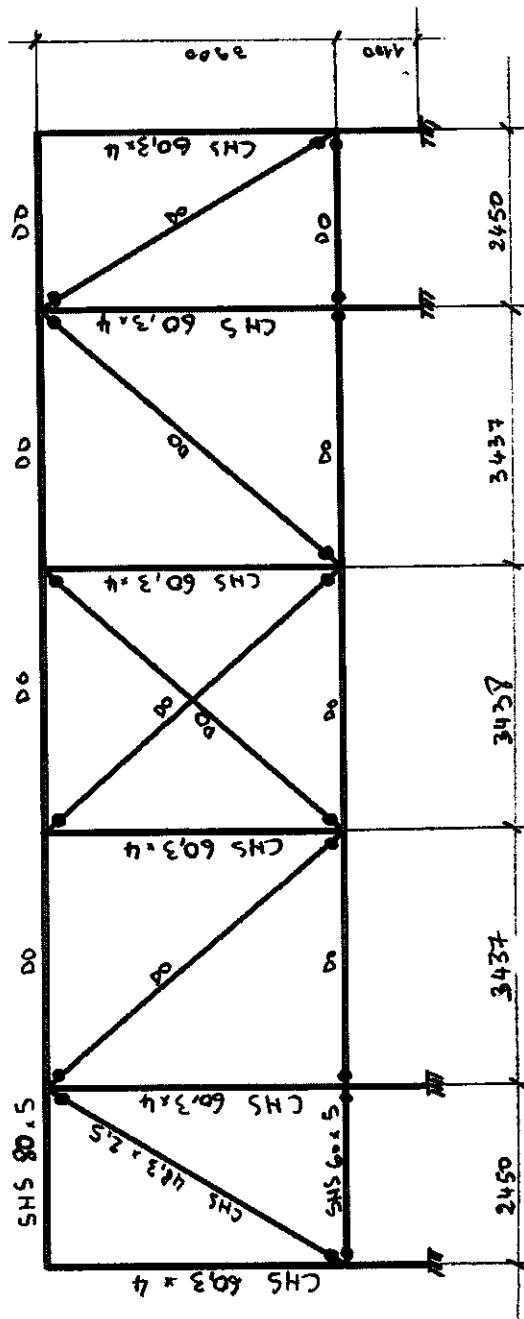
Key:
 ┌───────────┐ : Fixed column
 ● : Pinned Member

View:1 - Cases: 1 (Self Weight)

WALL TRUSS TYPE B

Note:

- Top Rail is continuous
- Low Rail is continuous between the two columns



atelier one

Project BRITISH MUSEUM

No. 1578/717

Date 01.02.87 By YB

Checked _____

Notes

ANALYSIS MODEL:

The steel structure was analysed using
the Robot computer program

The following pages describes the input
data into each load case.

atelier one

Project BRITISH MUSEUM

No. 15-78 17/8

Date 21-02-07 By YB

Checked _____

Notes

DEAD LOAD:

self weight of the steel work

Fabric weight: Assume $1 \text{ kg/m}^2 \rightarrow 0,01 \text{ kN/m}^2$
 $L = 3,9 \times 0,01 = 0,039 \text{ kN/m}$

Pretension from the fabric:

Assume wind applied to the fabric $= 1,5 \text{ kg/m}^2 = 0,015 \text{ kN/m}^2$

$$\text{Reaction} = \frac{0,015 \times 3,9}{2} = 0,03 \text{ kN}$$

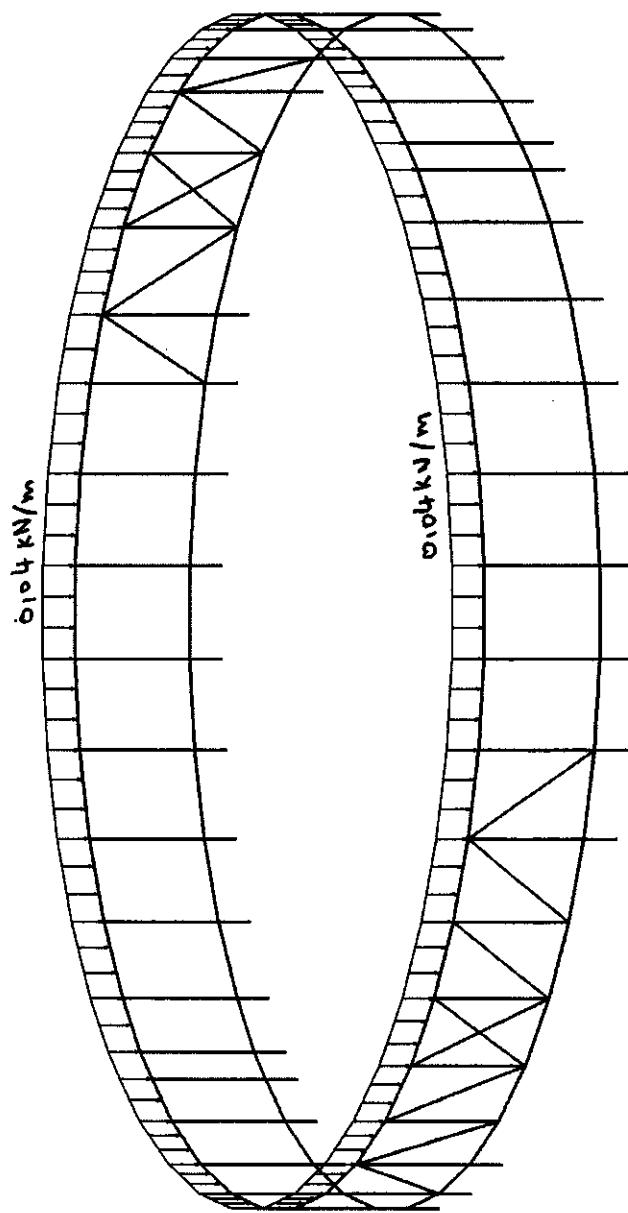
$$D.p = 75 \text{ mm}$$

$$R = \frac{L^2 + 4d^2}{8d} = \frac{3,9^2 + 4 \times 75^2}{8 \times 75} \\ = 25,388 \text{ mm}$$

$$\text{Tension} = 25,388 \times 1,5 = 38,1 \text{ kg/m}$$

$$\text{Tension} = 0,37 \text{ kN/m}$$

View - Cases: 2 (Fabric)

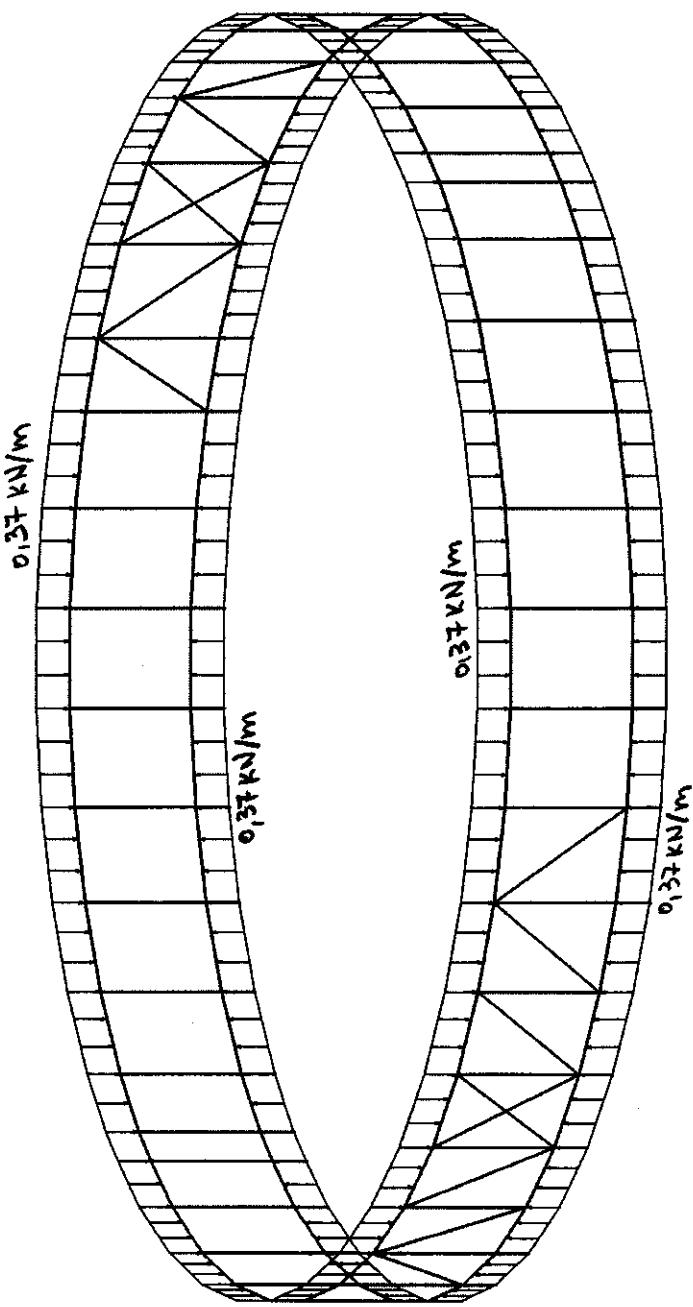
Cases: 2 (Fabric)
KN/m

BA ROBOT v 19.0.7
Author: Atelier One
Address:

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File: British Museum Fabric Wall.rtd
Project: British Museum Fabric Wall

View - Cases: 3 (Pretension)

**PRETENSION OF THE
FABRIC**



Case: 3 (Pretension)
 kN/m

Date : 01/02/07

Page : 1

atelier one

Project BRITISH MUSEUM

No. 1578 A/11

Date 01-02-07 By YB

Checked _____

Notes

Live Load:

- assume $0,74 \text{ kN/m}$ (no overcrowded areas)

Applied as UDL on 2 or 3 spans

Wind Load:

Assume $1,5 \text{ kg/m}^2 = 0,015 \text{ kN}$

The exhibition structure is indoor so the structure will not support natural wind load.

RHS

$$P = \frac{3,9}{2} \cdot 0,015 = 0,03 \text{ kN/m}$$

8
7
6

RHS

$$P = \frac{3,9 + 1,1}{2} \cdot 0,015 = 0,04 \text{ kN/m}$$

deck load

- Consider Wind load from 2 directions

Assume do not act at the same time.

Consider:

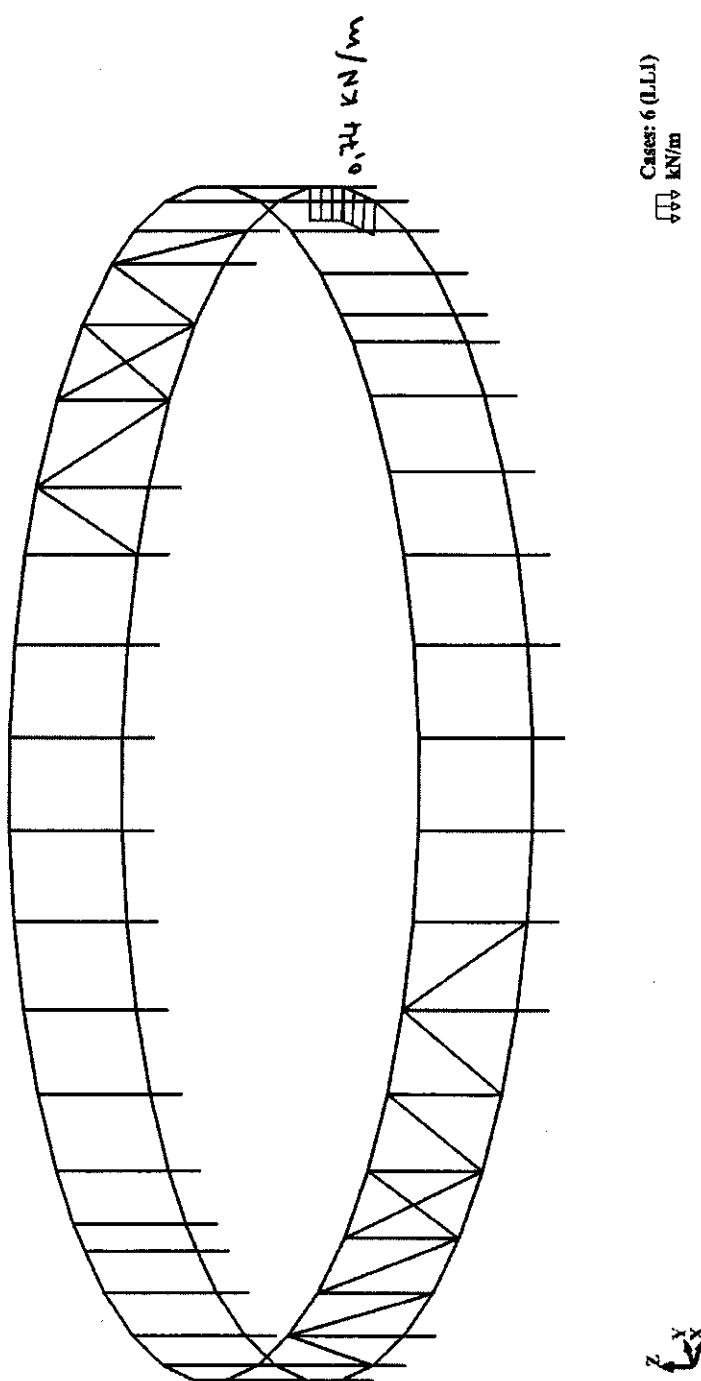
- Wind 1 Perpendicular to the Wall Truss

- Wind 2 Parallel to the Wall truss

1578/7/12

HAND RAIL LL1

View - Cases: 6 (LL1)



Cases: 6 (LL1)
0,74 kN/m

Date : 01/02/07

Page : 1

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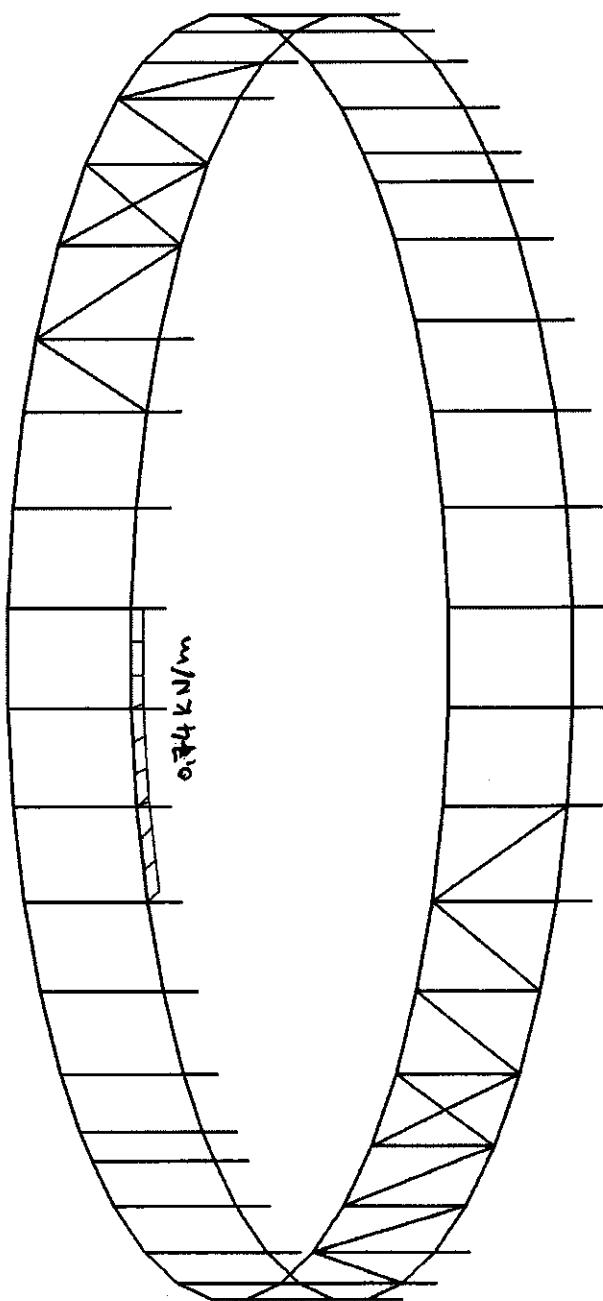
1578/7/13

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Address:

View - Cases: 7 (LL2)

HAND RAIL LL2



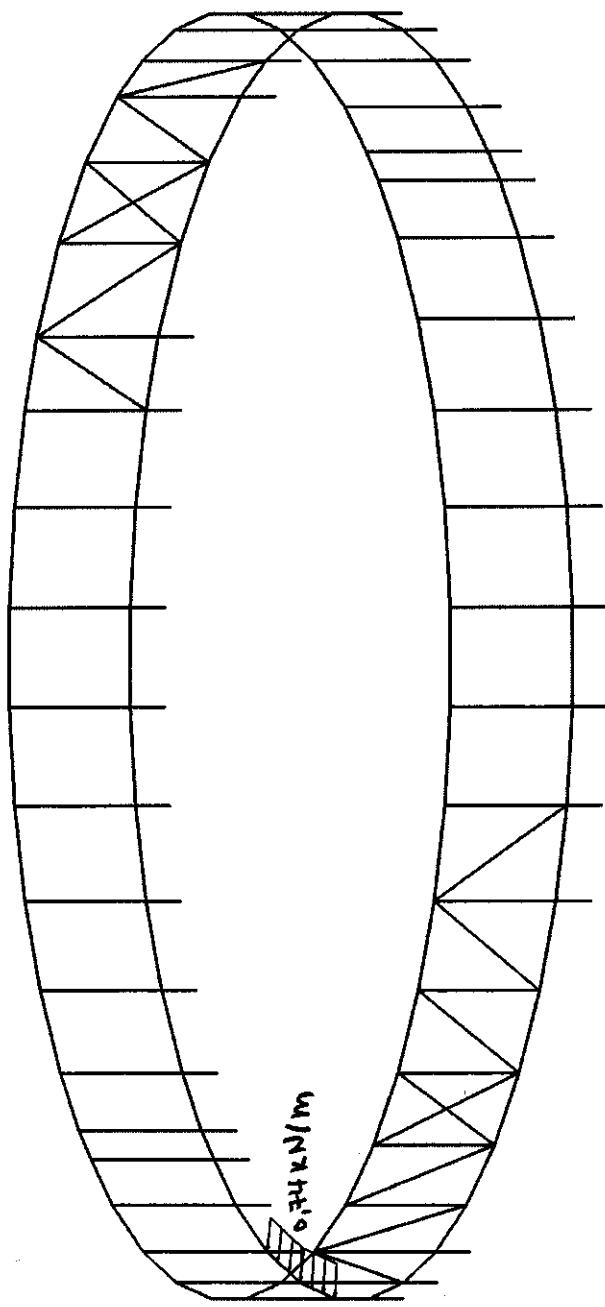
Cases: 7 (LL2)
↓↑ kN/m

Date : 01/02/07

Page : 1

View - Cases: 8 (LL3)

HAND RAIL LL3

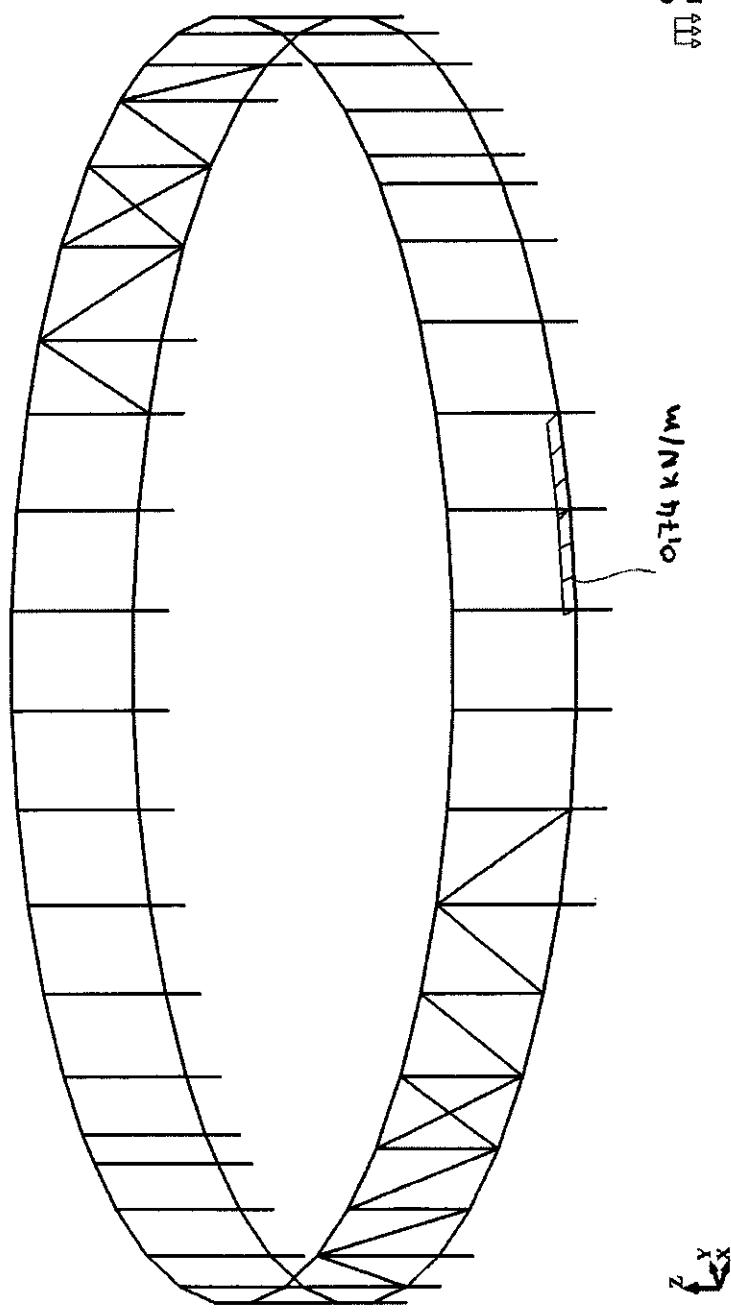
Cases: 8 (LL3)
↓↓↓ kN/mZ
Y
X

BAI ROBOT v 19.0.7
Author: Atelier One
Address:

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File: British Museum Fabric Wall.rtd
Project: British Museum Fabric Wall

View - Cases: 9 (LL4)

HAND RAIL LL4

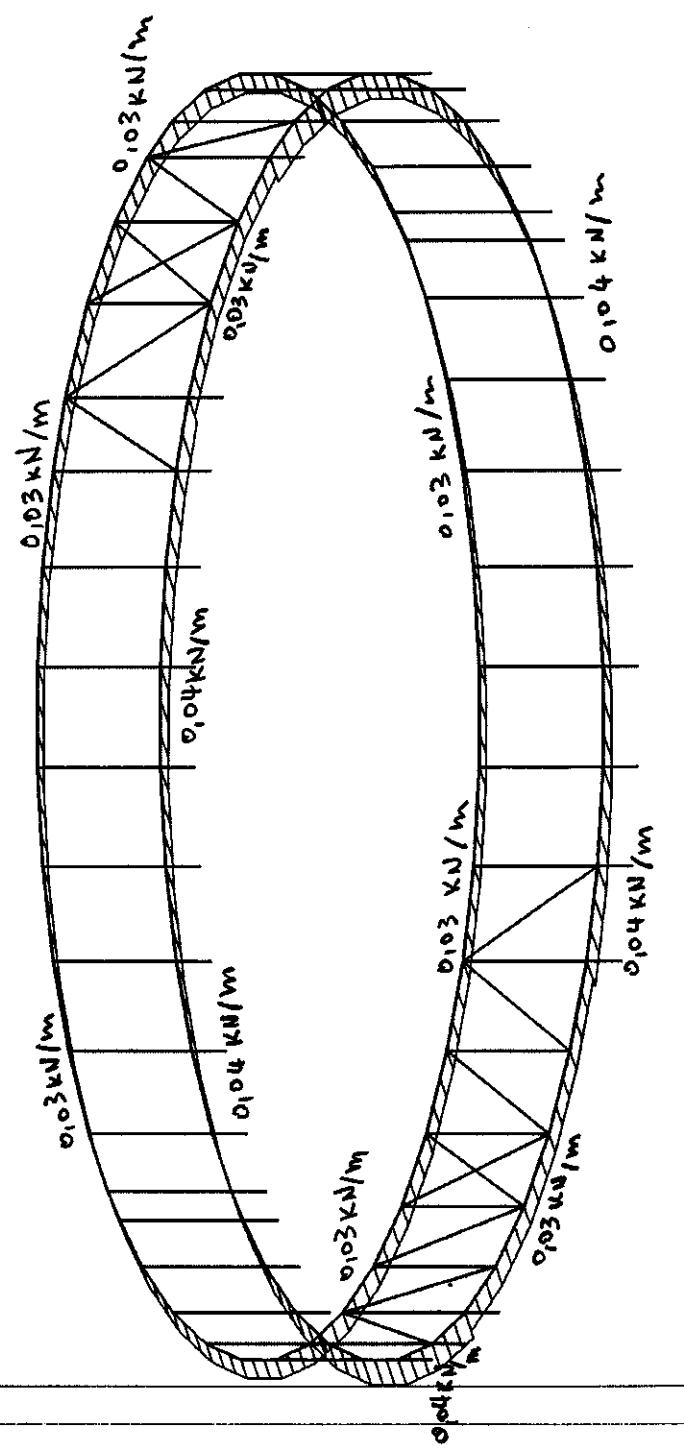


BAJ ROBOT v 19.0.7
Author: Atelier One
Address:

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File: British Museum Fabric Wall.rtd
Project: British Museum Fabric Wall

View - Cases: 4 (WIND1)

WIND 1

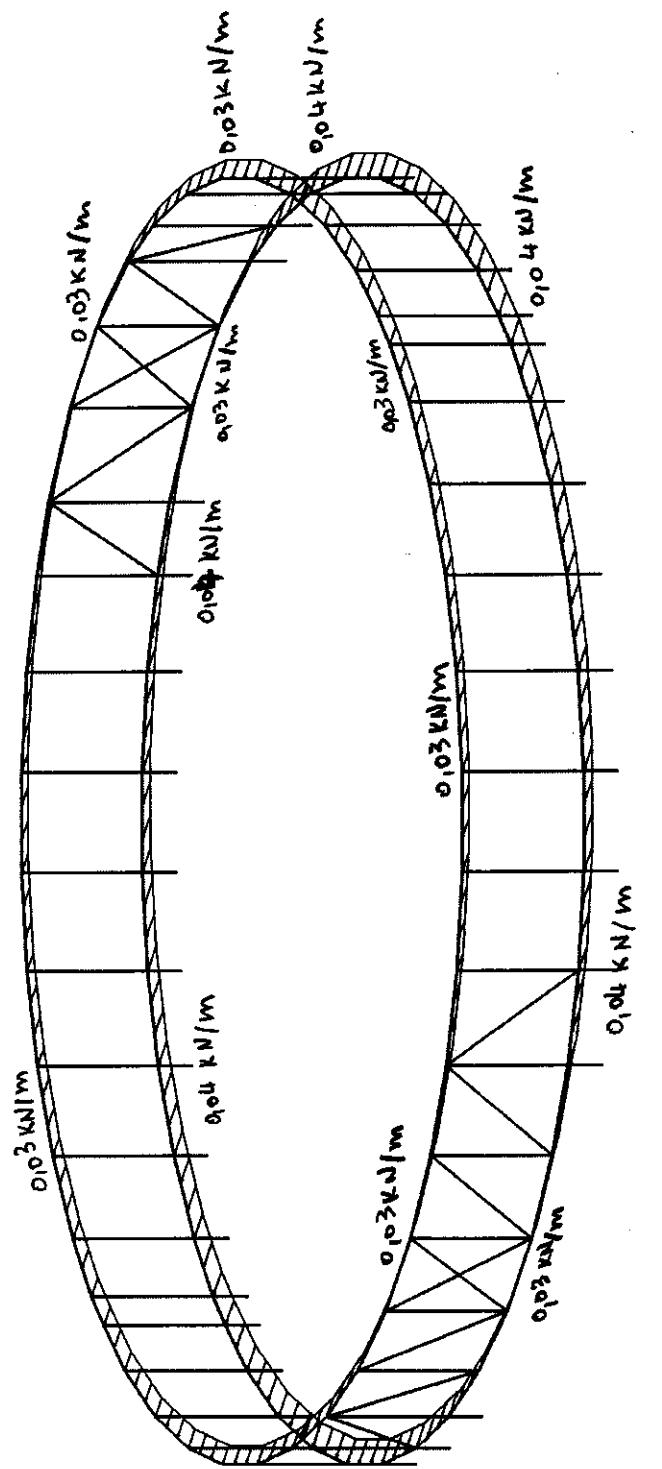
Cases: 4 (WIND1)
0.03 kN/mx
y
z

BA ROBOT v 19.0.7
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Address:

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File: British Museum Fabric Wall.rtd
Project: British Museum Fabric Wall

View - Cases: 5 (WIND2)

WIND 2



Cases: 5 (WIND2)
↙ ↘ ↗ ↙ kN/m



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File: British Museum Fabric Wall.rtd

Project: British Museum Fabric Wall

Loads:1

Info

1

Case	Case name	Nature	Analysis type
1	Self Weighth	dead	Static - Linear
2	Fabric	dead	Static - Linear
3	Pretension	dead	Static - Linear
4	WIND1	wind	Static - Linear
5	WIND2	wind	Static - Linear
6	LL1	live	Static - Linear
7	LL2	live	Static - Linear
8	LL3	live	Static - Linear
9	LL4	live	Static - Linear
50	1.0DL	dead	Linear Combination
51	1.0DL + 1.0LL1	dead	Linear Combination
52	1.0DL + 1.0LL2	dead	Linear Combination
53	1.0DL + 1.0LL3	dead	Linear Combination
54	1.0DL + 1.0LL4	dead	Linear Combination
55	1.0DL + 1.0WIND1	dead	Linear Combination
56	1.0DL + 1.0WIND2	dead	Linear Combination
61	1.0DL + 1.0LL1 + 1.0WIND1	dead	Linear Combination
62	1.0DL + 1.0LL2 + 1.0WIND1	dead	Linear Combination
63	1.0DL + 1.0LL3 + 1.0WIND1	dead	Linear Combination
64	1.0DL + 1.0LL4 + 1.0WIND1	dead	Linear Combination
65	1.0DL + 1.0LL1 + 1.0WIND2	dead	Linear Combination
66	1.0DL + 1.0LL2 + 1.0WIND2	dead	Linear Combination
67	1.0DL + 1.0LL3 + 1.0WIND2	dead	Linear Combination
68	1.0DL + 1.0LL4 + 1.0WIND2	dead	Linear Combination
101	1.4DL + 1.6LL1	dead	Linear Combination
102	1.4DL + 1.6LL2	dead	Linear Combination
103	1.4DL + 1.6LL3	dead	Linear Combination
104	1.4DL + 1.6LL4	dead	Linear Combination
105	1.0DL + 1.4WIND1	dead	Linear Combination
106	1.0DL - 1.4WIND1	dead	Linear Combination
107	1.0DL + 1.4WIND2	dead	Linear Combination
108	1.0DL - 1.4WIND2	dead	Linear Combination
111	1.4DL + 1.4WIND1	dead	Linear Combination
112	1.4DL - 1.4WIND1	dead	Linear Combination
113	1.4DL + 1.4WIND2	dead	Linear Combination
114	1.4DL - 1.4WIND2	dead	Linear Combination
121	1.2DL + 1.2LL1 + 1.2WIND1	dead	Linear Combination
122	1.2DL + 1.2LL2 + 1.2WIND1	dead	Linear Combination
123	1.2DL + 1.2LL3 + 1.2WIND1	dead	Linear Combination
124	1.2DL + 1.2LL4 + 1.2WIND1	dead	Linear Combination
125	1.2DL + 1.2LL1 + 1.2WIND2	dead	Linear Combination
126	1.2DL + 1.2LL2 + 1.2WIND2	dead	Linear Combination
127	1.2DL + 1.2LL3 + 1.2WIND2	dead	Linear Combination
128	1.2DL + 1.2LL4 + 1.2WIND2	dead	Linear Combination

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Project BRITISH MUSEUM

No. 1578/7/19

Date 01-02-07 By VB

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Notes

RESULTS:

Envelopes of peak forces and moments for each member type are shown over L₀

- These are for non-coexistent conditions

(i.e) Peak F_x (comp) = 10,59 kN [LC 112 Member 1048]

doesn't occur in same member on load case

as peak $M_y = 2,35 \text{ kNm}$

For checking members capacities, arrange data into co-existent conditions

(i.e) Extract all corresponding data for M_y, M_z for members and load cases where F_x occurs.

- Use this data for checking member capacities

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Project BRITISH MUSEUM

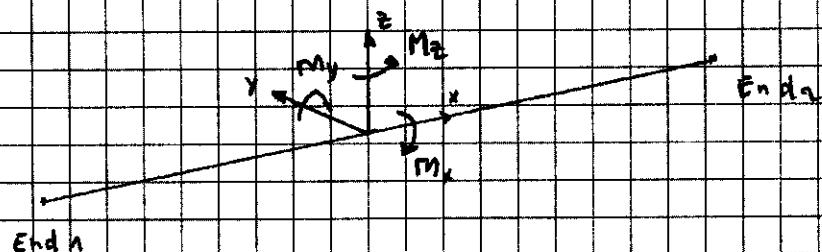
No. 157817/20

Date 01-02-07 By YB

Checked _____

Notes

Members Forces and Moments:



F_x = Axial Load (+ in Robot = compression)

F_y = Horiz Shear

F_z = Vert Shear

M_x = Torsion

M_y = Major Bending

M_z = Minor Bending

Forces - Cases: 50to56 61to68 101to108 111to114 121to128
Global extremes 1

- Cases: 50to56 61to68 101to108 111to114 121to128

Filtering	Bar	Case
Full list	100to141 200to210to9 50to56 61to	
Selection	1000to1030 103 50to56 61to68 1	
Total number	176	44
Selected number	78	35

- Cases: 50to56 61to68 101to108 111to114 121to128

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX	1.46	2.72	0.10	2.35	1.28	
Bar	1049	1012	1050	1019	1012	
Point	116	14	117	21	14	
Case	112 (C)	104 (C)	111 (C)	103 (C)	104 (C)	
MIN	-3.54	-2.72	-0.10	-2.35	-1.72	
Bar	1015	1019	1072	1012	1003	
Point	115	5	21	139	14	5
Case	113 (C)	101 (C)	111 (C)	104 (C)	121 (C)	

Forces - Cases: 101to108 111to114 121to128
 Global extremes
 1

- Cases: 101to108 111to114 121to128

Filtering	Bar	Case
Full list	100to141 200to21to9 50to56 61to	
Selection	100to141 101to108 111to1	
Total number	176	44
Selected number	42	20

- Cases: 101to108 111to114 121to128

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX	3.53	1.80	0.83	0.08	0.49	1.35
Bar	116	112	135	139	135	112
Point	118	113	137	139	137	510
Case	111 (C)	104 (C)	111 (C)	111 (C)	111 (C)	104 (C)
MIN	-3.19	-1.80	-0.83	-0.08	-0.46	-0.18
Bar	128	112	137	141	112	138
Point	129	112	138	141	610	139
Case	102 (C)	104 (C)	111 (C)	111 (C)	111 (C)	114 (C)

Forces - Cases: 101to108 111to114 121to128
Global extremes
1

- Cases: 101to108 111to114 121to128

Filtering	Bar	Case
Full list	100to141 200to210to56 61to	
Selection	200to234 236to210to108 111to1	
Total number	176	44
Selected number	42	20

- Cases: 101to108 111to114 121to128

	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX	2.40	0.28	1.30	0.28	0.48	0.68
Bar	240	216	238	239	236	240
Point	238	214	237	238	5/10	239
Case	111 (C)	111 (C)	114 (C)	111 (C)	112 (C)	111 (C)
MIN	-1.40	-0.28	-1.30	-0.28	-0.82	-0.65
Bar	233	239	236	216	205	215
Point	232	238	236	214	202	213
Case	126 (C)	111 (C)	113 (C)	111 (C)	111 (C)	112 (C)

TOP RAIL
BOX S SHS

Forces - Cases: 101to108 111to114 121to128
Global extremes
1

- Cases: 101to108 111to114 121to128

Filtering	Bar	Case
Full list	100to141 200to21to9 500to56 61to	
Selection	2000to2002 2001to08 111to1	
Total number	176	44
Selected number	14	20

- Cases: 101to108 111to114 121to128

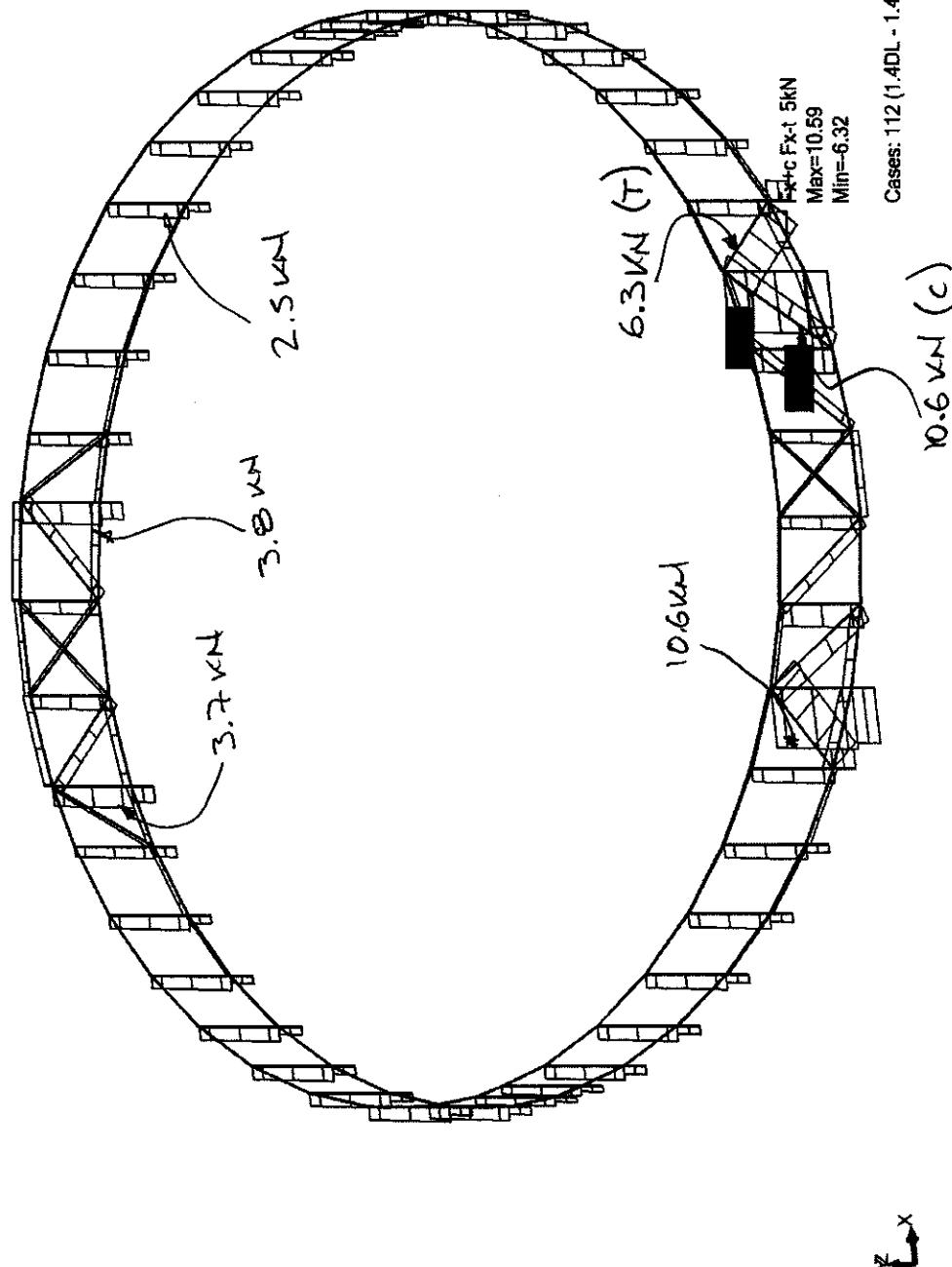
	FX (kN)	FY (kN)	FZ (kN)	MX (kNm)	MY (kNm)	MZ (kNm)
MAX	2.55	0.00	0.07	0.00	0.09	0.00
Bar	2015	2011	2000	2012	2001	2012
Point	134	213	138	141	5/10	214
Case	126 (C)	111 (C)	101 (C)	111 (C)	101 (C)	111 (C)
MIN	-6.32	-0.00	-0.07	-0.00	-0.00	-0.00
Bar	2011	2012	2000	2015	2014	2011
Point	213	141	236	232	119	115
Case	112 (C)	111 (C)	101 (C)	105 (C)	112 (C)	111 (C)

BRACING

+ B.3x2.5 cas

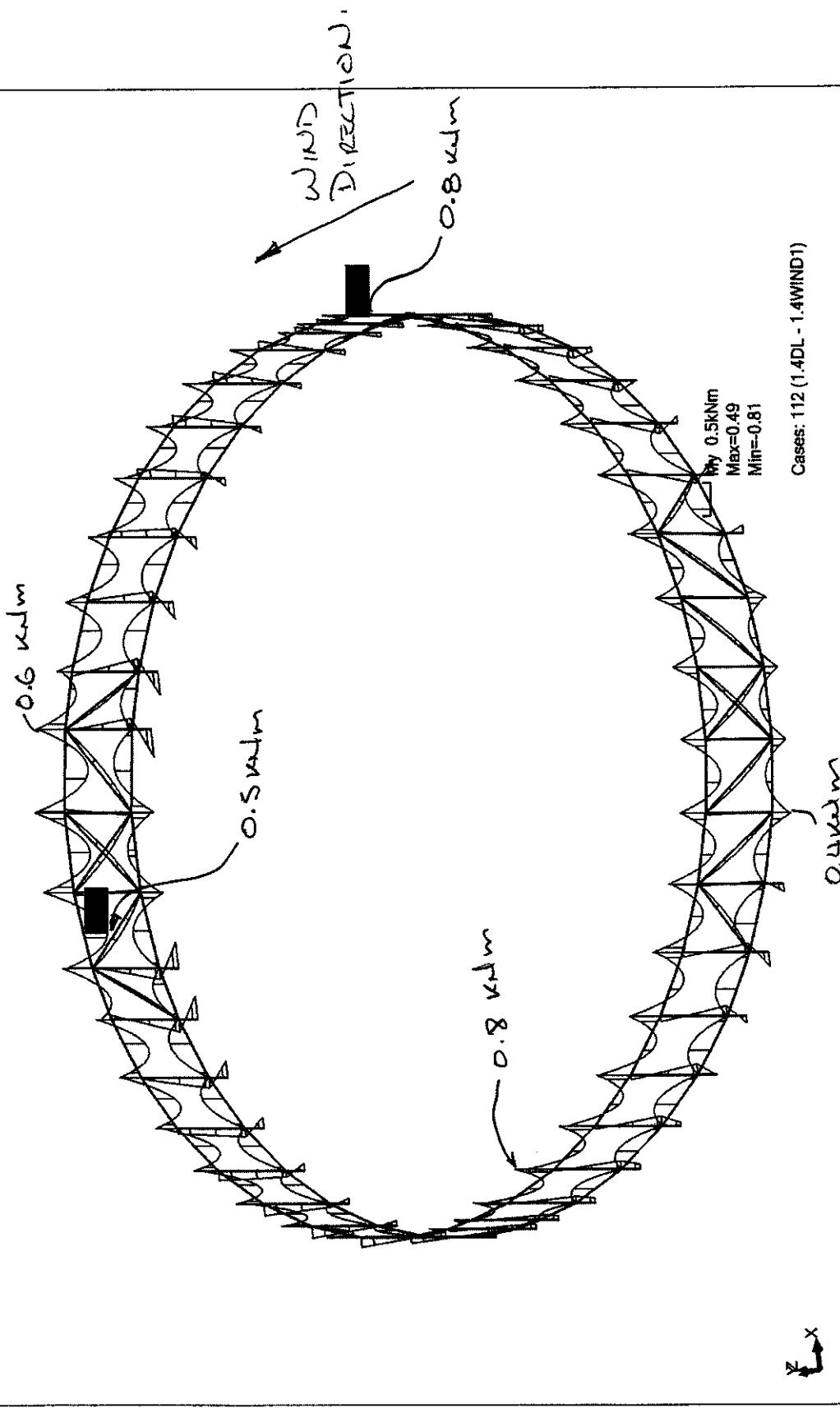
View - FX, Cases: 112 (1.4DL - 1.4WIND1)

Axial Force Diagram - LOAD COMBINATION 112 : 1.4DL - 1.4WIND1



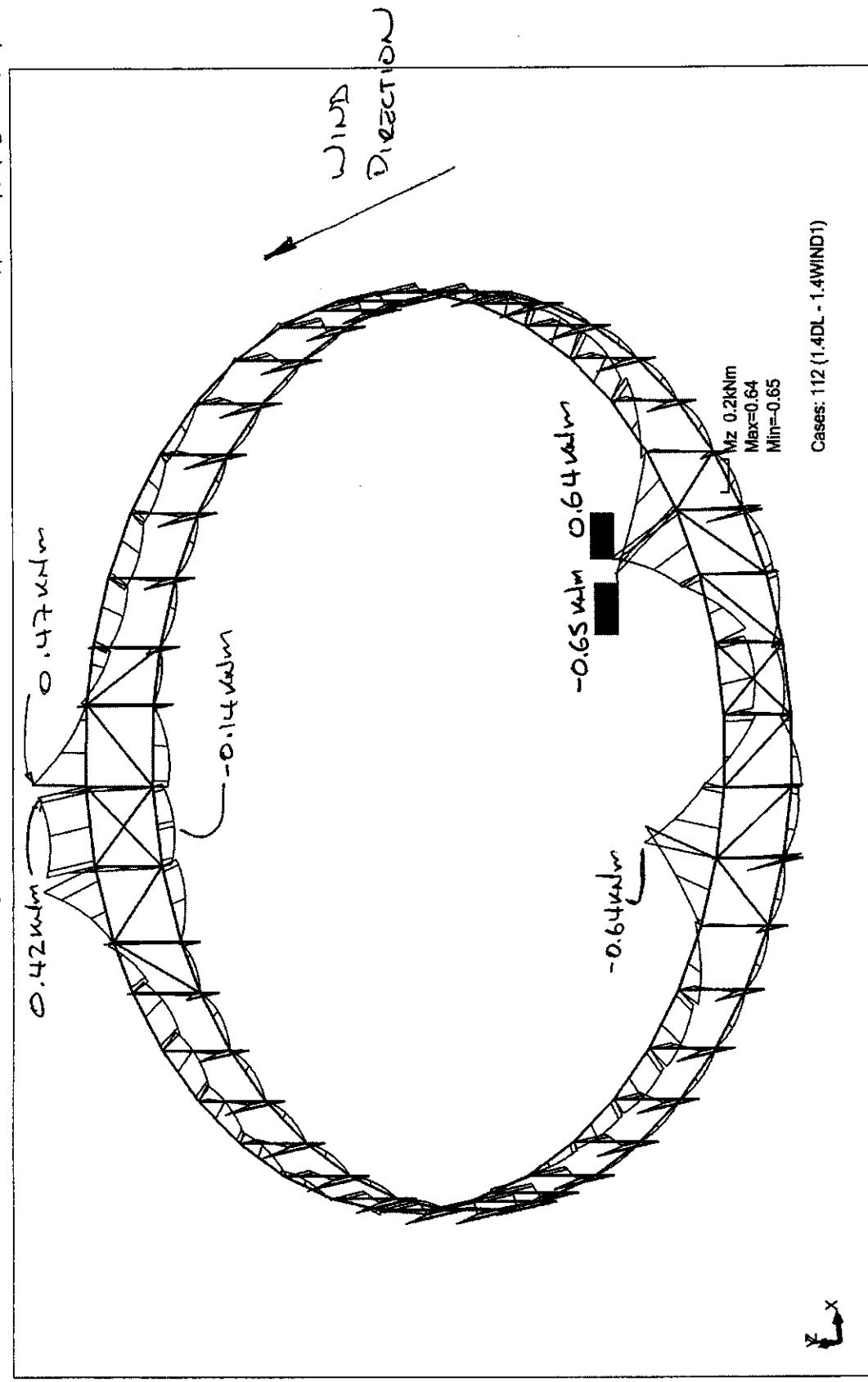
View - MY, Cases: 112 (1.4DL - 1.4WIND1)

BHD - major axis - LOAD COMBINATIONS 112: 1.4DL - 1.4WIND1



View - Mz, Cases: 112 (1.4DL - 1.4WIND1)

BMD - minor axis - LOAD COMBINATION 112 : 1.4DL - 1.4 WL



atelier one

Project BRITISH MUSEUM

No. 1578/7/28

Date 01-02-07 By YB

Checked _____

Notes

OVERALL BUCKLING CAPACITY:

Check all sections using the following criteria

$$\frac{F}{P_c} + \frac{M_y}{M_b} + \frac{M_z}{P_y Z_y} < 1$$

Where:

F is the applied axial load in the member

P_c is the compression capacity

M_y is the applied moment about the major axis

M_b is the buckling resistance moment capacity

M_z is the applied moment about the minor axis

P_y is the design strength.

Z_y is the elastic section modulus about minor axis

atelier one

Project BRITISH MUSEUM

No. 1578 17/29

Date 01.02.31 By YB

Checked _____

Notes

Columns Check:

Section: CMS 60.3 x 4

$$\rho_y Z_y = 3,3 \text{ kNm} \quad M_b = 4 \text{ kNm}$$

$$L_E = L = 5 \text{ m} \quad P_c = 21,9 \text{ kN}$$

case 1: Load 112 . 1.4 DL + 1.4 WIND 1

$$F_x = 16,59 \text{ kN} \quad M_y = -0,04 \text{ kNm} \quad M_z = -0,16 \text{ kNm}$$

$$\frac{16,59}{21,9} + \frac{0,04}{4} + \frac{0,16}{3,3} = 0,543 < 1 \quad \text{OK}$$

case 2: Load 104 1.4 DL + 1.6 LL 4

$$F_x = 136 \text{ kN} \quad M_y = -2,35 \text{ kNm} \quad M_z = 1,28 \text{ kNm}$$

$$\frac{136}{21,9} + \frac{2,35}{4} + \frac{1,28}{3,3} = 0,983 < 1 \quad \text{OK}$$

case 3: Load 121 1.2 DL + 1.2 LL 1 + 1.2 WIND 1

$$F_x = 1,20 \text{ kN} \quad M_y = -1,51 \text{ kNm} \quad M_z = -1,72 \text{ kNm}$$

$$\frac{1,20}{21,9} + \frac{1,51}{4} + \frac{1,72}{3,3} = 0,906 < 1 \quad \text{OK}$$

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Project BRITISH MUSEUM

No. 1578 17/30

Date 01-02-07 By YB

Checked _____

Notes

Lower Rail:

Section SWS 60x5

$$P_y z_y = 4,9 \text{ kNm} \quad M_b = 5,9 \text{ kNm}$$

Case 1: Load 1.11 : 1.4 DL + 1.4 WIND 1

$$F_x = 3,53 \text{ kN} \quad M_y = +0,37 \text{ kNm} \quad M_z = -0,05 \text{ kNm}$$

$$L_E = L = 3,041 \text{ m} \quad P_c = 101,7$$

$$\frac{3,53}{101,7} + \frac{0,37}{5,9} + \frac{0,05}{4,9} = 0,034 + 0,068 + 0,01 = \\ = 0,107 < 1 \quad \underline{\text{ok}}$$

Case 2: Load 1.11 : 1.4 DL + 1.4 WIND 1

$$F_x = 0,80 \text{ kN} \quad M_y = 0,49 \text{ kNm} \quad M_z = 0,03 \text{ kNm}$$

$$L_E = L = 3,43 \quad P_c = 81,7$$

$$\frac{0,80}{81,7} + \frac{0,49}{5,9} + \frac{0,03}{4,9} = 0,010 + 0,083 + 0,006 \\ = 0,099 < 1 \quad \underline{\text{ok}}$$

Case 3: Load 1.04 : 1.4 DL + 1.6 LL 4

$$F_x = -1,85 \text{ kN} \quad M_y = -0,46 \text{ kNm} \quad M_z = 135 \text{ kNm}$$

$$L_E = L = 3,041 \text{ m} \quad P_c = 101,7$$

$$\frac{-1,85}{101,7} + \frac{0,46}{5,9} + \frac{1,35}{4,9} = 0,006 + 0,078 + 0,276 \\ = 0,360 < 1 \quad \underline{\text{ok}}$$

atelier one

Project BRITISH MUSEUM

No. 1578/7/31

Date 2-2-07 By YB

Checked _____

Notes

Top Rail Check:

Section SNS X0x5

$$P_y Z_y = 9,4 \text{ kNm} \quad M_L = 11,5 \text{ kNm}$$

Case 1: Load M1 1,4 DL + 1,4 WIND 1

$$F_x = 2,40 \text{ kN} \quad M_y = -0,47 \text{ kNm} \quad M_z = 0,18 \text{ kNm}$$

$$L_e = L = 3,038 \quad P_c = 237,7 \text{ kN}$$

$$\frac{2,40}{237,7} + \frac{0,47}{11,5} + \frac{0,18}{3,4} = 0,016 + 0,041 + 0,019 \\ = 0,076 < 1 \text{ OK}$$

Case 2: Load M1 1,4 DL + 1,4 WIND 1

$$L_e = L = 3,04 \quad P_c = 237,7 \text{ kN}$$

$$F_x = 0,10 \text{ kN} \quad M_y = -0,82 \text{ kNm} \quad M_z = 0 \text{ kNm}$$

$$\frac{0,82}{11,5} = 0,071 < 1 \text{ OK}$$

Case 3: Load M1 1,4 DL + 1,4 WIND 1

$$F_x = 9,38 \text{ kN} \quad M_y = -0,47 \text{ kNm} \quad M_z = 0,68 \text{ kNm}$$

$$L_e = L = 3,04 \quad P_c = 237,7 \text{ kN}$$

$$\frac{9,38}{237,7} + \frac{0,47}{11,5} + \frac{0,68}{3,4} = 0,010 + 0,041 + 0,072 \\ = 0,123 < 1 \text{ OK}$$

atelier one

Project BRITISH MUSEUM

No. 1578/7132

Date 21-02-07 By VB

Checked _____

Notes

BRACING : CHS 48,3 x 2,5

Case 1: Load 112 · 1,4 DL + 1,4 WIN 1

$$F_x = -6,32 \text{ kN} \quad M_y = 0,00 \text{ kNm} \quad M_2 = 0,00 \text{ kNm}$$

$$L_e = L = 4,945 \quad P_c = 99,0 \text{ kN}$$

$$\frac{6,32}{99,0} = 0,064 < 1 \text{ ok}$$

Case 2: Load 112 · 1,2 DL + 1,2 LL 2 + 1,2 WIN 1

$$F_x = 2,55 \quad M_y = 0,00 \text{ kNm} \quad M_2 = 0,00 \text{ kNm}$$

$$L_e = L = 4,604 \text{ m} \rightarrow P_c = 8,6 \text{ kN}$$

$$\frac{2,55}{8,6} = 0,298 < 1 \text{ ok}$$

View:2 - Exact deformation(s), Cases: 52 (1.0DL + 1.0LL2)

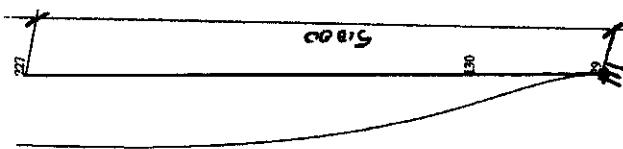
DEFLECTION

$$\delta_{max} = \sqrt{v_{x,max}^2 + v_{y,max}^2}$$

$$\delta_{max} = \sqrt{(18.2)^2 + (16.5)^2}$$

$$\delta_{max} = 24.56 \text{ mm}$$

$$\frac{L}{\delta_{max}} = \frac{5000}{24.56} = 203 > 180 \text{ ok}$$



Dia 10mm
Max=25.2

Cases: 52 (1.0DL + 1.0LL2)



atelier one

Project BRITISH MUSEUM

No. 1518/8/1

Date 02/07 By CJS

Checked _____

Notes

BRITISH MUSEUM

EXHIBITION LOAD ASSESSMENT

Check deck capacity & find loads
under heavily loaded areas.

BRONZE CHARIOT + 4 HORSES

Exhibition Wts,

Bronze chariot = 1061kg.

Horses = 340kg each.

Assume following distribution,

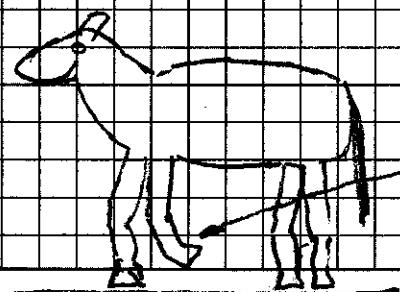
Chariot spread over 2 wheels,

Load/wheel = $\frac{1061}{2} = 530.5\text{kg}$.

Horse, $\frac{1}{2}$ load concentrated on one hoof

Peak leg load = $\frac{340}{2} = 170\text{kg}$

Check Horses - assume horse is walking.



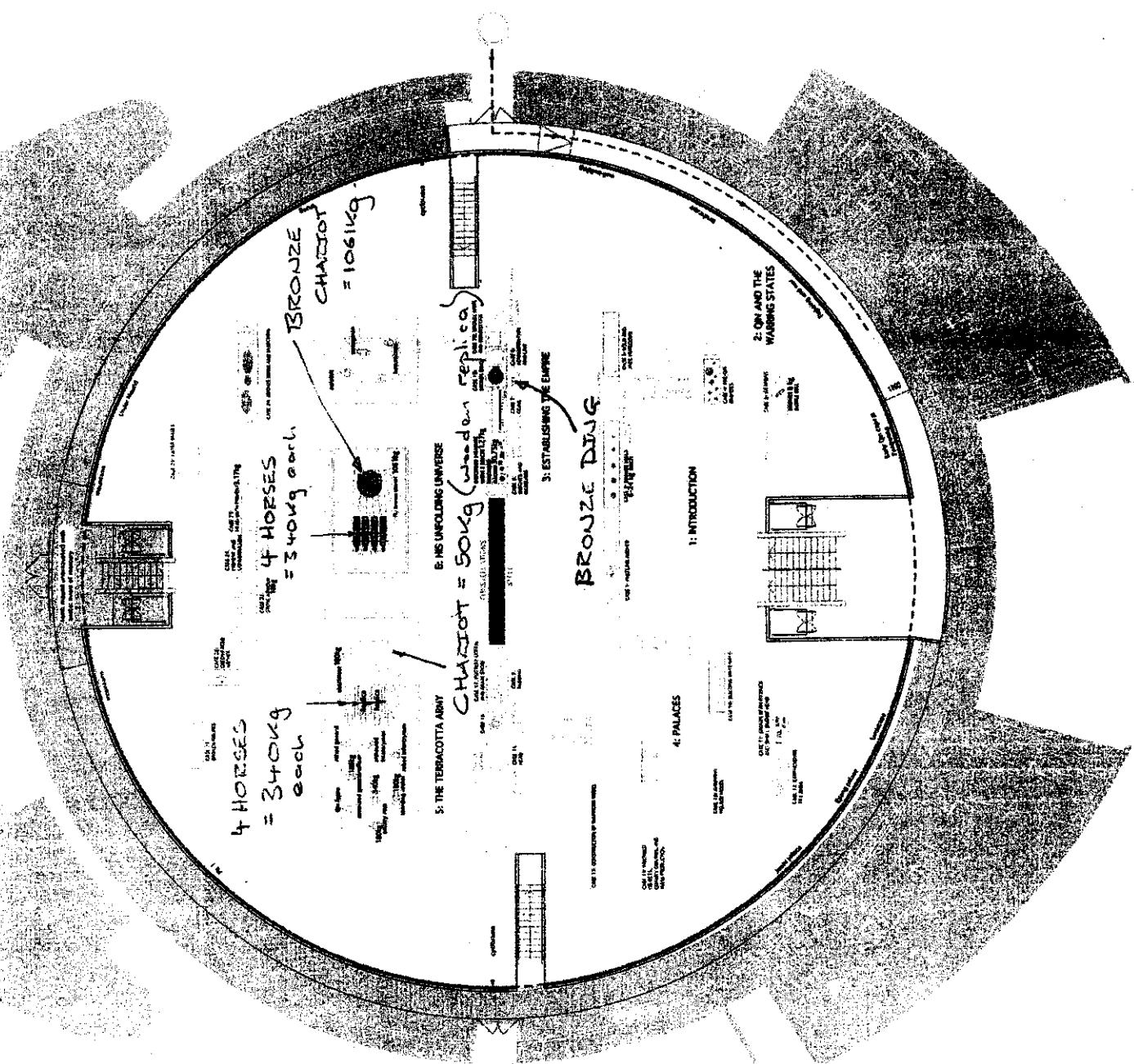
Ind. hoof ratio
contact with ground

NOTE:
Client requires
all 4 legs legs
on ground
Meeting
19/02/07

LAYOUT: FIRST EMPEROR

EXHIBITION PLAN WITH WEIGHTS
09 FEBRUARY 2007

NOTE: HAHN LOUDA SHOWCASES
APPROX. 780 KG EACH (700 x 700 x
1500mm HIGH)

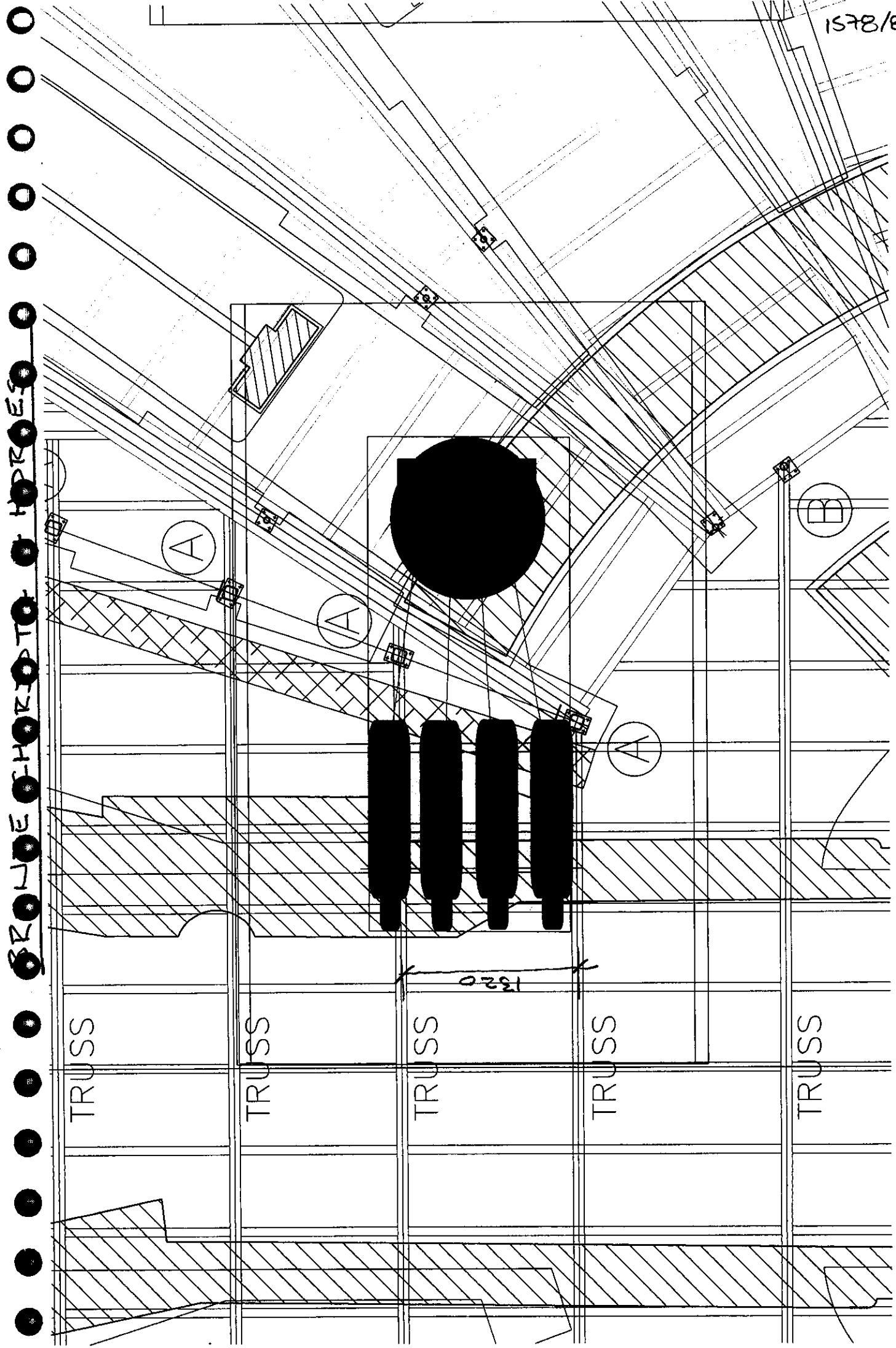


1578/012

metaphor

The British Museum
First Emperor
Project Address: The British Museum
Great Russell Street
London WC1B 3DG
Opening Date: 10 March 2007
Curator: Dr. Christopher Chipping
Designer: Studio Sainsbury
Photographer: Steve Williams
Page: 1

1578/B/3



atelier one

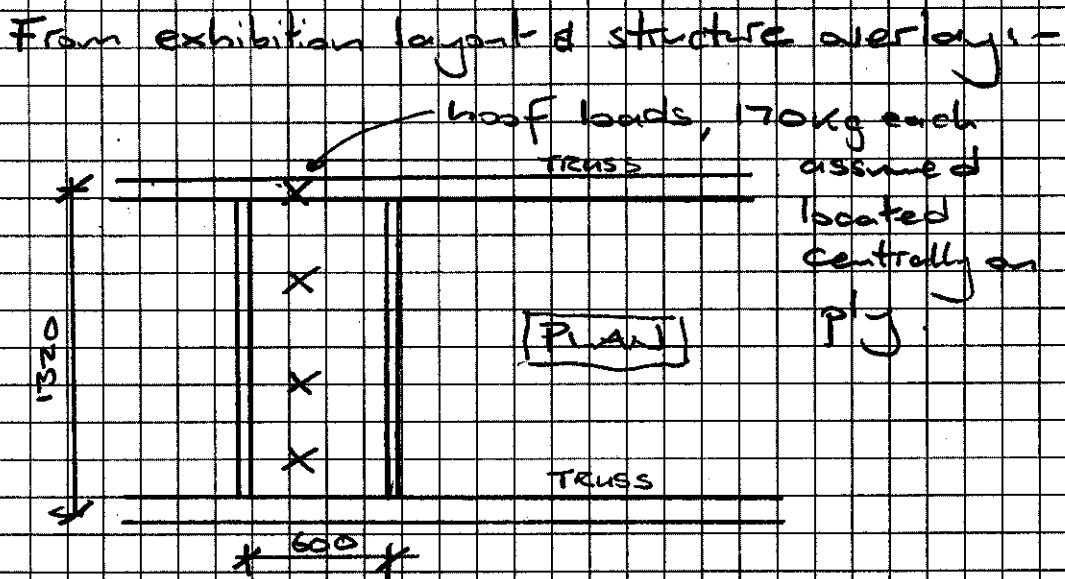
Project BRITISH MUSEUM

No. 1518/8/4

Date 02/07 By GSB

Checked _____

Notes



CHECK PLY - 25mm thickness

$$\text{Total load} = 4 \times 170\text{kg} = 680\text{kg.}$$

$$M = \frac{wl}{4} = \text{assume simply supported}$$

each side - conservative

$$M = \frac{6.8 \times 0.6}{4} = 1.02 \text{ kNm.}$$

$$Z = \frac{bd^3}{6} = \frac{1320 \times 25^2}{6} = 137500 \text{ mm}^3$$

For 2.5mm Canadian Douglas F1F

(commonly available & also weakest)

$$\sigma_{f/b} = 12.2 \text{ N/mm}^2$$

$$Mc = 137500 \times 12.2 \times 10^{-6} = 1.68 \text{ kNm}$$

$> 1.02 \text{ kNm}$

OK.

atelier one

Project BRITISH MUSEUM

No. 1578/8/S

Date. 02/01 By AB

Checked _____

Notes

Check defl.

$$I = \frac{bd^3}{12} = \frac{1320 \times 25^3}{12} = 1.719 \times 10^6 \text{ mm}^4$$

$$\delta = \frac{wL^3}{48EI} = \frac{6.8 \times 10^3 \times 600^3}{48 \times 4745 \times 1.719 \times 10^6} = 3.75 \text{ mm}$$

$$\text{allowable} = 0.003L = 0.003 \times 1320 = 3.96 \text{ mm} \\ > 3.75 \text{ mm}$$

$E_{\text{eff}} = 4745 \text{ N/mm}^2$

OK.

CHECK 175x65x1.4 PFC

$$\text{Span} = 1.32 \text{ m}$$

For simplicity assume all load concentrated
at centre, - conservative.

$$M = \frac{wl}{4} = \frac{6.8 \times 1.32}{4} = 2.24 \text{ kNm}$$

From previous, $M_c = 4.64 \text{ kNm} > 2.24 \text{ kNm}$

$$I = 196.7 \times 10^4 \text{ mm}^4$$

OK.

Check Defl.

$$S = \frac{6.8 \times 10^3 \times 1320^3}{48 \times 205000 \times 196.7 \times 10^4} = 0.8 \text{ mm}$$

$$\text{Span} = \frac{1320}{0.8} = 1650 > 360$$

OK.

atelier one

Project BRITISH MUSEUM

No. 1578/8/6

Date 02/07 By CTR

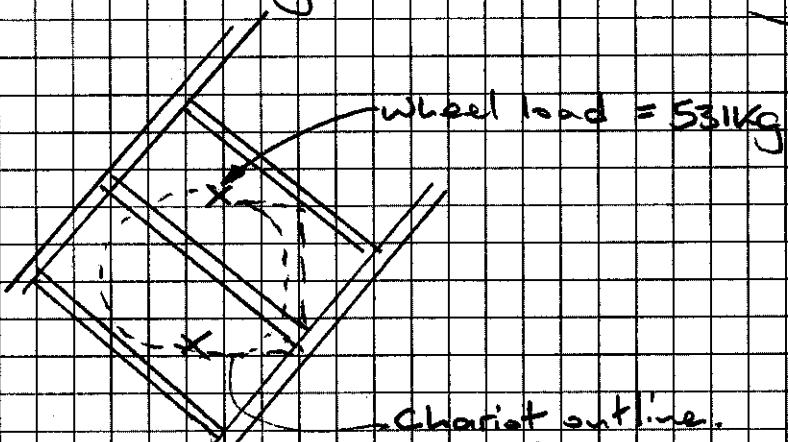
Checked _____

Notes

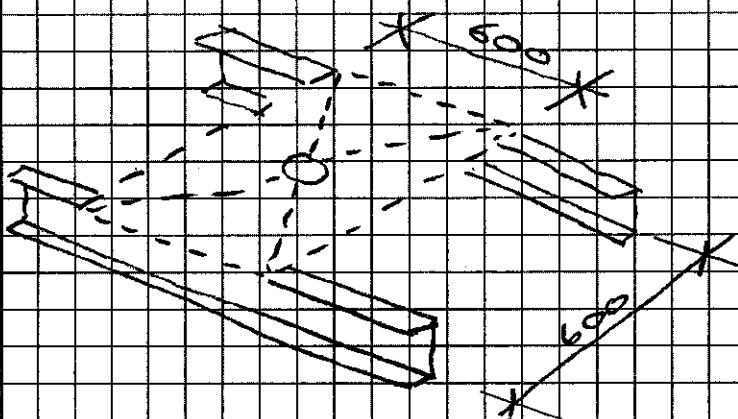
Check Chariot

Wheel load = 531kg

From exhibition layout & structure overlay:-



Assume following load distribution



$$Z = \frac{bd^2}{6} = \frac{600 \times 25^2}{6} = 62500 \text{ mm}^3$$

$$M_c = 12.2 \times 62500 = 0.763 \text{ kNm}$$

atelier one

Project

BRITISH MUSEUM

No. 1578/8/7

Date 02/07 By GJSB

Checked _____

Notes

$$M_{\text{applied}} = \frac{wL}{4} = \frac{5.31 \times 0.6}{4} = 0.80 \text{ kNm}$$

Note: if ply is continuous over

$$\text{support then } M < \frac{wL}{4} \approx \frac{wL}{6}$$

Check defl.

$$I = \frac{bd^3}{12} = \frac{600 \times 25^3}{12} = 781250 \text{ mm}^4$$

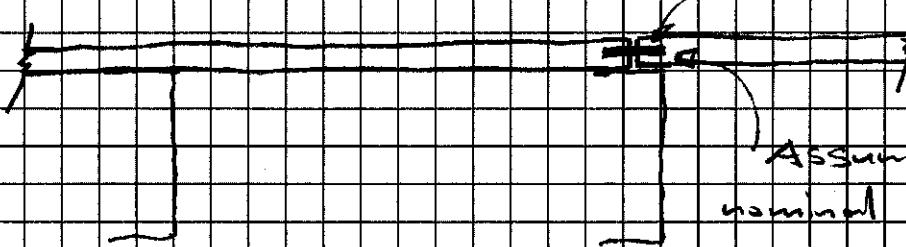
$$E = 474511 \text{ mm}^2$$

$$\delta = \frac{wL^3}{48EI} = \frac{5.31 \times 600^3}{48 \times 474511 \times 781250} = 6.4 \text{ mm}$$

If ply is continuous Then $\delta = 1.8 \text{ mm}$

$$\text{allowable} = 0.003 \times 600 = 1.8 \text{ mm}$$

From telecom with contractor, ply to be installed with biscuits



Assume only nominal continuity.

- Add additional PFC's under deck

atelier one

Project BRITISH MUSEUM

No. 1578 / 8 / 8

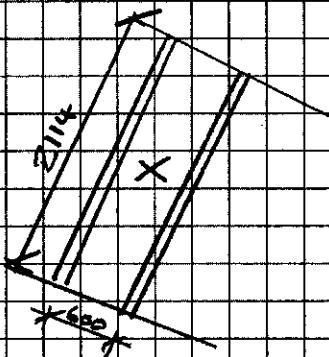
Date 02/07 By CJB

Checked _____

Notes

INDIVIDUAL WARRIOR

$$\text{Max wt} = 186 \text{ Kg} = 1.86 \text{ kN}$$



CHECK FLY - 25mm thick.

$$M = \frac{WL}{4} = \frac{1.86 \times 0.6}{4} = 0.279 \text{ kNm}$$

$$Z = \frac{bd^2}{6} = \frac{600 \times 25^2}{6} = 62500 \text{ mm}^3$$

$$\sigma_{\text{fly}} = 12.2 \text{ N/mm}^2$$

$$M_o = 62500 \times 12.2 \times 10^6 = 0.763 \text{ kNm}$$

$> 0.279 \text{ kNm}$

OK.

Check defl.

From previous, $I = 781250 \text{ mm}^4$, $E = 47451 \text{ N/mm}^2$

$$\delta = \frac{WL^3}{48EI} = \frac{1.86 \times 10^3 \times 600^3}{48 \times 47451 \times 781250} = 2.25 \text{ mm}$$

$$\text{Allowable} = 0.003 \times 600 = 1.8 \text{ mm}$$

atelier one

Project BRITISH MUSEUM No. 1578/8/9

Date 02/07 By GJB

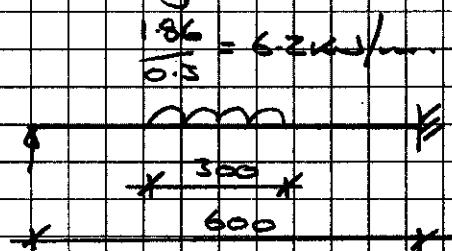
Checked _____

Notes

Re-check,

Min dims of washer = 300mm wide

Check following



From analysis, $S = 0.76\text{mm} < 1.8\text{mm}$

OK.

No add^u strengthening reqd

CHECK 175x65x1.4PFC

Span = 2.1m.

$$M = \frac{wL}{4} = \frac{1.86 \times 2.1}{4} = 0.98\text{kNm} < 4.64\text{kNm}$$

$$S = \frac{wL^3}{48EI} = \frac{1.86 \times 10^3 \times 2100^2}{48 \times 205000 \times 196.7 \times 10^6} = 0.89\text{mm}$$

$$\frac{\text{Span}}{S} = \frac{2100}{0.89} = 2360 > 360$$

OK.

atelier one

Project

BRITISH MUSEUM

No. 158 / 8/10

Date 02/07 By GJS.

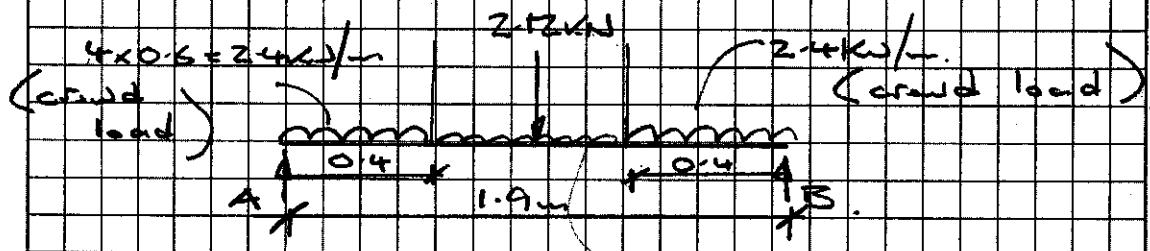
Checked

Notes

BRONZE DING.

1 No. Bronze bell = 212kg

Check PFC below,



Display case

$$S.W = 1.0 \text{ kN/m}$$

$$R_A = 2.4 \times 0.4 + \frac{1.0 \times 1.1}{2} + \frac{2.12}{2} = 2.57 \text{ kN}$$

→ midspan.

$$M = 2.57 \times 0.95 - (2.4 \times 0.4 \times 0.75) - \frac{(1.0 \times 0.75^2)}{2}$$

$$\approx 1.44 \text{ kNm} < 4.34 \text{ kNm}$$

OK.

atelier one

Project BRITISH MUSEUM

No. 1578/8/11

Date 02/07 By CJB

Checked _____

Notes

CHECK FWD LOADS

Horses

Marking plan shown over

$$\text{Total wt} = 4 \times 360 = 1440 \text{ kg}$$

$$\text{By inspection of plan, load carried by 2 h. Cols} = \frac{1440}{2} = 720 \text{ kg/col.}$$

- Check Col 1.

$$\text{Load applied, 2 h. horses} = 720 \text{ kg.}$$

$$\frac{1}{4} \text{ Chariot} = \frac{1061}{4} = 265 \text{ kg.}$$

restricted access area

$$= 3.5 \times 0.7 + 0.75 \times 4.0 = 5.45 \text{ m}^2.$$

$$\text{Load allowance} = 75 \text{ kg/m}^2, \text{ say}$$

$$\text{Total load} = 720 + 265 + (75 \times 5.45)$$

$$= 1393 \text{ kg} = 13.94 \text{ kN}$$

< 30 kN

OK.

- Check Col 2.

$$\text{Total area} = 1.5 \times 2.1 = 3.15 \text{ m}^2$$

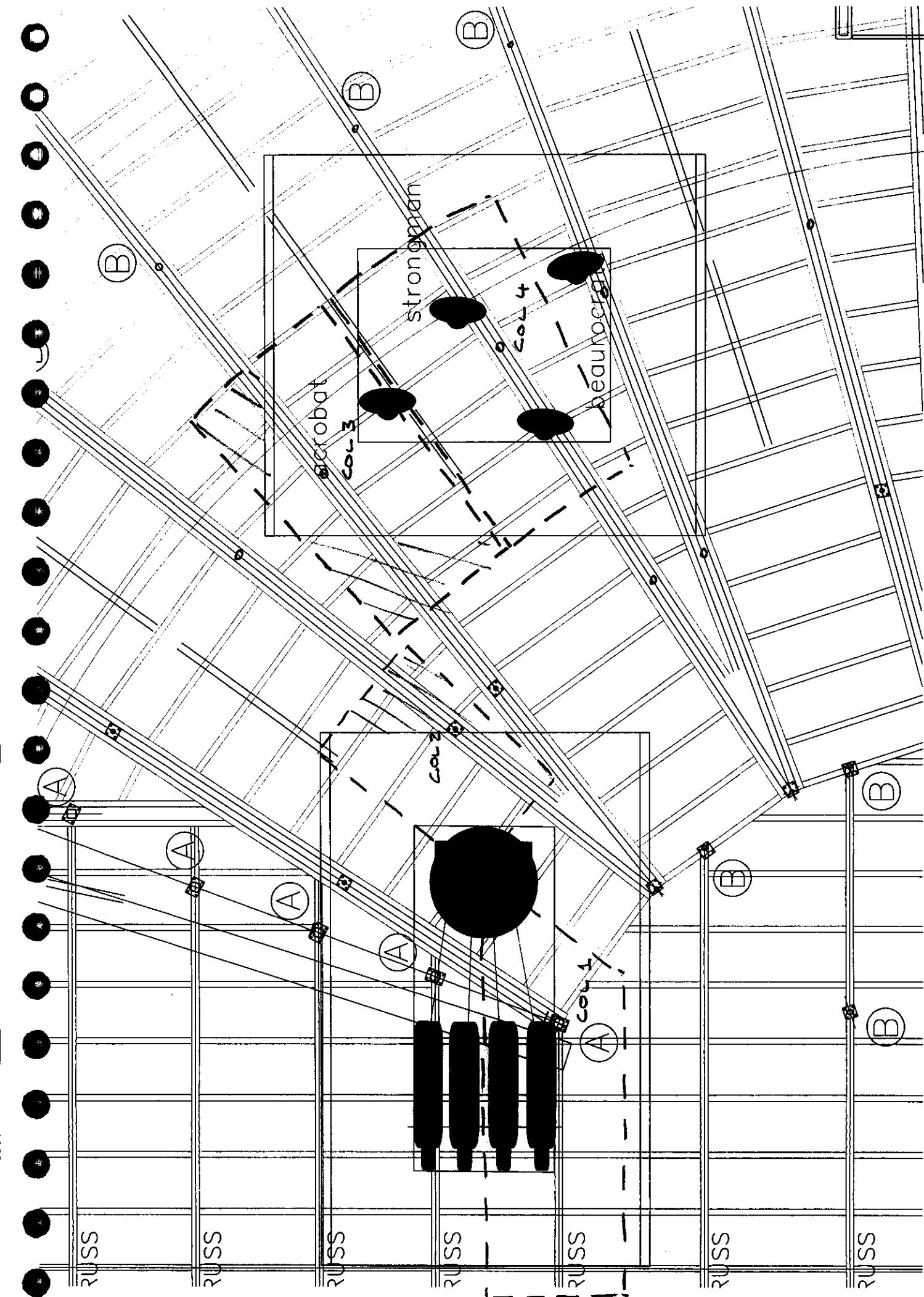
$$\text{Ground load} = 1.2 \times 1.3 \times \frac{1}{2} \times 400 = 312 \text{ kg.}$$

$$\text{Restricted access area} = 3.15 - \frac{(1.2 \times 1.3)}{2} = 2.37 \text{ m}^2$$

$$\text{Load allowance} = 2.37 \times 75 = 178 \text{ kg.}$$

COLUMN MACKING PLAN

1578/8/12



atelier one

Project BRITISH MUSEUM

No. 1578/B/13

Date 02/07 By CTS

Checked _____

Notes

$$\frac{1}{4} \text{ Chariot} = \frac{1061}{4} = 265 \text{ kg.}$$

$$\begin{aligned} \text{Total load} &= 312 + 178 + 265 = 755 \text{ kg} \\ &= 7.5 \text{ kN} < 30 \text{ kN OK.} \end{aligned}$$

- Check Cal 3

$$\text{Total area} = 2 \times 3.2 \text{ m} = 6.4 \text{ m}^2$$

$$\begin{aligned} \text{Crown load area} &= 1.5 \times 1.1 + 1.2 \times 1.3 \\ &= 3.2 \text{ m}^2. \end{aligned}$$

$$\text{Restricted access area} = 6.4 - 3.2 = 3.2 \text{ m}^2.$$

$$\text{Crown load} = 3.2 \times 400 = 1280$$

$$\text{Restricted access load} = 3.2 \times 75 = 240$$

$$1 \text{ m.s. warrior} = 186 \text{ kg.}$$

$$1706 \text{ kg.}$$

$$= 17 \text{ kN} < 30 \text{ kN OK.}$$

- Check Cal 4

$$\text{Total area} = 2 \times 3.2 = 6.4 \text{ m}^2$$

$$\text{Restricted access load} = 6.4 \times 75 = 480 \text{ kg.}$$

$$2 \text{ m.s. warriors} = 2 \times 186 = 372 \text{ kg.}$$

$$852 \text{ kg.}$$

$$= 8.5 \text{ kN} < 30 \text{ kN OK.}$$

Find 4 loads OK under exhibition loads